# GEODYN Documentation $_{\text{Volume }3}$

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## 1 INTRODUCTION

Specific problems to be solved by the GEODYN II program are defined by input cards. These cards fall into two categories: The Global Set and the Arc Set. The Global Set contains information which is common to all of the arcs being processed. Each Arc Set contains information defining its arc. Therefore one or more Arc Sets could be present in the GEODYN-II input.

The Global Set consists of four groups in the following order: The first group is the mandatory Run Description on three cards with no defined format. The second group is the set of Option cards which may be used to define and/or estimate conditions which are common to all the arcs being processed. The third group is the optional Station Coordinate Subgroup which may be used to alter GEODYN II's set of tracking stations. The last group in the Global Set is the mandatory Global Set Termination card.

The Arc Set also has four groups in this order: The first group is the mandatory Arc Description on three (or more) title cards well as mandatory cards to describe the reference coordinate system and time and spacecraft parameters in this arc. The second group is the set of Arc Option cards which may be specified to make use of GEODYN II's individual arc capabilities. The third group is the Data Selection/Deletion Subgroup which may be used to edit input observations. The Data Selection/Deletion Subgroup should not be present in orbit generation runs. In data reduction runs, the use of DATA and SELECT cards is mandatory (see individual card for explanation). The last group in the Arc Set is the mandatory Arc Set Termination card.

## 2 INPUT CARDS OVERVIEW

The Global Set begins with a mandatory subset of three cards on which the user may record any descriptive information concerning the job.

The Option cards which follow, are presented in the document in alphabetic order but not necessarily used in this order unless otherwise specified in the individual case. Information which applies to all arcs could be included in one of the following categories of input cards:

- 1. A forcing function description and/or condition input card such as:
  - cards which define the gravitational model
  - request application and/or modifications of the Earth or planet gravitational potential,
  - request application and/or modifications of the Earth tide model.
  - request estimation of the gravitational coefficients.
- 2. Observation modeling descriptive cards such as:
  - cards which require application of various models
  - modify and/or estimate parameters which describe the observation equation (polar motion, solid Earth tides, ocean tides and tectonic plate motion.)
- 3. Cards specifying models which describe tracking measurement corrections such as:
  - antenna or reflector offsets from the spacecraft center of gravity
  - choice of tropospheric refraction model
- 4. Cards which affect observation modeling.
- 5. Input cards controlling the format and the quantity of the listings and the printout.

The Global Set continues with the STAPOS subgroup. Input cards in this subgroup set conditions for solution, constraints and correlations pertaining to stations with available data. The STAPOS subgroup also includes cards containing station positions, geodetic information and planetary parameters for extra terrestrial stations. If the STAPOS subgroup is present it must be terminated with an ENDSTA card.

The Global Set of input cards ends with the mandatory card ENDGLB which also specifies the maximum number of Global parameter estimation iterations and the number of TITLE cards in each arc.

The Arc Set begins with a number of TITLE cards which are mandatory for each run on which the user may include any information pertaining the arc. Information to be included on the rest of the mandatory card subgroup is the coordinate system of integration, number of arc iterations, spacecraft information, start and end time of orbit integration and introduction of the components of orbital elements.

The next subgroup of the Arc Set is a group of Option cards which affects:

- 1. The solution itself with information such as:
  - editing multiplier
  - integration order/step size
  - apriori variance/covariance matrix for the satellite elements.
- 2. The observation equation with information such as:
  - measurement bias and
  - station timing bias
- 3. The force model with information such as application and/or adjustment of:
  - general acceleration
  - drag coefficient
  - solar radiation pressure

- 4. The output with information such as:
  - output of E or V-matrices
  - output required on specified units
  - $\bullet\,$  residual file output
  - normal point generation control

The next subgroup in the Arc Option cards is called the Data Selection subgroup and it is introduced by the DATA card. This card must be present with any arc requiring data reduction. The cards which follow DATA are very important to the user as an editing tool. One can specify Data Selection and Deletion time intervals, input of meteorological data, application of constant bias observation correction, request data preprocessing and specification of weighting and editing sigmas.

The Data Selection subgroup ends with an ENDARC card which also denotes end of the arc. In case more than one arc are required, the user should start with a new set of Arc TITLE cards.

A detailed description for each card in the Global and Arc Sets is given in the following section. That includes format and function description, default values and units and description of the consequences in case any of these cards is ommitted.

Finally the card ENDALL denotes the end of all arcs and the end of input cards.

# 2.1 A GUIDE TO GEODYN II INPUT CARDS

SUBJECT	CARD	DESCRIPTION	TYPE
ADJUSTMENT		Correlation between station cooordinates	STAPOS
CONTROL		Specify fixed stations	STAPOS
	CONSTADJ/	Adjusted stations/ Sigma for stations	STAPOS STAPOS
		Constrained stations control	STAPOS
		Normal point orbit adjustment control	ARC
		and a-priori variance input.	
	VARCOV :	A-priori var/covar matrix for the satellite elements.	ARC
	SIGMA :	Observation correction editing sigma	DATA
ALBEDO	ALBEDO :	Application of Earth radiation pressure	ARC
ALTIMETRY		Altimetry related input.	COMMON
	SSCOEF :	Spherical harmonics coefficients for	COMMON
		static sea surface topography model.	
	SSTOPO :	Max degree and order for static sea	COMMON
	OLOAD :	surface topography model.  Ocean loading using the expanded tide	COMMON
	ULUAD :	model.	COMMON
ANTENNA /	DELAY :	Transponder delay	COMMON
INSTRUMENT		Antenna offset/laser reflector offset	COMMON
	INSTRMNT:	Parameters for tracking instrument	STAPOS
ATMOSPHERIC EFFECT	DRAG :	Drag application / estimation of drag coefficient.	ARC
	ATMDEN :	Atmospheric density model.	COMMON
		Modify flux values.	COMMON
		Tropospheric model.	COMMON
	THRDRG:	Thermal Drag model (Lageos)	ARC
BIASES	EBIAS :	Pass by pass measurement biases	ARC
	MBIAS :	Measurement biases - Station	ARC
		timing biases.	
	PBIAS :	Pass by pass measurement or timing	ARC
	OBSCOB .	biases. Constant bias applied as observation	ARC
	obscon .	correction.	AILO
CONSTANTS	VLIGHT :	Speed of light	COMMON
COMPUTER	CYBG2E :	GEODYN IIE on CYBER (default).	COMMON
SELECTION		GEODYN IIE on IBM.	COMMON
SUBJECT	CARD	DESCRIPTION	TYPE
DATA	SELECT :	Data times/satellite/station/mtypes	DATA
SELECTION		Data times/satellite/station/mtypes	DATA
		Override met.data information	DATA
		Constant bias observation correction	DATA
		Data preprocessing.	DATA
	SIGMA :	Observation weighting and editing sigma.	DATA

GENERAL RELATIVITY	REL300	Relativistic corrections	ARC
GENERAL ACCELERATION	ACCEL	Application/adjustment of general acceleration.	ARC
nooddanniion	ACCELT	Application/adjustment of general acceleration in 3 dimensions.	ARC
	ACCEL9		ARC
GEOPOTENTIAL	EARTH	Modifies the Earth's gravitational potential/Earth's constants.	COMMON
	GEODETIC	Overrides the Earth's parameters for certain stations	STAPOS
	GCOEF	Modification/estimation of the coefficients in the Geopotential model	COMMON
	GCOEFS	Modification/estimation of C coeff.	COMMON
		Modification/estimation of S coeff.	COMMON
		Application of polar motion to the	COMMON
	GEOFUL	Earth's gravitational potential.	COMMON
	CDVEDO	-	COMMON
		Time depended gravity time intervals.	COMMON
	GRVIIM	Estimation of time dependent gravity coefficients.	COMMON
	GVDAND		COMMON
	GAPAND	Estimation of a range of geopotential coefficients.	COMMON
	DIANET	Third body gravitational potential.	COMMON
		Maximum degree of geopotential for	COMMON
	TOLS	variational equations.	COMMON
TNEEGDATION	DEEGVO		ADG
INTEGRATION INFORMATION	REFSIS	Specification of coordinate system for integration.	ARC
INFURNATION	EPOCH	Specification of times for integration	ARC
		Sets of satellites/times for	ARC
	DLAVE	integration	AILC
	STEP	Changes integration order and stepsize	ARC
	SIEF	changes integration order and stepsize	ARC
INTER -	TRKEDY	Defines tracking body	COMMON
PLANETARY		Third body gravitational potential	COMMON
		Planetary shape parameters for planetary	
	EXIMAGEO	stations.	SIAFUS
NORMAL	мормрт	Computation and output of normal naints	ARC
		Computation and output of normal points  Normal point orbit adjustment control	
POINTS	NURMPV	Normal point orbit adjustment control	ARC
SUBJECT	CARD	DESCRIPTION	TYPE
OUTPUT		Interrupts GIIS input listings	ARC
CONTROL	LIST	Negates effect of NOLIST	ARC
	OBSVU	Controls residual printout for all arcs,	ARC
		preprocessing correction output and	
	DADETI	measurement partials output	A D C
		Output partial derivative files	ARC
	PLATIM	Station coordinate printout at	ARC
		requested times. (Plate tectonics	
	D.D. = ***=====	option , see PLATE)	
	PRINTVU	Controls output of GIIS and GIIE on	ARC
		UNITO6.	

PUNCH : Requests punched output TERMVU : Output on UNITO9 (80 characters/line) ARC EMATRX : Output E matrix/V matrix ARC ORBFIL : Output trajectory file on specific units ORBTVU : Trajectory output on UNITO6 ARC RESIDU : Residual file output on UNIT19 ARC NORMPT : Computation and output of normal points ARC  PLATE PLATE : Tectonic plate motion. COMMON TECTONICS PLATIM : Reference date for station coordinates COMMON for tectonic plate motion.  POLAR MOTION POLE : Modifies polar motion and A1-UT1 COMMON values.  POLDYN : Dynamic polar motion. COMMON POLKF : Figure axis for dynamic polar motion. COMMON POLEUT : Modification/adjustment of true pole coordinates and A1-UT1 time differences  STATISTICAL EDIT : Editing multiplier/initial RMS. ARC CONTROL SIGMA : Observation correction editing sigma DATA  SIMULATIONS SIMDAT : SIMULATED DATA SUBGROUP ARC  SOFTWARE VECOPT : Vector optimization, core allocation COMMON OPTIMIZATION control.  SOLAR SOLRAD : Application/adjustment of solar ARC RADIATION  SPACECRAFT SATPAR : Spacecraft related parameters ARC RADIATION Control SLAVE : Sets of satellites/times for ARC integration ELEMS1 : Initial state vector (first 3 ARC components). ELEMS2 : Initial state vector (last 3 ARC
ORBFIL : Output trajectory file on specific units  ORBTVU : Trajectory output on UNITO6 RESIDU : Residual file output on UNIT19 ARC NORMPT : Computation and output of normal points ARC  PLATE PLATE PLATE : Tectonic plate motion. TECTONICS PLATIM : Reference date for station coordinates for tectonic plate motion.  POLAR MOTION POLE : Modifies polar motion and A1-UT1 COMMON values. POLLY : Dynamic polar motion. POLEUT : Modification/adjustment of true pole coordinates and A1-UT1 time differences  STATISTICAL EDIT : Editing multiplier/initial RMS. SIMULATIONS SIMDAT : SIMULATED DATA SUBGROUP  SOFTWARE OPTIMIZATION  SOLAR RADIATION  SOLAR RADIATION  SOLAR SOLRAD : Application/adjustment of solar and ARC radiation pressure.  SPACECRAFT SATPAR : Spacecraft related parameters ARC and
Units  ORBTVU : Trajectory output on UNITO6 RESIDU : Residual file output on UNIT19 ARC NORMPT : Computation and output of normal points ARC  PLATE PLATE : Tectonic plate motion.  COMMON TECTONICS PLATIM : Reference date for station coordinates for tectonic plate motion.  POLAR MOTION POLE : Modifies polar motion and A1-UT1 COMMON values. POLDYN : Dynamic polar motion. POLKF : Figure axis for dynamic polar motion. COMMON POLEUT : Modification/adjustment of true pole coordinates and A1-UT1 time differences  STATISTICAL EDIT : Editing multiplier/initial RMS. ARC CONTROL SIGMA : Observation correction editing sigma DATA  SIMULATIONS SIMDAT : SIMULATED DATA SUBGROUP ARC  SOFTWARE OPTIMIZATION  SOLAR RADIATION  SOLAR RADIATION  SOLAR RADIATION  SOLAR SOLRAD : Application/adjustment of solar radiation pressure.  SPACECRAFT RELATED PARAMETERS  ELEMS1 : Initial state vector (first 3
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PLATE PLATE : Tectonic plate motion. TECTONICS PLATIM : Reference date for station coordinates for tectonic plate motion.  POLAR MOTION POLE : Modifies polar motion and A1-UT1 COMMON values.  POLDYN : Dynamic polar motion.  POLET : Modification/adjustment of true pole coordinates and A1-UT1 time differences  STATISTICAL EDIT : Editing multiplier/initial RMS.  SIMULATIONS SIMDAT : SIMULATED DATA SUBGROUP ARC  SOFTWARE OPTIMIZATION COMMON control.  SOLAR RADIATION SOLRAD : Application/adjustment of solar radiation pressure.  SPACECRAFT RELATED PARAMETERS SLAVE : Sets of satellites/times for integration integration and souther the motion.  COMMON C
PLATE TECTONICS PLATIM: Reference date for station coordinates for tectonic plate motion.  POLAR MOTION POLE: Modifies polar motion and A1-UT1 COMMON values. POLDYN: Dynamic polar motion. POLKF: Figure axis for dynamic polar motion. POLEUT: Modification/adjustment of true pole coordinates and A1-UT1 time differences  STATISTICAL CONTROL SIGMA: Observation correction editing sigma DATA  SIMULATIONS SIMDAT: SIMULATED DATA SUBGROUP ARC  SOFTWARE OPTIMIZATION  SOLRAD: Application/adjustment of solar radiation pressure.  SPACECRAFT RELATED PARAMETERS ELEMS1: Spacecraft related parameters arc integration ELEMS1: Initial state vector (first 3 arc components).
TECTONICS PLATIM: Reference date for station coordinates for tectonic plate motion.  POLAR MOTION POLE: Modifies polar motion and A1-UT1 COMMON values. POLDYN: Dynamic polar motion. POLKF: Figure axis for dynamic polar motion. POLEUT: Modification/adjustment of true pole coordinates and A1-UT1 time differences  STATISTICAL EDIT: Editing multiplier/initial RMS. ARC CONTROL SIGMA: Observation correction editing sigma DATA  SIMULATIONS SIMDAT: SIMULATED DATA SUBGROUP ARC  SOFTWARE OPTIMIZATION Control.  SOLAR RADIATION  SOLAR SOLRAD: Application/adjustment of solar radiation pressure.  SPACECRAFT SLAVE: Spacecraft related parameters ARC ARC ARC integration PARAMETERS  ELEMS1: Initial state vector (first 3 ARC components).
POLAR MOTION POLE : Modifies polar motion and A1-UT1 COMMON values. POLDYN : Dynamic polar motion. POLKF : Figure axis for dynamic polar motion. COMMON POLEUT : Modification/adjustment of true pole coordinates and A1-UT1 time differences  STATISTICAL EDIT : Editing multiplier/initial RMS. SIGMA : Observation correction editing sigma DATA  SIMULATIONS SIMDAT : SIMULATED DATA SUBGROUP ARC  SOFTWARE OPTIMIZATION Control.  SOLAR RADIATION SOLRAD : Application/adjustment of solar radiation pressure.  SPACECRAFT RELATED SLAVE : Spacecraft related parameters ARC ARC ARC integration PARAMETERS ELEMS1 : Initial state vector (first 3 ARC components).
POLAR MOTION POLE : Modifies polar motion and A1-UT1 COMMON values.  POLDYN : Dynamic polar motion. POLKF : Figure axis for dynamic polar motion. COMMON POLEUT : Modification/adjustment of true pole coordinates and A1-UT1 time differences  STATISTICAL EDIT : Editing multiplier/initial RMS. SIGMA : Observation correction editing sigma DATA  SIMULATIONS SIMDAT : SIMULATED DATA SUBGROUP ARC  SOFTWARE OPTIMIZATION CONTROL : Vector optimization, core allocation COMMON control.  SOLAR RADIATION SATPAR : Spacecraft related parameters ARC radiation pressure.  SPACECRAFT SLAVE : Sets of satellites/times for integration ELEMS1 : Initial state vector (first 3 ARC components).
Values.  POLDYN: Dynamic polar motion. POLKF: Figure axis for dynamic polar motion. POLEUT: Modification/adjustment of true pole coordinates and A1-UT1 time differences  STATISTICAL EDIT: Editing multiplier/initial RMS. SIGMA: Observation correction editing sigma DATA  SIMULATIONS SIMDAT: SIMULATED DATA SUBGROUP ARC  SOFTWARE OPTIMIZATION Control.  SOLAR RADIATION SOLRAD: Application/adjustment of solar radiation pressure.  SPACECRAFT SATPAR: Spacecraft related parameters ARC ARC and and an application and application for an application and application
POLDYN : Dynamic polar motion. POLKF : Figure axis for dynamic polar motion. COMMON POLEUT : Modification/adjustment of true pole coordinates and A1-UT1 time differences  STATISTICAL CONTROL
POLKF : Figure axis for dynamic polar motion. POLEUT : Modification/adjustment of true pole coordinates and A1-UT1 time differences  STATISTICAL EDIT : Editing multiplier/initial RMS. ARC CONTROL SIGMA : Observation correction editing sigma DATA  SIMULATIONS SIMDAT : SIMULATED DATA SUBGROUP ARC  SOFTWARE OPTIMIZATION Control.  SOLAR RADIATION SOLRAD : Application/adjustment of solar radiation pressure.  SPACECRAFT RELATED SLAVE : Spacecraft related parameters ARC radiation integration ELEMS1 : Initial state vector (first 3 ARC components).
POLEUT: Modification/adjustment of true pole common coordinates and A1-UT1 time differences  STATISTICAL EDIT: Editing multiplier/initial RMS. ARC CONTROL SIGMA: Observation correction editing sigma DATA  SIMULATIONS SIMDAT: SIMULATED DATA SUBGROUP ARC  SOFTWARE OPTIMIZATION Control.  SOLAR RADIATION SOLRAD: Application/adjustment of solar radiation pressure.  SPACECRAFT RELATED PARAMETERS SATPAR: Spacecraft related parameters ARC integration ELEMS1: Initial state vector (first 3 ARC components).
STATISTICAL EDIT : Editing multiplier/initial RMS. ARC CONTROL SIGMA : Observation correction editing sigma DATA  SIMULATIONS SIMDAT : SIMULATED DATA SUBGROUP ARC  SOFTWARE OPTIMIZATION Control.  SOLAR RADIATION SOLRAD : Application/adjustment of solar radiation pressure.  SPACECRAFT RELATED PARAMETERS SLAVE : Sets of satellites/times for integration ELEMS1 : Initial state vector (first 3 components).
CONTROL SIGMA : Observation correction editing sigma DATA  SIMULATIONS SIMDAT : SIMULATED DATA SUBGROUP ARC  SOFTWARE VECOPT : Vector optimization, core allocation COMMON CONTROL.  SOLAR RADIATION SOLRAD : Application/adjustment of solar radiation pressure.  SPACECRAFT SATPAR : Spacecraft related parameters ARC RELATED SLAVE : Sets of satellites/times for ARC integration  ELEMS1 : Initial state vector (first 3 ARC components).
CONTROL SIGMA : Observation correction editing sigma DATA  SIMULATIONS SIMDAT : SIMULATED DATA SUBGROUP ARC  SOFTWARE VECOPT : Vector optimization, core allocation COMMON CONTROL.  SOLAR RADIATION SOLRAD : Application/adjustment of solar radiation pressure.  SPACECRAFT SATPAR : Spacecraft related parameters ARC RELATED SLAVE : Sets of satellites/times for ARC integration  ELEMS1 : Initial state vector (first 3 ARC components).
SOFTWARE VECOPT: Vector optimization, core allocation COMMON OPTIMIZATION control.  SOLAR SOLRAD: Application/adjustment of solar radiation pressure.  SPACECRAFT SATPAR: Spacecraft related parameters ARC RELATED SLAVE: Sets of satellites/times for ARC integration ELEMS1: Initial state vector (first 3 ARC components).
OPTIMIZATION control.  SOLAR SOLRAD: Application/adjustment of solar radiation pressure.  SPACECRAFT SATPAR: Spacecraft related parameters ARC RELATED SLAVE: Sets of satellites/times for ARC integration ELEMS1: Initial state vector (first 3 components).
OPTIMIZATION control.  SOLAR SOLRAD: Application/adjustment of solar radiation pressure.  SPACECRAFT SATPAR: Spacecraft related parameters ARC RELATED SLAVE: Sets of satellites/times for ARC integration ELEMS1: Initial state vector (first 3 components).
RADIATION radiation pressure.  SPACECRAFT SATPAR: Spacecraft related parameters ARC RELATED SLAVE: Sets of satellites/times for ARC PARAMETERS integration ELEMS1: Initial state vector (first 3 ARC components).
RADIATION radiation pressure.  SPACECRAFT SATPAR: Spacecraft related parameters ARC RELATED SLAVE: Sets of satellites/times for ARC PARAMETERS integration ELEMS1: Initial state vector (first 3 ARC components).
SPACECRAFT SATPAR: Spacecraft related parameters ARC RELATED SLAVE: Sets of satellites/times for ARC PARAMETERS integration ELEMS1: Initial state vector (first 3 ARC components).
RELATED SLAVE : Sets of satellites/times for ARC PARAMETERS integration ELEMS1 : Initial state vector (first 3 components).
PARAMETERS integration ELEMS1: Initial state vector (first 3 ARC components).
ELEMS1 : Initial state vector (first 3 ARC components).
components).
<u>-</u>
ELEMS2: Initial state vector (last 3 ARC
components).
VARCOV : A-priori variance-covariance for ARC
satellite elements
SUBJECT CARD DESCRIPTION TYPE
SPEED OF VLIGHT: Speed of light COMMON
SPEED OF VLIGHT: Speed of light COMMON LIGHT
LIGHT
LIGHT  STATIONS STAPOS: STAPOS SUBGROUP COMMON
STATIONS STAPOS: STAPOS SUBGROUP COMMON INFORMATION STATION COORDINATE CARD: Coordinates for stations STAPOS ADJUSTMENT CORREL: Correlation between station cooordinates STAPOS CONTROL FIXED: Specify fixed stations STAPOS
STATIONS STAPOS: STAPOS SUBGROUP COMMON INFORMATION STATION COORDINATE CARD: Coordinates for stations STAPOS ADJUSTMENT CORREL: Correlation between station cooordinates STAPOS CONTROL FIXED: Specify fixed stations STAPOS ADJUSTED: Adjusted stations/ Sigma for stations STAPOS
STATIONS STAPOS: STAPOS SUBGROUP COMMON INFORMATION STATION COORDINATE CARD: Coordinates for stations STAPOS ADJUSTMENT CORREL: Correlation between station cocordinates STAPOS CONTROL FIXED: Specify fixed stations STAPOS ADJUSTED: Adjusted stations/ Sigma for stations STAPOS ADJUSTMENT CONSTADJ/ STAPOS
STATIONS STAPOS: STAPOS SUBGROUP COMMON INFORMATION STATION COORDINATE CARD: Coordinates for stations STAPOS ADJUSTMENT CORREL: Correlation between station cooordinates STAPOS CONTROL FIXED: Specify fixed stations STAPOS ADJUSTED: Adjusted stations/ Sigma for stations STAPOS ADJUSTMENT CONSTADJ/ STAPOS CONTROL CONSTEND: Constrained stations control STAPOS
STATIONS STAPOS: STAPOS SUBGROUP COMMON INFORMATION STATION COORDINATE CARD: Coordinates for stations STAPOS ADJUSTMENT CORREL: Correlation between station cocordinates STAPOS CONTROL FIXED: Specify fixed stations STAPOS ADJUSTED: Adjusted stations/ Sigma for stations STAPOS ADJUSTMENT CONSTADJ/ STAPOS

TIDE MODELING	ETIDEN :	Colombo Earth tide model using normalized Legendre polynomials.	COMMON
HODELING	ETIDES :	Colombo Earth tide model using unnormalized Legengre polynomials.	COMMON
	H2LOVE :	Modification/adjustment of solid Earth tide coefficient of the 2nd kind.	COMMON
	L2LOVE :	Modification/adjustment of solid Earth tide coefficient of the 3rd kind.	COMMON
	TIDES :	Modifies Earth tide perturbations.	COMMON
		Applies ocean loading using the expanded	COMMON
		poletide model.	
	OTIDEN :	Ocean tides using the Colombo model	COMMON
		with normalized Legendre polynomials.	
	OTIDES :	Ocean tides using the Colombo model	COMMON
		with unnormalized Legendre polynomials.	
VARIANCE/ COVARIANCE	ADJUSTED:	Adjusted stations/ Sigma for stations	STAPOS
	NORMPV :	Normal point orbit adjustment control	ARC
	WAD GOV	and a-priori variance input.	ADG
	VARCUV:	A-priori var/covar matrix for the satellite elements.	ARC
	SIGMA :	Observation correction editing sigma	DATA

# 2.2 GLOBAL SET MANDATORY CARDS

## 2.2.1 TITLE

	+1	+	-23		+	4	+-	5	+	6	+	7	+	8
USER	SPECIFIED	JOB	DESCRIPTION	-	CARD	1								
USER	SPECIFIED	JOB	DESCRIPTION	-	CARD	2								
USER	SPECIFIED	JOB	DESCRIPTION	-	CARD	3								
	+0	+	-0+0		+	0	+-	0	+	0	+	0	+	0

COLUMNS FORMAT DESCRIPTION

DEFAULT VALUE & UNITS

1-80 10A8 User may specify on these three cards, information description of the job. These cards may also remain blank, but must always be present.

IF CARDS OMITTED: Run will abnormally terminate.

# 2.3 GLOBAL SET OPTION CARDS

# 2.3.1 ADELAY

ADELAY	19601101	1.0	1.00		
+0	-+0	+0	-+0	0+0-	+0
COLUMNS FORMA	T DESCRIPTION			DEFAULT VALU	JE & UNITS
1-6 A6	ADELAY: Introduces and/or estimate the of a S/C or a laser	e Antenna	Transponder	Delay	timeter
15-17 I3	Antenna number  If on columns 16-17  represents a laser i  will be treated a la	ID and the	e Antenna Tra	nsopnder Delay	
18-24 I7	Satellite ID				
25-44 D20.8	A priori value of the	ne Antenna	a Transponder	Delay	
45-59 D15.3	Sigma of the Antenna	a Transpor	nder Delay		
	LAY card content will e antenna number.	l be paire	ed with the O	FFSET card feat	uring
OFFSET 91	11606401		1.430712	-5.320D-06	1.419411
OFFSET 91	21606401		1.287648	-0.4826000	.1359708
OFFSET 91	31606401		-1.305229	-0.6616698	2.009962
OFFSET 91	41606401		1.417400	-0.6858000	.2482772
ADELAY	11606401	580.4808	1.00D	-20	
ADELAY	21606401	575.3643	1.00D	-20	
ADELAY	31606401	569.9368	1.00D	-20	
ADELAY	41606401	576.6929	1.00D	-20	
	ilometers; M = Meters ertz ; MH = Mega - Ho		=Seconds ;	M/S=Meters per	second

# 2.3.2 ALTIM

	1 1	+2+3+8
	0	+0+0+0
COLUMNS	FORMAT	DESCRIPTION DEFAULT VALUE & UNITS
1-6	A6	ALTIM - Controls various options connected with the altimetry measurement type.
7	I1	<pre>= 0 Dynamic portion of altimetry</pre>
		measurement partials wrt gravity coefficients will not be considered.
8	I1	= 0 Geometric portion of altimetry 0 measurement partials wrt gravity coefficients will be considered.
		= 1 Geometric portion of altimetry measurement partials wrt gravity coefficients will not be considered.
9	I1	= 0 GEODYN IIE will not replace a geoid 0 supplied from the data tape unless gravity coefficients are being adjusted.
		= 1 GEODYN IIE will replace a geoid supplied from the data tape (with the geoid implied by the gravity coefficients) even if gravity coefficients are not being adjusted.
		= 2 GEODYN will use an external grid to compute the mean sea surface. This computation will replace the Geodyn computed geoid and sea surface height. The external grid file is input on unit 29 of Geodyn IIE. [NOTE 4]
COLUMNS	FORMAT	DESCRIPTION DEFAULT VALUE & UNITS
10	I1	= 0 GEODYN IIE will not replace geometric 0 earth tides from the data tape unless H2 Love number is being adjusted. [NOTE 5]
		= 1 GEODYN IIE will replace geometric earth tides from the data tape even if H2 Love number is not being adjusted. [NOTE 5]
11	I1	= 0 GEODYN IIE will not replace geometric 0 ocean tides from the data tape. [NOTE 5]
		= 1 GEODYN IIE will replace geometric

ocean tides from the data tape. [NOTE	ocean	tides	from	the	data	tape.	ΓNOTE	5
---------------------------------------	-------	-------	------	-----	------	-------	-------	---

- = 2 GEODYN IIE will compute the geometric ocean tides using the GOT99.2 Ray model. [NOTE 3]
- 12 I1 = 0 GEODYN IIE will not replace sea surface 0 topography from the data tape. [NOTE 5]
  - = 1 GEODYN IIE will replace sea surface topography from the data tape. [NOTE 5]
- 13 Index to the altimetric observation modeling
  - = 0 Use the S/C normal height for modeling the observation.
  - = 1 Use the provided external attitude information to model the altimetric observation
  - =2 External attitude information is provided at observation times on the data records.

    (Ocean parameter records words 6-9)
- 14 I1 Total number of points for a dynamic crossover cols 18-19 stream after best time estimation
- 15 II Number of inner iterations after which the number on column 14 replaces the number on columns 18-19 and the number on columns 20-21 replace the number on columns 18-19
- 18-19 I2 Number of observations retained below each side of a crossover (used in dynamic crossover processing mtype 101)
- 20-21 I2 Degree of the polynomial used for dynamic crossover fitting.
- 22-23 I2 Second choice for degree of the polynomial for dynamic crossover fitting.
- 25-44 D20.8 A priori value for crossover radial separator. (0.01)
- 45-59 D15.3 Crossover partial editing level (cm) 10000
- 60-72 D13.1 Crossover distance editing level (cm) 0
- 73-80 D8.2 Maximum number of sltimeter bounce iterations
- NOTE [1]: It is possible to turn off either the dynamic portion of the Altimetry partials wrt to gravity coefficients or the geometric portion but not both.
- NOTE [2]: If geoid, earth tide, or ocean tide models are not supplied from the Altimetry data, GEODYN IIE will automatically calculate the models not supplied.

NOTE [3]: For the geometric ocean tide model correction using the GOT99.2 Ray tide model the analyst needs to include the following grid file in the directory where Geodyn IIE is executed:

Hp:

 $/ \verb|geod4/geodyn/SUPPORT/dat_raytide/got99_grid.dat|$ 

Cray:

/u2/z8sgp/geodyn/support/dat\_raytide/got99\_grid.dat

NOTE [4]: To run this option the external grid file must be ftp'd from unitree to the directory where Geodyn IIE is executed onto unit 29:

Cray file:

/u2/z8sgp/zms01/mssdata/mssh95.one16th.deg.craybin.Z

Hp file:

 $/\verb"u2/z8sgp/zms0l/mssdata/mssh95.one16th.deg.hpbin.Z$ 

NOTE [5]: For a better understanding of the functions invoked by the indices on columns 10 and 11 see the table below:

#### Col 10 FLAGS:

GEODYN		0	1
INPUT			
CARDS			
ETIDES H2 L2	NO Sigma	1. $\Delta T_{\rm SE}$ will be read from input tape 2. There are no dynamical direct effects	<ol> <li>GEODYN will compute ΔT<sub>SE</sub> using the Merit Standards for H2, L2, K1, and δΔT</li> <li>There are no dynamical direct effects</li> </ol>
ETIDES H2 L2	WITH Sigma	1. GEODYN will compute $\Delta T_{\rm SE}$ 2. Partials for H2 and L2 will be generated	1. GEODYN will compute $\Delta T_{\rm SE}$ 2. Partials for H2 and L2 will be generated

#### Col 11 FLAGS:

GEODYN INPUT CARDS		0	1
OTIDES	NO Sigma	1. $\Delta T_{\rm OC}$ will be read from input tape 2. $\Delta V$ will be computed from the ocean tide model	1. $\Delta T_{\rm OC}$ comes from the OTIDE cards 2. $\Delta T_{\rm OC}$ comes from the OTIDE cards
OTIDES	WITH Sigma	<ol> <li>ΔT<sub>OC</sub> will be read from input tape</li> <li>Dynamic partials will be generated for ocean tides</li> <li>NO geometric partials will be generated</li> </ol>	1. $\Delta T_{\rm OC}$ comes from the OTIDE cards 2. Dynamic partials will be generated 3. $\delta \Delta T_{\rm OC}$ will be read from the input tape

 $\Delta T_{ exttt{SE}}$  Correction for the solid Earth tides

 $\Delta T_{ t OC}$  Correction for the ocean tides

# 2.3.3 ALTIM2

ALTIM2	100	+2+5+5+	
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A 6	ALTIM2 - Controls various options connected with the altimetry measurement type.(CONTINUED)	
7-10	14	Defines perturbation value used to estimate numerical partials observations wrt satellite coordinates.	100 meters
12	I1		
13	I1		
14	I1		
15	I1		
16	I1		
18-19	12		
20-21	12		
22-23	12		
25-44	D20.8		
45-59	D15.3		
60-72	D13.1		
73-80	D8.2		

#### 2.3.4 ALTOUT

ALTOUT	-	+3+4+5+	
+	0+0	+0+	0+0
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A6	Requests creation and output of the altimetry geolocation file on unit 18.	
8	I1	<ul> <li>=0 No output.</li> <li>=1 Output the geolocation file at the first inner iteration of the first global iteration.</li> <li>=2 Output the geolocation file at the last inner iteration of the first global iteration.</li> <li>NOTES [2], [3]</li> </ul>	
9	I1	<pre>=0 No output. =1 Output the PDOUT file    at the first inner iteration    of the first global iteration. =2 Output the PDOUT file    at the last inner iteration    of the first global iteration.</pre>	
25-44 D	20.8	Mission reference date	YYMMDDHHMMSS
45-59 D	15.3	Mission reference fraction of seco	nds 0.SSSSSSS

NOTE [1]: The ALTOUT option applies only to altimetry or crossover data.

NOTE [2]: If dynamic crossover data type is being processed (mtype 101) and an ALTOUT 2 card is present, a geolocation file will be output at the last inner interation.

NOTE [3]: If Laser Altimeter modeling using external attitude is invoked and an ALTOUT 2 card is present, a geolocation file will be output at the last inner iteration.

IF CARD OMITTED: No Geolocation file will be output

#### 2.3.5 **ANTPHC**

---+---1---+---2---+---3----+----4----+---5----+---6----+----8
ANTPHC
----+---0---+---0---+---0---+---0

COLUMNS FORMAT

DESCRIPTION

DEFAULT VALUE & UNITS

0

1-6 A6 ANTPHC - Requests the input of tables with phase center variations (PCV) for GPS transmitter, ground station and user satellite receivers. The PCV corrections are input on unit 22. When this option is invoked columns 9-10 of the INSTRMNT card (STATION POSITION SUBGROUP) is used to identify the type of antenna used at each station. For satellite antennae the antenna number is obtained from columns 11-12 of the OFFSET card. As many as 99 PCV tables can be input simultaneously on unit 22. The table of corrections for each antenna must begin with a line that has 8 characters starting in column1: ANTNO=NN where NN is an integer from 1 through 99. After that there must be a header record. The format of header record is

'(I4,2F7.2,I4,2F7.2)'. These six numbers are I4: the number of zenith angles in the table

F7.2: the first zenith angle

F7.2: step of zenith angle

I4: the number of azimuth angles

F7.2: the first azimuth angle

F7.2: step of azimuth angle

The header record must be followed by the data record. The number of data must match the header record (the number of zenith angles times the number of azimuth angles). Each line contains 8 corrections (8D10.4).

15-17 I3

Measurement Type which the correction will be applied to. (This is only necessary when using the ANTPHC option for DORIS station antennae phase map correction, MTYPE=40)

The data should be stored in the order of azimuth vs zenith. For example, if there are NAZM azimuth angles and NZEN zenith angles for a table. The first NAZM numbers are for the first zenith angle. The (NAZM+1) to 2\*NAZM numbers are for the second zenith angle, etc. A sample table with ANTNO 28 is given below.

Note when ANTPHC is present and an antenna number specified in either OFFSET card or INSTRMNT card cannot be found from the external table file, PCV corrections will no be applied for that satellite or station.

See also INSTRMNT and OFFSET card

#### A sample PCV table: ANTN0=28 1.00 1 0.00 360.00 -.6100D - 02 - .5200D - 02 - .3300D - 02 - .1000D - 020 .1400D - 020 .3500D - 020 .4700D - 020 .4900D - 020 .4000D - 020 .4000D - 020 .4000D - 020 .4000D - 0200.4100D - 020.2800D - 020.8000D - 03 - .1000D - 02 - .2100D - 02 - .2100D - 02 - .1400D - .1400D - 02 - .1400D - .14-.1400D - 02 - .1400D - 02 --.1400D - 02 - .1400D - 02 --.1400D - 02 - .1400D - 02 --.1400D - 02 - .1400D - 02 --.1400D - 02 - .1400D - 02 --.1400D - 02 - .1400D - 02 --.1400D - 02 - .1400D - 02 --.1400D - 02 - .1400D - 02 --.1400D - 02 - .1400D - 02 --.1400D-02-.1400D-02-.1400D-02

#### 2.3.6 ANTPH2

1	1:	23	3	1	5+6	3+7	7+8
ANTPH2							
(	) + (	0(	)(	)+(	0(	)(	)+0

#### COLUMNS FORMAT

#### DESCRIPTION

DEFAULT VALUE & UNITS

1-6 A6 ANTPH2 - Requests application of GPS antenna phase correction by an external PCV file on unit 22. Multiple PCV tables can be input simultaneously on unit 22. Each table of corrections must start with a line beginning with either STANO or SATID, followed by the station number or satellite ID. (See Note 1)

Next, there needs to be a header record. The format of header record is '(I4,2F7.2,I4,2F7.2)'.

These six numbers are

I4: the number of zenith angles in the table

F7.2: the first zenith angle

F7.2: step of zenith angle

I4: the number of azimuth angles

F7.2: the first azimuth angle

F7.2: step of azimuth angle

The header record must be followed by the data record. The number of data must match the header record (the number of zenith angles times the number of azimuth angles). Each line contains 8 corrections (8D10.4).

The data should be stored in the order of azimuth vs zenith. For example, if there are NAZM azimuth angles and NZEN zenith angles for a table. The first NAZM numbers are for the first zenith angle. The (NAZM+1) to 2\*NAZM numbers are for the second zenith angle, etc. A sample table is given below.

Unit 22 should be structured with the SATID tables before the STANO tables, followed by the ANTNO tables. With this structure, both ANTPHC and ANTPH2 options can be used with one external file.

### NOTES:

#### Note 1

If there is more than one antenna for a station, than the STANO header will need an antenna number included following the station number. Include this antenna number on the INSTRMNT card. (If there is only one antenna, there is no need for an antenna number following the station number or on the INSTRMNT card.)

A sample of two PCV tables on unit 22:

#### SATID=7801041 0.00 1.00 1 0.00 360.00 -.1000D - 02 - .2600D - 02 - .1200D - 02 - .9000D - 030 .5000D - 030 .1400D - 020 .2000D - 0200.1700D - 020.5000D - 03 - .1000D - 03 - .6000D - 03 - .7000D - 03 - .6000D - 03 - .3000D - 030.2000D - 030.2000 $\tt 0.9000D-030.1800D-020.2900D-020.2000D-020$ 0.2900D - 020.2900D - 020.2000D - 020.20 $\tt 0.2900D-020.2000D-020.2000D-020.2000D-020.2000D-020.2000D-020.2000D-020.$ 0.2900D - 020.2900D - 020.2000D - 020.200.2900D - 020.2900D - 020.2000D - 020.200.2900D - 020.2900D - 020.2000D - 020.20 $\tt 0.2900D-020.2000D-020.2000D-020.2000D-020.2000D-020.2000D-020.2000D-020.$ $\tt 0.2900D-020.2000D-020.2000D-020.2000D-020.2000D-020.2000D-020.2000D-020.$ 0.2900D - 020.2900D - 020.2000D - 020.200.2900D-020.2900D-020.2900D-02 STANO=14200501 5.00 1 0.00 360.00 17 0.00 0.0000E + 00 - .9937E - 03 - .2876E - 02 - .5380E - 02 - .7822E - 02 - .1083E - 01 - .1299E - 01 - .1450E - 01 --.1550E - 01 - .1549E - 01 - .1489E - 01 - .1354E - 01 - .1161E - 01 - .9500E - 02 - .6603E - 02 - .2883E - 02 - .2883E - 02 - .2883E - 03 - .2883E - .2883E - .2883E - .2882E - .2882E0.1401E-02

#### 2.3.7 APLOAD

APLOAD	_	2+3+5+6+8 0+0+0+0
COLUMNS	FORMAT	DESCRIPTION DEFAULT VALUE & UNITS
1-6	A6	APLOAD - Requests application of atmospheric pressure loading on all sites featured in the IIS input APLFIL (UNIT 24)
		The information on the APLFIL may be obtained from three (3) different models:  1. The Leonid Petrov model

The Leonid Petrov model
 The Jean Paul Boy model
 The Tony Van Dam model
 Atmospheric Pressure Loading displacements are given at 6 hour intervals and refer to the site

are given at 6 hour intervals and refer to the site (Geodyn 4 digit site) number. This 4 digit site number is the same as the site number on the OLOAD option card.

IF CARD OMITTED: Atmospheric Pressure Loading will not be applied.

## 2.3.8 ATGRAV

+ ATGRAV	1+	25	-68
	0+	0+0+0+	-0+0
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A6	ATGRAV - Requests application of atmospheric effect on the gravity field. The presence of this card requests input data on Unit 18 in GEODYN IIS. [1] The data are times and coefficients describing the atmospheric models.	
7-8	12	Maximum degree of expansion. [2]	50
9-10	12	Maximum order of expansion. [2]	50
11-12	12	Indicator for linear interpolation	0
		<ul> <li>=0 Do not perform linear interpolation for the atmospheric coefficients. Apply the given values if time of observation falls six hours before or after the tweether the coefficients were computed.</li> <li>=1 Perform linear interpolation. Use the of atmospheric coefficient sets that and follow the observation time. For observations that fall before the first and after the last set perform linear extrapolation</li> </ul>	on ime pair preceed
13	I1	SWITCH to perform interpolation on C21 and S21 coefficients = 0 do not perform interpolation = 1 perform interpolation on C21 and S2	0
25-44	D20.8	YYMMDDHHMMSS.SS Start time of application	epoch start time
45-59	D15.3	YYMDDHHMMSS.SS End time of application	epoch stop
NOTE [1	]: GEODYN	IIS unit 18 is a binary file which contain	ns in each record.
	fo	ollowing coefficients apply and stream of DEGxORD coefficients in the following the stream of DEGxORD coefficients in the stream of DEGxORD coefficien	where the lowing sequence

S11 C20 C21 S21 C22 S22 C30 C31 S31 C32 S32 C33 S33

NOTE [2]: The expansion should be less than or equal to the gravity field expansion. If not, only the coefficients of degree and order less than the maximum degree and order of the gravity field will contribute.

GEODYN will accept up to 90x90 expansions for this option given at 3 hour intervals.

## 2.3.9 **ATMDEN**

	1	+2+3+4+5+6+7+8
ATMDEN	71	
		+0+0
COLUMNS	FORMAT	DESCRIPTION DEFAULT VALUE & UNITS
1-6	A6	ATMDEN - Specifies the static density model that will be used for atmospheric density calculations.
9-10	12	Model selection number. 71
		= 71 - Jacchia 1971 Model = 87 - French Drag Model [1] = 86 - MSIS Empirical Drag Model [2]
11-12	12	= 0 or 1 For Mars runs, the Culp-Stewart 1983 1 atmospheric density model is used  = 2 For Mars runs, the Stewart 1987 atmospheric density model is used  = 3 For Mars runs, the Mars Global Reference atmospheric density model is used (Marsgram)  = 4 For Mars runs, the revised Stewart atmospheric density model (by Lemoine) will be used  = 5 For Mars runs, DTM-MARS is used  = 0 or 1 For Venus runs, the JPL atmospheric density model is used  = 2 For Venus runs, the Hedin atmospheric density model is used
15-16	12	Controls choice of 3 hourly or daily Kp values and whether the solar flux average is for the end or midpoint of the 81 day averaging interval for atmospheric density model 87.
	Column	value is calculated with the moving average at the end point or midpoint of the averaging interval.
		<pre>= 0 End point of interval = 1 Midpoint of interval</pre>
COLUMNS	FORMAT	DESCRIPTION DEFAULT VALUE & UNITS
	Column	16 Controls whether 3 hourly or daily values of Kp are used.

25-44 D20.8 Controls the maximum height in Venus runs with JPL

= 0 3 hourly Kp values used [NOTE 1] = 1 Daily Kp values used Model and Hedin Model. Range is 1.0km - 1000000.0km

45-59 D15.3 If the DTM-MARS model is selected, this value gives the dust opacity (infrared wavelength)

NOTE [1]: Requires that 3 hourly Kp values be available in the BIH Tables file on UNITO2.

NOTE [2]: Works with both 3-hourly Kp values and 24-hour Kp values.

IF CARD OMITTED: The Jacchia 1971 density model will be used for atmospheric density calculations.

# 2.3.10 BINAST

BINAST	200. 1.00 400.0 0.10
1-6 A6	BINAST - Introduces a second planetary body to be integrated
25-44 D20.8	Value of GM
45-59 D15.3	Sigma for GM
60-72 D13.1 73-80 D8.2	Value of semi-major axis Sigma for semi-major axis

#### 2.3.11 CGMASS

CGMASS		+2+3+4+5+ .02 +0+0+0+	16	-5.0
COLUMNS	FORMAT	DESCRIPTION	DEFAULT V	ALUE & UNITS
1-6	A6	CGMASS - Defines new mass ratio and body-centered coordinates of center of gravity with respect to center of figure.		
7-14	18	Ratio of current spacecraft mass to value on SATPAR card in units of 1.0E-7.(DEFAULT=10000000 for a ratio of 1)	100000	00
15	I1	<pre>=0 No external CGMASS coordinates will be read. =1 Read external ascii CGMASS coordinates for this satellite. =2 Read external binary CGMASS coordinates for this satellites. [NOTE 2]</pre>	0	
18-24	17	Satellite ID (required)	0	
25-44	D20.8	Start Date and Time in YYMMDDHHMMSS. [NOTE 1]	0.	
45-59	D15.3	Body centered fixed coordinate of the center of gravity with respect to the center of the figure (X coordinate in meters).	0.	М
60-72	D13.1	Body centered fixed coordinate of the center of gravity with respect to the center of the figure (Y coordinate in meters).	0.	М
73-80	D8.3	Body centered fixed coordinate of the center of gravity with respect to the center of the figure (Z coordinate in meters).	0.	М
NOTE 1:	less that that	me provided on the CGMASS option must be equal han the beginning of the satellite arc. This he CGMASS is applied throughout the entire a ltiple CGMASS cards the CGMASS will be applicable.	will ensurc. If the	re

33

named as cgmassXXXXXXX, where XXXXXXX is the satellite ID number.

NOTE 2: The time dependent external CGMASS coordinates file should be

The data structure of  ${\tt cgmassXXXXXXX}$  is as following:

next CGMASS card starts.

Satellite ID

Header record: (4 real numbers)

Start time in UTC YYMMDDHHMMSS.SS format
Stop time in UTC YYMMDDHHMMSS.SS format
Time step in seconds
Data record:
Time tag
X coordinate
Y coordinate

Z coordinate

UNITS: KM = Kilometers; M = Meters ; S = Seconds ; M/S = Meters per second

IF CARD OMITTED: The ratio of current spacecraft mass to the value on the SATPAR card is 1.

# 2.3.12 COMCOF

COMCOF	+
COLUMNS FOR	MAT DESCRIPTION DEFAULT VALUE & UNITS
1-6 A6	COMCOF - Defines and/or requests adjustments of body-centered coordinates for center mass- center of figure OFFSET of an orbited celestial body.
25-44 D20	.8 x coordinate of the com-cof offset
45-59 D15	.3 y coordinate of the com-cof offset
60-72 D13	.1 z coordinate of the com-cof offset
73-80 D8.	3 Sigma for the com-cof coordinates

#### 2.3.13 CYBG2E

1	1 – – – – + – – – -	-2+	3	4	5	6	7+8
CYBG2E							
(	) +	-0	0	0	0	0	00

COLUMNS FORMAT

DESCRIPTION

DEFAULT VALUE & UNITS

1-6 A6 CYBG2E - Specifies that the interface file output by GEODYN IIS on unit 11 will be for a IIE run on a CYBER 200 series computer.

Information in the interface file will be treated as follows:

Double precision - left as  $64\ \mathrm{bit}$  IBM words to be converted on the CYBER

Integers - converted from 32 bit IBM words to 64 bit CYBER integer words

Logicals - converted from 32 bit IBM words to 64 bit CYBER logical words

Alpha-numeric - converted from EBCDIC to ASCII

IF CARD OMITTED: Interface file will be created for the Cyber computer. See IBMG2E option for outputting IBM interface file.

## 2.3.14 DELAY

	-1	-+	-2	-+	-3	-+	-4	-+	-5	+	-6	-+	7	8
DELAY														
	-0								-0		. 0 – – –		0	0

COLUMNS	FORMAT	DESCRIPTION	DEFAULT	VAL	JE &	UNITS
1-6	A6	DELAY - Defines transponder delay curve.				
7-14	18	Station number. (This field required if Satellite ID is missing. This field is ignored if Satellite ID is present.)		0		
15-17	13	Antenna number (optional if frequency is specified)		0		
18-24	17	Satellite ID (required)		0		
25-44	D20.8	Frequency in megahertz (No frequency implies all frequencies)		0.		МН
45-59	D15.3	Transponder delay.		0.		S
60-72	D13.1	Transponder delay rate as a function of rho dot in seconds per meter per second.		0.	S/	(M/S)
73-80	D8.3	Transponder delay variation as a function of rho dot squared in seconds per (meter per seconds) squared.		0.	S/(M	[/S)**2

UNITS: KM =Kilometers; M =Meters ; S =Seconds ; M/S=Meters per second H =Hertz ; MH =Mega-Hertz;

IF CARD OMITTED: No antenna transponder delay will be applied.

# 2.3.15 EARTH

+	1	+2+	-3+4+5+	68
EARTH O	_		3.986004359D+14 6378144.11	
+	0	+0+	-0+-	0+0
COLUMNS	FORMAT	DESCR	IPTION	DEFAULT VALUE & UNITS
1-5	A5	gravitational	ies the earth's potential and/or w earth constants.	
7	I1	Gravity model	initalization indicator.	0
		initializ	onal coefficients are ed from gravity model file. be changed by GCOEF, GCOEFC cards.	
		to zero e	onal coefficients are set xcept those which appear on DEFC, or GCOEFS cards which is card.	
8	I1	Gravity model	print control switch.	0
		1. lowest	o portions of model: t degree & order; and st degree & order.	
			tire gravity model. : : N/A	
9	I1		ly model checksum. : the contents of the this card:	0
			25-72 contain the or GM, ae,1/fe.	
			25-72 contain the or GM, ae, 1/fe.	
COLUMNS	FORMAT	DESCR	IPTION	DEFAULT VALUE & UNITS
15-17	13	used in the grant Minimum = 3.	e of coefficient to be ravitational model. Default depends on gravity unit 12.(See NOTES)	3
18-20	13	used in the grant minimum =2.	of coefficient to be ravitational model. Default depends on file on unit 12.	2
25-44	D20.8	Universal gra	vitational constant	3.9860064D14 M**3/S**2

times the mass of the Earth (GMe) or sigma value if column 9 = 1

- 45-59 D15.3 Semi-major axis of the Earth (Ae) or 6378138.0 M sigma value if column 9 =1
- 60-72 D13.1 Inverse of the Earth's flattening 298.255 (1/fe) or sigma value if column 9 =1

#### NOTES:

- [1] The default values for degree and order of the gravity field and the Earth constants are overridden by the values from the gravity model file on unit 12.
- [2] EARTH and PLANET are actually the same keyword card except that EARTH automatically supplies the planet number (0300) and does not have planetary position adjustment capability.

UNITS: KM =Kilometers; M =Meters ; S =Seconds ; M/S=Meters per second

IF CARD OMITTED: The complete geopotential field contained in the default gravity model file on UNIT 12 will be used.

### 2.3.16 ELEVWT

ELEVWT	_	+2+6+	·	
COLUMNS	FORMAT	DESCRIPTION DEFAULT	VALUE	& UNITS
1-6	A6	ELEVWT - Requests application of elevation dependent down weighting		
11-12	12	Option Indicator [NOTE 1]	0	
		If == 99, this card is used as a header card If.NE.99, this card is used as a data card		
13-14	12	Elevation angle	0	Deg
		<pre>If Option Indicator = 99     Maximum elevation angle at which the     observation weight will be decreased</pre>		
		If Option Indicator NE 99  Elevation angle which to apply down weighting scale factor		
15-17	13	Measurement Type (required on every card)	0	
25-44	D20.8	IF Option Indicator NE 99 Down-weighting scale factor	0.	

#### NOTES:

Note 1 A set of elevation dependent down-weighting may be given for each measurement type. Each set of down-weighting scale factors must begin with a header card. Following the header card, the down-weighting scale factors must be given in descending order at one degree intervals down to zero.

IF CARD OMITTED: Elevation dependent weighting will not be applied.

EXAMPLE: Elevation dependent downweighting for Measurement Type = 40 for elevation angles up to 15 degrees:

	1+	2+-	3	+4-	+	-5+	6-	+	7	-8
ELEVWT	9915040									
ELEVWT	15 40	0.	99500							
ELEVWT	14 40	0.	99000							
ELEVWT	13 40	0.	98500							
ELEVWT	12 40	0.	98000							
ELEVWT	11 40	0.	97500							
ELEVWT	10 40	0.	97000							
ELEVWT	09 40	0.	96500							
ELEVWT	08 40	0.	96000							
ELEVWT	07 40	0.	95500							
ELEVWT	06 40	0.	95000							

ELEVWT	05 40	0.94500
ELEVWT	04 40	0.94000
ELEVWT	03 40	0.93500
ELEVWT	02 40	0.93000
ELEVWT	01 40	0.92500
ELEVWT	00 40	0.92000
1	+2	

#### 2.3.17 EPHEM

---+---1----+----3----+----4----+---5----+---6----+----7----+----8
EPHEM 200
----+---0---+---0---+---0---+---0----+---0

COLUMNS FORMAT

DESCRIPTION

DEFAULT VALUE & UNITS

- 1-6 A6 EPHEM Tells GEODYN which ephemeris is being input on unit #1. This card is required to select the DE118 ephemeris and is optional for the DE96 and DE200 ephemerides.
- 15-17 I3 Ephemeris indicator as follows:

200

- = 96 DE96 ephemeris using mean equator and equinox of 1950 and Woolard's nutations series.
- =118 DE118 ephemeris using mean equator and equinox of 1950 coordinates and Wahr's nutations series.
- =200 DE200 ephemeris using mean equator and equinox of 2000 coordinates and Wahr series nutations. The selection of the DE200 ephemeris also causes different precession and Greenwich Mean Sidereal Time(GMST) calculations.

NOTE: GEODYN is unable to distinguish between the DE118 and DE200 ephemerides without this option card, but it is able to distinguish between the DE96 and the later DE118 and DE200 ephemerides. Thus this card is REQUIRED only when the DE118 ephemeris is selected. For the DE96 and DE200 ephemeris the program will determine which ephemeris is actually supplied on unit #1

IF CARD OMITTED: DE200 ephemeris is assumed input on UNIT 1. If a DE96 ephemeris is input on UNIT 1 and this option card is not present, GEODYN will reject the DE96 ephemeris and function properly.

### 2.3.18 EPHEM2

+1	+2	-3+	-4	-5+	-6+	78
EPHEM2	30					
	+0	-0+	-0+	-0+	-0+	0+0

COLUMNS FORMAT

DESCRIPTION

DEFAULT VALUE & UNITS

- 1-6 A6 EPHEM2 Introduces a supplementary planetary ephemeris for use in GEODYN to determine celestial body positions. The supplemenatry planetary ephemeris might include data from one or more celestial bodies.

  Anyone of the bodies can be used as the center of integration or just for force modeling purposes. [1]
- 8 II Sequence of the celestial body that will be used as center of integration.

  If this field is zero none of the celestial bodies in the supplementary ephemeris will be the center of integration.
- 15-17 I3 Number of celestial bodies in the supplementary planetary ephemeris.
- Note [1]: Use the global option SCBODY for the additional celestial bodies, physical parameter estimation and gravity field specification.

IF CARD OMITTED: GEODYN will not use a supplementary ephemeris.

# 2.3.19 ETIDEN

	1	+2+3+4 +1101 010 -0.033	-+5+ 0.0	6+ 10.	
ETIDEN	0	+0+0			10.
COLUMNS	FORMAT	DESCRIPTION		DEFAULT	VALUE & UNITS
1-6	A6	ETIDEN - Requests application adjustment of earth tides us Colombo tide model and normal Legendre polynomials. Man symmetry is included.	ing the lized		
9	I1	Indicates form of input coef	ficients.	0	
		0 = A and B coefficients.	[NOTE 1]		
10	I1	Indicates form of input sigm	as.	0	
		O = A and B coefficients	[NOTE 1]		
15-17	13	Sign (+1 or -1) of the Demos	Number [NOTE	2] +1	
18-20	13	MKH expansion arguments of t Demos number. [NOTE 2]	he	0	
		M = 0,1,2 K = 0,1,2 H = 0,1,2			
21-22	12	"J" expansion argument of De -9. LE. J .LE. 9 [NOTE 2]	mos number.	0	
23	I1	<pre>Indicates form of tidal ampl coefficent [ V (t) / V ba</pre>		0	
		<pre>0 = only main line, fixed c 1 = main line and sidebands     coefficient 2 = only main line, variabl</pre>	, variable		
COLUMNS	FORMAT	DESCRIPTION		DEFAULT	VALUE & UNITS
24	I1	Disturbing Body (B). [NOTE 2	]	0	
		<pre>0 = Moon and Sun 1 = Moon 2 = Sun</pre>			
25-44	D20.8	"A" coefficient. [NOTE 1]		0	. М
45-59	D15.3	"B" coefficient. [NOTE 1]		0.	М
60-72	D13.1	Sigma "A".		0.	М

NOTES:

[1] The "A" and "B" coefficients are related to amplitude and phase by the following formulas:

> A = K2 Love Number \* cos(phase) B = -K2 Love Number \* sin(phase)

[2] The "Demos Number" is given by:

Demos Number = IDSIGN\*(MKH\*1000+(J+10)\*10+B)

[3] There are 4 recognized main tidal lines and 16 associated sidebands. Their Demos numbers are:

 ${\tt SMKH} \quad {\tt JB} \mbox{ (See last page for S M K J B definitions)}$ 

Main Line 1: 1120 01

-1112 01, 1100 01, 1110 01, -1111 -21, 1101 21, 1111 21, Side Bands:

Main Line 2: 1120 02

Side Bands: 1101 22

1220 01 Main Line 3:

1201 21, 1210 01, 1211 21, 1200 01, -1212 01, -1211 -21, 1221 21 Side Bands:

-1212 01, -1211 -21,

Main Line 4: 1220 02 Side Bands: 1201 22

IF CARD OMITTED: The expanded earth tide model is not used. See TIDES card for tide model that will be used.

# 2.3.20 ETIDES

	1	+2+3+4 +1101 010 -0.033	-+5+ 0.0	-6+ 10.	10.
ETIDES	0	+0+0+0			
COLUMNS	FORMAT	DESCRIPTION		DEFAULT	VALUE & UNITS
1-6	The before IERS fol: Cole	ETIDES - Requests application adjustment of earth tides. The Earth Tide models available GEODYN. First implemented was Colombo tide model and unnorsolved the ETIDES cards involved and the ETIDES cards involved and Earth Tides. Immediately the ETIDES cards involved and ETIDES card	here are in s the malized d K  the ediately of the s page	Colombo	Model
9	I1	Indicates form of input coef	ficients.	C	)
		0 = A and B coefficients.	[NOTE 1]		
10	I1	Indicates form of input sigma	as.	C	1
		O = A and B coefficients	[NOTE 1]		
15-17	13	Sign (+1 or -1) of the Demos	Number [NOTE 2]	+1	
		This sign multiplies the quantum $-((2 - H) * \omega + (2 - 2H + J))$	-		
18-20	13	MKH expansion arguments of the Demos number. [NOTE 2]  M = 0,1,2  K = 0,1,2  H = 0,1,2	he	C	
21-22	12	"J" expansion argument of Der -9. LE. J .LE. 9 [NOTE 2]	mos number.	C	1
23	I1	<pre>Indicates form of tidal ampl coefficient [ V (t) / V bar ]</pre>		C	1
		<pre>0 = only main line, fixed c 1 = main line and sidebands     coefficient 2 = only main line, variable</pre>	, variable		
24	I1	Disturbing Body (B). [NOTE 2]	]	C	)
		0 = Moon and Sun			

1 = Moon2 = Sun

25-44 D20.8 "A" coefficient. [NOTE 1] 0. M

45-59 D15.3 "B" coefficient. [NOTE 1] 0. М

60-72 D13.1 Sigma "A". 0. Μ

D8.2 Sigma "B". 0. 73-80 М

#### NOTES:

[1] The "A" and "B" coefficients are related to amplitude and phase by the following formulas:

> A = K2 Love Number \* cos(phase) B = -K2 Love Number \* sin(phase)

[2] The "Demos Number" is given by:

Demos Number = IDSIGN\*(MKH\*1000+(J+10)\*10+B)

[3] There are 4 recognized main tidal lines and 16 associated sidebands. Their Demos numbers are:

SMKH JB (See last page for S M K H J B definitions)

Main Line 1: 1120 01

1110 01, -1111 -21, 1100 01, Side Bands: -1112 01,

1101 21, 1111 21, 1121 21

Main Line 2: 1120 02 Side Bands: 1101 22

Main Line 3: 1220 01

1201 21, 1210 01, 1211 21, 1200 01, -1212 01, -1211 -21, 1221 21 Side Bands:

-1212 01, -1211 -21,

Main Line 4: 1220 02 Side Bands: 1201 22

#### IERS EARTH TIDE MODEL

---+---1----+----3----+----8 ETIDES ---+---0---+---0 COLUMNS FORMAT DESCRIPTION DEFAULT VALUE & UNITS

- 1-6 ETIDES - Requests application and/or adjustment of IERS convention earth tides .
- 9 Forcing degree in tide operating potential (2 or 3). The order is the first integer in the Doodson number given in columns 18-24.

- Inteder number  $[0, \ldots, 3]$  gives phase increment after  $x\{pi/2\}$ . Useful when all tides use cosine factors with positive amplitudes.
- 15-17 I3 Integer Flag:
  - =0 no sidelines
  - =1 turn n sidelines with input tidal constituent.

=2 turn n sidelines with input tidal group.

- 18-24 I7 Doodson Number (six positive integers)
- 25-44 D20.8 "A" coefficient. 0. M
- 45-59 D15.3 "B" coefficient. 0. M
- 60-72 D13.1 Sigma "A". 0. M
- 73-80 D8.2 Sigma "B". 0. M

Suggested setup for Earth Tides

- $\hbox{ IF CARD OMITTED: The expanded earth tide model is not used. See TIDES card for tide model that will be used. } \\$

### 2.3.21 FANTIM

FANTIM	_	-2+	
COLUMNS	FORMAT	DESCRIPTION DEFAULT VALUE & UNIT	S
1-6	A6	FANTIM - Used for time dependent PHANTOM parameters (option FANTOM)	
7	I1	Index specifying the nature of the introduced global parameters:	
		<pre>= 1 Geometric model global parameters = 2 Force model global parameters</pre>	
25-44	D20.8	Epoch time for a time dependent model YYMMDDHHMMSS. [NOTE 1]	
45-59	D15.3	Start time for a time dependent option. The YYMMDDHHMMSS. end times are specified on the FANTOM cards.	
60-72	D13.1	Space for real information applicable to all the parameters in the group.	
73-80	D8.2	Space for real information applicable to all the parameters in the group	

## NOTES:

- [1] This time will be considered to be a reference time with respect to which linear rates or periodic terms are evaluated.
- [2] Only one FANTIM card per FANTOM parameter group is allowed.

 $\ensuremath{\mathsf{IF}}$  CARD <code>OMITTED:</code> And times are included on the <code>FANTOM</code> cards, the run will abnormally terminate.

### 2.3.22 FANTOM

FANTOM	_	2+3+8
+	0+	0+0+0
COLUMNS	FORMAT	DESCRIPTION DEFAULT VALUE & UNITS
1-6	A 6	FANTOM - Introduces the application and/or estimation of a group of global parameters [NOTE 3]
7	I1	Index specifying the nature of the introduced global parameters:
		<pre>= 1 Geometric model global parameters = 2 Force model global parameters</pre>
8	I1	Index specifying a specific direction for the parameter appearing on cols 25-44 [NOTE 1]
		= 0 All the parameters in the group are treated the same way (same partial formulation)
		<pre>= 1    = 2    To be specified by the user = 3   </pre>
9	I1	Index specifying the nature of the contents on cols $60-72$ and $73-80$
		<pre>= 0 These real fields are not used for the group of parametrs introduced. (Space will not be allocated for them in GEODYN). = 1 These real fields are used as they are described below = 2 These real fields may be used to include any type of real information (but time)</pre>
18-24	17	Satellite ID
25-44	D20.8	Parameter value
45-59	D15.3	Parameter sigma
60-72	D13.1	End time for application for this parameter. YYMMDDHHMMSS. (if the index on col 9 is 1) $[{\tt NOTE}\ 2]$
73-80	D8.2	Available space for extra real information.

### NOTES:

[1] Specify up to three (3) dimensions. Although this option is not required in order to solve for parameters in a 3-D space or solve for parameters of different significance (C and S coefficients for

example) it is helpful for EMATS and other types of labeling. Also parameters with the same index will be grouped together, facilitating that way the formation of partials.

- [2] Requires the presence of the FANTIM option.
- [3] This option provides the necessary allocation and links for parameterizaton. In order to use this option, the user must contact a GEODYN programmer for further additions of modeling and partials code.

# 2.3.23 FLUX

	_	+2+3+4+5+-	68
FLUX 1 FLUX	1	1372.5398	
+	0	+0+	0+0
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUES & UNIT
1-6	A6	FLUX - Adds and/or modifies FLUX (Solar and Magnetic) values in built-in tables.	
7	I1	Table printout control for flux, polar motion, A.1-UTC, and A.1-UT1 data	0
		= 0 Table values for 36 days will be printed.	
		= 1 All Table values used for this run will be printed.	
8	I1	Flux Kp indicator.	0
		= 0 FLUX Kp values will be obtained from columns 73-80 of this card.	
		= 1 FLUX Kp values will be calculated from Ap values.[NOTE 1]	
9	I1	Flux card indicator	0
		= 0 FLUX card used to modify flux values in built-in tables	
		= 1 FLUX card specifies solar flux at 1 AU, in the real field 25-44 [NOTE 2]	
25-44	D20.8	Date of FLUX values (in form YYMMDD). (Col. 9 = 0)	0.
		Solar flux at 1AU (Col. 9 = 1)	-22 1372.5398 10 W/M**2
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
45-59	D15.3	Value of SOLAR FLUX (F10.7 at 1 A.U). If blank, value in table will not be altered. [NOTE 3]	0. 10 W/M**2
60-72	D13.1	Uses input value for average solar flux, if left blank, average will not be altered.	-22 0. 10 W/M**2
73-80	D8.2	Value of magnetic flux Kp. If	

column 8 contains a 1 this input
value is

Kpi

Kp = log[ (sum (e) for i=1,8) / 8]

where Kpi are the three-hourly values of Kp.

#### NOTES:

When dates that are beyond the end of the flux tables are requested the average value of the last month is used. This is true for both solar and magnetic flux.

Solar and magnetic flux override requests are applied to the day specified in columns 25-44 only.

[1] Kp values will be obtained by cubic interpolation of the following table

Кр	Ap	Кр	Ap	Кр	Ap	Кр	Ap
0.	0	2.3333	9	4.6666	39	7.0000	132
0.3333	2	2.6666	12	5.0000	48	7.3333	154
0.6666	3	3.0000	15	5.3333	56	7.6666	179
1.0000	4	3.3333	18	5.6666	67	8.0000	207
1.3333	5	3.6666	22	6.0000	80	8.3333	236
1.6666	6	4.0000	27	6.3333	94	8.6666	300

REFERENCE: "Geomagnetic Indices," by Gordon Rostoker, Reviews of Geophysics and Space Physics, Vol. 10, No. 4, November 1972.

[2] This card can be used with or without the first flux card (i.e. col 9 = 0) If column 9 does not equal one, field one WILL NOT represent the solar flux at 1 AU. When col 9=1 this card CAN ONLY be used to specify the solar flux at 1 AU. A second card must be used to specify additional options.

The solar flux at 1 AU is divided by the speed of light to compute the solar radiation pressure on the satellite. Therefore, when specifying this value, one must insure that the speed of light is consistent.

- [3] The solar flux values input to GEODYN are the 1 A.U. values. GEODYN will scale the solar flux values to the actual earth-sun distance. The values printed out by GEODYN IIS are the scaled values.
- [4] Both FLUX Cards are accepted in the same run.

UNITS: KM =Kilometers; M =Meters ; S =Seconds ; M/S=Meters per second UNITS: DEG=Degrees ; RAD=Radians ; AS =Arc seconds; MAS=Milli-arc seconds UNITS: H =Hertz ; MH =Mega-Hertz; W =Watts

IF FIRST CARD OMITTED: Values from tables file on UNIT 2 will be used. IF SECOND CARD OMITTED: Default value of solar flux at 1 AU will be used.

# 2.3.24 FREEZE

FREEZE	_	+5+ +0+0+	
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A6	FREEZE: Allows the user to control observation editing and arc parameter contribution during various iteration	
7-8	12	Arc iteration after which the editing of observations will freeze.	999
9-10	12	Global iteration for which the option on columns 7-8 will apply	1
11-12	I2	Arc iteration before which the arc parameters will not contribute to the normal matrix. This option is usefulfor dynamic crossover processing where a number of inner iterations is needed for the best crossover location extimate	0

# 2.3.25 G2COFC

G2COFC	2+3+4+5 1	
COLUMNS FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6 A6 G	2COFC - C coefficients of the second asteroid situation.	asteroid in a binary
25-44 D20.8 V	alue of the C coefficient	
60-72 D13.1 S	igma of the C coefficient	
G2COFC 00	C (and G2COFS cards: 1000 1001	
	200014369265D-01	.001
G2COFC 00	2001 2002 .47033751D-02 1001	.001
	2001 2002	.001 .001

# 2.3.26 G2COFS

G2COFS 1	2+3+4+5 . 1	
COLUMNS FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6 A6 G2	COFS - S coefficients of the second asteroid situation.	asteroid in a binary
25-44 D20.8 Va	alue of the S coefficient	
60-72 D13.1 Si	gma of the S coefficient	
-	c (and G2COFS cards: .000	
G2COFC 002		.001
G2COFC 002 G2COFC 002 G2COFS 001	.47033751D-02	.001
G2COFS 002 G2COFS 002		.001

## 2.3.27 G2BOUT

1	1 +	2	3	4	5+6	5+7	7+8
G2BOUT							
(	)+	0	0(	)+(	0(	) + (	)+0

COLUMNS FORMAT DESCRIPTION

DEFAULT VALUES & UNIT

1-6 A6  ${\tt G2BOUT}$  - Requests the output of a  ${\tt GEODYN}$  II observation file on unit 20. This option is useful for GPS runs, in combination with the ELCUTOFF and the SATCUT input options. In these cases, GEODYN will output only observations which are not edited by the  ${\tt ELCUTOF}$  and the SATCUT specifications.

> Another useful application is the combination G2BOUT and the TRPOUT option.

### 2.3.28 GCMMOD

GCMMOD		+2+6+-	
COLUMNS			VALUE & UNITS
1-6	A6	GCMMOD - Requests application of the periodic Earth center of mass correction model.  The model is used to modify the position of tracking stations. Multiple periodic terms may be used. [NOTE 1]	
8	12	Option Indicator	0
		For Option Indicator = 1	
25-44 45-59	D20.8 D20.8	Refrence Year Period	0. 0. yr
		For Option Indicator = 2	
25-44 45-59	D20.8 D20.8	Amplitude of X component of correction Phase of X component of correction	0. mm 0. deg
		For Option Indicator = 3	
25-44 45-59	D20.8 D20.8	Amplitude of Y component of correction Phase of Y component of correction	O. mm O. deg
		For Option Indicator = 4	
25-44 45-59	D20.8 D20.8	Amplitude of Z component of correction Phase of Z component of correction	0. mm 0. deg
NOTEG			

#### NOTES:

1) Multiple periodic terms may be used to model the center of mass. The group of 4 GCMMOD cards must be included for each term. All terms must use the same reference year.

Each component of the model is calculated as follows:

signal = amplitude \* cos( (time - reference year)\*(2pi/period) - Phase )

IF CARD OMITTED: Earth center of mass model will not be applied

 ${\tt EXAMPLE:} \ \, {\tt The \ example \ below \ gives \ a \ annual \ periodic \ term}$ 

+	2+3+	4+	68
GCMMOD 1	1950.0	1.0	
GCMMOD 2	2.7	41.0	
GCMMOD 3	2.8	321.0	
GCMMOD 4	5.5	27.0	
+1+	2+3+	4+	68

# 2.3.29 GCOEF

GCOEF 2		+2+3+4+5+ 2 1 +0+0+0+	1.000000D-091.00D-09
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-5	A5	GCOEF - Modifies and/or requests the estimation of coefficients in the geopotential model.	
7	I1	Normalization indicator.	0
		= 0 values unnormalized C,S [See NOTES]	
		= 1 values normalized C,S	
		= 2 values ignored C,S	
15-17	13	Degree of C and S coefficients (N index). [NOTE 1]	0
18-20	13	Order of C and S coefficients (M index). [NOTE 1]	0
25-44	D20.8	A priori (or starting) value of C coefficient. [NOTE 2]	0.
45-59	D15.3	A priori (or starting) value of S coefficient. [NOTE 2]	0.
60-72	D13.1	Standard deviation or fractional uncertainty of C coefficient. Coefficient will not be adjusted if this field is zero.	0.
		If the value is less than or equal to 1.0D-5, it is used as the standard deviation	n.
		If the value is greater than 1.0D-5 (or (SQRT(1.0D-5) for J2) it is used as the fractional uncertainty, and the standard deviation is computed as the fractional uncertainty times the nominal value of the coefficient. [NOTE 1]	
73-80	D8.2	Standard deviation or fractional uncertainty of S coefficient. Coefficient will not be adjusted if this field is zero.	у О.
		If the value is less than or equal to 1.0D-5, it is used as the standard deviation	n
		If the value is greater than 1.0D-5 it is used as the fractional uncertainty, and the standard deviation is computed as the fractional uncertainty times the nominal value.	

NOTES:

[1] The denormalization subprogram in GEODYN underflows at degree 49 and order 48.

The geopotential coefficient denormalization equation used is:

$$C_n^m = \bar{C}_n^m \times D_n^m$$
$$S_n^m = \bar{S}_n^m \times D_n^m$$

where

$$D_n^m = \sqrt{(4n+2)\frac{(n-m)!}{(n+m)!}} \text{ for } m > 0$$
$$D_n^m = \sqrt{2n+1} \text{ for } m = 0$$

and

 $C_n^m$  ,  $S_n^m$  are un-normalized coefficients.  $\bar{C}_n^m$  ,  $\bar{S}_n^m$  are normalized coefficients.

- [2] If the degree and/or order on the GCOEF card is greater than the DEFAULT values [in the gravity file on unit 12], then the highest degree and/or order that is specified on the GCOEF card must be input on the EARTH card.
- [3] If it is desired that the starting values of geopotential coefficients be different than the a priori values for those coefficients then the GCOEF card should be used as follows:
  - o GCOEF cards requesting the desired coefficient adjustments should be included in the setup deck. The a priori values will be indicated on these cards.
  - For each coefficient pair for which the starting value is to be different than the a priori value, an additional GCOEF card should be in the setup deck. This GCOEF card will indicate the starting value and come later in the deck than the corresponding GCOEF card requesting adjustment.
- [4] Nominal values of geopotential coefficients are

$$J_n^m = \frac{10^{-5} D_m^n}{n^2}$$

IF CARD OMITTED: The geopotential used will be from the gravity model input on UNIT 12 of GEODYN-IIS unless modified by EARTH, GCOEFC, or GCOEFS cards.

# 2.3.30 GCOEFC

GCOEFC1	(	+2+3+4+5+ 00200048416600D-03 +0+0+0+	.00000001
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUES & UNIT
1-6	A 6	GCOEFC - Modifies and/or requests the estimation of C coefficients in the geopotential model.	
7	I1	Normalization indicator.[NOTE 1] = 0 values unnormalized = 1 values normalized = 2 values ignored	0
15-17	13	Degree of C coefficient (n index).	0.
18-20	13	Order of C coefficient (m index).	0.
25-44	D20.8	A priori (or starting) value of ${\tt C}$ coefficient.	0.
60-72	D13.1	Standard deviation or fractional uncertainty of C coefficient. Coefficient will not be adjusted if this field is zero.	0.
		If the value is less than or equal to 1.0D-5, it is used as the standard deviation	n.
		If the value is greater than $1.0D-5$ (or (SQRT( $1.0D-5$ ) for J2) it is used as the fractional uncertainty, and the standard deviation is computed as the fractional uncertainty times the nominal value of the coefficient. [NOTE 1]	

## NOTES:

[1] See GCOEF card for discussion of denormalization and nominal values.

IF CARD OMITTED: The geopotential used will be from the gravity model input on UNIT 12 of GEODYN-IIS unless modified by EARTH, GCOEF, or GCOEFS cards.

# 2.3.31 GCOEFS

GCOEFS		+2+3+4+5+6 +0+0+0		
COLUMNS	FORMAT	DESCRIPTION	DEFAULT	VALUE & UNITS
1-6	A 6	GCOEFS - Modifies and/or requests the estimation of S coefficients in the geopotential model.		
7	I1	Normalization indicator. [NOTE 1] = 0 values unnormalized = 1 values normalized = 2 values ignored		0
15-17	13	Degree of S coefficient (n index).		0
18-20	13	Order of S coefficient (m index).		0
25-44	D20.8	A priori (or starting) value of S coefficient.		0.
60-72	D13.1	Standard deviation or fractional uncertainty of S coefficient. Coefficient will not be adjusted if this field is zero.		0.
		If the value is less than or equal to 1.0D-5, it is used as the standard deviation		
		If the value is greater than 1.0D-5 it is used as the fractional uncertainty, and the standard deviation is computed as the fractional uncertainty times the nominal value of the coefficient. [NOTE 1]	1 <b>e</b>	

### NOTES:

[1] See GCOEF card for discussion of denormalization and nominal values

IF CARD OMITTED: The geopotential used will be from the gravity model input on UNIT 12 of GEODYN-IIS unless modified by EARTH, GCOEF, or GCOEFC cards.

### 2.3.32 GCOFCT

+	1+	2+3+4+5+	6+7+8
GCOFCT			
	0	0+0+	0+0
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A6	GCOFCT - Requests application or adjustment of the C gravity coefficient for the time period defined below.	
25 - 44	D20 0	Pogin time of time period gravity	

25-44 D20.8 Begin time of time period gravity

45-59 D15.3 End time of time period gravity

NOTE [1]: Time period gravity is introduced by groups of input cards where one group contains information about one discrete period of time. Therefore one needs as many groups as the number of discrete times where gravity coefficients are being replaced or adjusted. These groups follow the stream of the original gravity field coefficients and they have the following structure:

GCOFCT or GCOFST including beginning and end of time period, GCOEFC or GCOEFS respectively (as many as desired) with information about replacing value or adjustment and, GTPEND denoting the end of information for this time period.

### EXAMPLE:

GCOFCT			800801000000.0080080106	0000.00
GCOEFC1	2	0	500000000000D-03	1.0
GCOEFC1	3	1	0.300000000000D-05	1.0
GCOEFC1	4	1	600000000000D-06	1.0
GTPEND				
GCOFCT			800801070000.0080080109	0000.00
GCOEFC1	2	0	500000000000D-03	0.0
GCOEFC1	3	1	0.300000000000D-05	0.0
GCOEFC1	4	1	600000000000D-06	1.0
GTPEND				
GCOFST			800801000000.0080080106	0000.00
GCOEFS1	3	1	0.2500000000000D-06	0.0
GCOEFS1	4	1	550000000000D-06	1.0

IF CARD OMITTED: Time period gravity will not be applied.

# 2.3.33 GCOFST

GCOFST	+2+3+4+5	
COLUMNS FORM	AT DESCRIPTION	DEFAULT VALUE & UNITS
1-6 A6	GCOFST - Requests application or adjustment of the S gravity coef for the time period defined below	fficient
25-44 D20.	Begin time of time period gravit	у
45-59 D15.	End time of time period gravity	

NOTE [1]: See note on GCOFCT card.

IF CARD OMITTED: Time period gravity will not be applied.

# 2.3.34 GDYNEP

GDYNEP	+5+	
COLUMNS FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6 A6	GDYNEP - specifies a new Geodyn Internal Reference Time for this run.	
25-44 D20.8	New Geodyn Internal Reference Time in "YYMMDD". [NOTE 1]	410101 (JD = 2430000.5)

NOTE [1]: If YY = [51-99] the year is 19YY If YY = [00-50] the year is 20YY

IF CARD OMITTED: Geodyn will use its default reference time

#### 2.3.35 **GEOPOL**

---+---1----+----3----+----3----+----5----+----6----+----7----+----8
GEOPOL
----+---0---+---0----+---0----+---0

COLUMNS FORMAT

DESCRIPTION

DEFAULT VALUE & UNITS

0.

1-6

A6

GEOPOL - The GEOPOL option changes the Z axis used by GEODYN for computing latitude and longitude when evaluating spherical harmonic functions for gravity.

By default (without the GEOPOL card) the Z axis used by GEODYN for these computations is the Earth's rotation pole. However, when the GEOPOL option is selected, the Z axis used by GEODYN for these computations is the mean pole implied by polar motion series input on unit 2.

Note that in either case (GEOPOL selected or not), the Z axis used by GEODYN for gravity computations will not be the same as the Earth's figure axis. This can only be compensated by the use degree 2 order 1 terms in the gravity field which vary according to polar motion. The POLDYN and POLKF option cards can be used to model can be used to model these degree 2 order 1 terms.

Most users select the GEOPOL option if the the GEOPOL option is selected, then it is especially important to make use of the POLDYN and POLKF cards. That is because the Z axis implied by the GEOPOL option is further away from the figure axis than the default Z axis.

Also note that when the GEOPOL and POLDYN options are used, the user should place background degree 2 order 1 terms in the gravity field (see POLDYN card).

7 II If GEOPOL is used with the POLKF card,
Kf is normally +1/3. So if the GEOPOL
card is present, the program only allows
positive Kf, unless a "1" is input in
col 7 of the GEOPOL card.
If = 1, can have Kf < 0.
If not = 1, can only have Kf > 0.

IF CARD OMITTED: See above first paragraph.

### 2.3.36 GLBARC

+1	+2	-+3+	4	-5	6+8
GLBARC					
+0	+0	-+0+	0+	-0	0+0

### COLUMNS FORMAT DESCRIPTION

DEFAULT VALUE & UNITS

1-6 A6 GLBARC - Requests user selected arc parameter partials to be summed into the E-MATRIX at the end of the arc section.

This allows the specified arc parameters to be treated as the first global paramters in the E-MATRIX. At present only the drag parameters can be selected.[1]

7 I1 Selects drag parameters

- = 0 drag parameters will be selected
- = 1 turns off this option for the drag parameters
- 8-15 Additional parameters to be added in the future.
- NOTE [1]: This option only applies to E-MATRIX runs.

  The parameter is given a new E-MATRIX number. See

  Volume 5 for a description of the new E-MATRIX labels.

  labels.
- IF CARD OMITTED: This option will not be enabled For interplanetary runs this option will be enabled.

### 2.3.37 GMSUN

# 2.3.38 GPSMOD

GPSMOD	_		_				-							
COLUMNS	FORI	.MAT		DE	SCRIPT	CION					DEFAULT	VALUE	&	UNITS
1-6	A 6	(	PSMOI			GPSMOI								
7	I1		the j	arc i job w	n EMAT nversi ill te		un. Th e defa	e ind ult (	ex mu 0) me	st be			0	
8	I1	a con inputype an M proving sate hori	ombina it MTV 41 t MBIAS vide a ellite	piase ation (PE=8) for i for a bia sat to-ho than	s are of me 5 or a nput N the 4- s for ellite rizon one co	provide easurement of the combination of the combin	ed on ent ty nation 7. In measueg (st one w there ation.	the s pe 41 of m stead remen ation ay bi fore	etup and easur of p t typ -sate ases they	as 43 for ement rovid; e, we llite descr: may a	ing or ibe		0	

#### 2.3.39 GPSSHD

+	 3+	-4	-5	6+	-8
GPSSHD					
+	 0+	-0	-0+	0+	-0

COLUMNS FORMAT

DESCRIPTION

DEFAULT VALUE & UNITS

1-6 A6 GPSSHD - This option will instruct
GEODYN to produce yaw polynomial input for
GPS runs if this is not available. The
computation are being done in GEODYN
according to the Journal of Geodesy Paper
by Bar-Sever.

The GPSSHD option will turn any GPS data reduction run into an ORBGEN run automatically, regardless of what other cards are present.

Yaw polynomials will be computed and output on unit 48 of GEODYN IIE, in the form of YAWPOL group cards. These YAWPOL group cards may be then included in the global part of the setup for data reduction with the appropriate yaw modeling. NOTE [1]

25-44 D20.5 12 hrs + endarc time

YYMMDDHHMMSS

- 45-59 D15.3 Cutoff angle that helps save some computation deg 20 NOTE [2]
- NOTE [1]: The end of the GPSSHD group is denoted by the presence of the SHDEND card. In between the GPSSHD and the SHDEND option cards we enter the required YAWRAT cards with yaw rate and yaw rate rate information.
- NOTE [2]: GPS satellites are not completely shadowed, unless the beta angle of the orbit plane is about 150 (in absolute value) or less. In a GPSSHD run, GEODYN will compute the beta angle for each GPS satellite at EPOCH. If the absolute of the beta angle is greater than the cutoff (200 or the input on columns 45-59), for a particular GPS satellite at epoch, then that satellite will be excluded from all further consideration in the GPSSHD run.

# 2.3.40 GRIDS

# 2.3.40.1 GRIDS

---+---1----+----3----+----4----+----5----+----6----+----7----+----8
GRIDS
----+---0---+---0---+---0---+---0----+---0

COLUMNS FORMAT

DESCRIPTION

DEFAULT VALUE & UNITS

1-6 A 6 GRIDS - Introduces the use of grid elevation input files in IIE. These grid data files may be mean sea surface heights or land elevations from a given surface or origin. The names and the data for these grid files are provided on the following option cards. The GRIDS option ends with the ENDGRD option card. No more than 6 different grids are allowed as input. The grid files should be binary files consisting of variable length records. Each record contains REAL\*4 numbers representing the specified height, for all the grid points of the same latitude in an eastward direction. The first record begins at the highest latitude, so the first point in the file is located at the NW corner of the grid.

IF CARD OMITTED: yet GFILE cards are present, the IIS run will abnormally terminate

# 2.3.40.2 GFILE

+ GFILE	_	+2+3+4+5+ 2-80 0360 225.00 225.00	-68
		+0+0+	-0+0
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A6	GFILE - Supplies the name and input data for the grid file.	
7	I1	Index for grid data origin:	
8	I1	<ul> <li>=0 Elevations are given from the reference ellipsoid.</li> <li>=1 Elevations are given from the mean sea surface</li> <li>=2 Radii are given from the center of the planet</li> <li>Indicator of the data location on the grid file:</li> <li>= 0 The data are located on the grid intersection</li> <li>= 1 The data are located at the center of the grid.</li> </ul>	0
9-12	14	Number of data points in one record	no default
15-17	13	Grid size	(arcsec) No default
25-44	D20.8	Maximum latitude of the grid	(deg) 90
45-59	D15.3	Minimum latitude of thr grid	(deg) -90
60-72	D13.1	Maximum longitude of the grid counting from 0-360 degrees.	(deg) 360
22-24	D8.3	Minimum longitude of the grid	

IF CARD OMITTED: This model will not be used to compute the observations.

UNITS: M = Meters ; S = Seconds ; deg = Degrees

# 2.3.40.3 ENDGRD

+1	+2	+3	+4	-+	6+-	7+8
ENDGRD						
+0	+0	+	+	-++	0+-	00

COLUMNS FORMAT DESCRIPTION DEFAULT VALUE & UNITS

1-6 A6 ENDGRD - Denotes the end of the grid file sub-group.

# 2.3.41 GRVEPO

+1+2	-+3+8
GRVEPO	890101000000.
	-+0+0

COLUMNS FORMAT DESCRIPTION DEFAULT VALUE & UNITS

- 1-6 A6 GRVEPO Sets the epoch time for the time dependent gravity option. See option GRVTIM for more details.
- 25-44 D20.8 Epoch time in YYMMDDHHMMSS. for the time 0. dependent gravity model. Linear rates and periodic terms in the time dependent gravity model will be calculated from this time.

 ${\tt NOTES: See \ option \ GRVTIM \ for \ the \ time \ dependent \ gravity \ formulation.}$ 

UNITS: KM =Kilometers; M =Meters ; S =Seconds ; M/S=Meters per second MAS=Milliarc seconds

IF CARD OMITTED: Epoch time for time dependent gravity will be zero.

# 2.3.42 GRVTIM

GRVTIM1		+2+3+4+5+6+78 -2.5D-11
+	0	+0+0+0
COLUMNS	FORMAT	DESCRIPTION DEFAULT VALUE & UNITS
1-6	A6	GRVTIM - Requests application and/or adjustment of time dependent gravity coefficients. The geopotential coefficients may be given a linear rate and/or up to nine periodic terms. The epoch time for this option is set with the GRVEPO option.
7	I1	C or S geopotential coefficient designator.
		<pre>=1 C coefficient indicator. =2 S coefficient indicator.</pre>
8	I1	Defines the content of the fields in columns 25-80 as follows:
		Columns         Columns         Columns         Columns           25 - 44         45 - 59         60 - 72         73 - 80
		=1 Linear Rate Rate sigma [NOTE 2] =2 A B Period =3 A sigma B sigma
15-17	13	Degree of C and S coefficients (n index) 0
18-20	13	Order of C and S coefficients (m index) 0
21-22	12	# of period for this coefficient 1 (a number between 1-9)
25-44	D20.8	This field contains geopotential linear rate per year or rate sigma information as defined by the ues selected in columns 7 & 8.  col 8=1 Cdot or Sdot (linear rate).
COLUMNS	FORMAT	DESCRIPTION DEFAULT VALUE & UNITS
45-59	D15.3	This field contains geopotential rate sigma, amplitude (A), or amplitude (A) sigma information as defined by the values selected in columns 7 & 8.
		col 8=1 Standard deviation for Cdot or Sdot. A non zero sigma indicates

Standard deviation for amplitude (A). A non zero sigma indicates

col 8=2 Amplitude (A) of cosine term.

col 8=3

that this parameter will be adjusted.

that this parameter will be adjusted.

60-72 D13.1 This field contains geopotential amplitude (B), or amplitude (B) sigma information as defined by the values selected in columns  $7\ \&\ 8$ .

ι « ο.

col 8=2

Amplitude (B) of sine term.

col 8=3 Standard deviation for amplitude
(B). A non zero sigma indicates
that this parameter will be adjusted.

73-80 D8.3 This field contains the period for the periodic option. The frequency will be computed as frequency=2\*pi/period.

years

NOTE 1: See next page for the time dependent gravity formulation.

The cards specifying the apriori periodic terms MUST precede the cards which carry the accompanying sigma information, as in the following example. In the following example the coefficient C21 has two periodic terms.

GRVTIM12	2	0	0.0	0.0	1.0
GRVTIM13	2	0	1.0D-07	1.D-07	
GRVTIM12	2	1 1	0.0	0.0	1.0
GRVTIM13	2	1 1	1.0D-07	1.D-07	
GRVTIM12	2	1 2	0.0	0.0	1.0
GRVTIM13	2	1 2	1.0D-07	1.D-07	
GRVTIM12	2	2	0.0	0.0	1.0
GRVTIM13	2	2	1.0D-07	1.D-07	
GRVTIM22	2	2	0.0	0.0	1.0
GRVTIM23	2	2	1.0D-07	1.D-07	

NOTE 2: Coefficient rates should be normalized in case of application of time dependent gravity.

IF CARD OMITTED: Time dependent gravity will not be applied.

The formulation for the time dependent gravity implementation follows:

# 1#1 + Cdot(n,m)\*(t-t0) + 2#2\*(t-t0)) + 3#3\*(t-t0)) 4#4 + Sdot(n,m)\*(t-t0) + 5#5\*(t-t0)) + 6#6 \*(t-t0))

#### 7#7

- Normalized geopotential  ${\tt C}$  and  ${\tt S}$  coefficients.

Cdot, Sdot - Linear rates for the geopotential coefficients.

t0 - Start time for application of time dependent gravity. (See GRVEPO option for setting the start time.)

t - Current time. Time at which time dependent gravity will be computed.

Aci ,Asi - Amplitude of the cosine term for the C or S coefficient for application of a periodic gravity effect.

Bic ,Bsi - Amplitude of the sine term for the C or Scoefficient for application of a periodic gravity effect.

Omegai(n,m) - Frequency for the periodic gravity effect for the (n,m) coefficients. (Where Frequency=2\*pi/period)

8#8

# 2.3.43 GTPEND

GTPEND	_	2+5+	-	
COLUMNS	FORMAT	DESCRIPTION	DEFAULT	VALUE & UNITS
1-6	A6	GTPEND - Requests end of application of time period gravity		

IF CARD OMITTED: Abnormal termination will occur.

# 2.3.44 GXPAND

1-	+2	-+3+	4	56	+7	+8
GXPAND	02 36 01				1.0D-06	1.0D-06
0-	+0	-+0+	0	0+0	+0	+0

COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A 6	GXPAND - Requests estimation of a range of geopotential coefficients.	
11-14	14	Starting degree for geopotential estimation.	2
15-17	13	Ending degree for geopotential estimation. (Default = degree of geopotential specified on EARTH card.)	0
18-20	13	Starting order for geopotential estimation.	0
21-24	14	Ending order for geopotential estimation. (Default = degree of geopotential being generated.)	0
60-72	D13.1	Standard deviation of C coefficient. [NOTE 1]	0.
73-80	D8.2	Standard deviation of S coefficient. [NOTE 1]	0.

#### NOTES:

The a priori values of the coefficients will be obtained from the current geopotential model including all modifications to the model prior to this card.

[1] See GCOEF card for explanation of the standard deviation for the C and S coefficients.

# 2.3.45 H2LOVE

H2L0VE	-	+2+3+4+5+ .600 +0+0+0+	
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A6	H2LOVE - Requests that solid earth tide coefficients of the second kind be modified and/or adjusted.	
7	I1	<pre>If &gt; 0, print solid tide correction to station height. [NOTE 1]</pre>	
25-44	D20.8	A priori value of love number of the second kind to account for radial displacement, H2.	.600
45-59	D15.3	A priori sigma of H2. If greater than 0, requests adjustment of H2.	0.

NOTE [1]: Solid tide induced changes in station h

unit 98.

IF CARD OMITTED: Default value (H2 = 0.600) will be used.

# 2.3.46 I32G2E

+ I32G2E	1	+2+3+4+5+-	68
	0	+0+	0+0
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A 6	I32G2E - Specifies options for the interface and data file output by GEODYN IIS on unit 11 and 41 for a 32 bit integer machine.	
8	I1	Specifies whether binary or formatted interface and data files will be created by IIS.	0
		=0 Binary interface and data files will by created with the following attributes:	
		Information in the interface file will be treated as follows:	
		Floating point - 64-bit IBM word.	64-bit IBM word.
		Integers - IBM 32-bit word.	null 32-bit IBM word and 32-bit IBM word.
		Logicals - IBM 32-bit word.	null 32-bit IBM word and 32-bit IBM word.
		Alpha-numeric - EBCDIC characters.	EBCDIC characters.
		=1 formatted interface and data file wil be created.	
10	I1	Specifies 80 column or 132 column formatted interface and data files.	0
		=0 132 column formatted interface and data files will be created.	
		=8 80 column formatted interface and data files wil be created.	
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS

11-17 17 Core memory requested for IIS in MW. 2.4 MW

#### 2.3.47 I64G2E

---+---1---+---2---+---3----+---4----+---5----+---6----+---7----+---8 I64G2E ---+---0---+---0---+---0----+---0 COLUMNS FORMAT DESCRIPTION DEFAULT VALUE & UNITS 1-6 A6 I64G2E - Specifies options for the interface and data file output by GEODYN IIS on unit 11 and 41 for a 64 bit integer machine. Ι1 8 Specifies whether binary or formatted 0 interface and data files will be created by IIS. =0 Binary interface and data files will be created with the following attributes: Information in the interface file will be treated as follows: Double precision - left as 64 bit IBM words to be converted on the CYBER Integers - converted from 32 bit IBM words to 64 bit CYBER integer words Logicals - converted from 32 bit IBM words to 64 bit CYBER logical words

Alpha-numeric - converted from EBCDIC to ASCII

=1 formatted interface and data file will be created.

#### COLUMNS FORMAT DESCRIPTION

DEFAULT VALUE & UNITS

- 10 I1 Specifies 80 column or 132 column formatted interface and data files.
  - =0 132 column formatted interface and and data files will be created.
  - =8 80 column formatted interface and data files will be created.
- 11-17 17 Core memory requested for IIS in MW.

IF CARD OMITTED: Binary Interface and data files will be created for a

 $64\ \text{bit}$  integer machine. See I32G2E option for outputting an interface and data file for a 32 bit integer machine.

# 2.3.48 IAU200

+	2	+3	+4+-	5	6+	7+8
IAU200						
+		++	+0+-	0	0+	0+0

COLUMNS FORMAT

DESCRIPTION

DEFAULT VALUE & UNITS

1-6 A6

IAU200 - If this card is present GEODYN will use the new IAU-2000 standards for precession and nutation. New precession modules will be called in GEODYN IIE and the Chebyshev polymomial coefficients to produce the nutations DPSI and DEPS will be read from a special planetary ephemeris. The presence of the IAU200 option requires that presence of the new planetary ephemeris on input unit 1 of GEODYN IIS. and vice versa. If only one of the above conditions does not exist, GEODYN will terminate abnormally.

IF CARD OMITTED: GEODYN will aplly the old precession and nutation standards.

# 2.3.49 INTGCB

+1	-+2	-+3	+4+-	5+	-6+	-7+8
INTGCB						
	-+0	-+0	+0+-	0+	-0+	-0+0

# COLUMNS FORMAT DESCRIPTION

DEFAULT VALUE & UNITS

- 1-6 A6 INTGCB- In interplanetary mode this option allows the dual orbit determination of a satellite orbiting a planetary body along with the planetary body.
- 7 II If I=1, the orientation of the planetary (orbited) body will be computed from a force model starting from the initial conditions supplied on the PLANOR cards.
- IF CARD OMITTED: The orbit of the central planetary body can not be refined. In this case The trajectory of the central (orbited) planetary body must be supplied on either the standard planetary ephemeris or on the supplemental ephemeris. Also, without this card the orientation of the central body can not be modeled dynamically

# NOTES:

- 1. When this option is used the REFSYS card must request the Earth Equator and Equinox J2000 coordinate system for the reference.
- 2. When this option is used the elements of the planetary body must be supplied before (first) the elements of the artificial satellites. The elements of the planetary body are heliocentric J2000. The elements of the artificial satellites are planet centered J2000.
- 3. Just below is an example of the PLANET card section of the global setup when the INTGCB option is invoked. It is for a case when the orbit of Mercury is being estimated along with an artificial satellite. Note that the standard Mercury planet card has a VERY small GM value. Also note that the last PLANET card is the one being used for Mercury and that this card should also appear in the Gravity file (Unit 12).

INTGCB		
PLANET	10100	2.20317821893582D-13 2440000.0 10000000.000
PLANET	0200	3.24858598826460D+14 6051900.0 10000000.000
PLANET	0300	.39860044150000D+15 .637813630D+07 .2982564D+03
PLANET	0301	.49028004760000D+130.173800000D+070.4608295D+04
PLANET	0400	4.28283697739390D+13 3397000.0 154.409
PLANET	0500	1.26712767857796D+17 71492000.0 15.414
PLANET	0600	3.79406260611373D+16 60268000.0 10.208
PLANET	0700	5.79454900707187D+15 25559000.0 43.616
PLANET	0800	6.83653406387926D+15 24764000.0 58.543
PLANET	0900	9.81600887707005D+11 1151000.0 10000000.000
PLANET	9999	1.32712440017987D+20 696000000.0 10000000.000

4. Starting with version 1507.12, the numerically integrated ephemeris of the central planetary body that is produced when the INTGCB card is present can be replaced with the XEPHEM option. This allows for the use of dynamic orientation while still using an input trajectory for the orbited central planetary body. When this is done, the gravitational torques computed for the dynamic orientation will use the positions from the external ephemeris.

# 2.3.50 IBMG2E

+1	+2+	3+	4	5+	-6+	-78
IBMG2E						
+0	+0+	0+	0+	0+	-0+	-0+0

COLUMNS FORMAT DESCRIPTION

DEFAULT VALUE & UNITS

1-6 A6 IBMG2E - Specifies that the interface file output by GEODYN IIS on unit 11 will be for a GEODYN-IIE run on an IBM compatible computer.

> Information in the interface file will be treated as follows:

Floating point - 64-bit IBM word. 64-bit IBM word.

- IBM 32-bit word. null 32-bit IBM word Integers and 32-bit IBM word.

Logicals - IBM 32-bit word. null 32-bit IBM word and 32-bit IBM word.

Alpha-numeric - EBCDIC characters. EBCDIC characters.

IF CARD OMITTED: Interface file will be created for the Cyber computer.

# 2.3.51 INTCOM

+1 INTCOM	+2+3+4+5+6+7+8 890601. 890701.
+0	+0+0
COLUMNS FORM	AT DESCRIPTION DEFAULT VALUE & UNITS
1-6 A6	INTCOM: Turns on the intercomparison capability. This option allows the user to perform certain specific orbit generation runs. In these runs some assumptions are made, or certain perturbations only are turned on in order to study the output orbit and/or compare it with orbits generated by similar software packages.  The output orbit is writen on unit59 at a rate depending on the users request on the ORBTVU card and expressed in inertial True-of-Date Cartesian elements. [NOTE 2] The assumptions and/or supressions for each individual case are described under Case Indicator on columns 9-10.
9-10 I2	Case Indicator: Integer number which indicates a specific case. See NOTE 1 for description of the individual cases.
25-44 D20.8	Start time for 3-hourly kp values storage Epoch start in the form YYMMDD. (Applies on Case 15) time
45-59 D15.8	Stop time for 3-hourly kp values storage in Epoch stop the form YYMMDD. (Applies on Case 15) time
NOTE [1]: 1	Two body orbit generation (point masses) - Output mean of reference system - Supressed precession and nutation - No other forces applied. (CASE1A of the TOPEX intercomparison)
2	Two body - Output True-of-Date - Only precession and nutation applied No other forces applied.(CASE1B of the TOPEX intercomparison)
3	Two body - Output Pseudo body-fixed - Fixed true pole - Tests Greenwich hour angle and UT1 - Modify POLE cards accordingly.

4 Two body - Output body fixed - UT1=UTC - Tests polar motion - corrections will be applied to UT1-UTC - Modify POLE cards accordingly. (CASE1D of the TOPEX intercoimparison).

(CASE1C of the TOPEX intercomparison)

- 5 Two body Output full body-fixed Tests polar motion and UT1 combined. (CASE1E of the TOPEX intercomparison).
- 6 Case 2 application of J2 (CASE2 of the TOPEX intercomparison).

- 7 Case 2 + Point mass effects for other celestial bodies Modify PLANET cards accordingly. (CASE3A of the TOPEX intercomparison).
- 8 Case 7 + Indirect effect of the Earth's J2. (CASE3B of the TOPEX intercomparison).
- 9 Case 6 + Polar motion and UT1 (CASE4 of the TOPEX intercomparison).
- 10 Case 6 + Complete gravity model GEM-T1.
  Tests Gravity Model. (CASE5 of the TOPEX intercomparison).
- 11 Case 6 + Empirical along track accelerations Include necessary ACCEL cards. (CASE6 of the TOPEX intercomparison).
- 12 Case 6 + Solar radiation Use SOLRAD card. (CASE7 of the TOPEX intercomparison).
- 13 Case 10 + Simple solid Earth tide model Use correct input on the TIDES cards (CASE8 of the TOPEX intercomparison).
- 14 Case 10 + Complete Earth and Ocean tides. (CASE9 of the TOPEX intercomparison).
- 15 Case 6 + Drag application using DTM atmospheric model Constant fluxes are used FLUX and ATMDEN cards are necessary. Three hourly Kp values are used Include the right BIH tables on UNITO2. (CASE10 of the TOPEX intercomparison).
- 16 All perturbations in Cases 2-15 applied. (CASE11 of the TOPEX intercomparison).
- 17 Case 6 + Earth radiation Use ALBEDO card AND SOLRAD card. (CASE12 of the TOPEX intercomparison).
- 18 Case 6 + General relativity Use REL300 card. (CASE13 of the TOPEX intercomparison).

# NOTE [2]: OUTPUT FORMAT

#### A. HEADER RECORD

ITEM	FORMAT	DESCRIPTION
1-132	A 1	Description of the run. Contents of the second COMMON TITLE card.

B. EPHEMERIS RECORDS (Two records for each point)

#### First record

ITEM	FORMAT	DESCRIPTION
1	D22.16	Julian Date (integral part) (UTC)
2	D22.16	Fractional Julian Date (UTC)
3	D22.16	Greenwich Hour Angle

4-6	3D22.16	Satellite X Y Z elements (meters)
Second	record	
1-3	3D22.16	Satellite Xdot Ydot Zdot elements (meters/sec)
4	D22.16	Satellite geodetic latitude (instantaneous z axis) (degrees).
5	D22.16	Satellite East longitude (degrees).
6	D22.16	Satellite height above the reference ellipsoid (meters).

NOTE [3]: Sample UNITO5 setups for the orbit generation options exist in ZCDEP.TOPEX.UNITO5.DATA. The GLOBAL TITLE cards describe the individual cases.

IF CARD IS OMITTED: No Intercomparison will take place.

# 2.3.52 INVERT

+	-	+2+3+4+5+6+7+8
	0	+0+0+0
COLUMNS	FORMAT	DESCRIPTION DEFAULT VALUE & UNITS
1-6	A6	INVERT - Controls various matrix inversion routine options.
7	I1	= 0 Subroutine DSINV will be used 0 to invert normal matrices.
		= 1 Subroutine SYMINV will be used to invert normal matrices.
8	I1	= 0 Inverted normal matrices will not be 0 printed out.
		= 1 Inverted normal matrices will be printed out.
9	I1	= 0 Subroutine DSINV or SYMINV will be used 0 to invert NORMAL POINT pass by pass normal matrices with a priori information as supplied from NORMPV card.
		= 1 A generalized inverse scheme will be used to invert NORMAL POINT pass by pass normal matrices. A priori information supplied from the NORMPV card will be ignored.
10	I1	<pre>= 1 The SRIF (square root information filter) (NOTE 1)     option is invoked = 0 Default Inversion</pre>
NOTE 1:	accu: Squa: by g	A reliable and effective method for computing numerically rate estimates in problems where high precision is required. The root formulations increase numerical computation accuracy paranteeing positive-definiteness of the associated covariances by decreasing the condition numbers of the normal matrices.

#### 2.3.53 **IONCOR**

IONCOR		+2+3+8 +0+0+0+0
COLUMNS	FORMAT	DESCRIPTION DEFAULT VALUE & UNITS
1-6	A 6	IONCOR - Requests the application of the ionospheric correction based on the ionospheric model IR-95. [Ref. 1]
7	I1	<pre>Indicates the type of coefficients to be used in the model. [NOTE 1]</pre>
8	I1	<pre>Indicates the type of interpolation to be used for electron density profiles. 0 = 0 linear interpolation = 1 logarithmic interpolation (not tested yet)</pre>
NOTE [1]	URSI	<ul> <li>model recommended by the International Radio Consultative Committee (1967) (Ref. 1 p. 52).</li> <li>Model proposed by the International Union of Radio Science (1989) (Ref. 1 p. 52).</li> <li>sets of coefficient files must be included in the</li> </ul>

NOTE [2]: Frequencies used in the computation of the correction are obtained in one of two ways:

the year. The filenames are in the following format: and ursi\*.asc where \* is equal to the month plus ten.

in /users/geodyn/SUPPORT/dat\_ion95.

Option 1 - Frequencies provided. Nominal center or average (over forward and return) frequencies per link are included on columns 25-44 on the OFFSET input cards. In this case an OFFSET card must be included for each satellite in the run. Also frequencies must be input on all the OFFSET cards or option 2 will be invoked.

directory of the IIS execution. Files are located on geodesy2

There are 24 files, 12 files of each type, one for each month of

- Option 2 Frequencies computed in GEODYN, invoked only when one or more OFFSET cards do not provide frequencies, or OFFSET cards are not included in the setup.
- NOTE [3]: The earth orientation and flux tables used with the IONCOR option are special tables which include ionospheric indices. The analyst must use these tables with GEODYN IIS. However at this point TDF has not been modified to interpret the new tables format. With TDF runs the analyst should use the default GEODYN tables.
  - [Ref. 1] Dieter Bilitza, "International Reference Ionosphere 1990.", Nov. 1990. NSSDC/WDC-A-R&S 90-92.

# 2.3.54 J2SUN

J2SUN	_	2+3+4+5+6+7+ 2.11060885327268D-07 1.D-02 0+0+0+0+	
		DESCRIPTION DEFAULT VALUE &	
1-6	A6	2SUN - Introduces J2 of the SUN as a parameter	
25-44	D20.8	2 of the sun value	
45-59	D15.3	igma for J2	

# 2.3.55 L2LOVE

	1	+2+3+4+5+	-68
L2L0VE		.075	
	0	+0+0+0+	-0+0
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A6	L2LOVE - Requests that solid earth tide coefficients of the third kind be modified and/or adjusted.	
7	I1	<pre>If &gt; 0, print solid tide correction to station height. [NOTE 1]</pre>	
25-44	D20.8	A priori value of love number of the third kind to account for horizontal shearing. (L2)	.075
45-69	D15.3	A priori sigma of L2 . If greater than zero, requests adjustment of L2 .	0.

NOTE [1]: Solid tide induced changes in station height printed on unit 98.

IF CARD OMITTED: Default value (L2 = 0.075) will be used

# 2.3.56 LINK

LINK	0	+0+	-0+-	0+0
COLUMNS	FORMAT	DESCRIPTION	DEFAULT	VALUE & UNITS
1-6	A6	LINK - This option links two GEODYN parameters together. During the formation of the normal matrix, the partials of the second parameter will be folded into the partials of the first parameter. The		

---+---1----+----3----+----8

8-12 I5 ID number of the first adjusted parameter  $[\mathtt{NOTE1}]$ 

parameters.

13-17 I5 ID number of the second adjusted parameter [NOTE1]

NOTE 1: The adjusted parameter  ${\tt ID}$  number can be found in the  ${\tt IIS}$  parameter summary.

computed correction will adjust both

IF CARD OMITTED: No parameter linking will occur.

# 2.3.57 LIST

+1	+2	2	-3+	4	5	6+7	78
LIST							
0	+(	)+	-0+	0	0	0	)0

COLUMNS FORMAT DESCRIPTION

DEFAULT VALUE & UNITS

1-4 A4 LIST - If NOLIST option has been exercised earlier in Global Set options, then  ${\tt LIST}$ option will negate effects of NOLIST and cause GEODYN-IIS to resume listing of input from this point forward.

IF CARD OMITTED: Listings of input will terminate with 'NOLIST' or 'ENDALL' whichever is encountered first.

# 2.3.58 LNSTHR

	+3+	4+	68
LNSTHR			
	+	0+	00

COLUMNS FORMAT

DESCRIPTION

DEFAULT VALUE & UNITS

- 1-6 A6 LNSTHR is used to modify the model for the relativistic Lense-Thirring effect and/or solve for related parameters.
  - 1. The Lense-Thirring model introduce a paramter which multiplies the acceleration due to the Lense-Thirring effect in the form (1 + ).

$$\vec{A} = \alpha \frac{GM(1+KR)}{R^3} e^{-KR} \vec{R}$$

where:

 $\alpha$  - is the parameter

 ${\it G}$  - the gravitational constant

 ${\cal M}$  - the mass of the earth

K - 1/ $\lambda$ ;  $\lambda$  is the range of the force

 $ec{R}$  - is the position vector of the satellite

- 7 I1 1 data for mu  $(\mu)$  2 data for alpha  $(\alpha)$
- 18-24 I7 Satellite Id.
- 25-44 D20.8 Parameter value 0.D0
- 45-59 D15.3 Parameter sigma 0.D0
- 73-80 D8.2 If col7 = 2 this specifies the range of the force

IF CARD OMITTED: The above models will not apply to the force model.

#### 2.3.59 LOCALG

1	1 +	2	3+	4 + ,	5	6+7	7+8
LOCALG							
(	)+	0	0	0	0	0(	)+0

COLUMNS FORMAT

DESCRIPTION

DEFAULT VALUE & UNITS

- 1-6 A6 LOCALG Requests appliation and possibly estimation of gravity anomalies. Gravity anomalies [1] are regional (local) departures from the gravity field implied by the input set of spherical harmonics.

  The location of the gravity anomaly is specified on a second card [2].
- 7-14 I8 Gravity Anomaly number. This is a unique number specified by the user to identify the gravity anomaly parameter. No two gravity anomaly parameters may use the same gravity anomaly number. Information about the area of the gravity anomaly parameter is input on a grid [3]. The gravity anomaly number helps link up information from the grid to the gravity anomaly parameter.
- 18-24 I7 Grid Number [3]. Information about the gravity anomaly area is given on an external grid.

  The grid number identifies the grid which is being used. All gravity anomaly parameters in the same run must use the same grid.kk
- 25-44 D20.8 A priori value of the gravity anomaly at the center of the master block. [2]

mgals

- 73-80 D8.3 Standard deviation of gravity anomaly. A value > 0 results in estimation of gravity anomalies.
- Note [1] The parameters are free air anomalies on the surface of a sphere with radius (a\*a\*b)\*\*(1/3) where a is the semi major axis and b is the semi minor axis of the central body.
- Note [2] A card which is blank in COLs 1-6 must follow each LOCALG card. This second card contains information in COLs 7-66 about the location of the gravity anomaly.
- Note [3] When LOCALG parameters are present, IIS searches for a data set called SUBBLK. SUBBLK contains points which are assigned to the LOCALG parameters. The points represent subdivision areas of the LOCALG parameters and are used to numerically integrate the equations which upward continue the gravity anomaly to the satellite location.

SUBBLK is a formatted data set as follows:

- 7-14 I8 Gravity Anomaly number. This is a unique number for LOCALG parameters. At least one (preferably more than one) grid point will match each LOCALG parameter's gravity anomaly number.
- 18-24 I7 Grid Number. This is a number which uniquely identified the grid (all points on the grid have the same grid number.
- 25-49 D25.16 Geocentric latitude of grid point (degrees)
- 50-74 D25.16 Longitude of grid point (degrees)
- 75-99 D25.16 Area on unit sphere represented by the grid point

IF CARD OMITTED: Gravity anomalies will not be applied.

# SECOND CARD

-----

+	1+	-267	+8
+	0+	-0+0+0+0	+0
COLUMNS	FORMAT	DESCRIPTION DEFAULT VALUE	& UNITS
1-6	A6	Contains information about master block and its subdivisions.	
7-21	D15.2	Latitude of the center of the master block.	deg
22-36	D15.2	East longitude at the center of the master block.	deg
37-51	D15.2	Size of the master block in the direction of latitude.	deg
52-66	D15.2	Size of the master block in the direction of longitude.	deg
IF CARD	OMITTED:	And LOCALG was requested the program will terminate abnormally.	

# 2.3.60 LRARC

+1	+2	34	+5	+6	+8
LRARC 17 5 5	5				
					+0

COLUMNS	FORMAT	DESCRIPTION DEFAULT	VALUE &	UNITS
1-6	A 6	Requests input of laser retro-reflector range corrections on UNIT 8 [NOTE 1] (Currently, only applies to TOPEX)		
7-8	12	Number of incidence angles for which information is available on UNIT 8	17	
9-10	12	Maximum Order of Fourier Coefficients for X-coordinates	5	
11-12	12	Maximum Order of Fourier Coefficients for Y-coordinates	5	
13-14	12	Number of Station Detector Types in Range Correction Tables	14	
23-24	12	=1 This is the SARAL satellite and an LRA correct will be applied.	ion	

NOTE 1: This corrects the observation from the reflecting corner cube back to a fixed point on the spacecraft. It is also station dependent and is tied to the laser tracking station detector type specified on the INSTRMNT card. This must be used in conjunction with the OFFSET and CGMASS cards to get the complete range correction from the corner cube to the spacecraft CG.

Currently, this option can only be used for the TOPEX satellite.  $% \left\{ 1,2,\ldots,4\right\}$ 

IF CARD OMITTED: No lra range corrections will be read or used.

# 2.3.61 MLTDEM

COLUMNS FORMAT

+	2+	3+	4	5+	6+	78
ENDCON						
+0+	0+	0+	0+	0+	0+	0+0

DEFAULT VALUE & UNITS

1-6 I6 ENDCON: End of configuration

DESCRIPTION

Will be followed by an ENDSIM card if this is the last configuration. Will be followed by a SIMSAT card

if another configuration is requested.

 $\hbox{ IF CARD OMITTED: If simulated data generation (not replacement ) is invoked and this card is missing the run will } \\$ 

terminate.

#### 2.3.62 **NETCON**

---+---1---+---2---+---3----+---4----+---5----+---6----+---7----+---8 1.331 0.896 NETCON 4.789 ---+---0---+---0 COLUMNS FORMAT DESCRIPTION DEFAULT VALUE & UNITS 1-6 A6 NETCON - Introduces values for the right hand side equations and sigmas for network rotation and/or translation constraints 7 I1 = 1 The values represent station network translation constraints = 2 The values represent station network rotation constraints = 3 The values represent quasar network rotation constraints 8 = 1 The values represent RHS values = 2 The values represent sigmas 25-44 D20.8 RHS x compopnent value OR sigma D15.3 RHS y compopnent value OR sigma 45-59 60-72 D13.1 RHS z compopnent value OR sigma NOTE [1] : UNITS: KM =Kilometers; M =Meters ; S =Seconds ; M/S=Meters per second H = Hertz ; MH = Mega-Hertz; IF CARD OMITTED: No network translations or rotations wil be invoked

# 2.3.63 NOLIST

+	2	3+	4	5+	6+-	78
NOLIST						
+	0+	0	0+	0+	0+-	0+0

COLUMNS FORMAT DESCRIPTION

DEFAULT VALUE & UNITS

1-6 A6 NOLIST - Will cause GEODYN-IIS listings of input to be interrupted from point of NOLIST until next LIST option encountered.

IF CARD OMITTED: Listings will terminate with 'ENDALL'.

# 2.3.64 NONSNS

+1	+	-+3+	+4	56	5+8
NONSNS					
+0	+0	-+	+0+	0+0	)+0

COLUMNS FORMAT

# DESCRIPTION

DEFAULT VALUE & UNITS

1-6 A6

NONSNS - Shuts off the "No Shirt No Service" (NSNS) protection in a GPS data reduction run. Unless this card is present, NSNS protection is provided in a GPS data reduction run. When NSNS protection is in force, every block of GPS data that is not covered by an ambiguity bias (simple measurement bias) is edited.

IF CARD OMITTED: NSNS protection is provided.

# 2.3.65 NORMPT

NORMPTO		+2+3+4+5+ 120 +0+0+0+0+	
,	U		
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A6	Requests the computation and output onto unit 20 in GEODYN-IIE of metric data normal points on the last inner iteration of each data reduction arc during the last global iteration	S
7	I1	Indicates whether or not a normal point report should be printed on unit 15 by GEODYN-IIE.  O - Print a normal point report on unit 15.	rt 0
8	I1	1 - Do not print a normal point report.  Maximum number of local iterations to perform in editing each pass of data prior to formation of normal points.	m 5
9-10	12	Maximum number of physical buffers to be used in outputting normal points to unit 20.	d 2
11-14	14	Controls the maximum number of observations that will be used in one data block (minimum no. = 11, maximum = 9999, default value = 40 when EMATRX option is used)	1000
15-17	13	Number of integration steps over which interpolation may be simultaneously performed. Must be greater than or equal to integration order minus one.	Integration order minus one.
18-20	13	Maximum number of physical buffers per input observation block. This number must be great than or equal to the value used by the Tracking Data Formatter.	10 er
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
21-24	14	Normal point bin size in seconds. If a positive integer is not entered in this field normal points will not be generated.	None d,
25-44	D20.8	Normal point pass editing multiplier. (See Arc EDIT option for default)	Residual edit multipler.
45-59	D15.3	Normal point pass editing initial RMS. (See Arc EDIT option for default)	Editing RMS from previous Inner Iter.
73-80	D8.2	Normal point pass editing iteration convergence criterion. (See Arc EDIT)	Inner Iteration RMS convergence.

# NOTES:

[1] When in normal point generation mode, large numbers of parameters should not be estimated. This option also overrides columns 11-17 of the VECOPT card. Observation data to be input for normal point generation should be processed through the Tracking Data Formatter with the normal point option exercised in TDF.

Values in columns 11-14 and 15-17 of this card override values in these same columns on VECOPT option.

See also Arc Set option card NORMPV.

IF CARD OMITTED: Normal points will not be generated.

## 2.3.66 NUTATE

	1	-+2+3+45+6	-+8
NUTATE			
	0	-+0+0+0+0+	-+0+0
COLUMNS	FORMAT	DESCRIPTION DEFAU	JLT VALUE & UNITS
1-6	A6	NUTATE - Requests the application of nutation corrections that refer to the IAU-2000 model and/or used to modify and/or request adjustment of corrections to nutation components DPSI (nutation in longitude) and EPST (nutation in true obliquity).	
7	I1	<ul> <li>No parameter adjustment only nutation corrections since J2000 are applied. [NOTE 1]</li> <li>Parameter adjustment requested with no nutation corrections applied.</li> <li>Parameter adjustment requested and nutation corrections since J2000 are applied.</li> </ul>	
25-44	D20.8	Date in form YYMMDD.FFF of DPSI and EPST I correction values[1] to which this card applies. This date is for the midpoint of the interval specified in columns 25-44 of card two.	No default.
45-59	D15.3	DPSI (nutation in longitude) correction. See	e col. 8. (SA)
60-72	D13.1	EPST (nutation in true obliquity) correction. See	e col. 8. (SA)
[NOTE	200	se the NUTATE card with the new EOP IAU 000 tables to apply nutation corrections ince year 2000.	
SECOND (	CARD		
+	1	-+2+3+4556	-+8
		5.0 0.01 0.1	0.1
COLUMNS	FORMAT	T DESCRIPTION DEFAU	JLT VALUE & UNITS
1-23	23 X	Blanks must be present.	
24	I1	=1 the units for the below averaging interval is =0 the units for the below averaging interval is	
25-44	D20.8	Averaging period[1] in the above units for	No default.

DPSI and EPST values. The date in columns 25-44 of the first card is for the midpoint of this interval

45-59	D15.3 A	priori standard	deviation in	DPSI.	0.0	(SA)
60-72	D13.1 A	priori standard	deviation in	EPST.	0.0	(SA)
IF CARD	OMITTED:	The nutation co	rrection will	not be applied.		

### 2.3.67 OBSVU

---+---1---+---2---+---3---+---4----+---5---+---6---+---7---+----8
0BSVU 3 0010
----+---0---+---0---+---0---+---0

### COLUMNS FORMAT DESCRIPTION

DEFAULT VALUE & UNITS

0

- 1-5 A5 OBSVU Controls residual printout for all arcs.
- 7 I1 Residual printout controlled as follows:
  - or blank Indicates that
    residuals are requested on the
    first inner iteration of the first
    global iteration and the last inner
    iteration of the last global

iteration for all arcs. This is the same as =3 below.

- =1 Indicates that residuals are requested on the first inner iteration of the first global interaction for all arcs.
- =2 Indicates that residuals are requested on the last inner iteration of the last global iteration for all arcs.
- =3 Indicates that residuals are requested on the first inner iteration of the first global iteration and the last inner iteration of the last global iteration for all arcs.
- =4 Indicates residuals are requested on all iterations for all arcs.
- =5 Indicates no residuals are requested for any arc.

COLUMNS FORMAT DESCRIPTION

DEFAULT VALUE & UNITS

0

8 II = 1 Indicates preprocessing
corrections are to be written on
unit 16 for the iterations indicated
by column 7 and for the modulo
selected in columns 21-24.
Residuals must be printed on the first
iteration if the option is invoked.

### [see NOTE 1 below]

21-24 I4 Modulo number for printing of residuals, preprocessing corrections, and partials. Default is to print all residuals. A value N input here requests printing of every Nth residual.

NOTE[1]:

The numbering system that is used on the preprocessing printout on UNIT16 is related to the Header Record and partition number as described in Volume 5 Section 2 under Observation Corrections Record.

[eg. if the UNIT16 printout was for a range measurement and the number printed was 13 it would mean Header #1 partition #3 which would pertain to a dry tropospheric refraction correction for station #1. G means that the program computes and applies the correction. Absence of a "G" means the correction was provided by the data record.]

IF CARD OMITTED: Residuals are requested on the first inner iteration of the first global iteration and the last inner iteration of the last global iteration for all arcs. No preprocessing nor measurement partial files are created and all residuals

## 2.3.68 OFFADJ

OFFADJ		+3+5+	
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A6	OFFADJ - Provides for the adjust- ment of tracking point offset coordinates on a satellite. This card must be used in conjuntion with a OFFSET card. The OFFSET card describes the tracking point coordinate parameters. Columns 7-44 of the OFFADJcard must match columns 7-44 of the OFFSET card which corresponds to the tracking point coordinates for which the adjustment is desired.	
45-59	D15.3	Standard Deviation of X coordinate	(meter)
60-72	D13.1	Standard Deviation of Y coordinate	(meter)
73-80	D8.2	Standard Deviation of Z coordinate	(meter)

## 2.3.69 OFFSET

OFFSET		+2+3+4- 7806401 0. +0+0+0-	1.331	0.896	4.789
COLUMNS					VALUE & UNITS
1-6	A 6	OFFSET - Inputs body-cente coordinates for antenna of or laser reflector offset.			
7	I1	= 1 Indicates link 1 of of A second link (OFFSET2 = 2 Indicates link 2 of of [NOTE 1], [NOTE 2] = 3 Indicates link 3 of of [NOTE 1], [NOTE 2] = 7 Antenna orientation ve in the x direction = 8 Antenna orientation ve in the y direction = 9 Antenna orientation ve in the z direction [NOTE 6], [NOTE 7]	) must follow. fset.  fset.  ctor		
8	I1	= 9 Indicates external att used for link # 1 rota link # 2 and # 3 movab rotations. Must have e file available.  If this option is sele column 17 must be pres #'s 1-4 are valid.  This option overrides system specified in co Not applicable for OFF OFFSET1 card value is OFFSET3.  [NOTE 2]	tions and le antenna xternal attitude  cted an antenna # ent, only antenna  the coordinate lumn 15. SET2 or OFFSET3 ca	ard.	
9	I1	= 1 Indicates use of new m prepro word antenna id If this option is sele column 17 must be pres #'s 1-4 are valid. Not applicable for OFF OFFSET1 card value is OFFSET3.	enfication scheme cted an antenna # ent, only antenna	in	)
10	I1	[NOTE 3] = 1 Indicates use of OFFSE = 0 No OFFSET3 shall be		(	)

COLUMNS FORMAT DESCRIPTION

DEFAULT VALUE & UNITS

11-12 I2 Antenna type number used for phase map corrections

0

- O Position and velocity vectors
- 1 Position and sun vectors
- 2 TOPEX/Poseidon
- 3 SPOT-2
- 4 GPS
- 5 ERS-1
- 6 Mars observer Mapping Phase
- 7 Mars observer Cruise Phase
- 8 TDRSS
- 9 Magnetically stabilized S/C
- 10 GFO
- 11 TRMM
- 12 EUVE
- 13 ICESAT
- 14 ENVISAT
- 15 CRYOSAT-2
- 16 Leading GRAIL satellite
- 17 Trailing GRAIL satellite
- 18 HY2A
- 19 SARAL
- 20 SENTINEL

Not applicable for OFFSET2 or OFFSET3 card. OFFSET1 card value is used for OFFSET2.

16-17 I2 Antenna number (optional if frequency is specified). Mandatory if external attitude or new metric data antenna identification scheme is selected. This antenna number must match the appropriate external attitude file antenna number

Not applicable for OFFSET2 or OFFSET3 card. OFFSET1 card value is used for OFFSET2 and OFFSET3.

OR

50+ Laser Id if Laser Altimeter data are being processed. If the laser Altimeter Id is 1 then the user should put 51 on columns 16-17. This convention distinguishes from the antenna number and the antenna identification scheme in the GEODYN code.

### COLUMNS FORMAT DESCRIPTION DEFAULT VALUE & UNITS

- 18-24 I7 Satellite ID (required).

  Not applicable for OFFSET2 card.

  OFFSET1 card value is used for OFFSET2.
- 25-44 D20.8 Frequency in mega-hertz. Used to 0. MH discriminate between antennae when more than one antenna is on a satellite.

( Frequency = 0. implies that offset
applies to all antenna.)
Not applicable for OFFSET2 or OFFSET3 card.
OFFSET1 card value is used for OFFSET2.
(optional if antenna number and
new prepro ID scheme selected).

If the index in column 7 is equal to 7 then this field contains the first component of the orientation unit vector in the x direction

If the index in column 7 is equal to 8 then this field contains the first component of the orientation unit vector in the y direction

If the index in column 7 is equal to 9 then this field contains the first component of the orientation unit vector in the z direction

45-59 D15.3 Body centered fixed offset link 1 or link 2 or link 3 antenna from center of figure; x-axis.

O. M

If the index in column 7 is equal to 7 then this field contains the second component of the orientation unit vector in the  $\mathbf{x}$  direction

If the index in column 7 is equal to 8 then this field contains the second component of the orientation unit vector in the y direction

If the index in column 7 is equal to 9 then this field contains the second component of the orientation unit vector in the z direction

60-72 D13.1 Body centered fixed offset link 1 or link 2 or link 3 antenna from center of figure; y-axis.

O. M

If the index in column 7 is equal to 7 then this field contains the third component of the orientation unit vector in the x direction  $\mathbf{x}$ 

If the index in column 7 is equal to 8 then this field contains the third component of the orientation unit vector in the y direction

If the index in column 7 is equal to 9 then this field contains the third component of the orientation unit vector in the z direction

73-80 D8.3 Body centered fixed offset link 1 or link 2 or link 3 antenna from center of figure; z-axis.

O. M

OR Antenna Cutoff angle if the index on col. 7

is "9"

- NOTE [1]: OFFSET1 and OFFSET2 cards are a pair. OFFSET1 card must be followed by OFFSET2 card. If present, OFFSET3 must follow OFFSET2.
- NOTE [2]: See Volume 5 for a description of the external attitude file.

  Multiple link offset may only be used with Mars observer

  (Col 15 = 6) and external attitude. Using multiple links implies link # 1 is Spacecraft Body Fixed (SBF) Frame, and link # 2 is a movable link. Link 1 and CGMASS vector are in the same SBF frame Link # 2 gets rotaton from external attitude or internal attitude model. If q1 of quaternions equals -9999999 for any point in the block, then the internal model (selected in col. 15) will be used for the offset vector rotations. If no SBF to J2000 quaternions are found for the antenna number requested then the internal model (selected in col. 15) will be used for the offset vector rotations. If no movable antenna quaternions are present in the external attitude file only the 1st link will be applied with the SBF to J2000 quaternion information.
- NOTE [3]: See the antenna identification word in the metric data format description.
- NOTE [4]: The two vectors listed for each group are the initial vectors used to describe the satellite body fixed coordinate system.

  (options 0 and 1 only)

Position and velocity vectors - The X-axis is given by the velocity vector, the X-axis crossed with the position vector gives the Y-axis, and the X-axis crossed with the Y-axis gives the Z-axis.

Position and sun vectors - The Z-axis is given by the negative of the position vector, the Z-axis crossed with the sun vector gives the Y-axis, and the Y-axis crossed with the Z-axis gives the X-axis.

- NOTE [5]: See CGMASS for reference between center of figure/center of gravity.
- NOTE [6] : The default values for the antenna orientation vectors for GPS satellites
  - x = (1,0,0)
  - y = (0,1,0)
  - z = (0,0,1)

for user satellites

- x = (1,0,0)
- y = (0, -1, 0)
- z = (0,0,-1)
- NOTE [7]: If the index in column 7 is 7, 8 or 9 then the user needs to fill the satellite ID (cols 18-24) and the first 3 real fields with the antenna orientation in all 3 directions.

```
UNITS: KM = Kilometers; M = Meters ; S = Seconds ; M/S = Meters per second H = Hertz ; MH = Mega - Hertz;
```

 $\label{eq:card_omitted} \text{IF CARD OMITTED:} \quad \text{No antenna offset will be applied.}$ 

#### 2.3.70 OLOAD

The OLOAD option may be used for two different parameter estimation options in  ${\tt GEODYN}$ .

- 1. Application and Estimation of Ocean Loading
- 1. Application and Estimation of Center of Mass (COM) corrections and Earth Orientation and UT1 (EOP/UT1) corrections caused by high frequency tides.

The description below refers to Ocean Loading application and estimation. (For Tidal driven COM corrections and EOP/UT1 corrections please scroll down)

We attach here a reference by Richard Ray on Background to GEODYN Modeling of Diurnal/Semi

wc acca	ch hore	a reference by Michard May on Background to	GLODIN	noucii	ng or bru
OLOAD	3 7051	+0+0+0+0+0 27355500378 +.00378 +0+0+0+0	10.	·	10.
COLUMNS	-	DESCRIPTION	-	-	& UNITS
1-6	A6	OLOAD - Requests application and/or adjustment of ocean loading using the formulation described in NOTES [4, 5 and 6]. Also see NOTE [1]	ent		
7	I1	Print Options (See NOTE [3] below)			
8	I1	Direction specification. 1 - East, 2 - North, 3 - Vertical		3	
11-14	14	Site Number (See NOTE [8] below)			
18-24	17	Doodson number. (see NOTE [7] below)		0	
25-44	D20.8	"A" coefficient.	(	0.0	M
45-59	D15.3	"B" coefficient.	(	0.0	M
60-72	D13.1	Sigma "A".	(	0.0	
73-80	D8.2	Sigma "B".	(	0.0	

#### NOTES:

[1] In the second half of 2017, R Ray and D Rowlands scrubbed most aspects of tide modeling in GEODYN (solid Earth tide displacements for stations were not re-examined, although dynamic Earth tides, ETIDES were). The re-examination resulted in some code changes that were implemented in version 1802.0. Using the same OLOAD setup decks as in previous versions will result in slightly different ocean loading displacements due to a cleanup of sideband computations. Xp and YP computations have changed quite a bit (see note [10]. One should not attempt to model Xp and Yp with OLOADS prior to version 1802.0. Furthermore, some input changes are recommended on this web page.

- [2] NOTES [3 & 4] pertains to ocean loading corrections only. NOTES [5-8] pertain to ocean loading, center of mass variations and Earth orientation. NOTES [9-11] pertain to center of mass and Earth orientation
- [3] Ocean loading station lat, lon, height changes printed on unit 96.
- [4] GEODYN uses the following formula to compute the change in a component (East, North or Vertical) of position due due to the loading of the ith tide (a tide is specified by a Doodson number):

$$\label{eq:delta} \begin{split} & \text{delta}(\textbf{i},\textbf{t}) = \texttt{A}(\textbf{i}) \ * \ \text{cos}(\text{arg}(\textbf{i},\textbf{t})) \ + \ \texttt{B}(\textbf{i}) \ * \ \text{sin}(\text{arg}(\textbf{i},\textbf{t})) \end{split}$$
 The argument, arg(i,t), is mainly (see notes [5i and 6]) a linear combination of certain astronomical angles at time, t. The factors factors of the linear combination are dictated by the input Doodson number.

[5] In order to be consistent with the tidal community, for certain Doodson numbers, arg(i,t) is offset by an amount given in the table that follows just below NOTE that for the dynamic Earth and ocean tides (ETIDES and OTIDES) the angle by which the Doodson number is offset is defaulted to zero for all tides. It is up to the user specify a multiple of PI/2 to offset the dynamic tides.

```
Doodson OFFFSET TIDE
165555 PI/TWO ! K1
145555 -PI/TWO ! 01
163555 -PI/TWO ! P1
135655 -PI/TWO ! Q1
055565 PI
             ! Node tide
117655 -PI/TWO ! alpha1
125755 -PI/TWO ! 2Q1
127555 -PI/TWO ! sigma1
137455 -PI/TWO ! rho1
147555 PI/TWO ! tau1
153655 PI/TWO ! beta1
155455 PI/TWO ! M1 couplet line
155655 PI/TWO ! M1
157455 PI/TWO ! chi1
162556 -PI/TWO ! pi1
164554 PI/TWO ! S1 gravitational
164555 PI! S1 met line, not gravitational
164556 PI/TWO ! S1 gravitational
166554 PI/TWO ! psi1
167555 PI/TWO ! phi1
173655 PI/TWO ! theta1
175455 PI/TWO ! J1
183555 PI/TWO ! SO1
185555 PI/TWO ! 001
195455 PI/TWO ! upsilon1
253755 PI ! Gamma2
254556 PI ! alpha2
263655 PI ! lambda2
265455 PI ! L2
```

274554 PI ! R2 355555 PI ! M3

- [6] Starting with version 1802.0, alterations due to sidebands are automatically computed for every OLOAD input (whether for actual ocean loading or for tidally driven center of mass or EOP changes). Prior to version 1802.0, sideband computation was performed according to what was being sone with the same Doodson number in the dynamic tides. There were multiple problems with this approach, Sideband modeling slightly modify arg(i,t) (described above) and also scale the A and B coefficients. NOTE that sideband computations are also possible for the parameters associated with OTIDES and ETIDES cards, but for these the user must request these sideband computations, The OLOAD parameters use sideband option #1 as described on the OTIDES card.
- [7] Any number specified in columns 18-23 specified as a Doodson number having a leading digit greater than 6 will be interpreted as a retrograde tide having the same Doodson number but with 6 less in the leading digit. For example, 855555 is a retrograde M2 tide (255555).
- [8] In addition to describing parameters that model station displacements due to ocean loading, the OLOAD card can also describe parameters for the Earth's center of mass variations and also variations in Earth orientation parameters (Xp, Yp and UT1). Site number 1 is reserved for (indicates that the card pertains to) center of mass variations. Site number 2 is reserved for (indicates that the card pertains to) Earth orientation. For use of the OLOAD card for center of mass or Earth orientation, scroll down

UNITS: KM = Kilometers; M = Meters ; S = Seconds ; Kg = Kilograms
DEG = Degrees ; RAD = Radians ; AS = Arc seconds; MAS = Milli - arc seconds

IF CARD OMITTED: Ocean loading will not be used.

COLUMNS FORMAT DESCRIPTION DEFAULT VALUE & UNITS

- 1-6 A6 OLOAD Requests application and/or adjustment of Center of Mass OR Earth Orientation (EOP) and UT1 corrections driven by High Frequency Tides (See NOTE [1] and NOTE [9])
- 8 I1 Type of parameter

For column 14=1 1=X direction 2=Y direction 3=Z direction

For column 14=2

1=Xp and Yp (Polar Motion) 3=UT1

- 14 I1 Index defining the Group of Parameters
  - 1 Center of Mass correction (See NOTE[10])
    2 EOP/UT1 correction (See NOTE[11])

18-24	17	Doodson number. (See NOTES 5 and 6 above)	O COM/EOP/UT1
25-44	D20.8	"A" coefficient.	0.0 M /RAD/ S COM/EOP/UT1
45-59	D15.3	"B" coefficient.	0.0 M /RAD/ S
60-72	D13.1	Sigma "A".	0.0
73-80	D8.2	Sigma "B".	0.0

- [9] Suggested OLOLAD setups for center of mass and EOP are provide on this webpage. The corrections that result from these setups are also provided for a 10 day period in 1994. These corrections have been compared to corrections provided by other researchers.
- [10] The sense of the center of mass correction is that after it is added to coordinates given in the terrestrial reference frame of the stations, the coordinates are in a center of of the Earth's mass frame consistent with satellite orbits. GEODYN uses the following formula to compute the center of mass correction in a component (X, Y or Z) of position due to of the ith tide (a tide is specified by a Doodson number):

delta(i,t)=A(i) \* cos(arg(i,t)) + B(i) \* sin(arg(i,t))

The argument, arg(i,t), is mainly (see notes [5] and 6]) a linear combination of certain astronomical angles at time, t. The factors factors of the linear combination are dictated by the input Doodson number.

[11] The sense of EOP corrections is that when they are added to a smooth time series of EOP the resulting EOP series is closer to observed values.

UT1

The UT1 correction due to the ith tide (a tide is specified by a Doodson number) is computed according to the following formula:

delta(i,t)=A(i) \* cos(arg(i,t)) + B(i) \* sin(arg(i,t))

The argument,  $\arg(i,t)$ , is mainly (see notes [5i and 6]) a linear combination of certain astronomical angles at time, t. The factors factors of the linear combination are dictated by the input Doodson number.

Xp and Yp

 $\mbox{\tt Xp}$  and  $\mbox{\tt Yp}$  corrections are linked together. Richard Ray has prepared a PDF to give more background on this. The PDF can be download from this webpage.

The correction due to the ith tide for Xp and Yp is linked by a common pair of A and B coefficients.

The argument, arg(i,t), is mainly (see notes [5i and 6]) a linear combination of certain astronomical angles at time, t. The factors factors of the linear combination are dictated by the input Doodson number. This approach requires the use of both prograde and retrograde tides. See NOTE [7].

EXAMPLE SETUPS AND CORRECTIONS FOR EOP AND COM

EOP

### EXAMPLE SETUP FOR EOP OLOAD CARDS

(These coefficients are based on the ocean tide "Model C" from the paper by Chao, Ray, Gipson, Egbert, & Ma (JGR, 1996), with minor constituents inferred from a Munk-Cartwright Fourier series expansion of the tidal admittances. The tidal model is the same one that was adopted for the IERS-2010 Conventions, although the coefficients themselves are slightly different owing to various translations done by the IERS.)

These corrections are consistent with the suggested EOP input

When comparing our EOP coefficients to those tabulated by the IERS, there are some important differences. The main one is a different convention for the tidal arguments. In addition, the IERS explicitly tabulates all nodal sidelines. This must NOT be done for GEODYN, because GEODYN internally computes all needed lunar nodal modulations (as well as perigee modulations for M1 and L2). Thus, a GEODYN set of EOP coefficients will always be smaller than the corresponding table of IERS coefficients, at least when the same tidal constituents are being included. Please see the reference by Richard Ray on: Background to GEODYN Modeling of Diurnal/Semidiurnal Tidal Polar Motion

COM

#### EXAMPLE SETUP FOR COM OLOAD CARDS

(These coefficients are based on the  ${\tt GOT4.10c}$  ocean tide model, which is a small revision to the  ${\tt GOT4.10}$  model described by Ray (JGR-Oceans,

118, 4570, 2013), with the revision being a more rigorous accounting for tidal geocenter motion during the model development, as described by Desai & Ray (GRL, 41, 2454, 2014).)

These corrections are consistent with the suggested COM input - X component

These corrections are consistent with the suggested COM input - Y component

These corrections are consistent with the suggested COM input - Z component

IF CARD OMITTED: Tidal driven EOP/UTI or COM corrections will not be applied

## 2.3.71 OTCOEF

OTCOEF	1	+2+3+4+5+	68
+	0	+0+0+0	-0+0
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A6	Provides the apriori values for the S coefficients needed for the ocean tide model evaluation. This card applies to the ocean tide model that employs Proudman functions.	
7-8	12	Tidal constituent identification number	
		= 0 M2 = 1 K2 = 2 S2 = 3 N2 = 4 K1 = 5 O1 = 6 P1 = 7 Q1	
25-44	D20.8	A priori value for C coefficient.	0.
45-59	D15.3	A priori value for S coefficient.	0.
60-72	D13.1	Standard deviation for C coefficient. No adjustment if this field is zero.	0.
73-80	D8.2	Standard deviation for S coefficient. No adjustment if this field is zero.	0.
NOTE [1]		tream of OTCOEF cards should follow the OTMOD $ m e$ OTMOD Vol. 3).	card

 $\begin{tabular}{ll} \end{tabular} \begin{tabular}{ll} \end{tabular} \begin{tabular}{ll} \end{tabular} and OTMOD has been requested the run will terminate abnormally. \\ \end{tabular}$ 

## 2.3.72 OTIDEN

OTIDEN+	1 2 2	+2+3+4+5+ +1220 0110224744 +.0192629 +0+0+0+	10. 10.
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A6	OTIDEN - Requests application and/or adjustment of ocean tides using the Colombo tide model and normalized Legendre polynomials. M and K symmetry is taken into account.	
7-8	12	<pre>Indicates sign (+ or -) term in harmonic expansion.</pre>	+1
9	I1	Indicates form of input coefficients.	0
		O = A and B coefficients. [NOTE 1]	
10	I1	Indicates form of input sigmas.	0
		O = A and B coefficients. [NOTE 1]	
11-12	12	Degree of expansion.	0
13-14	12	Order of expansion.	0
15-17	13	Sign (+1 or -1) of the Demos Number. [NOTE	2] +1
18-20	13	MKH expansion arguments of the Demos number. [NOTE 2]	0
		M = 0, 1, 2 K = 0, 1, 2 H = 0, 1, 2	
21-22	12	"J" expansion argument of Demos number9. LE. J .LE. 9 [NOTE 2]	0
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
23	I1	<pre>Indicates form of tidal amplitude coefficient [ V (t) / V bar ] [NOTE 3]</pre>	0
		<pre>0 = only main line, fixed coefficient 1 = main line and sidebands, variable     coefficient 2 = only main line, variable coefficient</pre>	
24	I1	Disturbing Body (B). [NOTE 2]	0
		<pre>0 = Moon and Sun 1 = Moon</pre>	

2 = Sun

```
25-44
        D20.8 "A" coefficient. [NOTE 1]
                                                                  O. M
               (Amplitude)
45-59
       D15.3 "B" coefficient. [NOTE 1]
                                                                  0.
                                                                       Μ
               (Phase)
       D13.1 Sigma "A".
60-72
                                                                  0.
                                                                       М
               (Amplitude)
               Sigma "B".
73-80
       D8.2
                                                                  0.
                                                                       М
               (Phase)
```

#### NOTES:

[1] The "A" and "B" coefficients are related to amplitude and phase by the following formulas:

```
A = amplitude * cos(phase)
B = amplitude * sin(phase)
```

[2] The "Demos Number" is represented in the Parameter Label (Word #2) as follows:

[3] There are 30 recognized main tidal lines. Their Demos numbers are:

SMKH	JB	SMKH	JB	(See page 1 for S M K J B definitions)
				3 B delinitions/
-1020	02	-1020	11	
1001	-11	1120	11	
-1001	11	1101	11	
-1020	01	-1101	-11	
1120	01	1101	-11	
1120	02	-1101	11	
1101	01	-1120	01	
-1101	01	1220	21	
1101	02	1220	11	
-1101	02	1220	-11	
1101	00	1201	01	
1220	01	-1201	01	
1220	02	1201	02	
1001	-12	-1201	02	
-1001	12	1201	00	

Of these 30 main lines there are 18 distinct frequencies. The side bands associated with these 18 main line frequencies are:

```
Main Line 1: Ssa -1020 02
Side Bands:
                 1001 -22
Main Line 2: Mm 1001 -11, -1001 11
                                        1011 -11, 1021 -11,
1002 11, 1012 11
Side Bands:
                 -1021 11, -1011 11,
                 -1020 -11, -1010 -11,
Main Line 3: Mf -1020 01
                -1021 21, -1011 21, 1001 -21, 1021 -21, -1010 01, -1000 01,
Side Bands:
                                                     1011 -21,
                                                     1012 01
Main Line 4: 01
                 1120 01
                -1112 01, 1100 01, 1110 01, -1111 -21, 1101 21, 1111 21,
Side Bands:
Main Line 5: P1 1120 02
Side Bands:
                 1101 22
Main Line 6: K1 1101 01, -1101 01, 1101 02, -1101 02, 1101 00 Side Bands: -1121 01, -1111 01, 1111 01, 1121 01
Main Line 7: M2
                1220 01
Side Bands:
                 1201 21, 1210 01, 1211 21, 1200 01, -1212 01,
                 -1211 -21, 1221 21
                1220 02
Main Line 8: S2
Side Bands:
                 1201 22
Main Line 9: Sa
                 1001 -12, -1001 12
                -1020 -12
Side Bands:
Main Line 10: Mtm -1020 11
                  1001 -31, -1010 11,
Side Bands:
                                           1002 -11
Main Line 11: Q1 1120 11
Side Bands:
                 1100 11, 1110 11,
                                        1101 31
Main Line 12: M1 1101 11, -1101 -11
Side Bands:
             1110 -11, 1120 -11,
                                        -1111 -11, 1111 11, 1121 11
Main Line 13: J1 1101 -11, -1101 11
                -1111 11, 1111 -11, 1121 -11, -1120 -11, -1110 -11
Side Bands:
Main Line 14: 001 -1120 01
                 1101 -21, 1111 -21, -1110 01, 1102 01,
Side Bands:
Main Line 15: 2N2 1220 21
Side Bands:
                 1210 21
Main Line 16: N2 1220 11
Side Bands:
                 1210 11
Main Line 17: L2 1220 -11
Side Bands:
                 1210 -11, -1211 -11, 1201
                                                11, 1211 11,
                                                                  1221 11
Main Line 18: K2 1201 01, -1201 01, 1201 02, Side Bands: -1211 01, 1211 01, 1221 01
                                                 02, -1201 02, 1201 00
```

UNITS: KM =Kilometers; M =Meters ; S =Seconds ; Kg =Kilograms DEG=Degrees ; RAD=Radians ; AS =Arc seconds; MAS=Milli-arc seconds

IF CARD OMITTED: The expanded ocean tide model is not used. See TIDES card for tide model theat will be used.

## 2.3.73 OTIDES

OTIDES+	1 2 2	+2+3+4+5+ +1220 0110224744 +.0192629 +0+0+0+	10. 10.
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	The bef Doo fol Col	OTIDES - Requests application and/or adjustment of ocean tides. There are Ocean Tide models available in GEODYN. First implemented was the Colombo tide model and unnormalized Legendre polynomials. M and K symmetry is NOT included. presence of a RAYTID option ore the OTIDES cards invoques the dson Ocean Tides. Immediately lowing there is a description of the ombo model. At the end of this page re will be a description of the Doodson el.	Colombo Model
7-8	12	Indicates sign (+ or -) term in harmonic	+1
9	I1	expansion, for prograde or retrograde Indicates form of input coefficients.	0
		O = A and B coefficients. [NOTE 1]	
10	I1	Indicates form of input sigmas.	0
		O = A and B coefficients. [NOTE 1]	
11-12	12	Degree of expansion.	0
13-14	12	Order of expansion.	0
15-17	13	Sign (+1 or -1) of the Demos Number. [NOTE 2 This sign multiplies the quantity -((2 -H) * $\omega$ + (2 - 2H + J) * M + K * Omega	
18-20	13	MKH expansion arguments of the Demos number. [NOTE 2]	0
		M = 0,1,2 $K = 0,1,2$ $H = 0,1,2$	
21-22	12	"J" expansion argument of Demos number. -9. LE. J .LE. 9 [NOTE 2]	0
23	I1	<pre>Indicates form of tidal amplitude coefficient [ V (t) / V bar ] [NOTE 3]</pre>	0
		<pre>0 = only main line, fixed coefficient</pre>	

- 1 = main line and sidebands, variable
   coefficient
- 2 = only main line, variable coefficient
- 24 I1 Disturbing Body (B). [NOTE 2]
  - 0 = Moon and Sun
  - 1 = Moon
  - 2 = Sun
- 25-44 D20.8 "A" coefficient. [NOTE 1]

O. M

0

45-59 D15.3 "B" coefficient. [NOTE 1]

O. M

60-72 D13.1 Sigma "A".

O. M

73-80 D8.2 Sigma "B".

 $\mathsf{O}$  .  $\mathsf{M}$ 

### NOTES:

- [1] The "A" and "B" coefficients are related to amplitude and phase by the following formulas:
  - A = amplitude \* cos(phase)
  - B = -amplitude \* sin(phase)
- [2] The "Demos Number" is represented in the Parameter Label (Word #2) as follows:

\*\*\*\*\*\*\*\*\*\*\*\*\*\*\*\*\*\*

- \* HARMONIC \* MKH\*1000+(J+10)\*10+B \* DEMOS \*
- [3] There are 30 recognized main tidal lines.

Their	Demos	numbers	are:	

SMKH	JB	SMKH JB (See page 1 for S M K H J B definitions)
-1020	02	-1020 11
1001	-11	1120 11
-1001	11	1101 11
-1020	01	-1101 -11
1120	01	1101 -11
1120	02	-1101 11
1101	01	-1120 01
-1101	01	1220 21
1101	02	1220 11
-1101	02	1220 -11
1101	00	1201 01
1220	01	-1201 01
1220	02	1201 02
1001	-12	-1201 02
-1001	12	1201 00

Line 1: Bands:											
Line 2: Bands:		-1021	11,		11,	1011 1002	-11, 11,				
Line 3: Bands:		-1021		-1011 -1010		1001 -1000		1011 1012			
Line 4: Bands:		-1112		1100 1111				-1111	-21,		
Line 5: Bands:			02 22								
Line 6: Bands:								-1101 1121		1101	00
Line 7: Bands:		1201		1210 1221		1211	21,	1200	01,	-1212	01,
Line 8: Bands:											
Line 9: Bands:				-1001	12						
Line 10: Bands:				-1010	11,	1002	-11				
Line 11: Bands:			11 11,	1110	11,	1101	31				
Line 12: Bands:						-1111	-11,	1111	11,	1121	11
Line 13: Bands:						1121	-11,	-1120	-11,	-1110	-11
Line 14: Bands:				1111	-21,	-1110	01,	1102	01,	1112	01
Line 15: Bands:	2N2	1220 1210	21 21								
Line 16: Bands:		1220 1210	11 11								
Line 17: Bands:	L2			-1211	-11,	1201	11,	1211	11,	1221	11

Main Line 18: K2 1201 01, -1201 01, 1201 02, -1201 02, 1201 00 Side Bands: -1211 01, 1211 01, 1221 01

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This document contains the complete list of GEODYN OTIDE cards. If a Doodson number does not appear on the printed tables, then GEODYN cannot model the tide associated with that Doodson number.

UNITS: KM =Kilometers; M =Meters ; S =Seconds ; Kg =Kilograms
DEG=Degrees ; RAD=Radians ; AS =Arc seconds; MAS=Milli-arc seconds

IF CARD OMITTED: The ocean tide model is not used.

#### OCEAN TIDE MODELING

Ocean tide modeling in GEODYN is accomplished using the OTIDES option. The following explanation will assist the user in invoking specific tides and describes in more detail the implementation of the "tidal families" within GEODYN. The ocean tidal constituents are divided into two categories: Those major tides which (a) do and those which (b) do not have significant sideband constituents. The sidebands are long period modulations within a narrow band of the main tide lines. For the 18 tide lines identified by GEODYN as having significant sidebands, under the assumption that the admittances can be treated as identical for near-by frequencies, GEODYN used linear admittances to provide for the sideband tides within these so-called "tidal families" if requested. When column 23 is set to 1 on the OTIDES card, every degree and order element specified for the mainline tide causes inclusion of the same harmonic for all of the corresponding sideband tide lines in both the surface height and force modeling calculations. However, if the tide line is not recognized by GEODYN as having significant sideband contributions, a 1 in column 23 will be ignored. Since the linear admittances are computed using the ratio of the Doodson coefficients of the sidebands with respect to the mainline, the Doodson ceofficients shown in the following tables indicate the relative importance of the sideband terms within the total tidal family.

The following tables describe the ocean tidal constituents available within GEODYN. Tables 1, 1.1, 1.2, and 1.3 present tidal constituents which have no internal tables containing sideband terms. Therefore, even if the option flag in column 23 is set to 1, no sideband modeling will be invoked. Table 2. lists the most important tides commonly used in orbit modeling which fall within this category. Tables 3, 3.1, 3.2 and 3.3 present the 18 tidal families which can be invoked using the "family option" (column 23 set equal to 1 on the OTIDES card). Shown within these tables is the mainline tide, the default sidebands which are included, and the terms in the angular arguments defining the tide which modulate the mainline angular argument.

1. TIDES TREATED AS MAINLINE CONSTITUENTS:
NOTE: THESE TIDES HAVE NO PRE-DEFINED SIDEBAND CONSTITUENTS

## 1.1. LONG PERIOD BAND

DOODSONS	EARTH	DOODSONS	GEODYN
NUMBER	PERIOD	COEFFIC.	+MKH JB
(	(D=DAY,H=HOU	R)	ARGUMENT
058.5525	121.7532D	0.000007	1001-32
058.5545	121.7493D	0.004262	1020 12
059.5535	91.3127D	0.000173	1020 22
05X.5525	73.0505D	0.000006	1020 32
095.1455	6.8956D	-0.000003	1011 41
095.1555	6.8886D	0.000044	1001-41
095.1655	6.8817D	-0.000003	1011-41
095.3555	6.8594D	0.004014	1020 21
095.3655	6.8525D	0.001663	1010 21
095.3755	6.8456D	0.000156	1000 21
095.3855	6.8387D	-0.000003	1012-21
0X5.2555	5.4922D	0.000456	1020 31
0X5.2655	5.4877D	0.000189	1010 31
0X5.2755	5.4833D	0.000018	1000 31
0E5.1555	4.5794D	0.000047	1020 41
0E5.1655	4.5763D	0.000020	1010 41
0E5.1755	4.5732D	0.000002	1000 41

### 1.2 DIURNAL BAND

DOODSONS		DOODSONS		
NUMBER		COEFFIC.		
	(D=DAY, H=HO)	UR)	ARGUMENT	
105.9455	1.2751D	0.000022	1110 41	
105.9555	1.2749D	0.000114	1120 41	
115.8355	1.2190D	-0.000006	1100 31	
115.8455	1.2187D	0.000207	1110 31	
115.8555	1.2185D	0.001099	1120 31	
125.7355	1.1673D	-0.000056	1100 21	
125.7455	1.1671D	0.001823	1110 21	
125.7555	1.1669D	0.009669	1120 21	2Q1
125.9555	1.1661D	-0.000016	1101 41	
125.9655	1.1659D	-0.000003	1111 41	
155.4355	1.0357D	0.000060	1100-11	
155.4455	1.0355D	-0.001965	1110-11	
155.4555	1.0354D	-0.010427	1120-11	
155.6355	1.0350D	-0.000005	-1121-11	
155.6455	1.0349D	0.000864	-1111-11	
155.6555	1.0347D	-0.029805	1101 11	
155.6655	1.0346D	-0.005907	1111 11	
155.6755	1.0344D	0.000127	1121 11	
160.5585	1.0111D	0.000014	1120 32	
161.5575	1.0083D	0.000418	1120 22	
162.5565	1.0055D	0.010266	1120 12	$\pi_1$
162.5585	1.0055D	-0.000003	1101 32	
164.5545	1.0000D	-0.001467	1120-12	
164.5565	2 4.0000H	-0.004214	1101 12	$S_1$
166.5545	2 3.8693Н	-0.004214	1101-12	$\psi_1$
166.5565	2 3.8693Н	0.000063	-1120-12	
167.5535	2 3.8045H	-0.000106	1101-22	

167.5555	2 3.8045H	-0.007544	-1120 02	$\phi_1$
167.5556	2 3.8045H	-0.000002	-1110 02	
168.5525	2 3.7400H	-0.00003	1101-32	
168.5545	2 3.7400H	-0.000442	1120 12	
169.5535	2 3.6759H	-0.000018	-1120 22	
175.2555	2 3.1122H	0.00001	1120-31	
175.4355	2 3.1050H	-0.000005	-1121 11	
175.4455	2 3.1017H	0.000864	-1111 11	
175.4555	2 3.0985H	-0.029805	1101-11	$J_1$
175.4655	2 3.0952H			
175.4755	2 3.0919H	0.000127	1121-11	
175.6555	2 3.0847H	0.000449	1120-11	
175.6655	2 3.0815H	0.000288	-1110-11	
175.6755	2 3.0782H	0.000060	1102 11	
175.6855	2 3.0749H	0.000004	1112 11	
195.2455	2 1.5931H	0.00006	-1111 31	
195.2555	2 1.5903H	-0.000198	1101-31	
195.2655	2 1.5874H	-0.000039	1111-31	
195.4555	2 1.5782H	-0.003125	-1120 11	
195.4655	2 1.5754H	-0.002000	-1110 11	
195.4755	2 1.5725H	-0.000420	1102-11	
195.4855	2 1.5697H			
1X5.1555	2 0.9077H	-0.000016	1101-41	
1X5.1655	2 0.9050H	-0.000003	1111-41	
1X5.3555	2 0.8964H	-0.000417	-1120 21	
1X5.3655	2 0.8937H	-0.000267	-1110 21	
1X5.3755	2 0.8910H	-0.000056	1102-21	
1X5.3855	2 0.8884H	-0.000004	1112-21	
1E5.2555	2 0.2563H	-0.000047	-1120 31	
1E5.2655	2 0.2538H	-0.000030	-1110 31	
1E5.2755	2 0.2513H	-0.000006	1102-31	
	19.6543H	-00.0000005	5-1120 41	
1F5.1655	19.6519H	-00.000003	-1110 41	

# 1.3 SEMI-DIURNAL BAND

	DOODSON NUMBER	S	EARTH PERIO (H=HO	D	DOODSO: COEFFI		GEODYN +MKH JE ARGUMEN	
215.	9455	13.	4306H	-	0.000010		1210 41	
215.	9555	13.	4295H		0.000275		1220 41	
225.	8355	13.	1644H		0.00001		1200 31	
225.	8455	13.	1633H	-	0.000099		1210 31	
225.	8555	13.	1622H		0.002647		1220 31	
235.	7555	12.	9054H		0.023293		1220 21	2N2
270.	5585	12.	0495H		0.000035		1220 32	
271.	5575	12.	0329H		0.001006		1220 22	
274.	5545	11.	9836H	-	0.003535		1220-12	$R_2$
274.	5565	11.	9836H		0.000914		1201 12	
276.	5545	11.	9509H		0.000914		1201-12	
276.	5565	11.	9509H	-	0.000007	-	1220-12	
277.	5535	11.	9346H		0.000023		1201-22	
277.	5555	11.	9346H		0.000783	-	1220 02	
278.	5545	11.	9184H		0.000046	-	1220 12	
279.	5535	11.	9022H		0.000002	-	1220 22	
285.	2555	11.	7581H		0.000003		1220-31	
285.	4455	11.	7554H	-	0.000121	-	1211 11	

```
285.
     4555 11.
                7545H
                          0.006465
                                       1201-11
                                                 \eta_2
285.
     4655 11.
                7537H
                          0.002816
                                       1211-11
     4755
285.
           11.
                7528H
                          0.000305
                                       1221-11
                7510H
285.
     6555
                          0.000047
                                   - 1220-11
          11.
                7501H
                          0.000040
285.
     6655
           11.
                      _
                                   - 1210-11
285. 6755
                7493H
                                       1202 11
           11.
                          0.000013
285.
     6855
           11.
                7484H
                          0.000002
                                       1212 11
295. 3455
                5501H -
                          0.000010 - 1211 21
          11.
295. 3555
          11.
                5492H
                          0.000532
                                       1201-21
295. 3655 11.
                                       1211-21
                5484H
                          0.000232
295. 3755
           11.
                5476H
                          0.000025
                                       1221-21
                                   - 1220 01
295. 5555
          11.
                5458H
                          0.001686
295. 5655 11.
                5450H
                          0.001460
                                    - 1210 01
295. 5755
                                       1202 01
           11.
                5442H
                          0.000474
295. 5855
          11.
                5433H
                          0.000068
                                       1212 01
295. 5955
                5425H
                          0.00004
                                       1222 01
          11.
2X5. 2555 11.
                3510H
                          0.000043
                                      1201-31
2X5.
     2655
           11.
                3502H
                          0.000019
                                       1211-31
2X5. 2755
                          0.000002
          11.
                3494H
                                      1221-31
2X5. 4555
          11.
                3477H
                          0.000324
                                      -1220 11
2X5. 4655
          11.
                3469H
                          0.000281
                                      -1210 11
2X5. 4755
           11.
                3461H
                          0.000091
                                       1202-11
2X5. 4855
          11.
                3453H
                          0.000013
                                       1212-11
2E5. 1555
          11.
                1595H
                          0.000003
                                      1201-41
2E5. 1655 11.
                          0.00001
                                      1211-41
                1587H
2E5.
     3555
                1562H
                          0.000043
                                      -1220 21
           11.
2E5. 3655
                          0.000037
                                     -1210 21
          11.
                1555H
2E5. 3755
          11.
                1547H
                          0.000012
                                      1202-21
2E5. 3855
                1540H
                          0.000002
                                       1212-21
           11.
2F5. 2555
           10.
                9712H
                          0.000005
                                      -1220 31
2F5. 2655
           10.
                9704H
                          0.000004
                                      -1210 31
     2755
                9697H
2F5.
           10.
                          0.00001
                                      1202-31
```

2. IMPORTANT TIDES TREATED AS MAINLINE CONSTITUENTS:
NOTE: THESE TIDES HAVE NO PRE-DEFINED SIDEBAND CONSTITUENTS

TIDE	DOODSONS	EARTH	DOODSONS	GEODY	'N				
	NUMBER	PERIOD	COEFFIC.	+MKH	JB				
	()	Y=YEAR,H=HOU	JR)	ARGUME	NT				
						OMEG	MEAN	NODE	THETA
18.6 YR LUN	055.5655	18.8712Y	-0.065520	1011	01	0	0	1	0
9.4 YR LUN	055.5755	9.4356Y	0.000640	1021	01	0	0	2	0
T2 SOLAR	272.5565	12.0164H	0.024729	1220	12	2	3	2	2

3. TIDES WHICH CAN BE MODELED AS FAMILIES:
NOTE: THESE TIDES HAVE PRE-DEFINED SIDEBAND CONSTITUENTS

#### 3.1 LONG PERIOD BAND

		SA FAMI	LY			St	JN	
	DOODSONS	EARTH	DOODSONS	GEODYN				
	NUMBER	PERIOD	COEFFIC.	+MKH JB				
	(D	=DAY, Y=YEA	R)	ARGUMENT				
					OMEG	MEAN	NODE	THETA
MAINLINE	056.5545	1.0139Y	0.011733	1001-12	0	1	0	0
SIDEBAND	056.5565	1.0138Y	-0.000609	-1020-12	2	1	2	0

		SSA FAMI	LY			S	UN	
					OMEG	MEAN	NODE	THETA
MAINLINE	057.5555	182.6211D	0.072687	-1020 02	2	2	2	0
SIDEBAND	057.5535	182.6298D	0.000295	1001-22	0	2	0	0
		MM	FAMILY				MOOI	N
					OMEG	MEAN	NODE	THETA
MAINLINE	065.4555	27.5546D	0.082984	1001-11	0	1	0	0
SIDEBAND	065.4355	27.7797D	0.000053	-1021 11	0	1	2	0
SIDEBAND	065.4455	27.6667D	-0.005390	-1011 11	0	1	1	0
SIDEBAND	065.4655	27.4433D	-0.005390	1011-11	0	1	-1	0
SIDEBAND	065.4755	27.3330D	0.000053	1021-11	0	1	-2	0
SIDEBAND	065.6555	27.0925D	-0.004329	-1020-11	2	1	2	0
SIDEBAND	065.6655	26.9850D	-0.001793	-1010-11	2	1	1	0
SIDEBAND	065.6755	26.8783D	-0.000168	1002 11	2	1	0	0
SIDEBAND	065.6855	26.7724D	0.000004	1012 11	2	1	-1	0
		MF FAMI	LY			МО	ON	
					OMEG	MEAN	NODE	THETA
MAINLINE	075.5555	13.6608D	0.156567	-1020 01	2	2	2	0
SIDEBAND	075.3355	13.8333D	0.000004	-1021 21	0	2	2	0
SIDEBAND	075.3455	13.8053D	-0.000443	-1011 21	0	2	1	0
SIDEBAND	075.3555	13.7773D	0.006827	1001-21	0	2	0	0
SIDEBAND	075.3655	13.7494D	-0.000443	1011-21	0	2	-1	0
SIDEBAND	075.3755	13.7217D	0.0000443	1021-21	0	2	-2	0
			0.064864					
SIDEBAND	075.5655	13.6334D		-1010 01	2	2	1	0
SIDEBAND	075.5755	13.6061D	0.006086	-1000 01	2	2	0	0
SIDEBAND	075.5855	13.5789D	-0.000131	1012 01	2	2	-1	0
		MTM FAMI	LY			МО	ON	
					OMEG	MEAN	NODE	THETA
MAINLINE	085.4555	9.1329D	0.030112	-1020 11	2	3	2	0
SIDEBAND	085.2555	9.1849D	0.000551	1001-31	0	3	0	0
SIDEBAND	085.4655	9.1207D	0.012475	-1010 11	2	3	1	0
SIDEBAND	085.4755	9.1085D	0.001171	1002-11	2	3	0	0
3.2 DIURNAL	BAND							
		Q1 FAMI	LY			МО	ON	
	DOODSONS	EARTH	DOODSONS	GEODYN		_		
	NUMBER	PERIOD	COEFFIC.	+MKH JB				
		D=DAY,H=HOU		ARGUMENT				
	(	, 1100	/		OMEG	MEAN	NODE	THETA
MAINLINE	135.6555	1.1195D	0.072536	1120 11	-2	-3	-2	1
SIDEBAND	135.6355	1.1199D	-0.000420	1100 11	-2	-3	0	1
SIDEBAND	135.6455	1.1197D	0.013673	1110 11	-2	-3	-1	1
SIDEBAND	135.8555	1.1187D	-0.000198	1101 31	0	-3	0	1
,					·	-		
		01	FAMILY				MO01	
					OMEG	MEAN	NODE	THETA
MAINLINE	145.5555	1.0758D	0.377150	1120 01	-2	-2	-2	1

SIDEBAND	145.5255	1.0763D	0.000021	-1112 01	-2	-2	1	1
SIDEBAND	145.5355	1.0761D	-0.002186	1100 01	-2	-2	0	1
SIDEBAND	145.5455	1.0760D	0.071090	1110 01	-2	-2	-1	1
SIDEBAND	145.7455	1.0753D	0.000071	-1111-21	0	-2	1	1
SIDEBAND	145.7555	1.0751D	-0.002452	1101 21	0	-2	0	1
SIDEBAND	145.7655	1.0749D	-0.000486	1111 21	0	-2	-1	1
SIDEBAND	145.7755	1.0748D	0.000010	1121 21	0	-2	-2	1
		M1 FAMI	τV			МО	ΟN	
		III I AIII	<b>1</b> 1		OMEG	MEAN	NODE	THETA
MAINLINE	155.6555	1.0347D	-0.029805	1101 11	0	-1	0	1
GIDEDAND	155 4455	1 0255	0 001065	1110 11	0	4	4	4
SIDEBAND	155.4455	1.0355D	-0.001965	1110-11	-2	-1	-1	1
SIDEBAND	155.4555	1.0354D	-0.010427	1120-11	-2	-1	-2	1
SIDEBAND	155.6455	1.0349D	0.000864	-1111-11	0	-1	1	1
SIDEBAND	155.6655	1.0346D	-0.005907	1111 11	0	-1	-1	1
SIDEBAND	155.6755	1.0344D	0.000127	1121 11	0	-1	-2	1
		P1 FAMI	LY			S	UN	
					OMEG	MEAN	NODE	THETA
MAINLINE	163.5555	1.0027D	0.175093	1120 02	-2	-2	-2	1
SIDEBAND	163.5575	1.0027D	-0.000106	1101 22	0	-2	0	1
		K1 FAMI	LY			SUN/MO	ON	
					OMEG	MEAN	NODE	THETA
MAINLINE	165.5555	23.9345Н	-0.530078	1101 01	0	0	0	1
SIDEBAND	165.5355	23.9415H	-0.000066	-1121 01	0	0	2	1
SIDEBAND	165.5455	23.9413H	0.010506	-1111 01	0	0	1	1
SIDEBAND	165.5655	23.9310H	-0.071809	1111 01	0	0	-1	1
	165.5755	23.9310H 23.9274H	0.001541		0	-	-1 -2	
SIDEBAND	165.5755	23.92/4н	0.001541	1121 01	U	0	-2	1
		J1 FAMI	LY			MO		
					OMEG	MEAN	NODE	THETA
MAINLINE	175.4555	23.0985Н	-0.029805	1101-11	0	1	0	1
SIDEBAND	175.4455	23.1017H	0.000864	-1111 11	0	1	1	1
SIDEBAND	175.4655	23.0952H	-0.005907	1111-11	0	1	-1	1
SIDEBAND	175.4755	23.0919H	0.000127	1121-11	0	1	-2	1
SIDEBAND	175.6555	23.0847H	0.000449	-1120-11	2	1	2	1
SIDEBAND	175.6655	23.0815H		-1110-11	2	1	1	1
		001 FAMI	LY			MO	ON	
					OMEG	MEAN	NODE	THETA
MAINLINE	185.5555	22.3061H	-0.016249	-1120 01	2	2	2	1
SIDEBAND	185.3555	22.3189Н	-0.002452	1101-21	0	2	0	1
SIDEBAND	185.3655	22.3159H	-0.000486	1111-21	0	2	-1	1
SIDEBAND	185.5655	22.3030H	-0.010401	-1110 01	2	2	1	1
SIDEBAND	185.5755	22.3000H	-0.002186	1102 01	2	2	0	1
SIDEBAND	185.5855	22.2969H	-0.000144	1112 01	2	2	-1	1

3.3 SEMI-DIURNAL BAND

2N2 FAMILY

MOON

	DOODSONS NUMBER	EARTH PERIOD (H=HOUR)	DOODSONS COEFFIC.	GEODY +MKH ARGUME	JB				
MAINLINE	235.7555	12.9054Н	0.023293	1220	21	OMEG -2	MEAN -4	NODE -2	THETA 2
SIDEBAND	235.7455	12.9064H	-0.000869	1210	21	-2	-4	-1	2
		N2 FAMI	LY				МО	ON	
						OMEG	MEAN	NODE	THETA
MAINLINE	245.6555	12.6583H	0.174731	1220	11	-2	-3	-2	2
SIDEBAND	245.6455	12.6593H	-0.006518	1210	11	-2	-3	-1	2
		M2 FAMI	LY				МО	ON	
	055 555	40.40061	0 000507	1000	0.4	OMEG	MEAN	NODE	THETA
MAINLINE	255.5555	12.4206H	0.908507	1220	01	-2	-2	-2	2
SIDEBAND	255.5255	12.4234H	-0.00003	-1212	01	-2	-2	1	2
SIDEBAND	255.5355	12.4225H	0.000474	1200	01	-2	-2	0	2
SIDEBAND		12.4215H	-0.033892	1210		-2	-2	-1	2
SIDEBAND		12.4176H	-0.000010	-1211-		0	-2	1	2
SIDEBAND		12.4166H	0.000532	1201		0	-2	0	2
SIDEBAND		12.4157H	0.000232	1211		0	-2	-1	2
SIDEBAND	255.7755	12.4147H	0.000025	1221	21	0	-2	-2	2
		L2	FAMILY					MOON	
						OMEG	MEAN	NODE	THETA
MAINLINE	265.4555	12.1916H	-0.025118	1220-	11	-2	-1	-2	2
SIDEBAND	265.4455	12.1925H	0.000937	1210-	11	-2	-1	-1	2
SIDEBAND	265.6455	12.1887H	-0.000121	-1211-	11	0	-1	1	2
SIDEBAND	265.6555	12.1878H	0.006465	1201	11	0	-1	0	2
SIDEBAND	265.6655	12.1869H	0.002816	1211	11	0	-1	-1	2
SIDEBAND	265.6755	12.1860H	0.000305	1221	11	0	-1	-2	2
		S2 FAMI	T.Y				S	UN	
						OMEG		NODE	THETA
MAINLINE	273.5555	12.0000H	0.421776	1220	02	-2	-2	-2	2
SIDEBAND	273.5575	12.0000H	0.000023	1201	22	0	-2	0	2
		K2 FAMI	T.Y				SUN/MO	ON	
						OMEG	MEAN	NODE	THETA
MAINLINE	275.5555	11.9672H	0.114980	1201	01	0	0	0	2
SIDEBAND	275.5455	11.9681H	-0.001475	-1211	01	0	0	1	2
SIDEBAND	275.5655	11.9664H	0.034235	1211	01	0	0	-1	2
SIDEBAND	275.5755	11.9655H	0.003713	1221	01	0	0	-2	2
DOODSON	N OCEAN TIDE	MODEL							
	-1+2	+3	+4	+5-		+6-	+	7	+8
OTIDES	-0+0	+	+0	+0-		+0-	+	0	+0
	FORMAT			. 0				VALUE &	
		_		- 1					

139

OTIDES - Requests application and/or

1-6

A6

adjustment	οf	Doodson	ocean	tides	•	Colombo	0cean	Tides
------------	----	---------	-------	-------	---	---------	-------	-------

7-8	<pre>I2     Indicates sign term of the expansion: +1 = prograde tide</pre>		
	-1 = retrograde tide		
9	I1 Forcing degree in tide operating potential (2 or 3). The order is the first integer in the Doodson number given in columns 18-24.		
10	If Integer number [0,,3] gives phase increment after x{pi/2}. Useful when all tides use cosine factors with positive amplitudes.		
11-12	I2 Response degree of A,B coefficients.		
13-14	I2 Response order of A,B coefficients.		
15	<pre>I1    Paramater type index =0 Ocean tide coefficient =1 Atmospheric Tide coefficient</pre>		
16-17	I2 Integer Flag:		
	<ul><li>=0 no sidelines</li><li>=1 turn n sidelines with input tidal constituent.</li><li>=2 turn n sidelines with input tidal group.</li></ul>		
18-24	I7 Doodson Number (six positive integers)		
25-44	D20.8 "A" coefficient.	0.	M
45-59	D15.3 "B" coefficient.	0.	M
60-72	D13.1 Sigma "A".	0.	M

73-80 D8.2 Sigma "B".

O. M

### 2.3.74 OTMOD

1	1 +	-2	3	4	5	6+7	7+8
OTMOD							
(	) +	-0+	0	0	0	0(	)+0

COLUMNS FORMAT

DESCRIPTION

DEFAULT VALUE & UNITS

1-6 A6 Introduces the application/estimation of ocean tide model which employs Proudman functions evaluated at several points located on a uniform grid on the globe. The tidal height at a given location is evaluated by interpolation and it represents the tidal effect measured from the mean sea surface.

NOTE [1]: This card must be followed by a number of OTCOEF cards which represent the total number of Proudman functions.

IF CARD OMITTED: The above described tide model will not be applied.

### 2.3.75 PARFIL

	-1+2-	+3	+4	-+5+-	6+-	78
PARFIL	7375					20000.
	-0+0-	+0	+0	-+0	0+-	00

### COLUMNS FORMAT DESCRIPTION

DEFAULT VALUE & UNITS

0

- 1-6 A6 Requests output of measurement partial derivative file(s) on last inner iteration of each data reduction arc during first global iteration.
- 7 II Indicates whether each arc should open a new logical unit for output or continue writing to the same unit as the previous arc.
  - O Concatenate all arcs providing header information between each arc.
  - 1 CLOSE currently open output file at end of each arc and increment output unit number for start of each new arc.
  - 2 ENDFILE currently open output file at end of each arc. (applies only to IBM type operating environment for GEODYN-IIE)
- 11-12 I2 Indicates logical unit number (NU1) of first 90 segment of file. Maximum segment size is specified in columns 73-80.

  NOTE: 90 .LE. NU1 .LE. 99

If segment size is exceeded, the currently open logical unit will be CLOSED, the unit number will be incremented by one and output will continue.

COLUMNS FORMAT

DESCRIPTION

DEFAULT VALUE & UNITS

11-12 cont. If column 7 contains a one, then the currently open logical unit will be CLOSEd at the end of each arc, the unit number will be incremented by one and output will continue for the next arc.

If column 7 contains a two, then the currently open logical unit will be ENDFILEd at the end of each arc, the unit number will remain the same, and output will continue for the next arc.

13-14 I2 Maximum unit number (NU2) that will be opened Same as cols. for output of this partial derivative 11-12 file.

NOTE: NU1 .LE. NU2 .LE. 99

25-44 D20.8 Partial derivative file number to uniquely \$E\$-Matrix Number identify the file.

20000.

73-80 F8.1 Number of blocks (of 512 64-bit words) of disk space requested for each segment of the measurement partial derivative file.

A good estimate for the total disk space required for the entire measurement partial derivative file may be obtained by multiplying the number of weighted observations by the number of adjusted parameters and dividing by 512.

If outputing to tape, specification of a very large number here will prevent incrementing of unit number unless specified by column 7.

IF CARD OMITTED: No measurement partial derivative file will be generated.

## 2.3.76 PLANET

PLANET	300	+2+3+4+5+ 20 20	298.255
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A6	PLANET - Modifies the 3rd body gravitational potential and/or introduces new constants.	
7	I1	Gravity model initalization indicator.	0
		= 0 Gravitational coefficients are initialized from gravity model file. They may be changed by GCOEF, GCOEFC or GCOEFS cards.	
		> 0 Gravitational coefficients are set to zero except those which appear on GCOEF, GCOEFC, or GCOEFS cards which follow this card.	
8	I1	Gravity model print control switch.	0
		<pre>= 0 Print two portions of model:    1. lowest degree &amp; order; and    2. highest degree &amp; order.</pre>	
		= 1 Print entire gravity model. : : N/A = 2 Print only model checksum. :	
9	I1	= 2 Print only model checksum. :  Indicator of the contents of the remainder of this card:	0
		= 0 Columns 25-80 contain the values for GMb, Ab, 1/fp, 1/fe.	
		= 1 Columns 25-80 contain the (see NOTE sigmas for GMb, Ab, 1/fp, 1/fe.	[3])
10	I1	> 0 Indicates that this body should be excluded from force model calculation	ons
11-14	14	Indicates body for which gravity perturbation is to be modified.	no default
		Body numbers are as follows:	
		0100 Mercury 0200 Venus 0300 Earth 0301 Earth's Moon 0400 Mars	

		0700 Uranus 0800 Neptune 0900 Pluto 9999 Sun	
15-17	13	Maximum degree of coefficient to be used in the gravitational model.  Minimum =3. Default depends on gravity model file on unit 12.(See NOTE[1] and NOTE[2]).	3
18-20	13	Maximum order of coefficient to be used in the gravitational model.  Minimum =2. Default depends on gravity model file on unit 12.  (See NOTE[2]).	2
		The presence of degree and order in the two integer fields above automatically defines this planet as the center of integration	
21-22	12	Precession model indicator for Mars orbiters.  =0 The old precession model is used  =1 The new precession nutation model is used	old model
23	=5	Index to planetary constant option =0 1989 planetary constants for Venus =1 1991 planetary constants for Venus =2 2000 planetary constants for Venus 2000 IAU constants for Mars The Mars orientation model pliv-Yoder will be used	1989 Planetary constants for Venus 91 Mars IAU const.
	Kono	=6 2011 Margot Model for Mercury =7 IAU2000 for Mercury	2011 Margot Model
24	I1	<pre>Index to lunar orientation parameter option =0 values from the report /IAU/IAG/COSPAR =1 values drom the 403 Planetary Ephemeris</pre>	COSPAR report
25-44	D20.8	Universal gravitational constant times the mass of the Body (GMb), or sigma value if column 9 =1,	See table below for third body perturb. (M**3/S**2)
45-59	D15.3	Semi-major axis of the Body (Ab), or sigma value if column 9 =1,	See table below for third body perturb. (M)
60-72	D13.1	<pre>Inverse of the Body's polar flattening (1/fp), or sigma value if column 9 =1,</pre>	See table below for third body perturb.
73-80	D8.3	<pre>Inverse of the Body's equatorial flattenin (1/fe) or sigma value if column 9 =1.</pre>	g 0.

0600 Saturn

#### TABLE OF DEFAULT BODY CONSTANTS

Body	GMb	Ab	fp	fe
Mercury	2.203208000D+14			
Venus	3.248602000D+14	6.168810000D+06	0.0	0.0
Earth's Moon	4.902778000D+12	1.738090000D+06	0.37800000D-03	0.0
Mars	4.282845000D+13	3.380422000D+06	0.10500000D-01	0.0
Jupiter	1.267077000D+17	7.137155400D+07	0.66700000D-01	0.0
Saturn	3.792653500D+16	6.041128000D+07	0.10500000D 00	0.0
Uranus	5.780158500D+15			
Neptune	6.871307800D+15			
Pluto				
Sun	1.327125000D+20	6.96000000D+08	0.50000000D-04	0.0

#### NOTE[1]:

The default values for degree and order of the gravity field and the Body constants are overridden by the values from the gravity model file on unit 12.

EARTH and PLANET are actually the same keyword card except that EARTH automatically supplies the planet number (0300).

#### NOTE[2]:

All interplanetary runs need PLANET cards for Saturn and Jupiter for the relativity modelling.

#### NOTE[3]:

Only one set of (GMb, Ab, 1/fp, 1/fe) parameters can be adjusted. This set of parameters must be for the central body.

UNITS: M = Meters ; S = Seconds

#### IF CARD OMITTED:

The complete gravitational field contained in the default gravity model file on unit 12 will be used for the central body and the sun and moon will be used as third body perturbers. If the central body is other than Earth, then Earth will replace that body as a third body pertubation.

#### 2.3.77 PLANOR

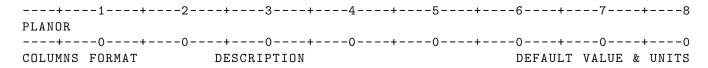
#### \*\*\* NOTE \*\*\*

The documentation below is for the latest version of GEODYN (version 1501 and later). Earlier versions have a different setup for applying the PLANOR option.

If using GEODYN version 1410, see PLANOR\_1410.

If using GEODYN version 1404 or earlier, see PLANOR\_1312\_AND\_EARLIER.

#### \*\*\*\*\*\*



1-6 A6 PLANOR - Introduces modification and/or estimation of planetary orientation parameters.

The planetary orientation parameters are used to describe the orientation of the central body at any epoch. The orientation of the body is given by three angles: the right ascension (RA) and declination (DEC) of the pole (with respect to the J2000 Earth equator and equinox coordinate system) and the W angle between the prime meridian of the body and the body's inertial IAU vector (the RA and DEC of the pole determine the inertial location of the IAU vector).

For the sun, the moon and the 9 major planets, GEODYN has default models for orientation. The PLANOR option allows the user to build a user defined planetary orientation model to repalce a default model or to define the orientation of a body not included on the DEXXX ephemeris. In versions prior to 1501 the user defined model is analytical with linear terms in RA and DEC. For W a quadratic model with an arbitrary number of periodic terms is possible.

Starting with version 1501 many upgrades have been made to the PLANOR capability. There are two types of models avaiable, analytical and dynamic. In analytical mode the RA and DEC now have periodic terms in addition to the linear terms. The analytical model may have an aribitrary number of periods. For each period there will be an Acos and Bsin parameter for each of the three angles, RA, DEC and W. The linear and quadratic terms must be referenced to a base epoch and this is the J2000 epoch (12 HR Jan 1, 2000, ET).

Also starting with the 1501 version the user may request a dynamic orientation model. The dynamic

orientation model is used when there is a "1" in column 7 of the INTGCB card. So, the dynamic orientation model is available only when the orbit of the central body is being numerically integrated. In dynamic mode there are 12 parameters. These are the three angles (RA, DEC and W) at the initial epoch of the arc, the derivatives wrt time of the three angles at the initial epoch of the arc and 6 moments of inertia of the central body with respect to the body fixed cartesian axes. In this mode the orientation of the axes is numerically integrated from the initial state at the initial arc epoch. The step size of the integration is the same as that chosen for the central body orbit numerical integration.

Just below is a list of all of the parameters available (some apply only to analytical mode or only to dynamic mode, for the units see NOTE 2):

- $\alpha$  = right ascension (analytical and dynamic mode)
- $d\alpha/dt$  = right ascension rate (analytical and dynamic mode)
- $\delta$  = declination (analytical and dynamic mode)
- $d\delta/dt$  = declination rate (analytical and dynamic mode)
- W = prime meridian
   (analytical and dynamic mode)
- dW/dt = prime meridian rate
   (analytical and dynamic mode)
- d2W/dt2 = prime meridian acceleration
   (analytical mode only)
- Ai = Coefficient to  $\cos \omega$ ti term for Right Ascension. (analytical mode only, 1501 version and later)
- Bi = Coefficient to  $\sin\omega$ ti term for Right Ascension. (analytical mode only, 1501 version and later)
- Ci = Coefficient to  $\cos \omega$  ti term for Declination. (analytical mode only, 1501 version and later)
- Di = Coefficient to  $\sin \omega ti$  term for Declination. (analytical mode only, 1501 version and later)
- Ei = Coefficient to  $\cos \omega$  ti term for Rotation. (analytical mode only)
- Fi = Coefficient to  $\sin \omega ti$  term for Rotation.

(analytical mode only)

i = 1, # of periodic terms

- Iyy = Moment of inertia about Y axis
   (dynamic mode only)
- Izz = Moment of inertia about Z axis
   (dynamic mode only)
- Ixy = XY axis moment
   (dynamic mode only)
- Iyz = YZ axis moment
   (dynamic mode only)
- Ixz = XZ axis moment
   (dynamic mode only)
- 7-8 I2 Index that allows estimation and modification of selected parameters.

#### ANALYTICAL MODEL:

- =10 right ascension
- =20 right ascension rate
- =30 declination
- =40 declination rate
- =50 prime meridian
- =60 prime meridian rate
- =70 prime meridian acceleration
- =11 Coefficient to  $\cos\omega$ ti term for Right Ascension
- =12 Coefficient to  $\sin\omega {
  m ti}$  term for Right Ascension
- =21 Coefficient to  $\cos\omega$ ti term for Declination
- =22 Coefficient to  $\sin\omega$ ti term for Declination
- =31 Coefficient to  $\cos\omega$ ti term for Rotation
- =32 Coefficient to  $\sin\omega$ ti term for Rotation
- PLANOR11 up to PLANOR32 as a group of 6 cards,

may be repeated as many times as the number of periodic terms. More than one parameter may be selected and modified. When only one parameter is selected (ex PLANOR10), then a value for this parameter may be specified on cols. 60-72 and a sigma may be specified on cols. 73-80. In this case all the other planetary orientation parameters (13 total) will be fixed using a sigma = 1.D-20.

#### DYNAMIC MODEL:

- =10 right ascension
- =20 right ascension rate
- =30 declination
- =40 declination rate
- =50 prime meridian
- =60 prime meridian rate
- =01 Moment of inertia about X axis
- =02 Moment of inertia about Y axis
- =03 Moment of inertia about Z axis
- =04 XY axis moment
- =05 YZ axis moment
- =06 XZ axis moment

For a theoretical description of the dynamic model, see volume 1 of the documentation, Section 8.16 - Orientation Initial Conditions - Variational Equations

- 17 Index for the planetary body used in a binary asteroid case.
  - =0 Default applies to one body
  - =1 First body (orbited body)
  - =2 Second body
- 25-44 D20.8 Period in earth seconds must be included on every PLANOR11, PLANOR21 and PLANOR31 card
- 60-72 D13.1 Value of parameter specified by index on GEODYN col. 7-8.

 $\alpha$  = degrees

 ${\rm d}\alpha/{\rm d} t$  = degrees/Julian centuries of 36525 days  $\delta$  = degrees  ${\rm d}\delta/{\rm d} t$  = degrees/Julian centuries of 36525 days  ${\rm W} = {\rm degrees}$   ${\rm dW/d} t$  = degrees/day

d2W/dt2 = degrees/day2

Ai = degrees

Bi = degrees

Ci = degrees

Di = degrees

Ei = degrees

Di = degrees

Ixx = kg\*m2

Iyy = kg\*m2

Izz = kg\*m2

Ixy = kg\*m2

Iyz = kg\*m2

Ixz = kg\*m2

# 73-80 D8.2 Sigma of parameter specified by index on col. 7. [NOTE 1]

NOTE[1]: When planetary orientation parameters are adjusted, the GEODYN output summary expresses them in the frame and units of adjustment. This is degrees of epoch J2000 and degrees/seconds for the rates.

#### NOTE [2]:

Units of PLANOR parameters in GEODYN

The PLANOR parameters are input to GEODYN via the PLANOR cards.

On the cards the units for the parameters are:

Right ascension  $\alpha$  = degrees

Right ascension rate  $d\alpha/dt$  = degrees/Julian century of 36525 days

Declination  $\delta$  = degrees

Declination rate  $d\delta/dt$  = degrees/Julian century of 36525 days

Prime meridian angle W = degrees Prime meridian angle dW/dt = degrees/day Prime meridian angle d2W/dt2 = degrees/day\*\*2

```
Ai Coefficient to costi term for RA = degrees
Bi Coefficient to sinti term for RA = degrees
Ci Coefficient to costi term for DEC = degrees
 Di Coefficient to sinti term for DEC = degrees
 Ei Coefficient to costi term for ROT = degrees
 Fi Coefficient to sinti term for ROT = degrees
 These values are in IIS (PLN subroutines) converted as follows and loaded
 into PARMV(IPVAL(IXXTRO...)) and PARMVO(IPVALO(IXXTRO..))
= radians/second \\ = radians \\ = radians \\ = radians/second \\ = radians/second \\ = radians/second \\ = radians/second \\ = radians \\ = radians/second **2 \\ = radians/second **3 \\ = rad
 Right ascension \alpha
                                                                                             = radians
 Bi Coefficient to sinti term
Ci Coefficient to costi term
Di Coefficient to sinti term
Fi Coefficient to costi term
                                                                                         = radians
 Ei Coefficient to costi term
                                                                                            = radians
 Fi Coefficient to sinti term
                                                                                            = radians
 The values in PARMV(IPVAL(IXXTRO...)) and PARMVO(IPVALO(IXXTRO..))
 are not used in the formations of the partials as in all instances the PLANOR
 parials are computed numerically and load into PMPA.
 The numerical partials are computed in degrees.
 For printing purposes (unit 6 in subroutine ESTIMG.f)
 the parameter values are converted to:
  Right ascension
                                                                                                = degrees
 Right ascension rate
                                                                                               = degrees/Julian century of 36525 days
 Declination
 Declination rate
                                                                                             = degrees/Julian century of 36525 days
 Prime meridian angle = degrees

Prime meridian angle rate = degrees/day

Prime meridian angle acceleration = degrees/day**2

Ai Coefficient to costi term = degrees
 Bi Coefficient to sinti term
                                                                                           = degrees
 Ci Coefficient to costi term
                                                                                           = degrees
 Di Coefficient to sinti term
                                                                                           = degrees
                                                                                           = degrees
 Ai Coefficient to costi term
 Fi Coefficient to sinti term
                                                                                              = degrees
 For output to the EMAT file the values are
  converted to the units of the partials
 Right ascension
                                                                                                = degrees
 Right ascension rate
                                                                                                = degrees/sec
 Declination
                                                                                               = degrees
 Declination rate
                                                                                               = degrees/sec
 Prime meridian angle
                                                                                             = degrees
 Prime meridian angle rate = degrees/sec
Prime meridian angle acceleration = degrees/sec**2
Ai Coefficient to costi term = degrees
 Ai Coefficient to costi term
                                                                                             = degrees
 Bi Coefficient to sinti term
 Ci Coefficient to costi term
                                                                                             = degrees
```

Di	Coefficient	to	sinti	term	=	degrees
Εi	Coefficient	to	costi	term	=	degrees
Fi	Coefficient	to	sinti	term	=	degrees

EXAMPLE SETUP	FOR BINARY	ASTEROIDS		
PLANOR1	1		0.0000	.1D-40
PLANOR2	1			.1D-40
PLANOR3	1		90.0000	.1D-40
PLANOR4	1			.1D-40
PLANOR5	1		0.0	.1D-40
PLANOR6	1		360.0	.1D-40
PLANOR7	1			.1D-40
PLANOR11	1	18973.4002		.1D-40
PLANOR12	1			.1D-40
PLANOR21	1	18973.4002		.1D-40
PLANOR22	1			.1D-40
PLANOR31	1	18973.4002		.1D-40
PLANOR32	1			.1D-40
PLANOR1	2		0.0	.1D-40
PLANOR2	2			.1D-40
PLANOR3	2		90.0000	.1D-40
PLANOR4	2			.1D-40
PLANOR5	2		0.0	.1D-40
PLANOR6	2		360.0	.1D-40
PLANOR7	2			.1D-40
PLANOR11	2	18973.4002		.1D-40
PLANOR12	2			.1D-40
PLANOR21	2	18973.4002		.1D-40
PLANOR22	2			.1D-40
PLANOR31	2	18973.4002		.1D-40
PLANOR32	2			.1D-40
PLANOR 1	2		1.460096E+15	
PLANOR 2	2		1.517356E+15	1.0D-40
PLANOR 3	2		1.640224E+15	1.0D-40PLANOR 4
2			6.630338E+10 1.0D-40	
PLANOR 5	2		2.955852E+10	1.0D-40
PLANOR 6	2		-2.040129E+11	1.0D-40

## 2.3.78 PLNOR2

PLNOR2			3+5 0.01D0 0+0+0	0.01D0	0.01D0	0.01D0
·	V	. 0 .			V	. 0
COLUMNS	FORMAT		DESCRIPTION	DEFAULT	VALUE	& UNITS
1-6	A6	PLNOR2	Modification and/or estimon planetary orientation All the parameters below and input at J2000.  (Continues from PLANOR)	parameters:		
7-8	12					
9	I1					
10	I1					
11-12	12					
13-14	12					
15-17	13					
18-20	13					
21-22	12					
23	I1					
24	I1					
25-44	D20.8	DXGE01	Used as an observation per in the creation of numeric observation partial derive (RA and DEC)	ical	0.01	LDO
45-59	D15.3	DXGEO2	Used as an observation poin the creation of numeriobservation partial deriv	ical	0.01	L DO
60-72	D13.1	DXFRC1	Used as an acceleration print the creation of explicanceleration partial derication Prime Meridian	cit numerical ivative.	0.0	01DO
73-80	D8.2	DXFRC2	Used as an acceleration print the creation of explicanceleration partial dericant (Rotation Prime Meridian)	cit numerical ivative.	0.0	)1D0

NOTES:

- [1] There must be a PLANOR card in the setup deck.
- [2] This card must be used in conjuction with columns 7-10 of the ALTIM2 card.

IF CARD OMITTED: If PLANOR adjustment has been invoked, default values will be used.

## 2.3.79 PLATE

PLATE	1	+2+3+4+5+	6+8
+	0	+0+0+0+	0+0
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-5	<b>A</b> 5	PLATE - Requests application of tectonic plate motion. A new tectonic plate can be added or values for the default model can be overridden.	
7-8	12	Plate number.	0
		=-2 Plate values from model NUVEL1 used.	
		=-1 Plate values from default model $AMO-2$ used.	
		= 0 Plate values from default model AM1-2 used.	
		=1-14 Plate values in default model will be replaced by values from columns 25-71 of this card.	
		>11 New plate number and values from columns 25-72 will be added to the model for this run.	
25-44	D20.8	Angular velocity of plate in degrees per million years.	O. DEG/MEGA-YRS
45-59	D15.3	Latitude of axis of rotation of plate in degrees north.	O. DEG
60-72	D13.1	Longitude of axis of rotation of plate in degrees east.	O. DEG

#### NOTES:

[1] The user should assign plate numbers on the station coordinate cards. (STAPOS subgroup) The plate numbers on the station coordinate correspond to the following plates.

Plate number	Plate
1	Africa
2	Antarctica
3	Arabia
4	Caribbean
5	Cocos
6	Eurasia

7	Australia
8	Nazca
9	North America
10	Pacific
11	South America
12	India
13	Juan de Fuca
14	Philippine

[2] See option card PLATIM to request station position printouts on selected dates.

IF CARD OMITTED: PLATE motion will not be applied.

#### 2.3.80 PLATIM

+		+2+3+4 810101.	-+5+ 820101.	-68
		+0+0		-0+0
COLUMNS	FORMAT	DESCRIPTION		DEFAULT VALUE & UNITS
1-6	A6	PLATIM - Sets the reference for the station coordinates plate motion and requests pr of station coordinates at sp dates. This option may only used in conjunction with the option.	for inting ecified be	
7 - 8 9 -10 11-12	I2 I2 I2	YY Year, month and day of MM ence date for station DD inates.		80 01 01
25-44	D20.8	First date for printing stat positions. (YYMMDD.)	ion	0.
45-59	D15.3	Next date for printing static positions. (YYMMDD.)	on	0.
60-72	D13.1	Next date for printing stations. (YYMMDD.)	on	0.
73-80	D8.2	Next date for printing static positions. (YYMMDD.)	on	0.

#### NOTES:

Any number of PLATIM cards may be included in your deck, but only one reference date may be used. If more than one reference date is found the last one will be used.

See option card PLATE to request application of the tectonic plate model. Inclusion of a PLATIM card without a PLATE card will NOT invoke the plate model.

IF CARD OMITTED: If PLATE motion is applied, the reference date will be 800101 and no additional station coordinate printouts will be made.

#### 2.3.81 PLMOON

----+---1----+---2---+---3----+---4----+---5---+---6---+---7----+---8
PLMOON 401 720000.0000000 10000.00
----+---1----+---3---+---4----+---5----+---6---+---7----+---8
COLUMNS FORMAT DESCRIPTION DEFAULT VALUE & UNITS

- 1-6

  A6

  PLMOON Requests application and or adjustment of planetary moon GM. This option can be selected for up to 5 moons. The moons must be orbiting the central body of the run. This option is not to be used for the following central bodies: Earth, Earth's moon, and the Sun. Since, the moons selected by this option are not contained in the JPL planetary ephemeris, the user must supply a Keplerian ephemeris for each of the moons. The format and file names for the ephemeris are described below.
- 12-14 I3 Indicates the moon for which this perturbation NO DEFAULTS should be applied.

Body numbers are in the following format: COM C - is central body of this run:

- 1 = Mercury
- 2 = Venus
- 4 = Mars
- 5 = Jupiter
- 6 = Saturn
- 7 = Uranus
- 8 = Neptune
- 9 = Pluto

M - number of this moon (up to 5 moons can be specified)

Example: Mars moon numbers could be:

Phobos - 401 Deimos - 402

25-44 D20.8 Universal gravitational constant times the mass M\*\*3/s\*\*2 of the moon

Default values exist for the following bodies:

Phobos - 401 - 720000.0 M\*\*3/s\*\*2

Deimos - 402 - 120000.0 M\*\*3/s\*\*2

60-72 D13.1 Sigma value for above moon GM

#### MOON EPHEMERIS INFORMATION:

For each moon specified in a run, an ephemeris must be supplied to the GEODYN IIE portion of the program. The file names used in the IIE setup have the following construction:

#### MOON EPHEMERIS FILE NAME = MON###

where ### is the particular moon number specified on the PLMOON card.

Thus, for a typical Mars run with 2 PLMOON cards (one for PHOBOS - 401 and

one for DEIMOS - 402) a moon ephemeris needs to be supplied for each moon. An example of the file names follows:

```
MON401 - IIE ephemeris file for Mars moon #1 (PHOBOS)
MON402 - IIE ephemeris file for Mars moon #2 (DEIMOS)
```

The moon ephemeris file has the following format:

```
HEADER RECORD -> F10.2 -> constant bias to be applied to mean anomaly values.

DATA RECORDS -> 1X, I6, I5, F10.6, F16.3, F15.11, F16.9, 3F17.9

WORD #1 -> I6 -> YYMMDD

WORD #2 -> I5 -> OHHMM

WORD #3 -> F10.6 -> SECONDS INCLUDING FRACTION

WORD #4 -> F16.3 -> SEMI-MAJOR AXIS (meters)

WORD #5 -> F15.11-> ECCENTRICITY

WORD #6 -> F16.9 -> INCLINATION (degrees)

WORD #7 -> F17.9 -> LONGITUDE OF THE ASCENDING NODE (degrees)

WORD #8 -> F17.9 -> ARGUMENT OF PERIGEE (degrees)

WORD #9 -> F17.9 -> MEAN ANOMALY (degrees)
```

IF CARD OMITTED: Planetary moon perturbations will not be applied.

## 2.3.82 PLNEPH

PLNEPH			_
		+2+3+4+5+6 DESCRIPTION DEF	AULT VALUE & UNITS
1-6	A 6	PLNEPH - Allows for correction to the ephemeris planet state vector by adjusting the orbital elements at an osculating epoch	
7-8	12	Indicator of type of orbital elements	2
		1 Keplerian [NOTE 2] 2 Brouwer-Clemence Set III	
9-12	14	Body identifier	none
		0100 = Mercury 0200 = Venus 0300 = Earth 0301 = Earth's Moon 0400 = Mars 0500 = Jupiter 0600 = Saturn 0700 = Uranus 0800 = Neptune 0900 = Pluto 9999 = Sun  0001 < 0002	
COLUMNS	FORMAT	DESCRIPTION DEF	AULT VALUE & UNITS
13-14	12	Element Indicator	none
		For each type of elements the following table indicates a number from 1 to 6	
21-24	14	Year of the osculating epoch: YYYY [NOTE 3]	the beginning time of the run
25-44	D20.8	$\begin{array}{c} {\tt MMDDHHMMSS.SS} \ \ {\tt of the osculating epoch.} \\ {\tt [NOTE 3]} \end{array}$	the beginning time of the run
45-59	D15.3	Element correction (for type and component in cols. 7-8 and 13-14)	0.0

for the osculating epoch

60-72 D13.1 Sigma for the value in cols. 45-59 0.0

73-80 D8.2

Note [1]: In order to adjust a particular element a non-zero sigma should be given for that element. Body ID is required on ALL PLNEPH cards.

Note [2]: Adjustment of Keplerian elements is not implemented at the present.

Note [3]: The user may specify a correction EPOCH later than the end of the arc but not earlier than the start time of the arc.

#### 2.3.83 POLDYN

POLDYN: There have been additions to the POLDYN capability starting with version 1802. A POLDYN card used with a version earlier than 1802 works as before. However, the explanations have changed.

----+---1----+---2----+---3----+---4----+---5---+---6---+---7----+---8
POLDYN 880901 0. 0. 3. 3. 3. ---+---0---+---0---+---0---+---0

COLUMNS FORMAT

DESCRIPTION

DEFAULT VALUE & UNITS

1-6 A6 POLDYN - Used to set options and define parameters associated with two force model effects and one surface deformation effect. The two force model effects are:

(1) variations in the C21 and S21 gravitational coefficients due to variations in the Earth's figure axis and (2) the ocean pole tide, The surface deformation effect is the solid Earth pole tide.

These three effects all depend on the crustfixed path of the Earth's spin axis and the
figure axis of the Earth.

The instantaneous location of the spin axis is
obtained from the polar motion series input
on unit 2. The figure axis however has a complex
motion of which ONLY the secular part is of
interest in the computation of the ocean pole
tide and solid Earth pole tide effects. The
secular component of the path of the instantaneous
spin axis is the same as the secular component
of the path of the figure axis. These are now
modeled as a line (linear path). The history
is sketched in the next paragraph.

Note that in the past, the IERS provided a series of a "filtered" version of their CO1 EOP series, and that was used to approximate the trajectory of the figure axis. After lengthy investigations and a dedicated workshop in July 2017, the IERS has adopted a conventional linear model to describe the secular trajectory of the figure axis. There are two IERS Conventions chapters that involve the new secular pole and products that use it: Ch. 6 (solid and ocean pole tides) and Ch. 7 where the secular pole is actually defined, and the site displacements due to various sources are computed. The final (online) versions of these two re-writes can be obtained by the public from:

Ch. 6 URL: http://iers-conventions.obspm.fr/2010/2010\_update/chapter6/icc6.pdf

Ch. 7 URL: http://iers-conventions.obspm.fr/2010/2010\_update/chapter7/icc7.pdf

It is very important to understand that for the computation of the rotational deformation (whether solid Earth or ocean pole tide), the relative coordinates {m1,m2} of the location of the instantaneous spin axis with respect to the location of the secular figure axis, MUST be obtained from the adopted linear model for the secular path of the figure axis and NOT from a filtered version of CO1 as it was done in the past. This will ensure that products derived by different entities on the basis of space geodetic data from ANY technique, i.e. SLR, VLBI, GNSS or DORIS, will all be consistent. The old IERS "mean pole"is now called the "secular pole" and it is derived from a simple linear fit of the CO1 series over the period 1900 to 2015. Under the new convention, the term "mean" is NOT allowed, since there is nothing that is the "mean" of anything in this realization. The functional form of the model is:

 $Xs(t) = Xs(t0) + Xs_{dot} * (t -t0)$ ; and

 $Ys(t) = Ys(t0) + Ys_{dot} * (t -t0)$ 

where "t" is time in years of 365.25 days and t0 = 2000.0 is the reference epoch for the new IERS secular pole model. The model adopted by the IERS can be realized as follows:

Xs(t0) = 55.00 [mas] $Xs_{dot} = 1.677 [mas/y]$ 

and:

Ys(t0) = 320.50 [mas] $Ys_dot = 3.460 [mas/y]$ 

Column 14 of the POLDYN card is devoted to defining the model used to approximate the figure axis. In addition to that, column 14 dictates how the C21 and S21 gravitational coefficients will be computed.

- 7 I1 If > 0, apply the ocean pole tide 0
- 8 I1 Only for versions 1802 and later OThis column allows the degree of the ocean pole tide model to be extended past 99 (see cols 9-10).
- 9-10 I2 The degree of the ocean pole tide model. 0
  The maximal degree is 60 for versions
  prior to 1802. For versions 1802 and
  later 360 is the max degree

I1 Describe the model used for the secular pole path (its use strongly recommended that the model described below in columns 15-80 be used)

14

0

Also define how the C21 and S21 gravitational coefficients will be computed. It is HIGHLY recommended that the NOTES (further below) be read before selecting an option on this column. The terminology used here is ONLY explained in the NOTES. Also note that options 0, 1, 2 and 3 be avoided. They are here only to explain previous conventions and GEODYN setup decks that may have used these.

- 0 = A linear model is used for the secular path of polar motion. The linear model can be defined in columns 15-80. Only the "second path" contribution to C21 and S21 is computed. The second path contribution is made using GEODYN's original formulation for the second path. This option can be used with or without the presence of a GEOPOL card.
- 1 = The IERS2010 model is used for the secular path
   of polar motion. This model is cubic for
   times from 1976 to 2010, and linear after 2010.
   Only the "second path" contribution
   to C21 and S21 is computed.
   The second path contribution is made using
   GEODYN's original formulation for the second
   path. This option can be used with or without
   the presence of a GEOPOL card.
- 2 = The IERS2010 model is used for the secular path of polar motion. This model is cubic for times from 1976 to 2010, and linear after 2010. Both the "first path" and "second path" contribution to C21 and S21 are computed. The second path contribution is made using IERS2010 equation 6.22. This option should only be used when a GEOPOL card is also present.
- 3 = The IERS2010 model is used for the secular path of polar motion. This model is cubic for times from 1976 to 2010, and linear after 2010. Only the "second path" path contribution to C21 and S21 is computed.

  The second path contribution is made using IERS2010 equation 6.22. This option should only be used when a GEOPOL card is also present.
- 4 = Available only in versions 1802 and later. A linear model is used to describe the secular motion of the figure axis. The linear model is defined in columns 15-80.

Both the "first path" and "second path" contribution to C21 and S21 are computed. The second path contribution is made using IERS2010 equation 6.22. This option should only be used when a GEOPOL card is also present.

5 = Available only in versions 1802 and later.

A linear model is used to describe the secular motion of the figure axis. The linear model is defined in columns 15-80.

Only the "second path" path contribution to C21 and S21 is computed.

The second path contribution is made using IERS2010 equation 6.22. This option should only be used when a GEOPOL card is also present.

15-20	16	Reference epoch "t0"for the linear model of the secular pole in YYMMDD format.	Epoch t EPOCH	
25-44	D20.8	<pre>Xs(t0) value corresponding to the secular pole coordinate at the epoch t0 specified in columns 15-20 (intercept of the linear model for Xs)</pre>	0.	MAS
45-59	D15.3	Ys(t0) value corresponding to the secular pole coordinate at the epoch t0 specified in columns 15-20 (intercept of the linear model for Ys)	0.	MAS
60-72	D12.8	<pre>Xs_dot drift rate for the secular pole model specified in columns 25-44 (slope of the the linear model for Xs)</pre>	0.	MAS/Yr.
73-80	D8.3	Ys_dot drift rate for the secular pole model specified in columns 45-59.	0.	MAS/Yr.

#### Notes:

# Background Geophysics

The ellipsoidal shape of the Earth is caused by Earth rotation. As the rotation axis of the Earth wanders (polar motion), the figure axis of the Earth's ellipsoid changes location too. The wandering of the Earth's figure axis can not quite keep up with the rotation axis. All three effects covered by the POLDYN card are connected with this lagging of the figure axis. To compute where the figure axis is, it is necessary to have a model for the secular path of the Earth's rotation axis. That is because the instantaneous figure axis is about 2/3 of the distance from the actual rotation axis to the secular path of the the rotation axis (closer to the secular path pole). Note that the secular path of the rotation axis is the same as the secular path of the figure axis. So, for the three effects connected with the POLDYN card, the secular pole path model is compared to the location of the actual pole (given by the polar motion series input on unit 2).

## Models for the Secular Path of the Pole

GEODYN allows two secular path models: a linear model for the secular path and the IERS2010 model which is cubic for times from 1976 to 2010, and linear after 2010. Once the model for the secular path is set, then the ocean pole tide and the solid Earth tide deformation models are also set (other than the fact that the user must specify the degree to which the ocean pole tide will be modeled).

# C21 and S21

The remainder of this note is devoted to an explanation of the third effect connected with the POLDYN card, the computation of the C21 and S21 gravitational coefficients. At this point it is worthwhile remembering that C21 and S21 values mainly reflect the separation between two axes: the instantaneous figure axis and the Earth fixed Z axis used to compute the gravity field (there are also some small deformation effects).

Connection Between C21, S21 and GEOPOL

The previous paragraph sets up the connection between the POLDYN card and the GEOPOL card. Most modern GEODYN setup decks include a GEOPOL card. If a GEOPOL card is not present, then the Z axis of the gravity field is the same as the Earth's rotation axis. In this case, the Z axis of the gravity field is not truly Earth fixed. In this case, however, the Z axis of the gravity field is not separated very much from the figure axis. If a GEOPOL card is present, then the Z axis of the gravity field is truly Earth fixed. It is the same Z axis as is used by stations. The separation between the gravity Z axis and the rotation axis is large. The figure axis is near the spin axis, so the separation is very roughly the size of polar motion.

Some of the options given on Col 14 of the POLDYN card are sensible only if a GEOPOL card is present in the deck. Each option on col 14 spells out whether GEOPOL is required or not.

Path 1 and Path 2 to the Figure Axis (from the Z axis of the gravity field)

It is clear that the choice of a model for the secular path of the rotation axis along with the choice of using a GEOPOL card are needed for determining the separation of the figure axis from the Z axis of the gravity field. It is important to keep in mind that the figure axis is about 2/3 of the distance from the actual rotation pole to the secular path pole. Also, C21 and S21 are very roughly (ignoring some deformation effects) given by:

C21 = SQRT(3)\*X\*C20 - X\*C22 + Y\*S22S21 = -SQRT(3)\*Y\*C20 - Y\*C22 - X\*S22 Where the gravitational coefficients are normalized and X and Y are angles in radians. X proceeds west (like X polar motion) from the Z axis of the gravity field to the figure axis and Y proceeds south (like Y polar motion) from the the Z axis of the gravity field.

Note that one can break the computation of C21 and S21 into multiple sub-paths that connect from the Z axis of the gravity field to the figure axis. Each path would have its own X and Y angle. It is only necessary that the sub-paths form a connected complete path.

There are two cases depending on the GEOPOL card (or the lack of one):

Case 1 (the most likely) GEOPOL option has been selected:

The path from the Z axis of the gravity field to the figure axis is broken into two paths:

Path (1): Gravity Z axis to secular path pole

then

Path (2): secular path pole to instantaneous figure axis

If the the contribution from the first path is computed analytically, it would be given by:

Depending on the setting in Col 14, this part1 contribution may be computed by  ${\tt GEODYN}$ . If it is not computed by  ${\tt GEODYN}$ , then it must be modeled and/or estimated using  ${\tt GCOEF}$  and  ${\tt GRVTIM}$  cards.

The contribution from the second path can be modeled using one of two methods depending on how Col 14 is set:

Method (1) for second path contribution is GEODYN's original formulation:

```
C21_part2 = SQRT(3)*C20*Kf*(xp - xp_sec)
S21_part2 = -SQRT(3)*C20*Kf*(yp - yp_sec)
```

where Kf is defined on the POLKF card and should be about 1/3 , The polar motion values are in radians.

Note that when  ${\tt GEODYN}$ 's original formulation for the contribution from the second path is chosen, the contribution from the first path is NOT computed. In this case the user MUST model and/or estimate C21\_part1 and S21\_part1 with a combination of GCOEF and GRVTIM cards.

Method (2) for second path contribution is according to IERS2010 equation 6.22:

```
C21_part2 = -1.333E-9*(m1 + 0.0115*m2)

C21_part2 = -1.333E-9*(m2 - 0.0115*m1)

m1 = xp-xp_sec

m2 = -(yp-yp_sec)
```

Here the polar motion values are in seconds of arc.

Note that when the IERS2010 formulation for C21 and S21 is selected, C21\_part1 and S21\_part1 may be computed by GEODYN depending on Col 14.

The two ways of computing the second path contribution are roughly equivalent except that the IERS2010 approach uses Love numbers and takes into account deformation. The factor of 1.333E-9 has approximately the same factors built into it as GEODYN's original approach (plus a conversion from seconds of arc to radians).

Remember that both approaches for the second path contribution depend on the fact that the second path starts at the secular path pole and heads in the direction of the instantaneous rotation axis.

This is why the factor 1/3 is built into the second path contribution. If the full path went through the actual pole, the second path would start at the actual pole and head towards the secular path pole. The absolute value of the 1/3 factor would change to 2/3 and either the sign of the factor would flip or equivalently m1 would be xp\_sec-xp and m2 would change to -(yp\_sec-yp), Also the computation for the first path would depend on actual polar motion, not the secular path values. It probably makes more sense to use the secular path because the more delicate second path computation (which involves deformation) is smaller when one starts at the secular path, At any rate, this is way the IER2010 conventions have it.

```
Case 2 GEOPOL option has NOT been selected (this is given only to interpret old setups):
```

Nowadays, this would be a rare case. If no GEOPOL card is present, then there is no first path (the overall path would start close to the figure axis) and no need for the C21\_part1 and S21\_part1 computations mentioned in Case 1 (GEOPOL).

As far as C21\_part2 and S21\_part2 go, the IERS210 formulation:

```
C21_part2 = -1.333E-9*(m1 + 0.0115*m2)

C21_part2 = -1.333E-9*(m2 - 0.0115*m1)

m1 = xp-xp\_sec
```

```
m2 = -(yp-yp_sec)
```

can not be used without some modification. It has a factor of 1/3 built into it somewhere. This would need to be changed to -2/3 because the path from the gravity Z axis starts at the actual rotation pole. The modified version of this is NOT available in GEODYN.

The only sensible selection for C21 and S21 computation when there is no GEOPOL card is GEODYN's original formulation:

```
C21\_part2 = SQRT(3)*C20*Kf*(xp - xp\_sec)
S21\_part2 = -SQRT(3)*C20*Kf*(yp - yp\_sec)
```

where Kf is defined on the POLKF card and should be about -2/3,

Using GCOEF and GRVTIM cards When Col 14 Excludes the Part 1 Contribution

When  ${\tt GEODYN}$  computes the part 1 contribution to C21 and S21 the following equation is used:

```
C21\_part1 = SQRT(3)*xp\_sec*C20 - xp\_sec*C22 + yp\_sec*S22
S21\_part1 = -SQRT(3)*yp\_sec*C20 - yp\_sec*C22 - xp\_sec*S22
```

To have GCOEF and GRVTIM cards mimic this the user will need to evaluate the above equations at the epoch of the GRVTIM cards to get the values used on the GCOEF cards for the C21 and S21 coefficients. The rates used on the GRVTIM cards will come from differentiating the secular path model for the pole and then evaluating:

```
C21_part1_dot = SQRT(3)*xp_sec_dot*C20 - xp_sec_dot*C22 + yp_sec_dot*S22
S21_part1_dot = -SQRT(3)*yp_sec_dot*C20 - yp_sec_dot*C22 - xp_sec_dot*S22
```

IF CARD OMITTED: Dynamic polar motion will not be applied.

## 2.3.84 POLE

POLE 0	_	+2+3+4+5+ 780903. 176. +0+0+0+	421.	17	.0842
COLUMNS	FORMAT	DESCRIPTION	DEFAULT	VALUE &	UNITS
1-5	A5	POLE - Modifies polar motion and/or A.1-UT1 values obtained from the BIH tables file on unit 2.			
7	I1	Indicator of whether X of the pole, Y of the pole or $A.1-UT1$ (or any combination of the 3) are to be replaced by the values on this card.			
		Col 7 Values to be replaced			
		= 0			
25-44	D20.8	Date of pole/A.1-UT1 value. (YYMMDD.HH) This must correspond exactly with the BIH date that is being overridden.		0.	
45-59	D15.3	Value of X of the pole.		0.	MAS
60-72	D13.1	Value of Y of the pole.		0.	MAS
73-80	D8.2	Value of A.1-UT1.		0.	S
		ometers; M = Meters ; S = Seconds ; M rees ; RAD=Radians ; AS = Arc seconds; M		-	

IF CARD OMITTED: Values from tables file on UNIT 2 will be used.

## 2.3.85 POLEUT

POLEUT23	3	+2+3+- 860104. +0+0+-	22.72530	0.1890	0.1639							
COLUMNS	FORMAT	DESCRIPTION		DEFAUL	T VALUE & UNITS							
1-6	A6	POLEUT - Used to modi adjustment of true po and A.1-UT1 time diff choses option 4 on co and/or adjustment of	ele coordinates erences. If the use 1.7, then modificat	tion								
7	I1	<ul> <li>No parameter adjust</li> <li>Parameter adjust</li> <li>Third card not parameter</li> <li>Pole adjustment</li> <li>circle passing to by the longitude</li> <li>25-44 of the third</li> <li>Adjusted parameter</li> <li>provided on the</li> </ul>	provided on the third card in this group.									
8	I1	POLEUT card.  = 1 Pole values are A.1-UT1 obtained  = 2 Pole values obta A.1-UT1 obtained		es; ard;	0							
25-44	D20.8	Date in form YYMMDD. F values[1] to which the date is for the midpo specified in columns	is card applies. The $int$ of the $interval$	his	default.							
45-59	D15.3	A.1-UT1 value[1].		See	col. 8. (S)							
60-72	D13.1	X of pole[1].		See	col. 8. (SA)							
73-80	D8.2	Y of pole[1].		See	col. 8. (SA)							
SECOND (												
		+2+3+- 5.0	0.01	0.1	0.1							
+	0	+0+0+-	0+0	+	0+0							

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DEFAULT VALUE & UNITS

COLUMNS FORMAT DESCRIPTION

1-23	23X	Blanks must be present.
24	I1	=1 the units for the below averaging interval is hours 0 =0 the units for the below averaging interval is days
25-44	D20.8	Averaging period[1] in the above units for No default. pole and A.1-UT1 values. The date in columns 25-44 of the first card is for the midpoint of this interval
45-59	D15.3	A priori standard deviation in A.1-UT1. 0.0 (S)
60-72	D13.1	A priori standard deviation in X of pole. 0.0 (SA)
73-80	D8.2	A priori standard deviation in Y of pole. 0.0 (SA)
THIRD C		
+	1	+2+3+5+6+7+8
+	0	+0+0+0
1-24	24X	Blanks must be present.
25-44	D20.8	Longitude of great circle constraint for See col. 7. (DEG) X and Y of pole.
25-44 45-59	D20.8	
		X and Y of pole.  A priori cross correlation between A.1-UT1 0.0 Unitless
45-59	D15.3	X and Y of pole.  A priori cross correlation between A.1-UT1 0.0 Unitless and X of pole.  A priori cross correlation between A.1-UT1 0.0 Unitless
45-59 60-72	D15.3 D13.1 D8.2 CARD	X and Y of pole.  A priori cross correlation between A.1-UT1 0.0 Unitless and X of pole.  A priori cross correlation between A.1-UT1 0.0 Unitless and Y of pole.  A priori cross correlation between X of pole 0.0 Unitless
45-59 60-72 73-80 FOURTH	D15.3 D13.1 D8.2 CARD	X and Y of pole.  A priori cross correlation between A.1-UT1 0.0 Unitless and X of pole.  A priori cross correlation between A.1-UT1 0.0 Unitless and Y of pole.  A priori cross correlation between X of pole 0.0 Unitless
45-59 60-72 73-80 FOURTH	D15.3 D13.1 D8.2 CARD	X and Y of pole.  A priori cross correlation between A.1-UT1 0.0 Unitless and X of pole.  A priori cross correlation between A.1-UT1 0.0 Unitless and Y of pole.  A priori cross correlation between X of pole 0.0 Unitless and Y of pole.
45-59 60-72 73-80 FOURTH	D15.3 D13.1 D8.2 CARD	X and Y of pole.  A priori cross correlation between A.1-UT1 0.0 Unitless and X of pole.  A priori cross correlation between A.1-UT1 0.0 Unitless and Y of pole.  A priori cross correlation between X of pole 0.0 Unitless and Y of pole. 2+3+456
45-59 60-72 73-80 FOURTH	D15.3 D13.1 D8.2 CARD	X and Y of pole.         A priori cross correlation between A.1-UT1
45-59 60-72 73-80 FOURTH 1-24	D15.3 D13.1 D8.2 CARD	X and Y of pole.  A priori cross correlation between A.1-UT1

IF CARD OMITTED: DEFAULT VALUES GIVEN ABOVE ARE USED.

- NOTE: [1] Values provided in columns 45-80 of first card are differenced with interpolated table values for specified date (cols. 25-44) and differences are applied as a discrete offset over entire averaging interval (second card cols. 25-44).
- NOTE: [2] Normally (GEODYN default) the EOP at a given time t are computed in GEODYN from the table values using biquadradic interpolation. When option 4 on col. 7 s chosen then EOP rates are requested and the model changes. In this case to compute the EOP at a given time t, we use the linear model

UNITS: DEG-DEGREES; S-SECONDS; SA-SECONDS OF ARC

IF CARD OMITTED: DEFAULT VALUES GIVEN ABOVE ARE USED.

#### 2.3.86 POLKF

+	2+3+	4+-	5+-	6+	8
POLKF	66667				
+	0	0+-	0+-	0+	0+0

0.

COLUMNS FORMAT DESCRIPTION DEFAULT VALUE & UNITS

1-6 A6 POLKF requests application and or adjustment of the Kf factor described in the POLDYN documentation (see Note(1)).

25-44 D20.8 Kf factor. [NOTE 1,2] -2./3.

45-59 D15.3 Standard deviation for Kf.

NOTE[1]: This option is tightly connected to the GEOPOL and POLDYN options. The documentaion for the GEOPOL option should be read first, then the documentation for the POLDYN card should be read.

NOTE[2]: If a GEOPOL card is present, Kf must be > 0. If Kf < 0 is desired, use GEOPOL1. See GEOPOL card documention.

UNITS: KM =Kilometers; M =Meters ; S =Seconds ; M/S=Meters per second MAS=Milliarc seconds

IF CARD OMITTED: The default value of -2./3. will be used for the figure axis scale factor. Dynamic polar motion will not be applied unless the POLDYN option is selected.

#### 2.3.87 POLTID

	2+	3+	4	5+	-6+	78
POLTID						
	0+	0+	0+	0+	-0+	0+0

COLUMNS FORMAT DESCRIPTION

DEFAULT VALUE & UNITS

1-6 A6 POLTID - Request application of pole tide on station positions. THIS REQUIRES THE DEFINITION OF A MODEL FOR THE MEAN PATH OF POLAR MOTION. This model must be specified on the POLDYN card. Unfortunately, versions prior to 1802 will proceed to compute an invalid POLTID correction without a POLDYN card.

IF CARD OMITTED: The pole tide correction will not be applied.

## 2.3.88 PRNTVU

PRNTVU	222	+2+3+4+5+		•
·	· ·			
COLUMNS	FORMAT	DESCRIPTION	DEFAULT	VALUE & UNITS
1-6	A 6	PRNTVU - Requests suppression of output of selected GEODYN-IIS and GEODYN-IIE printout items to unit 6.		
7-8	12	Specifies either the number of lines per page to be output on unit 6. 6 - 61 lines per page. 8 - 81 lines per page. >20 - specific number of lines per page.		61
9-17	9I1	For columns 9-17 the following applies:		
		Blank or zero - Default will apply 1False GEODYN-IIS output suppressed 2True Requests GEODYN-IIS output.	l.	
9	I1	Simple list of GEODYN-IIS setup.	2	TRUE.
10	I1	Interpretive list of GEODYN-IIS setup.	1	FALSE.
11	I1	Observation block selection report.	1	FALSE.
12	I1	Gravity model coefficients.	2	TRUE.
13	I1	Global parameter values and sigmas.	2	TRUE.
14	I1	Arc parameter values and sigmas.	2	TRUE.
15	I1	Sea surface topography.	1	FALSE.
16	I1	Ocean Tide Model.	2	TRUE.
17	I1	Reserved for GEODYN-IIS.		
COLUMNS	FORMAT	DESCRIPTION	DEFAULT	VALUE & UNITS
18-24	711	For columns 18-24 the following applies:		
		Blank or zero - Default will apply 1False GEODYN-IIE output suppressed 2True Requests GEODYN-IIE output.	l.	
18	I1	Simple list of GEODYN-IIS setup.	2	TRUE.
19	I1	Values of estimated E-biases.	1	FALSE.

20	I1	E-matrix labels in Summary Page.	1FALSE.
21	I1	Adjusted station baselines.	1FALSE.
22	I1	Correlations for adjusted parameters.	2TRUE.
23	I1	Shadow crossing.	1FALSE.
24	I1	Reserved for GEODYN-IIE.	

 $\label{eq:card_omitted} \mbox{IF CARD OMITTED: Defaults apply.}$ 

## 2.3.89 PUNCH

PUNCH	_	+2+3+4+5+	
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A6	PUNCH - Requests punched output on unit 7 for complete setup deck or for adjusted parameters only.	
8	I1	Punched output control.  = 0 Requests that complete setup deck be punched with all estimated parameters replaced by the adjusted values.  = 1 Requests that only updated parameters be punched.	0
9	I1	<pre>Iteration control for punched output  = 0 Requests that punching occur on last inner of each global iteration  = 1 Requests that punching occur on all iterations</pre>	0

IF CARD OMITTED: No punched output will be generated.

### 2.3.90 RAYTID

	-1+	2	+	3+-	4	-+5-	+	-6+-	7	-+8
RAYTID										
	-0+		+	0+-	0	-+0-	+	-0+-	0	-+0

COLUMNS FORMAT DESCRIPTION

DEFAULT VALUE & UNITS

1-6 A6 RAYTID - Request application of the IERS convention Earth Tides and the Doodson Ocean Tide model

IF CARD OMITTED: The default tide models will be applied. See ETIDES and OTIDES options.

#### 2.3.91 REFRAC

---+---1---+---2---+---3---+---4----+---5----+---6----+---7----+---8 REFRAC 1 ---+---0---+---0---+---0----+---0 COLUMNS FORMAT DESCRIPTION DEFAULT VALUE & UNITS 1-6 A6 REFRAC - Specifies the tropospheric model to be used for tropospheric corrections to the observations. 8 Ι1 Model selection number = 0 Hopfield Model = 1 Marini Murray Model [NOTE 1] = 2 VLBI/GPS = 3 GPS Model using Niell mapping functions = 4 Marini Murray for laser data Model 03 for nonlaser data = 5 Mendez (Porto) model for laser data = 6 GPT is used to calculate pressure and temperature; Hopfield model is used to calculate the apriori hydrostatic zenith delay; GMF is the mapping function. = 7 Option 7 is the same as option 6 except that Saastamoinen model is used to calculate the apriori hydrostatic zenith delay. = 8 VMF1 [NOTE 3] [NOTE 2] 9 0 Ι1 Correction Component Applied For all models 0-4This correction pertains to the wet and dry correction carried on the 3rd and 4th word of the observation corrections record. These corrections will be added to the sum of observation corrections which later on in GEODYN will be subtracted from the residual. = 0 Wet and Dry = 1 Wet Only = 2 Dry Only (!NOTE!) For the time being (GEODYN version 0812) ONLY the index=0 is true to the description. As soon as the problem will be fixed, this comment will be removed. \*\*(For Mendez Model only)\*\* =0 Use Mapping Function dependent on latitude, height and temperature =1 Use Mapping Function with no dependence on meteorological data 10 Scale Factor Adjustment Ι1 0 For all models 0-7= 0 K\*(Wet + Dry)

= 1 K\*Wet Only

11	I1	Wet Component Mapping Function (Model 2 only) = 0 Chao = 1 CFA2.2	0
12	I1	Dry Component Mapping Function (Model 2 only) = 0 Chao = 1 CFA2.2	0
13	I1	SWITCH FOR PRINT OF TROP EBIAS AND ZENITH PATH DELAYS = 0 No print out = 1 Print zpds and trop EBIAS (UNIT 400)	0
15-17	13	Measurement type	

#### NOTES:

[1] The Marini Murray model is only available for laser range data. If the Marini Murray model is selected (column 8 =1) and data other than laser range is present, the Hopfield model will be used for all non laser range data and the Marini Murray model will be used for the laser range data.

[2]	MODELS	NON-LASER	LASER
	0	HOPFIELD	HOPFIELD
	1	HOPFIELD	MARINI-MURRAY
	2	VLBI/GPS	VLBI/GPS
	3	GPS/NEIL	GPS/NEIL
	4	GPS/NEIL	MARINI-MURRAY
	5	HOPFILED	MENDEZ
	6	GMF/HOPFIELD	GMF/HOPFIELD
	7	GMF/SAASTAMOINEN	GMF/SAASTAMOINEN
	8	VMF1	VMF1

[3] VMF compressed data files may be found on xyz2 in the following directory:

/users/geodyn/SUPPORT/dat\_FILES/dat\_VMF/VMF\_GRID\_FILES
The user must select the VMF file that covers their arc.
The name of the VMF file features the covered dates.
The VMF files usually cover several months and have adequate overlap to cover all date ranges.

The presence of an ascci file with the name 'orography\_ell.txt' IS REQUIRED in the directory where GEODYN IIE is executed. The file may be found in the same directory as above /users/geodyn/SUPPORT/dat\_FILES/dat\_VMF/VMF\_GRID\_FILES OR on this documentation main page under the name : VMF OROGRAPHY FILE (MEAN GRID VALUES)

When using the VMF refraction model in GEODYN IIS will produced an interface file with the name 'vmf\_interface\_file' which will contain the data in the time range requested by the user.

If any of the two files , vmf\_interface\_file, and orography\_ell.txt are not present or not created, GEODYN will terminate with a

comprehensive error message.

IF CARD OMITTED: Hopfield model will be used for non laser range data.

(!NOTE!) The default below is not accurate as of GEODYN 0812. When the default for laser data is fixed, we will remove this comment.

Marini Murray model will be used for laser range data.

#### 2.3.92 REL300

	1	-2+	3+-	4	5+	6+-	7+	8
REL300								
+	0	-0+	0+-	0+	0+	0+-	0+	0

COLUMNS FORMAT DESCRIPTION

DEFAULT VALUE & UNITS

1-6 A6 REL300 - Requests that relativistic corrections be applied for earth orbiting satellites. This includes general relativistic light time corrections to the measurement model, general relativistic point mass accelerations, and the Lense-Thirring effect, and the relativistic coriolis force.

UNITS: KM =Kilometers; M =Meters ; S =Seconds ; M/S=Meters per second MAS=Milliarc seconds

IF CARD OMITTED: Relativistic corrections will not be applied.

#### 2.3.93 RELPPN

	1	+2	3	+	4	5	-+6-	+		-+8
RELPPN				1	.D+00		1.D-30		1.D+00	1.D-30
	0	+0	+	+(	0+-	0	-+0-	+	0	-+0
COLUMNS	FORMAT		DESCRIPTION				DE	FAUL	T VALUE	& UNITS
1-6	A6	RELPPN	- Introduces	the 1	PPN rel	lativity	model.	The	third-bo	dy

1-6 A6 RELPPN - Introduces the PPN relativity model. The third-body perturbation of each body (Sun, planets and asteroids) is derived from the isotropic, parametrized post-Newtonian (PPN) n-body metric.

The acceleration of the spacecraft and the integrated body due to interaction with other point masses follows the formulation reported in Eq. 27 of:

"The Planetary and Lunar Ephemerides DE430 and DE431" by Folkner et al. 2014.

There are two parameters associated with this model, Gamma and Beta
[NOTE 1]
Use RELPPN only for Interplanetary satellites [NOTE 2]

25-44 D20.8 Value of Gamma

45-59 D15.3 Sigma for Gamma

60-72 D13.1 Value of Beta

73-80 D8.2 Sigma for Beta

- NOTE 1. The presence of both RELPPN and REL300 in a GEODYN setup sets RELPPN to be the used model. GEODYN will not double count the relativistic effect.
- NOTE 2. RELPPN cannot be used for Earth orbiting satellites because of reference frame computation issues.

#### 2.3.94 SATCUT

	2+3+	4	5+6	8
SATCUT				
	0+	0+	00	0

COLUMNS FORMAT DESCRIPTION

is 0

DEFAULT VALUE & UNITS

1-6 A6 SATCUT - Gives the user the ability to set an elevation cut off angle for satellite to satellite links in measurements. The cut off angle for satellite to satellite links is reckoned from the horizon of the lower satellite. The default value

## 2.3.95 SCBODY

	1	+2+3+4+5+	-6+	78
SCBODY	ı		0 1	, , ,
	0	+0+0+0	-0+-	0+0
COLUMNS	FORMAT	DESCRIPTION	DEFAULT	VALUE & UNITS
1-6	A6	SCBODY - Allows the user to overwrite values and request adjustment of physical parameters of a celestial body featured in the EPHEM2 global option. (Supplementary Planetary Ephemeris)		
7	I1	<pre>Gravity model initialization indicator. = 0 Gravitational coefficients are    initialized from the gravity model    file.</pre>		
		The option to modify the coefficients using GCOEF, GCOEFC, and GCOEFS cards is not yet applicable.		
		> 0 (See PLANET card) Not Available yet.		
8-10	13	Celestial body sequence in the supplementary planetary sequence.		
11-17	17	Celestial body ID.		
18-20	13	Max degree of coefficient to be used in the gravitational model.		
21-23	12	Max degree of coefficient to be used in the gravitational model.		
24	I1	<pre>Indicator of the contents of the real fields (col 25-80). = 0 Columns 25-80 contains the values of     GM, A, 1/fp, 1/fe</pre>		
		= 1 Columns 25-80 contains the sigmas of GM, A, 1/fp, 1/fe		
25-44	D20.8	Universal gravitational constant times the mass of the body (GM) or sigma value if column $7 = 1$		
45-59	D15.3	Semi-major axis of the body (A) or sigma value if column $7 = 1$		
60-72	D20.8	Inverse of the body's polar flattening $(1/f)$ or sigma value if column 7 = 1	p)	
73-80	D20.8	<pre>Inverse of the body's equatorial flattening (1/fp) or sigma value if column 7 = 1</pre>		

## 2.3.96 SHDEND

1-	+2	+3	4-	+5-	+6	+8
SHDEND						
+0-	+0	+0	+0-	+0-	+0	+0

COLUMNS FORMAT DESCRIPTION DEFAULT VALUE & UNITS

1-6 A6 SHDEND - Denotes the end of the GPSSHD group

IF CARD OMITTED: Abnormal termination

#### 2.3.97 SSCOEF

+1	+	-2-	+3+-	4	+	-5+	6	+7-	+8
SSCOEF2	2	1					1.00	00000D-0	91.00D-09
+0	+	-0-	+-	0	+	0+	0	+0-	+0

•	O		
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A6	SSCOEF - Modifies and/or requests the estimation of spherical harmonic coefficients in static sea surface topography model.	
7	I1	Normalization indicator [NOTE 1]	1
		<pre>= 1 values normalized C,S = 3 values unnormalized C,S [See NOTES]</pre>	
15-17	13	Degree of C and S coefficients (N index).	0
18-20	13	Order of C and S coefficients (M index).	0
25-44	D20.8	A priori (or starting) value of C coefficient. [NOTE 2]	0.
45-59	D15.3	A priori (or starting) value of S coefficient. [NOTE 2]	0.
60-72	D13.1	Standard deviation of C coefficient. No adjustment if this field is zero.	0.
73-80	D8.2	Standard deviation of S coefficient. No adjustment if this field is zero.	0.

#### NOTES:

[1] The denormalization subprogram in GEODYN underflows at degree 49 and order 48.

The geopotential coefficient denormalization equation used is:

$$C_n^m = \bar{C}_n^m \times D_n^m$$
$$S_n^m = \bar{S}_n^m \times D_n^m$$

where

$$D_n^m = \sqrt{(4n+2)\frac{(n-m)!}{(n+m)!}} \text{ for } m > 0$$
  
 $D_n^m = \sqrt{2n+1} \text{ for } m = 0$ 

and

 ${\cal C}_n^m$  ,  ${\cal S}_n^m$  are un-normalized coefficients.

 $ar{C}_n^m$  ,  $ar{S}_n^m$  are normalized coefficients.

The denormalization subprogram in GEODYN underflows at degree 49 order 48.

- [2] If the degree and order on the SSCOEF card must be less then or equal to the maximum degree and order as input on the SSTOPO card.
- [3] If it is desired that the starting values of coefficients be different than the a priori values for those coefficients then the SSCOEF card should be used as follows:
  - o SSCOEF cards requesting the desired coefficient adjustments should be included in the setup deck. The a priori values will be indicated on these cards.
  - o For each coefficient pair for which the starting value is to be different than the a priori value, an additional SSCOEF card should be in the setup deck. This SSCOEF card will indicate the starting value and come later in the deck than the corresponding SSCOEF card requesting adjustment.

IF CARD OMITTED: Static Sea Surface Topography model will be null.

#### 2.3.98 SSTCOF

1		2	3	4	5	6+8	3
SSTCOF							
0	(	0+	0+	0	0(	0+(	)

COLUMNS FORMAT DESCRIPTION DEFAULT VALUE & UNITS

- 1-6 A6 SSTCOF Provides the apriori values for the coefficients needed for sea surface topography height evaluation.
- 25-44 D20.8 A priori value for coefficient.
- 60-72 D13.1 Standard deviation for coefficient. 0.

  No adjustment if this field is zero.
- NOTE [1]: A stream of SSTCOF cards should follow the SSTMOD card (See SSTMOD Vol. 3).
- $\begin{tabular}{ll} \textbf{IF CARD OMITTED:} & \textbf{And SSTMOD has been requested the run will terminate} \\ & \textbf{abnormally.} \\ \end{tabular}$

## 2.3.99 SSTMOD

+ SSTMOD	1	+2+3+4+5+	-6+	78
+	0	+0+0+0+	-0+	0+0
COLUMNS	FORMAT	DESCRIPTION	DEFAULT	VALUE & UNITS
1-6	A6	SSTMOD - Introduces the application/ estimation of a sea surface topography model which employs a number of functions evaluated at several points located on a uniform grid on the globe. The sea surface topography height at a given location is evaluated by interpolation and represents the height from the geoid to the mean sea surface. The user may input the grid file either in IIS or in IIE. For large grid files IIE input is recommended.		
7-8	12	= 0 Function file input from IIE [2]	0	27,28
		= 1 Function file input from IIS [3]		21
9-10	12	Input file number for the Proudman function SST coefficients. Recommended units are 45-49		
11-12	12	Number of real variables on one record of the above input file. Each record should include all the coefficients that describe one function in the following order: constant term periodic terms -		
13-14	12	Internal geodyn number that describes the function model		
		<pre>1 = constant term only 8 = periodic terms for one frequency 9 = periodic terms for two frequencies 10 = periodic terms for three frequencies</pre>		
15-17	13	Index for printing a coefficient report on unit 6 of the GEODYN IIS output		
		=0 No report will be printed		
		=1 A coefficient report will be printed		
25-44	D20.8	Period for model 8 (sec)		
45-59	D15.3	Period for model 9 (sec)		
60-72	F13.1	Period for model 10 (sec)		

- 73-80 F8.1 Editing factor for boundary locations editing. This must be a real number between 0.D0 and 1.D0.
- NOTE [1]: This card must be followed by a number of SSTCOF cards which represent the total number of functions evaluated per grid point.
- NOTE [2]: Files input from IIE:
  Two files (units 27 and 28) must be brought into the directory
  where GEODYN will execute. These files which are specific to the
  machine on which they are used, represent the direct access
  function file (27) and the matching grid file (28). See Vol. 5
  Sect. 2.10 for a description of the files
- NOTE [3]: File input from IIS: unit 21

  See Vol. 5, Section 2.10 for description of the file.
- $\begin{tabular}{ll} \textbf{IF CARD OMITTED:} & \textbf{The above described Sea surface topography model will} \\ & \textbf{not be applied.} \\ \end{tabular}$

# 2.3.100 SSTOPO

SSTOPO	-	+2+3+4+5+	
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A6	SSTOPO - Defines the altimetric sea surface topography model maximum degree and order and the epoch time for the time dependent terms.	
8	I1	Sea surface topography model print control switch.	2
		<pre>= 1 Print entire model. = 2 Print only model checksum.</pre>	
15-17	13	Maximum degree of coefficient to be used in the spherical harmonic expansion of the sea surface topography model.	3
18-20	13	Maximum order of coefficient to be used in the spherical harmonic expansion of the sea surface topography model.	2
25-44	D20.3	Epoch time in YYMMDDHHMMSS.SS for the time dependent sea surface topography linear rate and periodic terms.	

IF CARD OMITTED: Sea surface topography computations will not be performed.

## 2.3.101 SSTPRD

+ SSTPRD	1	+2+6	-+8
	0	+0+0+	-+0
COLUMNS	FORMAT	DESCRIPTION DEFA	ULT VALUE & UNITS
1-6	A6	SSTPRD - Modifies and/or requests the estimation of the periodic sine and cosine coefficients for the C or S spherical harmonic coefficients in sea surface topography model. Defines the periodic frequency.	
7	I1	Normalization indicator	1
		<pre>= 1 values normalized C,S = 3 values unnormalized C,S</pre>	
8	I1	Coefficient type	0
		= 1 Coefficient of Cosine (C) = 2 Coefficient of Sine (S)	
9	I1	<pre>Frequency/Period indicator (frequency=2*pi/period)</pre>	1
		= 1 Annual (365.2524 days) = 2 Semi-annual (182.6262 days) = 3 Seasonal (91.3131 days) = 4 Monthly (30.4377 days) = 5 To be defined in columns 10-14	
10-12	13	Integer number of days in period	0
13-14	12	Fractional number of days in period	0
15-17	13	Degree of C or S coefficients	0
18-20	13	Order of C and S coefficients	0
25-44	D20.8	A priori value of the cosine coefficient (a)	0.
45-59	D15.3	A priori value of the sine coefficient (b)	0.
SSTPRD		+2+6 +0+0+0+0	
COLUMNS	FORMAT	DESCRIPTION DEFA	ULT VALUE & UNITS
60-72	D13.1	Standard deviation of the cosine coefficient. No adjustment if this fields is zero.	0.
73-80	D8.2	Standard deviation of the sine coefficient (b). No adjustment if this field is zero.	0.

## $\boldsymbol{2.3.102}\quad \mathbf{SSTTIM}$

SSTTIM		+2+3+4+5+	
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A6	SSTTIM - Modifies and/or requests the estimation of the mean and secular spherical harmonic coefficients in sea surface topography model.	
7	I1	Normalization indicator	1
		<pre>= 1 values normalized C,S = 3 values unnormalized C,S</pre>	
8	I1	Coefficient type	0
		= 1 Coefficient of Cosine (C) = 2 Coefficient of Sine (S)	
15-17	13	Degree of C or S coefficients	0
18-20	13	Order of C and S coefficients.	0
25-44	D20.8	A priori (or starting) value of the mean Cbar or Sbar coefficient.	0.
45-59	D15.3	A priori (or starting) value of the Cdot or Sdot linear rate coefficients.	0.
60-72	D13.1	Standard deviation of the mean Cbar or Sbar coefficent. No adjustment if this field is zero.	0.
73-80	D8.2	Standard deviation of the secular Cdot or Sdot coefficent. No adjustment if this field is zero.	0.

NOTE: Card must be used in conjunction with SSTOPO card. Card cannot be used in conjunction with SSCOEF card.

## 2.3.103 STIERS

+1	-+2	+3	+4	+5	+6	-+8
STIERS						
+0	-+0	+	+0	+0	+0	-+0+0

COLUMNS FORMAT DESCRIPTION

DEFAULT VALUE & UNITS

1-6 A6 STIERS - Request application of the IERS (DEHANT) model for solid earth tides

IF CARD OMITTED: The default default simple model will be applied.

## 2.3.104 TARGET

TARGET	_	+2+5+		
COLUMNS	FORMAT	DESCRIPTION	DEFAULT	T VALUE & UNITS
1-6	A 6	TARGET - Introduces coordinates of a station on a planet other than the earth.		
7	I1	Type of planet centered fixed coordinates	0	
		<pre>=0 - X,Y,Z (Cartesian) =1 - latitude,longitude,radius (Spherical) =2 - axis, longitude, height (Cylindrical)</pre>		M degrees or M degrees or M
18-24	17	Target ID number.		
25-44	D20.8	X (meters), $\varphi$ (latitude in degrees) or A (a	xis in 1	meters)
45-59	D15.3	Y (meters) or $\lambda$ (west longitude in degrees)		
60-72	D13.1	Z (meters), r (radius in meters) or H (heigh	ht in me	eters)

# 2.3.105 TARSIG

TARSIG	1+	2+3+4+5+6+7+8
+	0+	0+0+0
COLUMNS	FORMAT	DESCRIPTION DEFAULT VALUE & UNITS
1-6	A 6	TARSIG - Introduces sigmas for the coordinates of a planetary station (Target).
18-24	17	Target ID number.
25-44	D20.8	sigma of X (meters), $\phi$ (latitude in degrees) or $A$ (axis in meters)
45-59	D15.3	sigma of Y (meters) or $\lambda$ (west longitude in degrees)
60-72	D13.1	sigma of Z (meters), r (radius in meters) or H (height in meters)

IF CARD OMITTED: And measurement types 37, 38, 47 or 48 are being processed planetary target coordinates will not be solved for.

## 2.3.106 TERMVU

TERMVU	222	+2+3+4+5+		
·	Ü			v
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNI	TS
1-6	A6	TERMVU - Requests output of selected IIE printout items to unit 9 where the linesize is 80 characters.		
		For columns 9-18 the following applies:		
		Blank or zero - default will apply 1False means no output 2True requests output		
9	I1	Requests output of GEODYN-IIS setup.	1FALSE.	
10	I1	Iteration control .	2TRUE.	
		= 1 Requests items selected by columns 11-15 be output on each iteration.		
		= 2 Requests items selected by columns 11-15 be output on last inner iteration of each global iteration.		
11	I1	Requests output of observation residuals.	1FALSE.	
12	I1	Requests output of station residual summary.	1FALSE.	
13	I1	Requests output of observation type residual summary.	2TRUE.	
14	I1	Requests output of E-BIAS adjustment.	1FALSE.	
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNI	TS
15	I1	Requests output of arc parameter adjustments.	2TRUE.	
16	I1	Requests output of summary page.	2TRUE.	
17	I1	Requests output of global parameter adjustments.	2TRUE.	
18	I1	Requests output of arc global update.	2TRUE.	

IF CARD OMITTED: Default values will be used to control output to UNIT 9.

## 2.3.107 TIDES

	1	+2+3+4+5+	6	+7	+8
TIDES 2	0	. 29			
+	0	+0+0+0+	0	+0	+0
COLUMNS	FORMAT	DESCRIPTION	DEFAU	LT VALUE	& UNITS
1-5	K2 may	TIDES - Modifies Earth tide perturbation and requests application of pole tide. The default model for standar es is the one that applies K2 amplitude, phase and K3 amplitude. An extended model also be used in GEODYN. The description lows the default model	КЗ	amp1/K2 phase	phase/
7-8	12	Indicates which tidal coefficient to be modified.		0	
		VALUE COEFFICIENT 20 K2 amplitude 21 K2 phase 30 K3 amplitude			
		A value of 0 sets all coefficients (K2 amplitude, K2 phase and K3 amplitude) to zero. [1]			
9-10	12	A value of 1 means that the pole tide is applied to correct the station positions. (stations only)		0	
25-44	D20.8	Value of tidal parameter being modified. Defaults:			
		COEFFICIENT VALUE K2 amplitude 0.29 K2 phase 2.5 degrees K3 amplitude 0.0		0.29 [2] 2.5 [2] 0.	DEG
45-59	D15.3	Standard deviation of tidal coefficient. If non-zero value is specified, tidal parameter will be adjusted.		0.	
TIDES	2 0	+2+3+4+5+ 2 00.30190	1.0		1.0
1-5	<b>A</b> 5	TIDES - Description of the extended model			
7-8	12	Must be zero to assume the extended model.			
11-12	12	Forcing degree			

- 13-14 I2 Forcing order
- 21-22 I2 Receiving degree
- 23-24 I2 Receiving order
- 25-44 D20.8 "A" coefficient. 0.
- 45-59 D15.3 "B" coefficient. 0.
- 60-72 D13.1 Sigma "A". [NOTE 3] 0.
- 73-80 D8.2 Sigma "B". [NOTE 3] 0.
- NOTE [1]: It is sufficient to turn on the pole tide in one card (even if more TIDES option cards are included in the setup).
- NOTE [2]: These default values apply only to earth tracking stations. In case of other planetary stations the default values are zero.
- NOTE [3]: Convention for adjusting standard tides:
- 1. If there is no sigma on any TIDES card, there will be no adjustment of the standard tides.
- 2. If there is a sigma in one of the TIDES cards, GEODYN applies that sigma to the parameter, and all the other parameters will be assigned a sigma= 1.D-20
- UNITS: KM =Kilometers; M =Meters ; S =Seconds ; M/S=Meters per second DEG=Degrees ; RAD=Radians ; AS =Arc seconds; MAS=Milli-arc seconds
- IF CARD OMITTED: Default values given above are used, and pole tide will not be applied.

# 2.3.108 TOLS

TOLS	-	+2+3+4+5+-	
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A6	TOLS - Requests changes in integration tolerances.	
11-14	13	Maximum degree of geopotential used in computing the variational equations.	12 Except = maximum degree of geo- potential whenever E-matrix requested.

IF CARD OMITTED: Defaults apply.

#### 2.3.109 TRKBDY

---+---1----+----3----+----4----+---5----+---6----+----7----+----8
TRKBDY 300
----+---0---+---0---+---0---+---0----+---0

COLUMNS FORMAT

DESCRIPTION

DEFAULT VALUE & UNITS

1-6 A6 TRKBDY - Sets the planet number of the tracking body, i.e. body on which are located the tracking stations.

11-14 I4 Indicates body number for the tracking body.

Body numbers are as follows:

0100 Mercury 0200 Venus 0300 Earth

0301 Earth's Moon

0400 Mars 0500 Jupiter 0600 Saturn 0700 Uranus 0800 Neptune 0900 Pluto 9999 Sun

UNITS: M = Meters ; S = Seconds

IF CARD OMITTED: Then Earth (0300) is used as the tracking body.

### 2.3.110 TRPOUT

1-	+	-2+	3+	4	-+5	-+6	-+7	+8
TRPOUT	300							
0-	+	-0	0+		-+0	-+0	-+0	+0

COLUMNS FORMAT DESCRIPTION

DEFAULT VALUE & UNITS

1-6 A6 TRPOUT - Requests estimation of tropospheric biases. The estimated tropospheric biases will be added to the original observation, and the sum of the observation corrections.

> The new osbservation and sum of observations corrections will then be output on unit 20 on a GEODYN II binary data file. This option works only with the presence of the global option G2BOUT.

#### 2.3.111 VLIGHT

VLIGHT		299792500.				
+	0	+0+	0+	0	-+0	
COLUMNS	FORMAT	DESCRIPTION	DEFAULT	VALUE	& UNITS	
1-6	A6	VLIGHT - Modifies and/or requests adjustment of speed of light.				
25-44	D20.8	Current best value of speed of light.[NOTE 1] If = 0, value will default to a priori speed of light.	29979	2458.	M/S	
45-59	D15.3	A priori value of speed of light. [NOTE 1]	29979	2458.	M/S	
60-72	D13.6	Standard deviation of speed of light.	0.		M/S	

### NOTES:

[1] Speed of light values will be checked for reasonableness and will be ignored if different from default value by more than 1 part per million.

If greater than zero speed light will

IF CARD IS OMITTED: Default value will be used.

be adjusted.

## 2.3.112 **VECOPT**

VECOPT	5 40	+2+3+4+5+- 10	20000.
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A 6	VECOPT - Vector optimization and core allocation control option.	
7	I1	Controls vector optimization for the chaining of partials and the formation of the normal matrix.	<pre>0 if no EMATRX card, 2 if EMATRX card in   setup.</pre>
		=0 Optimizes over the number of observations	
		>0 Optimizes over the number of estimated parameters.	
		>1 Forms normal matrix by partitions.[3]	
9-10	I2	Indicates number of scratch units available for storage of partials when normal matrix is formed by partitions. Limit of 20.	1
		First scratch unit will be 51.	
		Subsequent scratch units will increment by one up to 70.	
11-14	14	Controls the maximum number of observations that will be used in one data block (minimum no. = 11, maximum = 9999) [1,2].	1000 if no EMATRX card, 40 if EMATRX card in setup.
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
15-17	13	Number of integration steps over which interpolation may be simultaneously performed. Must be greater than or equal to integration order minus one.	Integration order minus one [2].
		Increasing this number allows more observations per block [2].	
		This is of particular advantage when MBIAS, PBIAS or EBIAS options are invoked along with local pass editing (see EDIT card column 7).	
		This options consumes considerably more memory and is therefore disadvantageous	

to large E-Matrix generation runs.

25-44 D20.8 Maximum amount of memory available for 2E job. For CRAY runs this would be the same amount of memory requested on your QSUB card. For Cyber runs 5100000 64 bit words is suggested.

7000000 64 bit words

73-80 F8.1 Number of blocks (of 512 64-bit words) of disk space requested for each normal matrix partitioned formation scratch file.

20000.

A good estimate for the total disk space required for all of these scratch files may be obtained by multiplying the number of weighted observations plus twice the number of E-biases by the number of adjusted parameters and dividing by 512.

#### NOTES:

- [1] The Tracking Data Formatter program outputs a maximum of 9999 obs. per block. GEODYN-IIS may reduce the number of observations in a block but not increase it. By reducing the number of observations in a block, memory requirements are decreased. This may be required when a large number of parameters are estimated.
- [2] Values in columns 11-14 and 15-17 are automatically provided by NORMPT option.
- [3] Partitioning is automatically requested for E-matrix runs regardless of the value indicated in column 7.

IF CARD OMITTED: Default optimizations will be used.

#### 2.3.113 XEPHEM

XEPHEM 2		-2+3+4 920324000000.	920325000000.	-6+	78
+	0+	-0+0+0		-0+	0+0
COLUMNS	FORMAT	DESCRIPTION		DEFAULT	VALUE & UNITS
1-6	A6	XEPHEM - Indicates that ephemeris will be provid the measurement model ro determining the orbits.	ed for use in utines for		
7-8	12	Number of satellite ephe on external ephemeris. T satellites are included record of the ephemeris	he ID's for the in the header	l	1
11-14	14	Size of internal buffers satellite ephemeris info			300
15-16	12	Unit number for the file external ephemeris. This in IIE. IIS does not ver exists, nor does it veri actually contains the nu or time periods specifie if the information is in	unit is read or ify that the file fy that the file mber of satellited. IIE will term	nly Le e ces	66
25-44	D20.8	Start time for the exter form: YYMMDDHHMMSS. Star least three time steps p point for the 10 point h	t time must be a	at st observ	
45-59	D15.3	End time for the externa form: YYMMDDHHMMSS. End five time steps after th the 10 point hermitian i	time must be at e last observati	least on for	

- Note [1]: The external ephemeris file contains a cartesian trajectory for some or all of the satellites in the run. All satellites in the run are still specified after the REFSYS card with each satellite requiring a SATPAR, an EPOCH and two ELEMS cards. A 10th order hermitian interpolator is used to interpolate the satellite trajectory to the epochs required for measurement modeling.
- Note [2]: In most cases the coordinate system used for the trajectories on the external ephemeris file is the True of Date coordinate system associated with the central body with the origin at the center of mass of the central body. However starting with version 1507.12, the XEPHEM option can be used to input and replace the trajectory of the central body when the INTGCB capability is being used. When the XEPHEM option is being used in conjunction with the INTCB option, the trajectory coordinate system is in the True of Reference coordinate system (usually J2000 Earth Equator

and Equinox) with the origin at the center of mass of the sun.

- Note [3]: When using the XEPHEM option the user should remember that although GEODYN is using the input ephemeris for measurement modeling, the trajectory and associated force model partials are still being computed and folded into the normal equations, To avoid a degraded least square solution, all force model parameters associated with a satellite on the external ephemeris should be tightly constrained.
- Note [4]: IIE uses a double buffering scheme where each buffer is (6 \* No. of satellites \* Buffer size) 64 bit words. In general, the larger the buffer the more efficient the run.
- Note [5]: To determine the start time for the external ephemeris note that the integrator evaluates the first point 5 integration steps before the listed epoch time. The external ephemeris start time must be at least 3 time steps prior to this point. So, if the integration step size is Tstep, and the interval on the external ephemeris is Textint, then the external ephemeris start time must be less than Tstart - 5\*Tstep - 3\*Textint. To determine the stop time for the external ephemeris note that the integrator may go 2 integration steps beyond the epoch stop time. The external ephemeris stop time must be at least 5time steps after this point. So, if the integration step size is Tstep, and the interval on the external ephemeris is Textint, then the external ephemeris stop time must be greater than Tstop + 2\*Tstep + 5\*Textint.

IF CARD OMITTED: Satellite ephemeris is obtained from integration of orbit.

#### EXTERNAL EPHEMERIS

The external ephemeris file is a binary file consisting of a header record and any number of data records. The total number of data records is optional, but it must be sufficient to at least cover the data times in the GEODYN run (+3 steps before the first data point and 5 steps after the last data point for the 10 point hermitian interpolation). Following is a description of the header and data records.

HEADER RECORD. (Record Length, RECL = 6\*nsat+1) \_\_\_\_\_

#### Word(s)

- Number of satellite ephemerides in this file (nsat)
- Start time of ephemeris data in UTC yymmddhhmmss.
- Time interval between ephemeris points. (Step size)
- 4 -> nsat+3 - Satellite ID's of the satellites in this file in the

order that the satellite ephemeris data is provided in the data records. (i.e. the first satellite ID must belong to the first set of six satellite elements in the data record)

nsat+4 -> RECL - Set to zero. Not currently used.

DATA RECORDS. (Record Length, RECL = 6+nsat+1)

-----

The time between data records must correspond exactly to the time interval specified in word 3 of the header record. The satellite elements provided in each data record must be in the same order as the satellite ID's specified in the header record. The satellite elements provided must be true of data cartesian elements.

Word(s)

1 - Time in seconds from start time specified in the header record. For the first data record this must be zero seconds.

2 -> 7 - Six satellite elements for satellite number 1.

 $6*(nsat-1)+2 \rightarrow 6*nsat+1 - Six satellite elements for satellite number nsat$ 

## 2.3.114 YARKOV

YARKOV	_	+2+3+8
COLUMNS	Ü	
1-6	A6	YARKOV - Requests the computation and application of the Yarkovsky/ Schach effect on small spinning satellites such as LAGEOS and LAGEOS II
7	I1	<pre>Index for GEODYN "TYPE" of run  =0 GEODYN will do an ORBGEN and     will produce a binary file     (UNIT 67) containing the     necessary shadowing information.  =1 Data reduction run where GEODYN     will read UNIT 67 and will     compute the Yarkovsky     accellerations</pre>
8	I1	<pre>Index for model to be applied: =0 Rapid spin case =1 Slow rotating case</pre>
18-24	17	Satellite ID
25-44	D20.8	The amplitude of the perturbative acceleration.
45-59	D15.3	Thermal relaxation time

### 2.3.115 YAWPOL

+	1+	-2+6	-+8		
YAWPOL		920324000000. 920325000000.			
	0+	-0+0+0+0	-+0+0		
COLUMNO	FODMAT	DECORIDATION	IIT WATTE & UNITE		
CULUMNS	FORMAT	DESCRIPTION DEFAU	ULT VALUE & UNITS		
		Introduces a group of option cards with Yaw polynomial information. This group may be used for GPS satellites in order to control the Yaw that GEODYN will compute. The end of the YAWPOL group is defined by the global card YPLEND.			
1-6	A6	YAWPOL - Following the blank YAWPOL card, the Yaw polynomial information cards, have the following format:			
1-7	17	Satellite ID. NOTE [1]			
10-22	F13.1	ET start time.	SEC. from 2430000.5		
24-36	F13.1	ET stop time.	SEC. from 2430000.5		
39-49	F11.6	a0 of the quadratic polynomial	degrees		
52-62	F11.6	a1 of the quadratic polynomial	degrees		
65-76	F11.6	a2 of the quadratic polynomial	degrees		
79	A 1	It is important that column 79 be filled with the letters S, P or N. These are just comments but it helps GEODYN IIS realize that the card is a YAW polynomial card. NOTE [2]			
80	A 1	<pre>1 - first half of maneuver 2 - second half of maneuver N - noon</pre>			
Note [1]: All polynomials should be grouped by satellite ID and time order.					
Note [2]: Input to colums 79-80 are interpreted as follows:  S1 = first half of shadow maneuver  S2 = second half of shadow maneuver  P1 = first post shadow maneuver (not always present)  P2 = second post shadow maneuver (usually present)					

P2 = second post shadow maneuver (usually present)
NN = Noon maneuver (only present for very low beta angles)

### 2.3.116 YAWRAT

YAWRAT 2 500	2+3+4+5+ 920324000000. 920325000000. 0+0+0+	
COLUMNS FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6 A6	YAWRAT - Provide the maximum yaw rates, and yaw rate rates for each satellite in the GPSSHD group	
8-24 17	Satellite ID.	
25-34 D20.5	Yaw rate value	deg/sec
24-36 F13.1	Yaw rate rate value	deg/sec2

### 2.3.117 YPLEND

1	+2	-+3+4	-+5678
YAWPOL 2	600	920324000000.	920325000000.
0	+0	-+0	-+0+0

COLUMNS FORMAT DESCRIPTION DEFAULT VALUE & UNITS

1-6 A6 YPLEND - Denotes the end of the YAWPOL group.

IF CARD OMITTED: Abnormal termination.

## 2.3.118 POSITION CARD GROUP

## 2.3.118.1 STATION COORDINATE SUBGROUP

### 2.3.118.1.1 STAPOS

STAPOS	1+-	2+3+4+5+6	-+/8
+	0+-	0+0+0+0+	-+0+0
COLUMNS	FORMAT	DESCRIPTION DEFA	ULT VALUE & UNITS
1-6	A 6	STAPOS - Introduces station position sub-group cards. The station position sub-group contains also optional station linear velocity information.	
7	I1	Coordinate system for adjustment of stations and indicator that all stations are to be adjusted except those explicitly excluded using the FIXED subgroup card.	0
		= 0 Indicates no adjustment, or adjustment of individual stations will be indicated explicitly by use of ADJUSTED cards.	
		> 0 Indicates that all stations for which tracking data exists will be adjusted unless explicitly overridden by use of a FIXED card. (Columns 25-72 must be a non zero)	
		Coordinate system for adjustment:  = 1 Geodetic - Phi, Lambda, Height  = 2 Cartesian - X,Y,Z  = 3 Cylindrical - S.A.D, Lambda, Z  = 4 Spherical - Phi, Lambda, R	
8	I1	Station geodetic file indicator. (Default is for geodetics file to be read prior to reading station coordinates following this STAPOS card)	0
		= 0 Indicates stations will be read from the station geodetic file on unit 16.	
		> 0 Indicates that no stations from the station geodetics file are to be used.	
9	I1	> 0 Value in columns 73-80 overrides ELCUTOFF in Station Geodetics File and sets new default until overridden with ELCUTOFF following this card.	
11-14	14	Maximum number of configurations which data is available in this run. [NOTE 1]	50
25-44	D20.8	Sigma for first station component as indicated by value in column 7.	0.

(must be a non zero if col 7>0)

45-59	D15.3	Sigma for second station component	0.
		as indicated by value in column 7.	
		(must be a non zero if col 7>0)	

- 60-72 D13.1 Sigma for third station component 0. as indicated by value in column 7. (must be a non zero if col 7>0)
- 73-80 D8.2 Elevation cutoff override of Station Geodetics File (see column 9).

#### NOTES:

[1] A configuration in GEODYN II is defined as a unique combination of station-satellite-measurement type.

The valid STAPOS subgroup cards are listed below

ADJUSTED - Turns on station adjustment

CORREL - Station correlations

CONSTADJ - Start of constrained stations
CONSTEND - End of constrained stations
FIXED - Turns off station adjustment

GEODETIC - Defines station geodetic information EXTRAGEO - Specifies planetary shape parameters ELCUTOFF - Sets station elevation cutoff angle

 ${\tt INSTRMNT} \quad {\tt -} \quad {\tt Sets} \quad {\tt station} \quad {\tt antenna} \quad {\tt and} \quad {\tt operating} \quad {\tt frequency} \quad {\tt information} \quad$ 

STATION COORDINATE CARDS - Station positions

STAVEL - Station velocities

TIMVEL - Reference time for station velocities SIGVEL - Sigmas for adjusted station velocities

STATL2 - Station earth tide L2 STATH2 - Station earth tide H2 ENDSTA - End of STAPOS subgroup

The format of all cards in the subgroup is as follows:

A8,2I1,I2,I8,3D15.6,I5,10

There are two categories of stations:

- 1. Those adjusted (i.e., those which follow an ADJUSTED card).
- 2. Those not adjusted (i.e., those which follow a FIXED card). Within the set of stations which are adjusted there are those which are constrained to move together. A set of stations constrained to move as a set can be defined as that set which is adjusted and which follows a CONSTADJ card and precedes either a CONSTEND card or a FIXED card or an ENDGLB card or the next CONSTADJ card.

The conditions stated on an ADJUSTED card or a FIXED card are imposed until another ADJUSTED or FIXED card is encountered or an ENDSTA card is

encountered.

The conditions stated on a CORREL card apply only to adjusted stations and are imposed until another CORREL card is encountered. (Default correlations are zero).

The conditions stated on an ELCUTOFF card are imposed until another ELCUTOFF card is encountered. (Default - ELCUTOFF is zero degrees.)

The conditions stated on an  ${\tt INSTRMNT}$  card are imposed until another  ${\tt INSTRMNT}$  card is encountered.

The conditions stated on a GEODETIC card or an EXTRAGEO card are imposed until another GEODETIC or EXTRAGEO card is encountered.

Station Location cards follow the appropriate cards of other types necessary to specify the conditions that are to apply to those station locations.

Station Location cards without values in the location data fields imply that the location data is to be obtained from the Station Geodetics file but that the other conditions indicated by the preceding STAPOS Subgroup cards apply.

The STAPOS card can indicate the adjustment of all tracking stations not explicitly excluded by the FIXED card. This includes stations obtained from the Station Geodetics file if this file has not been specifically excluded. If the adjustment of all stations has been requested using the STAPOS card, then the ADJUSTED card may still be used to specify the conditions of adjustment of particular subgroups of station coordinates. Unless specified otherwise (using GEODETIC or EXTRAGEO) all stations are considered to be located on the earth (see Keyword Option Card EARTH).

Station velocity cards (STAVEL) should follow immediately the station location cards.

It is required that the station velocity group has the following sequence:

STATION POSITION CARD STAVEL TIMVEL SIGVEL

If no velocity cards are present,  ${\tt GEODYN}$  will apply zero velocities to the stations, therefore extra velocity parameters will appear in the  ${\tt GEODYN}$  parameter summary printout.

Considering the station position and station velocity combinations,  ${\tt GEODYN}$  will handle:

- a. Fixed positions/ fixed velocities
- b. Adjusted positions/ fixed velocities
- c. Fixed positions/ adjusted velocities
- d. Adjusted positions/ adjusted velocities

No correlations are provided for station velocities.

Extra attention should be given at a setup with constraint stations. The station velocity cards in this case should follow immediately the master station and not the constrained station (station with data).

Cards like ADJUSTED and FIXED do not affect station velocities. Only the presence of SIGVEL controls the adjustment of station velocities.

The STATL2 (or STATH2) cards should follow the station velocity group cards or if there are no velocity cards this card must follow the station coordinate cards, for example:

STATION POSITION CARD (STAVEL) (TIMVEL) (SIGVEL) STATL2 STATH2

STATL2 and STATH2 cards are independent of each other. The user can apply or or solve for L2 or H2 or both per station. If the STATL2 (or STATH2) card is not present for an individual station the solid earth tide corrections for this station will either use the value from the L2LOVE card (or H2LOVE) if they are present or the default GEODYN value if they are not present.

### 2.3.118.1.2 ADJUSTED

						_
ADJUSTEI	)2	10.	3+4+ 10.		10.	
+	0	+0+(	)+	0+	0+0-	+0
COLUMNS	FORMAT	DESCRIF	PTION		DEFAULT VALU	JE & UNITS
1-8	A8	-	nests adjustment oving this card up			
9	I1	Coordinate syst of station.	em for adjustment	;	0	
		= 2 = 3	Phi,Lambda,Height K,Y,Z S.A.D.,Lambda,Z Phi,Lambda,R			
21-35	D15.6	Sigma Phi or Si	igma X or Sigma S.	A.D.	0.	S or M
36-50	D15.6	Sigma Lambda on	Sigma Y		0.	S or M
51-65	D15.6	Sigma Height on	s Sigma Z or Sigma	a R	0.	М
UNITS:		· ·	eters ; S =Sec adians ; AS =Arc		-	

IF CARD OMITTED: Stations will not be adjusted unless adjustment of all stations requested by inclusion of non-zero sigmas on STAPOS card.

### 2.3.118.1.3 CONSTADJ

+1	-+2+-	3+-	4	5+	-6+	-78
CONSTADJ						
+0	-+0	0+-	0+	0+	-0+	-0+0

COLUMNS FORMAT DESCRIPTION

DEFAULT VALUE & UNITS

1-8 A8 Adjusted stations which follow this card are constrained together.

CONSTADJ is negated by CONSTEND, FIXED, next CONSTADJ or ENDSTA.

An ADJUSTED card must preceed the CONSTADJ card.

IF CARD OMITTED: Stations will not be constrained.

### 2.3.118.1.4 CONSTEND

+1+	2+	3+	4	5+	6+	78
CONSTEND						
+	0+	0+	0	0+	0+	00

COLUMNS FORMAT DESCRIPTION

DEFAULT VALUE & UNITS

1-8 A8 CONSTEND - Turns of constrained stations.
Negates CONSTADJ card.

# 2.3.118.1.5 CORREL

CORREL	_	.831	.917	.983	-6+8 -0+0
COLUMNS	FORMAT	DESCRIPTION			DEFAULT VALUE & UNITS
1-6	A6	CORREL - Provides for between station coordinates. Uses the same system as specified last ADJUSTED card.	rdinate compon- e coordinate		
21-35	D15.6	Correlation between and 2.	components 1		0.
36-50	D15.6	Correlation between and 3.	components 1		0.
51-65	D15.6	Correlation between and 3.	components 2		0.

 $\label{eq:card_omega_relations} \mbox{ IF CARD OMITTED:} \quad \mbox{No correlations between the station components exist.}$ 

#### 2.3.118.1.6 ELCUTOFF

---+---1---+---2---+---3---+---4----+---5---+---6---+---7---+----8
ELCUTOFF 10.
----+---0---+---0---+---0---+---0

COLUMNS FORMAT DESCRIPTION DEFAULT VALUE & UNITS

1-8 A8 ELCUTOFF - Specifies elevation angle below which data will automatically be edited.

21-35 D15.6 Elevation cutoff angle. 0. DEG

IF CARD OMITTED: Elevation cutoff of zero degrees is used.

### 2.3.118.1.7 EXTRAGEO

EXTRAGE	0	+2+3+4+5+ 0300 6378145.0 298.255 +0+0+0+	
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-8	A8	EXTRAGEO - Specifies planetary shape parameters for stations which follow.	
17-20	14	Planetary body number:         Mercury       0100         Venus       0200         Earth       0300         Earth's Moon       0301         Mars       0400         Jupiter       0500         Saturn       0600         Uranus       0700         Neptune       0800         Pluto       0900         Sun       9999	0
21-35	D15.6	Semi-major axis for planet.	6378138. M
36-50	D15.6	Inverse of polar flattening of the planet.	298.255
51-65	D15.6	Inverse of equatorial flattening of the planet.	0.
UNITS:		<pre>lometers; M = Meters ; S = Seconds ; grees ; RAD=Radians ; AS = Arc seconds;</pre>	

IF CARD OMITTED: Default values for earth apply.

### 2.3.118.1.8 FIXED

+1	+	-2	-+	3+	4	+	-5+	6	7+8
FIXED									
+0	+	-0	-+	0	0		-0+	0	00

COLUMNS FORMAT DESCRIPTION

DEFAULT VALUE & UNITS

1-8 A8 FIXED - Used to indicate that for the stations that follow this card there will be no adjustment. This card is used to negate adjustment requests by either the STAPOS or ADJUSTED cards.

> To negate the FIXED card another ADJUSTED card may be used.

IF CARD IS OMITTED: No negation of adjustment will occur.

### 2.3.118.1.9 GEODETIC

+	2	+4	-5	-6+	-78
GEODETIC	6378145.0	298.255			
+	0	+0	-0+	-0+	-0+0

COLUMNS FORMAT DESCRIPTION DEFAULT VALUE & UNITS

1-8 A8 GEODETIC - Specifies Earth ellipsoid parameters for stations which follow.

21-35 D15.6 Earth semi-major axis in meters. [NOTE 1] 0. M

36-50 D15.6 Inverse of the flattening (1/f) of the earth. [NOTE 1] 0.

#### NOTES:

[1] Defaults will be obtained in the following order:

Default values built into GEODYN IIS
Values obtained from Gravity Model File
Values obtained from EARTH card
Values from GEODETIC card

UNITS: KM =Kilometers; M =Meters ; S =Seconds ; M/S=Meters per second DEG=Degrees ; RAD=Radians ; AS =Arc seconds; MAS=Milli-arc seconds

IF CARD OMITTED: Default values from gravity file on UNIT 12 or from an earth option card will apply.

+	1	2		-3	++	5	-+6-	+-	7	+
INSTRMNT			1.0	0	0. ++	0			0	
+	0	0		-0			-+0-	+-	0	+
COLUMNS F	ORMAT		DESCR	IPTION			DE	FAULT	VALUE	& UNIT
1-8 A	8		MNT - Spe ument par		tracking					
9-10 I	2	been GEODY	invoked) N will us	se the	NTPHC optio antenna nu vided on un	mber		0		
			other	wise						
			na mount OT been :		(If ANTPHC	option		3		
		= 2 = 3 = 4 = 5 = 6 = 7	Displacer station l Station l unit 23	n- Sout Elevati Le-Decl VLBI A nent ap height height [NOTE 2 change	on Lination Antenna Mou Oplied to read from 2] es read fro					
13-16 I	4	Laser	Tracking	g Stati	on Detecto	r Type		1		
		=1 =2 =3 =4 =5 =6 =7 =8 =9	MOBLAS -8 TLRS -3 MLRS MOBLAS -7 TLRS -1 TLRS -2 Bar Giyyo Orroral MTLRS -2	(Monum (Quinc (FT. I (Green	kala, HI) nent Peak,	CA)				
		=10 =11	tations: Herstmono Zimmerwal Borowiec Grasse	Ld	ıd					

```
=14 MTLRS-1
                =15 Helwan
                =16 Beijing
                =17 Chang Chun
                    Katsively, Ukraine
                    Riga, Latvia
                =18 Postsdam
                    Dunaovcy, Ukraine
                    Maidanak, Uzbekistan
                    Evpatoria, Ukraine
                    Komsomolsk, Russia
                    Balkash, Russia
                    Simeiz, Ukraine
                =19 Santiago, Cuba
                    Metsahovi, Finland
               SPAD Stations:
                =20 Herstmonceux
                    Graz
                    Simosato
                    MTLRS-1
                    FTLRS
               Matera Station:
                =21 Matera
21-35
        D15.6 Antenna axis displacement in meters.
                                                                  0.
                                                                              Μ
36-50
       D15.6 Nominal received wavelength in microns.
                                                                        M*1.0D-6
                                                                  0.
               ( 0. indicates nominal wavelength will
                 be used from observation file as
                 supplied by the Tracking Data
                 Formatter Program.)
51-65
       D15.6 Turn around factor (TRF).
                                                                  1.
               (TRF=Wavelength trans./wavelength rec.)
NOTES:
        Inclusion of an INSTRMNT card which is blank except for columns 1-8
        negates any previous INSTRMNT card and reverts back to the station
        complement defaults.
       A "7" in column 10 requests reading the station height offsets from
        unit 23. The following quantities are read:
         Col
         1-6
                 IYMD
                            Ι6
                                    - year, month, day (YYMMDD)
         7-10
                 IHM
                                    - hour, minute (HHMM)
                            Ι4
        11-20
                 SEC
                            F10.1
                                    - seconds
        21-22
                                     - not used
        23-30
                NSTA
                            18
                                    - station number
```

San Fernando, Spain

If the file connected to unit 23 is not present or is empty, no station

D24.16 - height offset

36-59

AOFF

height offsets are applied. Otherwise, the station height offset is chosen as the value on the record whose time is less than or equal to the current time. A time before the start of the table takes the value of the first offset for that station; a time after the end of the table takes the last value in the table.

[3] A "8" in column 10 requests reading the station pressure values from unit 23. The following quantities are read:

Col			
1-6	IYMD	I6	- year, month, day (YYMMDD)
7-10	IHM	14	- hour, minute (HHMM)
11-20	SEC	F10.1	- seconds
21-22			- not used
23-30	NSTA	18	- station number
36-59	pmean	D24.16	<ul> <li>pressure offset (mbar)</li> </ul>
60-83	scale		- pressure scale factor

The reading of unit 23 and the assignment of values is done as for option 7. (See NOTE [2] above)

IF CARD IS OMITTED: No instrument correction will be applied.

#### 2.3.118.1.11 STATH2

STATH2	_	+2+3+4+5+	
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A6	STATH2 - Requests application or estimation of solid earth tide coefficient of the second kind for this station only	
13-20	18	Station number	
21-35	D15.6	A priori value of the coefficient	value from H2LOVE
36-50	D15.6	Sigma of the coefficient. If this field is zero the coefficient will not be estimated.	

NOTE: A STATH2 card must follow the station velocity card group, or if there are no velocity cards this card must follow the station coordinate card, if solid earth tide of the second kind correction, other than the one specified on the H2LOVE card is to be applied.

IF CARD OMITTED: Earth tide for the preceeding station will be applied using the coefficient from the H2LOVE card. If H2LOVE card is not present GEODYN will use the default value.

#### 2.3.118.1.12 STATL2

STATL2	-	+2+3+4+5+	
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A 6	STATL2 - Requests application or estimation of solid earth tide coefficient of the third kind for this station only	
13-20	18	Station number	
21-35	D15.6	A priori value of the coefficient	value from L2LOVE
36-50	D15.6	Sigma of the coefficient. If this field is zero the coefficient will not be estimated.	

NOTE: A STATL2 card must follow the station velocity card group, or if there are no velocity cards this card must follow the station coordinate card, if solid earth tide of the third kind correction, other than the one specified on the L2LOVE card is to be applied.

IF CARD OMITTED: Earth tide for the preceeding station will be applied using the coefficient from the L2LOVE card. If L2LOVE card is not present GEODYN will use the default value.

### 2.3.118.1.13 STATION COORDINATE CARDS

STALAS	00007	2+3+4+5+6 0631130719.3758	
COLUMNS	FORMAT	DESCRIPTION DEFA	ULT VALUE & UNITS
1-8	A8	Station name. no	default
9	I1	Coordinate system indicator:	0
		= 0 GEODYN will determine if input is geodetic or cartesian.	
		<pre>= 1 Geodetic - Phi,Lambda,Height = 2 Cartesian - X,Y,Z = 3 Cylindrical - S.A.D,Lambda,Z = 4 Spherical - Phi,Lambda,R</pre>	N / A
10 I1 N		<pre>ints Index =0 : No participation =1 : The station participates in net translati</pre>	
11-12	12	< 99 is Plate number. = 99 apply thermal deformation for antenna	0
13-20	18	Station number.	0
21-35	D15.6	<pre>First coordinate:   (+ or -) DDMMSS.SSS   (+ or -) XXXXXXX.XXX meters</pre>	0.
36-50	D15.6	Second coordinate:  DDDMMSS.SSS (+ or -)XXXXXXX.XXX meters	0.
51-65	D15.6	Third coordinate: (+ or -) XXXXXXX.XXX meters	0.
67-70	14	Site number only for stations involved in ocean loading (same as in columns 13-20)	0
71-80	A10	Available for comments.	

IF CARD OMITTED: Stations coordinates will be obtained from geodetic file on UNIT 16.

### 2.3.118.1.14 STAVEL

STAVEL	-	+2+3+4+5+	· ·	
COLUMNS	FORMAT	DESCRIPTION	DEFAULT	VALUE & UNITS
1-6	A 6	STAVEL: Introduces station velocities in a 3-D Cartesian coordinate system.		
13-20	18	Station number		
21-35	D15.6	First component of velocity	0	M/S
36-50	D15.6	Second component of velocity	0	M/S
51-65	D15.6	Third component of velocity	0	M/S

 ${\tt NOTE}$  : A STAVEL card must follow the equivalent station coordinate card if velocities other than zero are to be applied.

UNITS: KM =Kilometers; M =Meters ; S =Seconds ; M/S=Meters per second H =Hertz ; MH =Mega-Hertz; YR=Years

 $\begin{tabular}{lll} \textbf{IF CARD OMITTED:} & \textbf{Station velocities for the preceeding stations are assumed to} \\ & \textbf{be zero.} \\ \end{tabular}$ 

#### 2.3.118.1.15 TIMVEL

---+---1----+----3----+----4----+---5----+---6----+----7----+----8
TIMVEL
----+---0----+---0----+---0----+---0

COLUMNS FORMAT DESCRIPTION DEFAULT VALUE & UNITS

1-6 A6 TIMVEL: Introduces a reference time for the station velocities.

13-20 I8 Station number

21-35 D15.6 Reference time in YYMMDD. EPOCH START TIME

NOTE: A TIMVEL card corresponding to a certain station should follow its station coordinate and STAVEL card.

UNITS: KM = Kilometers; M = Meters ; S = Seconds ; M/S = Meters per second H = Hertz ; MH = Mega-Hertz;

#### 2.3.118.1.16 SIGVEL

SIGVEL	-	+2+3+4+5+		
COLUMNS	FORMAT	DESCRIPTION	DEFAULT	VALUE & UNITS
1-6	A6	SIGVEL: Requests adjustment of the velocities for the preceeding station.		
13-20	18	Station number		
21-35	D15.6	Sigma for the first velocity component	0	M/S
36-50	D15.6	Sigma for the second velocity component	0	M/S
51-65	D15.6	Sigma for the third velocity component	0	M/S

 ${\tt NOTE}$  : A SIGVEL card must follow the equivalent STAVEL and TIMVEL card if velocity adjustment for the preceeding station is requested.

UNITS: KM = Kilometers; M = Meters ; S = Seconds ; M/S = Meters per second H = Hertz ; MH = Mega-Hertz; YR = years

 $\hbox{ IF CARD OMITTED:} \quad \hbox{Velocities will be applied on the preceeding station but no velocity adjustment will take place. } \\$ 

### 2.3.118.1.17 ENDSTA

+1	-+2	-+3	-+4	-+5	-+6	-+7	-8
ENDSTA							
+0	-+0	-+0	-+0	-+0	-+0	-+	-0

COLUMNS FORMAT DESCRIPTION

1-8 A8 ENDSTA - Ends STAPOS Subgroup.
This card is required if a STAPOS card is present.

# 2.3.118.2 QUASAR COORDINATE SUBGROUP

### 2.3.118.2.1 QUAPOS

QUAPOS	0+0	· ·	-	· ·		·	· ·
COLUMNS	FORMAT	DESCRIPTION			DEFAULT	VALUE	& UNITS

1-6 A6 QUAPOS - Introduces quasar position sub-group cards.

IF CARD OMITTED: VLBI data processing will abnormally terminate.

### 2.3.118.2.2 QUASAR COORDINATE CARD

0016+731	4000	01945.786329 732730.0180 -0+0+0+	3 0.015 0.020			
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS			
1-8	A8	Quasar name				
9-16	18	Quasar ID				
20 I1 Ne	20 I1 Net Constraints Index =0 : No participation =1 : The quasar participates in net rotation					
25-44	D20.8	Quasar right ascension	HHMMSS.SSSSS			
45-59	D15.3	Quasar declination +	/- DDMMSS.SSSS			
60-72	D13.1	Quasar right ascension sigma	seconds			

IF CARD OMITTED: VLBI data processing will abnormally terminate

73-80 D8.2 Quasar declination sigma arc seconds

## 2.3.118.2.3 ENDQUA

+1	+2	+3	+4	+5	-+6	-+8
ENDQUA						
+0	+0	+0	+0	+0	-+0	-+00

COLUMNS FORMAT DESCRIPTION DEFAULT VALUE & UNITS

1-6 A6 ENDQUA - Denotes the end of Quasar coordinate subgroup

IF CARD OMITTED: VLBI data processing will abnormally terminate.

## 2.3.119 GLOBAL SET TERMINATOR

#### 2.3.119.1 ENDGLB

ENDGLB	-	+2+3+4+5+ +0+0+0		·	· ·
COLUMNS	FORMAT	DESCRIPTION	DEFAULT	VALUE &	UNITS
1-6	A6	ENDGLB - Indicates the end of the global set option cards. This card is MANDATORY in all runs.			
7-8	12	Convergence criteria for the global iterations in percent. [NOTE 1]	(	02	%
9	I1	Minimum number of global iterations.	:	1	
10	I1	Maximum number of global iterations	:	1	

#### NOTES:

[1] Global convergence is defined as follows:

ABS(RMS P - RMS C)/RMS C \* 100 .LE. Convergence Criteria

#### Where:

RMS P = RMS previous global iteration RMS C = RMS current global iteration .LE. = Less than or equal

On the first global iteration the value for the previous  $\ensuremath{\mathtt{RMS}}$  is defaulted to 1000.

 $\label{eq:card_omitted} \mbox{IF CARD OMITTED: Run will abnormally terminate.}$ 

# 2.4 ARC SET MANDATORY CARDS

#### 2.4.1 ARC SET MANDATORY OVERVIEW

The ARC mandatory section includes three lines for the user to provide appropriate comments about the ARC, a REFSYS line to establish the inertial coordinate system that will be used for satellite orbit integration, a SATPAR line to define the satellite ID and cross sectional area and mass, an EPOCH line to define the epoch time for the satellite elements, and an ELEMS1 and ELEMS2 line to define the satellite elements.

The first eight lines of the ARC Set for each arc must be in the specific order shown below:

```
Arc title 1
Arc title 2
Arc title 3
REFSYS
SATPAR --
EPOCH | These four lines are repeated for
ELEMS1 | each satellite in the arc
ELEMS2 --
```

If more than one satellite is used in an arc than the last four lines are repeated (SATPAR, EPOCH, ELEMS1 ELEMS2) for each satellite.

### 2.4.2 TITLE

+1+	2+3+4+5+	-6+	7	8
USER SPECIFIED JOH	B DESCRIPTION - CARD 1			
USER SPECIFIED JOH	B DESCRIPTION - CARD 2			
USER SPECIFIED JOH	B DESCRIPTION - CARD 3			
+	0+0+0	-0+-	0+	0
COLUMNS FORMAT	DESCRIPTION	DEFAULT	VALUE & UI	NITS
1-80 10A8	User may specify on these three cards,			

1-80 10A8 User may specify on these three cards, information description of the arc.

These cards may also remain blank, but must always be present.

IF CARDS OMITTED: Run will abnormally terminate.

### 2.4.3 **REFSYS**

REFSYS	2	2+5+ 830901000000. 0+0+0+	
	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A 6	REFSYS - Specifies the coordinate system of integration and the number of arc iterations.	
7-8	12	Maximum number of arc iterations during the first global iteration.	1
9	I1	Minimum number of arc iterations during the first global iteration.	1
10	I1	Maximum number of arc iterations during global iterations after the first.	1
11	I1	Coordinate reference system	0
		O True of reference date and time 1 Mean of J2000 [1]	
21-26	16	Year, month, day of reference date (YYMMDD	). 0
27-30	14	Hour, minute of reference time (HHMM).	0
31-40	D10.8	Seconds of reference time (SS.SSSSSS).	0.

#### NOTE:

[1] The DE200 Ephemeris is needed when the J2000 reference system is selected. No reference date is needed in cols. 21-40 if J2000 is selected.

IF CARD OMITTED: Run will abnormally terminate.

### **2.4.4** SATPAR

SATPAR		2+3+ 76039010.28274 0+0+	411.000		
COLUMNS		DESCRIPTION	-0+		VALUE & UNITS
1-6	A6	SATPAR - Specifies sification number (Sates sectional area and maximum degree of gracoefficients to be confor this spacecraft.	ellite ID), cross ass. Also specifies avitational		
7	I1	0 - Obtain average S, area and S/C mass 25-59 of this cas 1 - Obtain same from file.	s from columns rd.	0	
8	I1	O - No tables of S/C solar radiation from S/C parameter	to be obtained : er file. :	0	
		<ul> <li>1 - Only drag cross used from S/C parameters</li> <li>2 - Only solar radiation S/C parameters</li> <li>3 - Both drag cross radiation tables parameters file.</li> </ul>	rameter file. : tion tables used : ers file. : section and solar :	N/A	
9	I1	Twelve parameter startidentifier. If this there must be one most the unit 5 setup to the fisrt satellite 1 and for the second satellites forming a indices on col. 9, the ELEMS1 and ELEMS2 case and the twelve second ination parameter with the index 1 is cards featuring the and the second SATPANELEMS cards featuring (Ref. NOTE [2]). Any cards for these two the sigmas of the 12 parameters.	index is present re SATPAR card in form the pair. For the index must be 2. For the pair with these ne following rds contain 6 (six) tate vector rs. The first SATPA followed by ELEMS mid-point parameter R is followed by g baseline paramete following VARCOV satellites feature	R s,	
10-11	12	Spacecraft attitude of the control o	_	0	

```
6 - Mars Observer Mapping Phase
                   7 - Mars Observer Cruise Phase
                   8 - TDRSS
                   9 - Magnetically stabilized S/C
                  10 - GFO
                  11 - TRMM
                  12 - EUVE
                  13 - ICESAT
                  14 - ENVISAT
                  15 - CRYOSAT-2
                  16 - Leading GRAIL satellite
                  17 - Trailing GRAIL satellite
                  18 - HY2A
                  19 - SARAL
                  20 - SENTINEL
12
        Ι1
                  Requests external attitude for this
                  satellite:
                                                                   0
                   0 - No external attitude
                   9 - External attitude (See Volume 5
                       for a description of the external
                       attitude file)
13
        Ι1
                  Local gravity application switch:
                                                                  0
                   0 - Consider gravity anomaly, surface:
                       density and geoid height local
                       gravity terms to be significant
                                                                N/A
                       for this spacecraft.
                   1 - Consider local gravity terms to
                       be insignificant for this
                       spacecraft.
14
        Ι1
                  Requests external thermal acceleration
                  for this satellite:
                                                                   0
                   O - No external thermal acceleration
                   1 - External thermal acceleration (See
                       Volume 5 for a description of the
                       external acceleration file)
15-17
                  Maximum degree of gravitational coefficients
                                                                  Max. of Model
        T.3
                  to be considered significant for this S/C.
18-24
                  Satellite ID (satellite ID is required).
        17
                  S/C cross sectional area.
25 - 44
        D20.8
                                                              See col. 7
                                                                           M**2
45-59
        D15.3
                  S/C mass.
                                                              See col. 7
                                                                           Κg
UNITS:
        KM =Kilometers; M =Meters
                                     ; S =Seconds
                                                        ; Kg =Kilograms
        DEG=Degrees ; RAD=Radians
                                      ; AS =Arc seconds; MAS=Milli-arc seconds
NOTES:
 [1]
         If more than one satellite is used in an arc than the four lines
         SATPAR, EPOCH, ELEMS1 ELEMS2 - are repeated for each satellite.
         SATPAR
         EPOCH
                        | These four lines are repeated for
         ELEMS1
                        | each satellite in the arc
```

ELEMS2 --

[2] A full description for the baseline representation of the initial epoch state parameters may be found in the following reference:

"Short-arc analysis of intersatellite tracking data in a gravity mapping mission", D.D.Rowlands, R.D.Ray, D.S.Chinn, F.G.Lemoine, Journal of Geodesy (2002)76:307-316

IF CARD OMITTED: Run will abnormally terminate

# 2.4.5 EPOCH

	1+	2+5+	
EPOCH		830901 830901	830903
+	0+	0+0+0+	-0+0
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A6	EPOCH - Specifies epoch time for satellite elements ,optional start time for orbit integration, and mandatory stop time for the run. The stop time is used by GEODYN to determine ephemeris and flux data requirements.	
21-26	16	Year, month and day of epoch (YYMMDD).	0
27-30	14	Hours and minutes of epoch (HHMM).	0
31-40	D10.8	Seconds of epoch (SS.SSSSSS).	0.
41-46	16	YYMMDD of start. : Default is epoch time : If specified as other	
47-50	14	HHMM of start. : than epoch, orbit wil : be integrated to this	l cols. 27-30
51-60	D10.8	Seconds of start.: time and epoch reset : to this time before a : measurement processin	cols. 31-40 ny
61-66	16	YYMMDD of stop. : A STOP TIME IS	0
67-70	14	HHMM of stop. : MANDATORY :	0
71-80	D10.8	Seconds of stop. : FOR ALL ARCS	0.

NOTE: All times specified on the ORBTVU and the ORBFIL card must fall between the start time and the stop time specified on the EPOCH card.

# 2.4.6 SLAVE

+	1+	2+3-	+	4	-5	+	-6+-	7	+-	8
SLAVE		7603901								
+	0+	0+0-	+	0+	-0	+	-0+-	0	+-	0
COLUMNS	FORMAT	DESCRIPT	ION				DEFAULT	VALUE	& U	NITS
1-5	A5	SLAVE - Used	in plac	e of EPOCH	d card	to				
		indicate that	this sa	tellite be	elongs	to				
		a set of sate	llites.							
24-30	17	Satellite ID	of Maste	r Satellit	te for	this	<b>;</b>			
		set. The Mast	er Satel	lite must	exist	and				
		the SATPAR, EF	•	•						
		for the Maste								
		defined prior	•	•	•					
		and ELEMS2 ca			e disc	ussio	n			
		of slave sate	IIItes b	erow.						

# 2.4.7 ELEMS1

ELEMS1		2+3+ 8986390.0780	8283790.5802	-150249	98.2249
	0+	0+	0+0	+0+	0+0
COLUMNS	FORMAT	DESCRIPTION		DEFAULT	VALUE & UNITS
1-6	A6	ELEMS1 - Introduces components of the othe S/C.		or	
7	I1	Form of elements  0 - Either Cartes  1 - Cartesian  2 - Keplerian  3 - Non-singular  5 - Cartesian non  6 - Keplerian non	Keplerian -elliptic	0	
8	I1	Coordinate system o	f input elements	NOTE(2) 0	
11-14	14	Central body for in  0100 - Mercury 0200 - Venus 0300 - Earth 0301 - Earth's mo 0400 - Mars 0500 - Jupiter 0600 - Saturn 0700 - Uranus 0800 - Neptune 0900 - Pluto 9999 - Sun		300	
COLUMNS	FORMAT	DESCRIPTION		DEFAULT	VALUE & UNITS
21-40	D20.14	Element 1 see Tab	le on next page	0	М
41-60	D20.14	Element 2 see Tab	le on next page	0	M or none
61-80	D20.14	Element 3 see Tab	le on next page	0	M, DEG or none

UNITS: KM =Kilometers; M =Meters ; S =Seconds ; Kg =Kilograms DEG=Degrees ; RAD=Radians ; AS =Arc seconds; MAS=Milli-arc seconds

NOTE: [1] GEODYN II expects the elements to be referred to the IAU system regardless of which planet the S/C is orbiting about.

 ${\tt NOTE:}$  [2] The coordinate system of input elements should be the same system

as specified on column 11 of the REFSYS option.

#### 2.4.8 ELEMS2

+-	2+3+	-4+5+	-6+8
ELEMS2	849.44502691749	-1854.4317908133	-5301.4527479779
+-	0+	-0+	-0+0

COLUMNS	FORMAT	DESCRI	PTION				DEFAULT	VALUE	& UNITS
1-6	A6	ELEMS2 - Incomponents the S/C.							
	21-40	D20.14	Element 4	see	Table	below	0	M/S, or n	
	41-60	D20.14	Element 5	see	Table	below	0	M/S, or n	
	61-80	D20.14	Element 6	see	Table	below	0	M/S	or DEG

#### ELEMENTS DEFINITION

I	NERTIAL CARTESIAN ELEMENTS		NG KEPLERIAN MENTS	NON SINGULAR KEPLERIAN ELEMENTS
,,,,,,,,,,,,,	,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,	,,,,,,,,,,,	,,,,,,,,,,,,,	,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,
' Element 1'	x (meters)	, A	(meters) ,	A (meters)
,		,	,	,
'Element 2'	y (meters)	, E	,	E*cos(w+OMEGA)
,		,	,	,
' Element 3'	z (meters)	, I	(degrees) ,	E*sin(w+OMEGA) '
,		,	,	,
' Element 4'	x (meters/sec)	, OMEGA	(degrees) ,	sin(I/2)*sin(OMEGA) '
,	•	,	,	,
' Element 5'	y (meters/sec)	, w	(degrees) ,	sin(I/2)*cos(OMEGA) '
,		,	,	,
' Element 6'	z (meters/sec)	, M	(degrees) ,	M+w+OMEGA (degrees) '

NOTE: [1] GEODYN II expects the elements to be referred to the IAU system regardless of which planet the S/C is orbiting about.

- 2.5 ARC SET OPTION CARDS
- 2.5.1 ACCELERATION SUBGROUP

#### 2.5.1.1 ACCEL

			_		
ACCEL 3		7603901	+4+5+ 1.80D-12	-6+	78
			+0+	-0+-	0+0
COLUMNS	FORMAT	DESCRIPTION		DEFAULT	VALUE & UNITS
1-6	A 6		s application and/or neral acceleration.		
7	I1	6 - GPS Y BIAS	(R) (R x V) (V)	0	
8	I1	Indicates order o	of acceleration	0	
		0 - Ga 1 - Ga  2 - Ga			
18-24	17	Satellite ID (red	quired).	0	
25-44	D20.8	A priori value of coefficients.	acceleration	0	M/S**2 or M/S**3 or M/S**4
COLUMNS	FORMAT	DESCRIPTION		DEFAULT	VALUE & UNITS
45-59	D15.3	time dependent ac wanted. To select eration the end t	in this field.	0	
60-72	D13.1	A priori sigma of coeff	acceleration ficient.	0	M/S**2 or M/S**3 or M/S**4
NOTE[1]		ods may be specifi	ed only for the highest	order c	

Time periods may be specified only for the highest order coefficients used (i.e. if Ga and Ga dot are used, only Ga dot can have time intervals). When using time periods the number of ACCEL cards required is equal to the number of time periods plus the order (order = the number in column 8 +1). The Ga, Ga dot, and/or Ga double dot without a time interval are used for times beyond the end of the the last interval specified on the ACCEL cards. See following example.

EXAMPLE: ACCEL cards for time dependent general accelerations (highest order - Ga ;three time intervals; 4 cards required)

+-	1	2+-	4	5+	-6+	-78
ACCEL	30	7603901	1.80D-12			
ACCEL	30	7603901	1.82D-12	840901000000.		
ACCEL	30	7603901	1.84D-12	840902000000.		
ACCEL	30	7603901	1.86D-12	840903000000.		
+-	0+	0+-	0	0+	-0+	-0+0

 $\begin{tabular}{ll} WARNINGS: If duplicate time periods are specified the run will terminate \\ in IIS with a warning message. \\ \end{tabular}$ 

The first time period that exceeds the end time of the run (see EPOCH card) will be accepted. All other end times exceeding the end time of the run will be ignored.

IF CARD OMITTED: General accelerations are not applied

# 2.5.1.2 ACCLRM

+ ACCLRM	1+	2+3+4+5+6	+8
	0+	0+0+0+0	+0+0
COLUMNS	FORMAT	DESCRIPTION DEFAU	LT VALUE & UNITS
1-6	A6	ACCLRM - Introduces the accelerometer data group. Options included in this group are ACCBIA and DYNSEL The option ACCEND denotes the end of the accelerometer data group.	
25-44	D20.8	Time tolerance for matching extarnal acceleration file times with interation times. This optio is used with the dynamic acceleration mode (DYNSEL)	S
UNITS:		eters; M = Meters ; S = Seconds ; Kg = Kil es ; RAD=Radians ; AS = Arc seconds; MAS=Mil	=

### 2.5.1.3 AXES

+	1+	2+3+5+6+7	-+8
AXES	9 9		
+	0+	0+0+0+0+0	-+0
COLUMNS	FORMAT	DESCRIPTION DEFAULT VALUE	& UNITS
1-6	A6	AXES: allows the user to manipulate the way input values from the accelerometer are used in GEODYN The program ingests X, Y & Z accelerometer components both in geometric and dynamic modes	
		columns 7-9 are used to denote which axes will be used in GEODYN for dynamic accelerometry	
		columns 10-12 are used to allow the change of the sign of accelerometry components in the dynamic mode	
		columns 13-15 are used to allow the change of the sign of accelerometry components in the geometric mode	
7	I1	=0 The X component of accelerometry will replace the geodyn computed =9 The X component of accelerometry will not replace the geodyn computed	0
8	I1	=0 same as above for Y component =9	0
9	I1	=0 same as above for Z component =9	0
10-12	3I1	<pre>=9 changes the sign implied by the GFZ document   for X,Y or Z axis for dynamic accelerometry =0 does not change the sign</pre>	0
13-15	3I1	<pre>=9 changes the sign implied by the GFZ document   for X, Y or Z axis for geometric accelerometry =0 does not change the sign</pre>	0

#### NOTE:

A 9 in column 9 of the AXES card will delete Z axis for dynamic accelerometry.

A 9 in column 14 would cause the Y axis to have the opposite sign from the sign implied by the GFZ document for geometric accelerometry.

IF CARD OMITTED: The axes will have the signs implied by GFZ document.

2.5.1.4	ACCBIA	TIP	TO	0407.00	١

ACCBIA12		2000039	-3+4+5+ 000807000000.00000808000000.0 5635.00			
	0+	0+	-0+	-0+-	0	-+0
COLUMNS	FORMAT	DESCR	IPTION	DEFAULT	VALUE	& UNITS
1-6	A 6		quests application and/or of accelerometer data biases			
7	I1	Type of Bi	as			
		1 - Abso 2 - Scal 3 - Timi				
8	I1	Component	of Bias			
		1 - Tota 2 - X co 3 - Y co 4 - Z co	mponent			
9	I1	Order of B	ias			
			ar			
18-24	17	Satellite	ID (required).	0		
25-44	D20.8	Begin time	for the accelerometer bias	YYMMDDHI	HMMSS.S	S
45-59	D15.3	End time f	or the accelerometer bias	YYMMDDHI	HMMSS.S	S
60-72	D13.1	A priori v	alue of accelerometer bias	0		
73-80	D8.2	A priori s	igma of accelerometer bias	1	.D-25	
COLUMNS	FORMAT	DESCR	IPTION	DEFAULT	VALUE	& UNITS
1-6	6 X	BLANKS mus	t be present			
25-44	D20.8	the model	the two periodic terms in associated with the time pecified on the previous			S
25-44 NOTE [1]	D20.8	the model interval s	associated with the time		<b>.</b>	S

NOTE [1] The times on the ACCBIA cards should be input in advanced time order for each satellite. If the time order is not observed, GEODYN IIS will terminate abnormally.

UNITS: KM = Kilometers; M = Meters ; S = Seconds ; Kg = Kilograms

DEG=Degrees ; RAD=Radians ; AS =Arc seconds; MAS=Milli-arc seconds

IF CARD OMITTED: If the ACCBIA card is omitted, no biases will be applied to the accelerometer data. If the ACCBIA card is present, but the second BLANK card is missing the program will terminate abnormally.

ACCBIA12 ACCBI2	21	2+3+4+5+ 2012001	00 04
COLUMNS		DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A6	ACCBIA: Requests application and/or adjustment of accelerometer data biases	
7	I1	Type of Bias	
		<ul><li>1 - Absolute Bias</li><li>2 - Scale Bias</li><li>3 - Timing Bias</li></ul>	
8	I1	Component of Bias	
		<ul> <li>1 - Total acceleration</li> <li>2 - X component</li> <li>3 - Y component</li> <li>4 - Z component</li> </ul>	
9	I1	Order of Bias	
		<pre>1 - Constant 2 - Linear 3 - Quadratic 4 - Periodic (sinwt) 5 - Periodic (coswt)</pre>	
18-24	17	Satellite ID (required).	0
25-44	D20.8	Begin time for the accelerometer bias	YYMMDDHHMMSS.SS
45-59	D15.3	End time for the accelerometer bias	YYMMDDHHMMSS.SS
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A6	ACCBI2	
25-44	D20.1	A priori value of accelerometer bias	0.
45-59	D15.3	Period for the two periodic terms in the model associated with the time interval specified on the previous card.	S
73-80	D8.2	A priori sigma of accelerometer bias	1.D-25

IF CARD OMITTED: If the ACCBIA card is omitted, no biases will be applied to the accelerometer data. If the ACCBIA card is present, but the second BLANK card is missing the program will terminate abnormally.

# 2.5.1.6 DYNSEL

DYNSEL	2	2+3+4+5+ 2000039	)
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A 6	DYNSEL - Defines the times for which GEODYN will use external accelerations to drive the satellite. During these any accelerometer biases or attitude parameters become force model parameters.	
18-24	17	Satellite ID (required).	0
25-44	D20.8	Begin time for using external accelerometer data	YYMMDDHHMMSS.SS
45-59	D15.3	End time for using external accelerometer data.	YYMMDDHHMMSS.SS
IF CARD	OMITTED:	The satellite will be integrated as usual	and there will be

IF CARD OMITTED: The satellite will be integrated as usual and there will be  ${\sf no}$  external acceleration contribution.

# 2.5.1.7 SURFRC

SURFRC1	111111 20000	
+	0+0	+0+0
COLUMNS	FORMAT	DESCRIPTION DEFAULT VALUE & UNITS
1-6	A6	During dynamic accelerometry periods, accelerations from the external acceleration file are added to the accelerations that GEODYN computes for a satellite.  Usually, the accelerations on the external acceleration file come from an accelerometer on board the satellite.  Accelerometers measure forces that act on the surface of a satellite (like solar radiation and atmospheric drag).  During periods of dynamic acceleration, the SURFRC card allows a user to prevent the surface forces that GEODYN usually computes from internal models from the being added to the total satellite acceleration. This prevents double counting of forces during dynamic acceleration periods.  When there is a "1" in any of the columns from 7-13, during periods of dynamic acceleration GEODYN will not add the acceleration from the internal model associated with that force to the total satellite acceleration.
7	I1	Index for General Acceleration
8	I1	Index for Drag
9	I1	Index for Solar Radiation
10	I1	Index for Albedo
11	I1	Index for Topex Thermal Radiation.
12	I1	Index for Topex Louver Forces
13	I1	Lageos Thermal Drag
18-24	17	Satellite ID

NOTE: THE SURFRC cards should always be the last in the group before the  ${\tt ENDACC}$  option

 $\hbox{ IF CARD OMITTED: During dynamic acceleration periods, all surface accelerations computed from GEODYN internal models will be added to the total satellite acceleration. } \\$ 

# 2.5.1.8 ENDACC

ENDACC	1+	-2+3+4+5+6+7+8
	0+	-0+0+0+0
COLUMNS	FORMAT	DESCRIPTION DEFAULT VALUE & UNITS
1-6	A 6	ENDACC - Denotes the end of the

accelerometer date sub-group

#### 2.5.2 ACCELT

+	1+	2+3	-+4	5+	-6+	78
ACCELT3		7603901 0+0	1.80D-12	0+	-0+	0+0
COLUMNS	FORMAT	DESCRIPTIO	N		DEFAULT	VALUE & UNITS
1-6	A6	ACCELT - Requests application and/or adjustment of general accelerations in three dimensions. This is a special option that may be used to simulate thrusting by applying accelerations in the alongtrack, crosstrand radial directions at the same time. SPECIAL CARE is required in using this option because the input must be in a very specific order. (See the sample set				
7	I1	Acceleration ty 1 - Radial 2 - Crosstrac 3 - Alongtrac	k		0	
18-24	17	Satellite ID (r	equired).		0	
25-44	D20.8	A priori value	of acceleration	١.	0.	M/S**2
45-59	D15.3	End time of the this accelerati in the form YYM	on coefficient			
COLUMNS	FORMAT	DESCRIPTIO	N		DEFAULT	VALUE & UNITS
60-72	D13.1	A priori sigma	of acceleration	ı	0.	M/S**2 or M/S**3 or M/S**4
NOTES:						
[1]	order s	tion is not user hown in the exam	ple on the next	page. All	of the o	cards

pertaining to the alongtrack acceleration come first, followed by the cards pertaining to the crosstrack acceleration, followed by the cards pertaining to the radial accelerations. The times on the alongtrack cards are the only times that are used, however, times are required on the crosstrack and radial cards. The times must be in ascending order and all times specified must be within the run times specified on the EPOCH card or GEODYN IIS will throw them out.

UNITS: KM =Kilometers; M =Meters ; S =Seconds ; Kg =Kilograms

DEG=Degrees ; RAD=Radians ; AS =Arc seconds; MAS=Milli-arc seconds

IF CARD OMITTED: General accelerations are not applied.

### Sample ACCELT Setup

Following is a set of ACCELT cards used to apply two thrusting type of accelerations (thrusting in the sense that the accelerations are zero prior to and after the acceleration interval).

+1	+3	+
ACCELT3	7603901	0.0
ACCELT3	7603901	0.0 911007020000.00
ACCELT3	7603901	1.000E-06911007020200.00
ACCELT3	7603901	0.0 911007040000.00
ACCELT3	7603901	1.000E-06911007040200.00
ACCELT3	7603901	0.0 911007060000.00
ACCELT2	7603901	0.0 911017020000.00
ACCELT2	7603901	1.000E-06911017020200.00
ACCELT2	7603901	0.0 911017040000.00
ACCELT2	7603901	1.000E-06911017040200.00
ACCELT2	7603901	0.0 911017060000.00
ACCELT1	7603901	0.0 911027020000.00
ACCELT1	7603901	1.000E-06911027020200.00
ACCELT1	7603901	0.0 911027040000.00
ACCELT1	7603901	1.000E-06911027040200.00
ACCELT1	7603901	0.0 911027060000.00
+1	+3	+

Notice that the first card does not have a time period specified and that the times for the crosstrack (2) and radial (1) cards have been increased by 10 days. This will require that the end time specified on the EPOCH card be 20 days later.

Note that when this option is used, GEODYN IIE will print the thrusting acceleration vector in m/(sec\*\*2) at integration steps where the thrusting vector has a magnitude of larger than 1.0E-7(m/sec\*\*2)

# 2.5.3 ACCEL9

ACCEL99	9	7603901 1.80D-12		
+	0+	0+0+0+0	-0+-	0+0
COLUMNS	FORMAT	DESCRIPTION	DEFAULT	VALUE & UNITS
1-6	A6	ACCEL9 - Requests application and/or adjustment of general acceleration.		
9	I1	Indicates direction of 9 parameter gener acceleration	al 0	
		1 - Along Track ((R x V) x R) 2 - Cross Track (R x V) 3 - Radial (R)		
10	I1	Indicates type of 9 parameter general acceleration parameter	0	
		1 - Cosine coefficient (A) 2 - Sine coefficient (B) 3 - Constant (C)		
11-12	12	Indicates if this card will be used to specify period end time for 9 parameter general acceleration option.  See NOTE[2] for details.	0	
		99 - Card DOES specify 9 parameter general acceleration end time.		
13	I1	Thrust Index		
		<pre>0 - Acceleration 1 - Acceleration Thrust</pre>		
18-24	17	Satellite ID (required).	0	
COLUMNS	FORMAT	DESCRIPTION	DEFAULT	VALUE & UNITS
25-44	D20.8	A priori value of acceleration coefficients.	0	M/S**2 or M/S**3 or M/S**4
45-59	D15.3	This field must be left blank unless the time dependent acceleration option is wanted. To select time dependent acceleration the end time of the period for which this acceleration coefficient applies is input in the form YYMMDDHHMMSS.SS in this field.	0	
60-72	D13.1	A priori sigma of acceleration	0	. M/S**2

coefficient. or M/S\*\*3 or M/S\*\*4

#### NOTE[1]:

When choosing the nine parameter general acceleration model, a card specifying the period endtime (99 in Cols.11-12) must precede all other cards. A priori values for the coefficients in this period must follow subsequently. One no longer has to include all nine cards for each coefficient in a period.

EXAMPLE: ACCEL cards for time dependent 9 parameter general accelerations

	1	+2+	3+6	+8
ACCEL9	99	7603901	851230010000.00	
ACCEL9	11	7603901	1.1D-11	1.0D0
ACCEL9	12	7603901	2.2D-11	1.0D0
ACCEL9	21	7603901	4.4D-11	
ACCEL9	22	7603901	5.5D-11	1.0D0
ACCEL9	99	7603901	860101000000.00	
ACCEL9	31	7603901	7.7D-11	
ACCEL9	32	7603901	8.8D-11	
	0	+0	0+0+0	+0

UNITS: KM =Kilometers; M =Meters ; S =Seconds ; Kg =Kilograms DEG=Degrees ; RAD=Radians ; AS =Arc seconds; MAS=Milli-arc seconds

WARNINGS: If duplicate time periods are specified the run will terminate in IIS with a warning message.

The first time period that exceeds the end time of the run

(see EPOCH card) will be accepted. All other end times exceeding the end time of the run will be ignored.

IF CARD OMITTED: General accelerations are not applied

# 2.5.4 ACOEF

ACOEF	-	2+3+4+5+	
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A 6	ACOEF - Coefficients for Albedo or Emissivity Models.	
7-8	12	<pre>Indicator of Albedo or Emissivity model   1 = Albedo   0 = Emissivity</pre>	
9-10	12	Degree of coefficient	
11-12	12	Order of coefficient	
25-44	D20.8	C - Coefficient value	0
45-59	D15.3	S - Coefficient value	0

The coefficients above should be unnormalized.

NOTE [1]: For a setup example see note on ALTIME card.

#### 2.5.5 ALBEDO

ALBEDO	2	2+5+5+	·	,
COLUMNS	FORMAT	DESCRIPTION	DEFAULT	VALUE & UNITS
1-6	A 6	ALBEDO - Requests application of earth or other planet radiation pressure		
7-8	12	# of rings	2	
9-10	12	Model indicator	0	
		<pre>0 = default 1 = model provided below [Note 3]</pre>		
11-12	12	TUM model used if greater then zero (only applicable to GPS satellites) [Note 4]	0	
25-44	D20.8	ALBEDO constant [Note 2]		
45-59	D15.3	Emissivity constant [Note 2]		

NOTE [1]: Solar radiation must be turned on when using ALBEDO.

NOTE [2]: If the albedo constant is zero, then the spherical harmonic albedo Earth model (currently zeroeth first and second degree zonal harmonics) will be used.

If the emissivity constant is zero, then the spherical harmonic emissivity model (same as above) will be used.

NOTE [3]: For an example setup of this case look at the ALTIME card.

NOTE [4]: To use the TUM model, the following files must be present in the directory that IIE is executed:

EMITM001 EMITM002 EMITM003 EMITM004 EMITM005 EMITMO06 EMITM007 EMITM008 EMITM009 EMITMO10 EMITM011 EMITM012 REFLM001 REFLM002 REFLM003 REFLM004 REFLM005 REFLM006 REFLM007 REFLM009 REFLM010 REFLM011 REFLM012

These files can be found in the following directory on the  $\ensuremath{\mathtt{xyz2}}$  machine:

/Users/geodyn/SUPPORT/dat\_FILES/dat\_TUM/Earth\_Radiation\_Pressure

IF CARD OMITTED: Earth radiation pressure will not be applied.

# 2.5.6 **ALTIME**

+	1+	2+3	-+4+-	5+	6+-	78
ALTIME	0+	0+0	-+0+-		0	
COLUMNS F		DESCRIPTION		0		VALUE & UNITS
1-6 A	6	ALTIME - time in albedo and emis	time input card for and emissivity			
7-8 I	2	Albedo or Emissi 1 = Albedo 0 = Emissivity	vity indicat	or		
11-12 I	2	Maximum degree o	of model foll	owing this o	card	
13-14 I	2	Maximum order of	model follo	wing this ca	ard	
21-22 I	2	North latitude b	oundary for	polar caps		
23-24 I	2	South latitude b	ooundary for	polar caps		
25-44 D20	0.8	Begin solar long	gitude			degrees of sed since dnight
45-59 D1	5.3	End solar longit	ude			degree of sed since dnight
60-72 D13	3.1	Albedo or emissi	vity constan	t for north	pole	
73-80 D8	.2	Albedo or emissi	vity constan	t for south	pole	
NOTE [1]:	emissi	II input decks h vity coefficients following direct	for all sol			ls
	/geod4	/dmoore/alb				
	alb.br			alb.br205		
	alb.br alb.br			alb.br225 alb.br250		
	alb.br			alb.br274		
	emiss. emiss. emiss. emiss.	/dmoore/emiss br010 emiss.br090 br030 emiss.br110 br050 emiss.br126 br070 emiss.br130 br084 emiss.br150	emiss.br170 emiss.br185 emiss.br205	emiss.br274 emiss.br290 emiss.br310	<del>[</del> ) )	r350
NOTE [2]:	Exampl	e setup:				
ALBEDO 6 ALTIME 0 ACOEF 0 ACOEF 0	0 3 3 0 0 1 0	0.9887231	168.00 2000000D+00 .000000D-02	185. 916294200D-0		

ACOEF	0	2	0		-0.507099500000D+00
ACOEF	0	2	1		-0.916294000000D-01 .297368800D-01
ACOEF	0	2	2		0.117757000000D-00 .618260100D-02
ACOEF	0	3	0		0.614876100000D-01
ACOEF	0	3	1		0.677822100000D-01 .334011800D-02
ACOEF	0	3	2		0.219367100000D-02613645100D-02
ACOEF	0	3	3		-0.396645400000D-02 .406565900D-03
ALTIME	1	0	6	0	168.00 185.0
ACOEF	1	1	0		0.239637500000D-03
ACOEF	1	2	0		-0.945492860000D-01
ACOEF	1	3	0		0.708928660000D-01
ACOEF	1	4	0		-0.803240640000D-01
ACOEF	1	5	0		0.452530570000D-01
ACOEF	1	6	0		0.354310220000D-01

# 2.5.7 ALTWVL

ALTWVL	1+2	+3+4+5+	6+	7	+8
	0+0	+0+0+0	0+	0	+0
COLUMNS	FORMAT	DESCRIPTION	DEFAULT	VALUE	& UNITS
1-6	A 6	ALTWVL Provides the wavelenth of the altimeter laser. If altimetry data are being processed and this option is invoked, the value of the wavelength will override that provided on the data records.			microns
18-24	17	Satelltite ID.			
25-44	D20.8	Wavelength in microns.			

IF CARD OMITTED: The wavelength provided by the data will be used.

# 2.5.8 ATITUD

+	1+2	+3+4+5+	6+-	7	+8
	0+0	+0+	0+-	0	+0
COLUMNS	FORMAT	DESCRIPTION	DEFAULT	VALUE	& UNITS
1-6	A 6	Requests application and/or adjustment of attitude angles (roll, pitch, yaw).			
7	I1	<pre>Index defining the attitude angle 1 - Roll 2 - Pitch 3 - Yaw</pre>			
8	I1	Index defining the term of the polynomial.  0 - constant term  1 - Coefficient of linear term  2 - Coefficient of quadratic term  3 - Coefficient of amplitude  4 - Coefficient of phase			
10-11	13	Indicates if this card will be use to specify end time for application 99 - Card DOES specify application end time.  00 - Card provides a-priori values	n.		
12-14	13	and sigmas for parameters.  If cols 10-11 are 99 then this 3 dinteger represents the sequence of rotations for this period of time the S/C ex: 123 means first apply roll, the pitch, last yaw. 312 means first ayaw, then roll, last pitch.  If cols 10-11 are 00 this field must be filled with the laser numb For the S/C this number should be (See example).  =999 This attitude parameter is a model parameter. It works only in combination with the DYNSEL option	for  men  mpply  mer.  modern  force		
COLUMNS	FORMAT	DESCRIPTION	DEFAULT	VALUE	& UNITS
15-17	13	If cols 10-11 are 99 then this 3 digit number represents the sequence of rotations for the individual lasers and for this period of time All the individual lasers will have the same sequence of rotation. Also in this case a			

of rotation. Also in this case a

combination of the numbers 1,2,3 should be used.

18-24 I7 Satellite ID.

25-44 D20.8 If cols 10-11 are 00 this field contains the a-priori value of the coefficient.

If cols 10-11 are 99 this field

must be empty.

45-59 D15.3 If cols 10-11 are 99 this field contains the end time of application of the parameters that follow.

If cols 10-11 are 00 this field must be empty. The period that ends at the specified time, begins at the previous end time or at EPOCH time. The end time is in the form: YYMMDDHHMMSS.SS.

60-72

If cols 10-11 are 00 this field contains the standard deviation of the coefficient. If this field is zero, GEODYN will assign a very small standard deviation to

-30

the parameter  $1 \times 10$  . If cols 10-11 are 00 this field contains the period of the S/C

The example below requests adjustment of roll, pitch and yaw (complete five term polynomial: constant, linear, quadratic, amplitude and phase terms) with a sigma 1.D0 for each parameter. The S/C (id=9252010) has a laser instrument (id=001) and a solution is requested for both S/C and laser instrument at time periods:

1. EPOCH - 930219031600.00

 $2. \ 930219031600.00 \ - \ \ 930219111600.00$ 

The sequence of rotations for the S/C is: ROLL-PITCH-YAW The sequence of rotations for the laser is: ROLL-PITCH-YAW

ATITUDE	9912312	239205201	9302190316	500.00
ATITUD10	000	9205201	+.01000000000D-02	.100000D+01
ATITUD11	000	9205201	+.020000000000D-05	.100000D+01
ATITUD12	000	9205201	+.03000000000D-11	.100000D+01
ATITUD13	000	9205201	+.04000000000D-11	.100000D+01
ATITUD14	000	9205201	+.050000000000D-11	.100000D+01
ATITUD20	000	9205201	+.04000000000D-02	.100000D+01
ATITUD21	000	9205201	+.05000000000D-05	.100000D+01
ATITUD22	000	9205201	+.06000000000D-11	.100000D+01
ATITUD23	000	9205201	+.07000000000D-11	.100000D+01
ATITUD24	000	9205201	+.08000000000D-11	.100000D+01
ATITUD30	000	9205201	+.070000000000D-02	.100000D+01
ATITUD31	000	9205201	+.08000000000D-05	.100000D+01
ATITUD32	000	9205201	+.09000000000D-11	.100000D+01
ATITUD33	000	9205201	+.10000000000D-11	.100000D+01

```
000
ATITUD34
                 9205201 +.110000000000D-11
                                                                .100000D+01
ATITUD10
           001
                 9205201 +.110000000000D-02
                                                                .100000D+01
                 9205201 +.120000000000D-05
ATITUD11
           001
                                                                .100000D+01
           001
                 9205201 +.130000000000D-11
                                                                .100000D+01
ATITUD12
                 9205201 +.140000000000D-11
ATITUD13
           001
                                                                .100000D+01
                 9205201 +.150000000000D-11
ATITUD14
           001
                                                                .100000D+01
ATITUD20
           001
                 9205201 +.140000000000D-02
                                                                .100000D+01
           001
                 9205201 +.150000000000D-05
                                                                .100000D+01
ATITUD21
           001
                 9205201 +.160000000000D-11
ATITUD22
                                                               .100000D+01
           001
                 9205201 +.170000000000D-11
ATITUD23
                                                                .100000D+01
ATITUD24
           001
                 9205201 +.180000000000D-11
                                                                .100000D+01
ATITUD30
           001
                 9205201 +.170000000000D-02
                                                               .100000D+01
ATITUD31
           001
                 9205201 +.180000000000D-05
                                                                .100000D+01
           001
                 9205201 +.190000000000D-11
ATITUD32
                                                                .100000D+01
                                                                .100000D+01
ATITUD33
           001
                 9205201 +.200000000000D-11
           001
                 9205201 +.210000000000D-11
ATITUD34
                                                                .100000D+01
         991233219205201
                                              930219111600.00
ATITUD
ATITUD10
           000
                 9205201 +.010000000000D-02
                                                                .100000D+01
           000
                 9205201 +.020000000000D-05
                                                                .100000D+01
ATITUD11
ATITUD12
           000
                 9205201 +.030000000000D-11
                                                                .100000D+01
           000
ATITUD13
                 9205201 +.040000000000D-11
                                                                .100000D+01
ATITUD14
           000
                 9205201 +.050000000000D-11
                                                                .100000D+01
                 9205201 +.010000000000D-02
ATITUD20
           000
                                                                .100000D+01
           000
                 9205201 +.020000000000D-05
ATITUD21
                                                                .100000D+01
                 9205201 +.030000000000D-11
           000
ATITUD22
                                                                .100000D+01
           000
                 9205201 +.040000000000D-11
ATITUD23
                                                                .100000D+01
           000
                 9205201 +.050000000000D-11
ATITUD24
                                                               .100000D+01
ATITUD30
           000
                 9205201 +.010000000000D-02
                                                               .100000D+01
           000
                 9205201 +.020000000000D-05
ATITUD31
                                                                .100000D+01
                                                                .100000D+01
ATITUD32
           000
                 9205201 +.030000000000D-11
           000
                 9205201 +.040000000000D-11
ATITUD32
                                                                .100000D+01
                                                                .100000D+01
ATITUD42
           000
                 9205201 +.050000000000D-11
                 9205201 +.010000000000D-02
ATITUD10
           001
                                                                .100000D+01
ATITUD11
           001
                 9205201 +.020000000000D-05
                                                                .100000D+01
                 9205201 +.030000000000D-11
                                                                .100000D+01
ATITUD12
           001
           001
                 9205201 +.040000000000D-11
ATITUD13
                                                                .100000D+01
                 9205201 +.050000000000D-11
ATITUD14
           001
                                                                .100000D+01
           001
                 9205201 +.010000000000D-02
ATITUD20
                                                                .100000D+01
ATITUD21
           001
                 9205201 +.020000000000D-05
                                                                .100000D+01
ATITUD22
           001
                 9205201 +.030000000000D-07
                                                                .100000D+01
ATITUD23
           001
                 9205201 +.040000000000D-07
                                                                .100000D+01
           001
                 9205201 +.050000000000D-07
                                                                .100000D+01
ATITUD24
           001
                 9205201 +.010000000000D-11
ATITUD30
                                                               .100000D+01
ATITUD31
           001
                 9205201 +.020000000000D-05
                                                                .100000D+01
           001
                 9205201 +.030000000000D-07
ATITUD32
                                                                .100000D+01
           001
                 9205201 +.040000000000D-07
                                                               .100000D+01
ATITUD33
ATITUD34
           001
                 9205201 +.050000000000D-07
                                                                .100000D+01
```

In the same example as above, if one wishes to solve only for Yaw constant and linear terms and for the same periods of time the input reduces to :

```
991231239205201
ATITUDE
                                              930219031600.00
ATITUD30
           000
                 9205201 +.070000000000D-02
                                                                .100000D+01
           000
ATITUD31
                 9205201 +.080000000000D-05
                                                                .100000D+01
ATITUD30
           001
                 9205201 +.170000000000D-02
                                                                .100000D+01
                 9205201 +.180000000000D-05
ATITUD31
           001
                                                                .100000D+01
ATITUD
         991233219205201
                                              930219111600.00
           000
                 9205201 +.010000000000D-02
                                                                .100000D+01
ATITUD30
```

ATITUD31	000	9205201	+.020000000000D-05	.100000D+01
ATITUD30	001	9205201	+.010000000000D-11	.100000D+01
ATITUD31	001	9205201	+.020000000000D-05	.100000D+01

In the above case GEODYN will assign zero values and sigmas =1.D-30 to all the other parameters in the model. In general, if only one parameter in the model has been assigned a sigma, GEODYN will use a full model parameterization scheme with tight sigmas for all the other parameters.

Here is an example of Attitude parameters that will be considered as force model parameters. This capability is possible only along with the DYNSEL accelerometer option.

ATITUD	9999999	92000039		0	00080800000	0.00
ATITUD10	000	2000039	0.1010000			1.0
ATITUD11	000	2000039	0.0000000			1.0
ATITUD12	000	2000039	0.0000000			
ATITUD13	000	2000039	0.0000000			
ATITUD14	000	2000039	0.0000000			
ATITUD20	000	2000039	0.0150000			1.0
ATITUD21	000	2000039	0.0000000			1.0
ATITUD22	000	2000039	0.0000000			
ATITUD23	000	2000039	0.0000000			
ATITUD24	000	2000039	0.0000000			
ATITUD30	000	2000039	0.0000000			
ATITUD31	000	2000039	0.0000000			
ATITUD32	000	2000039	0.000000			
ATITUD33	000	2000039	0.0000000			
ATITUD34	000	2k00039	0.0000000			

IF CARD OMITTED: Attitude modeling will not be invoked in GEODYN

# 2.5.9 **CAMERA**

CAMERA		2+3+5+6+8 0+0+0+0
COLUMNS	FORMAT	DESCRIPTION DEFAULT VALUE & UNITS
1-6	A 6	CAMERA - Requests application and/or adjustment of camera distortion parameters
7	I1	Parameter Index (1-8)
		<pre>1-5 : Focal length/Distortion parameter           (Adjustable) 6 : Center Pixel Coordinates           (Unadjustable) 7-8 : K-Matrix parameter           (Unadjustable)</pre>
18-24	17	Camera ID number
25-44 45-59		FOR PARAMETER INDICES 1-5 Parameter value (a-priori) Parameter sigma
25-44 45-59		FOR PARAMETER INDEX 6 First Center Pixel Coordinate Second Center Pixel Coordinate
25-44 45-59 60-72		FOR PARAMETER INDEX 7 K-MATRIX(1) K-MATRIX(2) K-MATRIX(3)
25-44 45-59 60-72		FOR PARAMETER INDEX 8 K-MATRIX(4) K-MATRIX(5) K-MATRIX(6)

# EXAMPLE: CAMERA cards to define all camera parameters:

1-	+	3+-	5+6	+8
CAMERA1	2007043	150.07328	0.0004D-03	
CAMERA2	2007043	000000D-06	0.04308D-03	
CAMERA3	2007043	000000D-06	0.04308D-03	
CAMERA4	2007043	000000D-06	0.04308D-03	
CAMERA5	2007043	000000D-06	0.04308D-03	
CAMERA6	2007043	512.5D0	512.5D0	
CAMERA7	2007043	71.4286D0	0.0000D0	0.0D0
CAMERA8	2007043	0.00386D0	-71.44341D0	0.0D0
0-	+(		0+0	+0+0

# 2.5.10 CON9PA

CONSPA		0+0+	
COLUMNS		DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A 6	Introducing constraints for time dependent 9 parameter general acceleration	
7	I1	Direction of 9 parameter general acceleration	
		<pre>1 = along track 2 = cross track 3 = radial</pre>	
8	I1	Type of 9 parameter general acceleration	
		<pre>1 = cosine coefficient 2 = sine coefficient 3 = constant</pre>	
9	I1	[2] Indicator used to indicate if the normal matrix will be updated in the last inner iteration according to constraints:	
		<ul> <li>0 - IIE will NOT update the normal matrix before writing EMATRIX file.</li> <li>1 - IIE will update the normal matrix before writing EMATRIX file.</li> </ul>	,0,
18-24	17	Satellite ID.	
25-44	D20.3	Begin time of adjacent [1] periods to be constrained.	YYMMDDHHMMSS.SS
45-59	D15.3	End time of adjacent periods to be constrained.	YYMMDDHHMMSS.SS
60-72	D13.1	Weight assigned to adjacent periods	
73-80	D8.3	Correlation time.	sec
NOTE [1]	periods exclude	ne CON9PA card is needed to constrain a number of ACCEL9. If some periods from the Acced from constrained adjustments, use as managed and sets of continuous adjacent periods	CCEL9 stream are ny CON9PA cards as

NOTE [2]: If a value other than 0 or 1 is present in column 9 GEODYN IIS will stop and show an error message.

#### 2.5.11 CONBIA

+	1+	2+5+	-68
COLUMNS		0+0+0+0 DESCRIPTION	O+O+O DEFAULT VALUE & UNITS
1-6	A 6	Introducing constraints for time dependent tropospheric biases.	
7-14	18	Station ID	
15-17	13	Type of bias to be constrained	
		=500 for tropospheric biases	
25-44	D20.3	Begin time of adjacent [2] periods to be constrained.	YYMMDDHHMMSS.SS
45-59	D15.3	End time of adjacent periods to be constrained.	YYMMDDHHMMSS.SS
60-72	D13.1	Weight assigned to adjacent periods.	
73-80	D8.3	Correlation time.	sec

- NOTE [1]: The CONBIA option is station dependent, therefore the station ID is required.
- NOTE [2]: Only one CONBIA card is needed to constrain a number of adjacent periods from MBIAS. If some periods from the MBIAS stream are excluded from constrained adjustments, use as many CONBIA cards as the number of sets of continuous adjacent periods.

## 2.5.12 **CONDRG**

CONDRG	-	2+5+5	
COLUMNS	-	0+0+0+0+ DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A6	Introducing constraints for time dependent drag.	
9	I1	[2] Indicator used to indicate if the normal matrix will be updated in the last inner iteration according to constraints:	
		<ul> <li>0 - IIE will NOT update the normal matrix before writing EMATRIX file.</li> <li>1 - IIE will update the normal matrix before writing EMATRIX file.</li> </ul>	,0,
18-24	17	Satellite ID.	
25-44	D20.3	Begin time of adjacent [1] periods to be constrained.	YYMMDDHHMMSS.SS
45-59	D15.3	End time of adjacent periods to be constrained.	YYMMDDHHMMSS.SS
60-72	D13.1	Weight assigned to adjacent periods.	
73-80	D8.3	Correlation time.	sec

- NOTE [1]: Only one CONDRG card is needed to constrain a number of adjacent periods for time dependent DRAG. If some periods from the DRAG stream are excluded from constrained adjustments, use as many CONDRG cards as the number of sets of continuous adjacent periods.
- NOTE [2]: If a value other than 0 or 1 is present in column 9 GEODYN IIS will stop and show an error message.

## 2.5.13 CONSOL

+ CONSOL	1+	2+3+4+5+	-68
	0+	0+0+0	-0+0
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A 6	Introducing constraints for time dependent solar radiation.	
18-24	17	Satellite ID.	
25-44	D20.3	Begin time of adjacent [1] periods to be constrained.	YYMMDDHHMMSS.SS
45-59	D15.3	End time of adjacent periods to be constrained.	YYMMDDHHMMSS.SS
60-72	D13.1	Weight assigned to adjacent periods.	
73-80	D8.3	Correlation time.	sec

NOTE [1]: Only one CONSOL card is needed to constrain a number of adjacent periods for time dependent SOLRAD. If some periods from the SOLRAD stream are excluded from constrained adjustments, use as many CONSOL cards as the number of sets of continuous adjacent periods.

## 2.5.14 CONSPL

	1	2+3+-	45+	-67-	+8
CONSPL	20001304	* · <del>-</del>	0.1	0.1	0.1
	O+ FORMAT		0+	-O+O- DEFAULT VALU	
1-6	A 6	CONSPL - Input in parameters constra	formation for VLBI splints.	ine	
7-14	18	Station Number.		0	
15-17	13	Bias Type. See lis this option descri	t of bias types after ption.	0	
25-44	D20.8	Standard deviation	for degree 0		
45-59	D15.3	Standard deviation	for degree 1		
60-72	D13.1	Standard deviation	for degree 2		
73-80	D8.2	Standard deviation	for degree 3		
UNITS:			; S = Seconds ; K ; AS = Arc seconds; M	•	seconds

IF CARD OMITTED: No constraints will be applied to the spline type biases.

## 2.5.15 DSTATE

+ DSTATE	1+	2+5+6-	8
	0+	0+0+0+0	+0
COLUMNS	FORMAT	DESCRIPTION	EFAULT VALUE & UNITS
1-6	A 6	DSTATE - Requests estimation of an instantaneous state difference at a specific time. This capability is extremely useful with thrusts	
7	I1	Dstate Component	
		1 = x	
		2 = y	
		3 = z	
		4 = x	
		5 = y	
		6 = z	
8	I1	Coordinate system of input indicator	
		<ul><li>=1 Inertial TOR coordinate system</li><li>=2 Position and velocity are in the satellite along track, cross track and radial directions.</li></ul>	
18-24	17	Satellite ID.	0
25-44	D20.8	Dstate value	0.0
45-59	D15.3	Dstate time (YYMMDDHHMMSS.SS)	
60-72	D13.1	Dstate sigma	1.D-25
NOTE 1:	should the sat cards. Not all present GEODYN	ATE option requires that all the data from of the input together and in advanced time order ellites must be in the same order as on the 6 components of the DSTATE at a specific time of the user does not wish to solve for them will use a tight sigma for those components sent in the estimation process	r. Also all SATPAR input ime must be 1, however
NOTE 2:	The DST	ATE option is a force model option, but no is	

are associated with the DSTATE parameters. The DSTATE parameters

simply cause the COWELL numerical integration sums to be altered. There are two integration scenarios: INTGCB option and no INTGCB option (see INTGCB card). The user needs to be

aware of these two scenarios when using DSTATE parameters.

- NOTE 2a: When no INTGCB card is present only the orbit of the artificial satellite (and not the orbited body) is being numerically integrated. In this mode, the DSTATE option requires the use of multi-rate integration. Multi-rate integration is accessed by using in columns 60-72 of the STEP card. Note that unlike most applications that require multi-rate integration, the DSTATE option does NOT require the use of the HRATEF card.
- NOTE 2b: When an INTGCB card is present, both the orbit of the artificial satellite well as the orbited body are being numerically integrated. In this mode, multi-rate integration is not possible and prior to version 1507.11 DSTATE modeling was not available together with INTGCB. Starting with version 1507.11 DSTATE modeling without multi-rate integration is possible when using the INTGCB capability. In this mode, the DSTATE option requires that a 1 second step size is used. If a DSTATE epoch occurs in-between integration steps, the DSTATE event will be modeled at the nearest integration step.

## 2.5.16 DRAG

		2+3+4+5+6 7654321 2.3	-+7	+8
+	0+	0+0+0+0	-+0	+0
COLUMNS	FORMAT	DESCRIPTION DEFA	ULT VAL	UE & UNITS
1-6	A 6	DRAG - Modifies drag application and/or requests estimation of drag coefficients.		
8	I1	Indicates order of drag coefficients:	0	
		0 - Cd		
		1 - Cd		
		2 - Cd		
18-24	17	Satellite ID (required).	0	
25-44	D20.8	Drag coefficient.(For Cd and Cd the units are per second and per second squared)	0.	none or 1/S or 1/S**2
45-59	D15.3	This field must be left blank unless the time dependent drag option is wanted. To select time dependent drag specify the end time of the period for which this acceleration coefficient applies in the form YYMMDDHHMMSS.SS in this field. See NOTE[1] for more details.	0.	
60-72	D13.1	Standard deviation of drag coefficient.		none or 1/S or 1/S**2

### NOTE[1]:

Time periods may be specified only for the highest order coefficients used (i.e. if Cd and Cd dot are used, only Cd dot can have time intervals).

When using time periods the number of drag cards required is equal to the number of time periods plus the order (order = the number in column 8 + 1).

The first DRAG card with an end time covers the time period from epoch to the specified end time. Subsequent drag cards with end times cover the time interval from the preceeding end time to the current end time. In this manner continuous time intervals are guaranteed from epoch to the last specified end time. If orbit integration continues beyond the end of the last specified drag time interval, the Cd ( or Cd dot, or Cd double dot) without a time interval is used. A Cd (or Cd dot or Cd double dot) without a time interval is MANDATORY when using this time dependent option. See example below.

EXAMPLE: Drag cards for using time dependent drag coefficients.

(highest order - Cd dot; two time intervals; 4 cards required)

+-	1-	+2	+3	+	4	5	-+	-6	-+	-7	-+	-8
DRAG	0	7654321	2.3									
DRAG	1	7654321	.00002									
DRAG	1	7654321	.000021		84	090200	0000.					
DRAG	1	7654321	.000022		84	090300	0000.					
+-	0-	+	+0	+(	)	0	-+	-0	-+	-0	-+	-0

 $\begin{tabular}{ll} WARNINGS: If duplicate time periods are specified the run will terminate \\ in IIS with a warning message. \end{tabular}$ 

The first time period that exceeds the end time of the run (see EPOCH card) will be accepted. All other end times exceeding the end time of the run will be ignored.

UNITS: KM = Kilometers; M = Meters ; S = Seconds ; Kg = Kilograms

DEG=Degrees ; RAD=Radians ; AS =Arc seconds; MAS=Milli-arc seconds

IF CARD OMITTED: Drag is applied with a default Cd of 2.3

## 2.5.17 EBIAS

1	+2	+3	+4	+5	+6	-+8
EBIAS	51	1.				
		+	+0	+	+	-+0+0

COLUMNS	FORMAT	DESCRIPTION	DEFAULT	VALUE	&	UNITS
1-6	A6	EBIAS - Requests adjustment of pass by pass measurement biases.				
7-14	18	Station Number. [2]	0			
15-17	13	Bias Type. See list of bias types after this option description. [2]	0			
18-24	17	Satellite ID. [2]	0			
45-59	D15.3	Start date and time for generating pass by pass biases (YYMMDDHHMMSS). See NOTE	0			
60-72	D13.1	End date and time for generation of pass by pass biases (YYMMDDHHMMSS). See NOTE				

- NOTE [1]: If the start and end dates are left blank electronic biases will be generated for the entire data span.
- NOTE [2]: The station number, satellite ID and bias type are required.

  Blanks and /or zeroes will not work.
- NOTE [3]: If the measurement type is larger than 99 (3 digit) use EBIASM card to specify it.

### GEODYN-II E-BIAS TYPES

The  ${\tt GEODYN-II}$  E-bias types include all of the  ${\tt GEODYN-II}$  measurement types as well as other types which are an extension of the simple measurement bias.

At present, the only E-bias types defined include simple constant measurement biases. Other future E-bias types will include linearly varying measurement biases and observation scale biases.

E-bias type numbers are assigned as follows:

BIAS DESCRIPTION	BIAS TYPE NUMBER	SATELLITE DEPENDENT	STATION DEPENDENT
simple measurement biases	measurement type	no o	r yes
measurement bias varying linearly with time	meas. type + 100	no o	r yes
measurement scale bias	meas. type + 200	no o	r yes

tropospheric refraction scale bias

500 no or yes

## 2.5.18 EBIASM

EBIASM	100	2+3+4+5+		,
COLUMNS	FORMAT	DESCRIPTION	DEFAULT	VALUE & UNITS
1-6	A 6	EBIASM - This card is used when the measurement type is larger than 99 (3 digit). The 3 digits in columns 15-17 will overwrite the measurement type in columns 16-17 of the PBIAS card. Requests adjustment of pass by pass measurement biases.		
15-17	13	Measurement Type.	0	
IF CARD	OMITTED:	Measurement type will be read from cols 3	16-17 of	the

EBIAS card.

### 2.5.19 EDIT

EDIT		3.5 200. 0+0+0+	
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A 6	EDIT - Specifies editing multiplier and initial RMS value.	
7	I1	Special editing switch.	0
		<ul> <li>= 0 Normal editing performed.</li> <li>= 1 Perform local pass editing on 1st iteration.</li> <li>= 2 Perform local pass editing on all iterations.</li> </ul>	
25-44	D20.8	Editing multiplier. This number remains constant for all iterations in the arc.	3.5
45-59	D15.3	Editing RMS. Initial RMS value used for iteration number 1. Subsequent iteration use the RMS from the previous iteration	
60-72	D13.1	Electronic bias edit level.	Value in cols. 25-44 times value in cols. 45-59.
73-80	D8.2	Arc convergence criteria.	.02

#### NOTES:

[1] The editing level for each iteration is determined as follows:

Edit level = Edit multiplier \* Edit RMS

- [2] The "ratio to sigma" for each observation is compared to this edit level and "ratio to sigmas" greater than the edit level are eliminated from the current iterations solution.
- [3] The electronic bias edit level defaults to the same edit level as described above. (edit multiplier \* edit RMS)

## NOTES: (continued)

- [4] Time periods may be specified only for the highest order coefficients
- [5] Convergence on first iteration GEODYN II will converge on the first inner iteration if the RMS on the EDIT card is within 2% of the first iteration RMS
- [6] E-bias Edit level For E-bias editing the E-bias edit level is given by default

as the edit multiplier \* RMS from the EDIT card. If the "electronic bias edit level" from columns 60-72 of the EDIT card is non zero, then the value from column 60-72 is used as the edit level (Not the value in column 60-72 times anything else)

IF CARD OMITTED: The default edit level of 700 (3.5\*200) is used.

## 2.5.20 EMATRX

+	1+	2+3+4+5+	-68		
EMATRX	1	123456.			
+	0+	0+0+0+0	-0+0		
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS		
1-6	A6	EMATRX - Requests output of an E-matrix file on unit 71 and on option a V-matrix file on unit 80. Also automatically requepartitioned formation of normal equations			
8	I1	> 0 indicates a V-matrix file will also be output.	0		
9	I1	<pre>E-matrix print options 0 - E-matrix is not printed. 1 - E-matrix header and label     information is printed. 2 - Complete E-matrix is printed.</pre>	0		
15-17	13	Number of processors to be used for an EMAT run			
25-44	D20.8	E-matrix number. This number may be any number greater than 0.	0.		
45-59	D15.3	Stop date for V-matrix output in form YYMMDDHHMMSS.	0.		
60-72	D13.1	Output interval in seconds for the V-matr If a V-matrix has been requested this number must be greater than zero in order to get a V-matrix at regular times. If the V-matrix is requested at measurement time this number should be greater than negatione and less than or equal to zero. V-matrix partials will be output in the treference coordinate system.	ue s, ve		
73-80	D8.2	Sigma used in the case of a Reduced Norm case. (NOTE 1)	al Matrix		
NOTE 1: GEODYN considers the adjusted parameters as a group of parameters where all adjust with their assigned sigma value.  Some parameters are often tightly constrained and do not change with the adjustments. This scenario may produce very large normal matrices. With this option any GEODYN parameter for which the sigma is less than a certain constant, will not be contributing in the normal matrix.					

DEG=Degrees ; RAD=Radians ; AS =Arc seconds; MAS=Milli-arc seconds

UNITS: KM = Kilometers; M = Meters ; S = Seconds ; Kg = Kilograms

IF CARD OMITTED: No E-matrix of V-matrix will be output.

SEE ALSO: VECOPT option in Global Set.

NOTE: [1] The input elements for the S/C in an interplanetary run must refer to the IAU system otherwise V-matrix values will be erroneous. See ELEMS1 and ELEMS2 cards.

## 2.5.21 FANTIM

+1	2	3+	4	5+6	68
FANTIM					
	0	0	)+(	0(	0+0

COLUMNS	FORMAT	DESCRIPTION DEFAULT VALUE &	UNITS
1-6	A 6	FANTIM - Used for time dependent PHANTOM parameters (option FANTOM)	
7	I1	Index specifying the nature of the introduced arc parameters:	
		<pre>= 1 Geometric model arc parameters = 2 Force model arc parameters</pre>	
25-44	D20.8	Epoch time for a time dependent model YYMMDDHHMM [NOTE 1]	ISS.
45-59	D15.3	Start time for a time dependent option. The YYMMDDHHMM end times are specified on the FANTOM cards.	ISS.
60-72	D13.1	Space for real information applicable to all the parameters in the group.	
73-80	D8.2	Space for real information applicable to all the parameters in the group	

## NOTES:

- [1] This time will be considered to be a reference time with respect to which linear rates or periodic terms are evaluated.
- [2] Only one FANTIM card per FANTOM parameter group is allowed.

IF CARD OMITTED: And times are included on the FANTOM cards, the run will abnormally terminate.

## CG OFFEST PARAMETERS USING FANTIM:

This option uses FANTOM cards to model the center of gravity offset in the x, y and z components in the satellite body-fixed frame.

### FANTIM

-----

col. 7	1 - means arc geometric FANTOM parameters	
col. 25-44	Epoch time for a time dependent model [NOTE 1]	YYMMDDHHMMSS.
col. 45-59	Start time for a time dependent option. The end times are specified on the FANTOM cards.	YYMMDDHHMMSS.

See ARC FANTOM OPTION FOR THE FANTOM SPECIFICATIONS

## 2.5.22 FANTOM

+ FANTOM	1+	2+3+8
+	0+	0+0+0
COLUMNS	FORMAT	DESCRIPTION DEFAULT VALUE & UNITS
1-6	A6	FANTOM - Introduces the application and/or estimation of a group of arc parameters [NOTE 3]
7	I1	Index specifying the nature of the introduced arc parameters:
		<pre>= 1 Geometric model arc parameters = 2 Force model arc parameters</pre>
8	I1	Index specifying a specific direction for the parameter appearing on cols 25-44 [NOTE 1]
		= 0 All the parameters in the group are treated the same way (same partial formulation)
		<pre>= 1    = 2    To be specified by the user = 3   </pre>
9	I1	Index specifying the nature of the contents on cols 60-72 and 73-80
		<pre>= 0 These real fields are not used for the group of parametrs introduced. (Space will not be allocated for them in GEODYN). = 1 These real fields are used as they are described below = 2 These real fields may be used to include any type of real information (but time)</pre>
18-24	17	Satellite ID
25-44	D20.8	Parameter value
45-59	D15.3	Parameter sigma
60-72	D13.1	End time for application for this parameter. YYMMDDHHMMSS. (if the index on col 9 is 2) [NOTE 2]
73-80	D8.2	Available space for extra real information.

## NOTES:

[1] Specify up to three (3) dimensions. Although this option is not required in order to solve for parameters in a 3-D space or solve

for parameters of different significance (S and S coefficients for example) it is helpful for EMATS and other types of labeling. Also parameters with the same index will be grouped together, facilitating that way the formation of partials.

- [2] Requires the presence of the FANTIM option.
- [3] This option provides the necessary allocation and links for parameterizaton. In order to use this option, the user must contact a GEODYN programmer for further additions of modeling and partials code.

IF CARD OMITTED: No extra model or parameters will be included in the  ${\tt GEODYN}$  calculations

#### CG OFFEST PARAMETERS USING FANTIM:

This option uses FANTOM cards to model the center of gravity offset in the x, y and z components in the satellite body-fixed frame.

#### FANTOM

-----

- col. 7 1 means arc geometric FANTOM parameters
- col. 8 component number (x = 1, y = 2, z = 3)
- col. 9 must be = 2 to specify col. 60-72 contains the sinusoidal period
- col. 18-24 Satellite ID
- col. 25-44 Parameter value
- col. 45-59 Parameter sigma [NOTE 2]
- col. 60-72 Period for the sinusoidal function in sec.

### NOTES:

- [1] This time will be considered to be a reference time with respect to which linear rates or periodic terms are evaluated.
- [2] The parameter sigma must always be non-zero for this model. If a FANTOM parameter to be adjusted, set the sigma to a very small positive number.

It is assumed that you will have 15 FANTOM cards, one for each of the parameters in the The cgmass offset is modeled in the satellite body fixed frame as:

#### where:

a(i) is the value on the ith FANTOM card
period(i) is the period on the ith FANTOM card

The periods on cards 1, 6 and 11 are not used.

An example set of FANTOM cards is:

FANTIM1	9	0802140000.	9080214000	00.	
FANTOM112	8032001	0.00	1.00	E-02	0.
FANTOM112	8032001	0.00	1.00	E-02	3600.
FANTOM112	8032001	0.00	1.00	E-02	3600.
FANTOM112	8032001	0.00	1.00	E-02	7200.
FANTOM112	8032001	0.00	1.00	E-02	7200.
FANTOM122	8032001	0.00	1.00	E-02	0.
FANTOM122	8032001	0.00	1.00	E-02	3600.
FANTOM122	8032001	0.00	1.00	E-02	3600.
FANTOM122	8032001	0.00	1.00	E-02	7200.
FANTOM122	8032001	0.00	1.00	E-02	7200.
FANTOM132	8032001	0.00	1.00	E+02	0.
FANTOM132	8032001	0.00	1.00	E-02	3600.
FANTOM132	8032001	0.00	1.00	E-02	3600.
FANTOM132	8032001	0.00	1.00	E-02	7200.
FANTOM132	8032001	0.00	1.00	E-02	7200.

# 2.5.23 FNBURN SUBGROUP

FLRATE2

FNBURN 9601101 200402082905.0000 .100000D+20
COLUMNS FORMAT DESCRIPTION DEFAULT VALUE & UNITS  1-6 A6 FNBURN - Introduces the application and/or estimation of finite burn parameters  FNBURN is the first card in a group of cards which also contains the cards THRUST, FLRATE, THRRTA, THRDEC, and end ends with the card ENDBRN.  Each of the cards above is described separately.  An example of the FNBURN group is given below (NOTE 1)  18-24 I7 Satellite ID
COLUMNS FORMAT DESCRIPTION DEFAULT VALUE & UNITS  1-6 A6 FNBURN - Introduces the application and/or estimation of finite burn parameters FNBURN is the first card in a group of cards which also contains the cards THRUST, FLRATE, THRRTA, THRDEC, and end ends with the card ENDBRN. Each of the cards above is described separately. An example of the FNBURN group is given below (NOTE 1)  18-24 I7 Satellite ID
1-6 A6 FNBURN - Introduces the application and/or estimation of finite burn parameters FNBURN is the first card in a group of cards which also contains the cards THRUST, FLRATE, THRRTA, THRDEC, and end ends with the card ENDBRN. Each of the cards above is described separately. An example of the FNBURN group is given below (NOTE 1)  18-24 I7 Satellite ID
estimation of finite burn parameters  FNBURN is the first card in a group of cards which also contains the cards THRUST, FLRATE, THRRTA, THRDEC, and end ends with the card ENDBRN. Each of the cards above is described separately. An example of the FNBURN group is given below (NOTE 1)  18-24 I7 Satellite ID
FNBURN is the first card in a group of cards which also contains the cards THRUST, FLRATE, THRRTA, THRDEC, and end ends with the card ENDBRN. Each of the cards above is described separately. An example of the FNBURN group is given below (NOTE 1)  18-24 I7 Satellite ID
which also contains the cards THRUST, FLRATE, THRRTA, THRDEC, and end ends with the card ENDBRN. Each of the cards above is described separately. An example of the FNBURN group is given below (NOTE 1)  18-24 I7 Satellite ID
THRDEC, and end ends with the card ENDBRN.  Each of the cards above is described separately.  An example of the FNBURN group is given below (NOTE 1)  18-24 I7 Satellite ID
Each of the cards above is described separately. An example of the FNBURN group is given below (NOTE 1)  18-24 I7 Satellite ID
18-24 I7 Satellite ID
25-44 D20 8 Dato in VVMMDDUUMMGG GGGG HT
OF-NA DOU 8 Doto in AAMDDAMMGG GGGG NA
25 44 μ20.0 ματε τη ιιμημημημημορ. 2020 01
45-59 D15.3 Parameter sigma
45-59 D15.3 Parameter sigma
[NOTE 1]:
FNBURN 1606401 181015170000.3990 1.000000D-50
THRUST1 1.0922824000D+02 5.282000D-50
THRUST2 4.1985467000D+00 1.000000D-50
THRUST3 -3.9539867000D-01 1.000000D-50
THRUST4 1.5605763000D-02 1.000000D-50
THRUST5 -2.1846148000D-04 1.000000D-50
FLRATEO 1.4917600000D+03 1.000000D-50
FLRATE1 5.3764724000D-02 2.301200D-55
FLRATE2 0.00000000D+00 1.00000D-50
FLRATE3 0.00000000D+00 1.00000D-50
FLRATE4 0.00000000D+00 1.00000D-50
THRRTA1 8.3447575000D+01 4.332500D-55 THRRTA2 0.00000000D+00 1.00000D-50
THRRTA2 0.00000000D+00 1.00000D-50 THRRTA3 0.0000000D+00 1.00000D-50
THRRTA4 0.00000000D+00 1.00000D-50
THRRTA5 0.000000000D+00 1.00000D-50
THRDEC1 7.5908608000D+00 4.332500D-55
THRDEC2 0.00000000D+00 1.00000D-50
THRDEC3 0.00000000D+00 1.00000D-50
THRDEC4 0.00000000D+00 1.00000D-50
THRDEC5 0.00000000D+00 1.00000D-50
ENDBRN1 3.000000000D+01 1.284000D-55
ENDBRN 1.000000E-50
ENDBRN 1.000000E-50
ENDBRN 1.000000E-50
FNBEND 1606404 181015170030 3000 1 00000D 50
FNBURN 1606401 181015170030.3990 1.000000D-50 THRUST1 8.6930216000D+02 4.114100D+01
THRUST2 0.000000000D+00 1.00000D+01
THRUST3 0.000000000D+00 1.00000D+01 THRUST3 0.00000000D+00 1.00000D+01
THRUST4 0.000000000D+00 1.00000D+01
THRUST5 0.000000000D+00 1.00000D+01
FLRATEO 1.4901440000D+03 1.000000D-50
FLRATE1 3.6994717000D-01 1.792400D-55

0.000000000D+00 1.00000D-50

FLRATE3	0.00000000D+00	1.00000D-50
FLRATE4	0.00000000D+00	1.00000D-50
THRRTA1	8.3501789000D+01	2.731100D-55
THRRTA2	0.00000000D+00	1.00000D-50
THRRTA3	0.00000000D+00	1.00000D-50
THRRTA4	0.00000000D+00	1.00000D-50
THRRTA5	0.00000000D+00	1.00000D-50
THRDEC1	7.7018506000D+00	2.731100D-55
THRDEC2	0.00000000D+00	1.00000D-50
THRDEC3	0.00000000D+00	1.00000D-50
THRDEC4	0.00000000D+00	1.00000D-50
THRDEC5	0.00000000D+00	1.00000D-50
ENDBRN1	2.2436700000D+02	7.479700D-01
ENDBRN		1.00000E-50
ENDBRN		1.000000E-50
ENDBRN		1.000000E-50
FNBEND		

#### 2.5.23.2 THRUST ----+---1----+---2----+---3----+---4----+---5----+---6----+---7----+---8 THRUST1 0.5 .100000D+20 ---+---0---+---0 COLUMNS FORMAT DESCRIPTION DEFAULT VALUE & UNITS THRUST - Defines the thrust coefficients' 1 - 6A6 7 Ι1 Order of coefficient 25 - 44D20.8 A-priori value (newtons/sec\*\*order) 45 - 59D15.3 Parameter sigma Sample FNBURN group setup featuring THRUST FNBURN 1606401 181015170000.3990 1.00000D-50 THRUST1 1.0922824000D+02 5.282000D-50 THRUST2 4.1985467000D+00 1.00000D-50 THRUST3 -3.9539867000D-01 1.00000D-50 THRUST4 1.5605763000D-02 1.00000D-50 THRUST5 -2.1846148000D-04 1.00000D-50 FLRATEO 1.491760000D+03 1.00000D-50 FLRATE1 5.3764724000D-02 2.301200D-55 FLRATE2 0.00000000D+00 1.00000D-50 **FLRATE3** 0.000000000D+00 1.00000D-50 FLRATE4 0.00000000D+00 1.00000D-50 THRRTA1 8.3447575000D+01 4.332500D-55 THRRTA2 0.000000000D+00 1.00000D-50 THRRTA3 0.000000000D+00 1.00000D-50 THRRTA4 0.00000000D+00 1.00000D-50 THRRTA5 0.00000000D+00 1.00000D-50 THRDEC1 7.5908608000D+00 4.332500D-55 THRDEC2 0.00000000D+00 1.00000D-50 THRDEC3 0.00000000D+00 1.00000D-50 THRDEC4 0.00000000D+00 1.00000D-50 THRDEC5 0.00000000D+00 1.00000D-50 ENDBRN1 3.000000000D+01 1.284000D-55 ENDBRN 1.00000E-50 ENDBRN 1.00000E-50 ENDBRN 1.00000E-50 FNBEND FNBURN 1606401 181015170030.3990 1.00000D-50 THRUST1 8.6930216000D+02 4.114100D+01 THRUST2 0.000000000D+00 1.00000D+01 THRUST3 0.000000000D+00 1.00000D+01 0.000000000D+00 THRUST4 1.00000D+01 THRUST5 0.00000000D+00 1.00000D+01 FIRATEO 1.4901440000D+03 1.00000D-50 FLRATE1 3.6994717000D-01 1.792400D-55

1.00000D-50

1.00000D-50

1.00000D-50

2.731100D-55

1.00000D-50

1.00000D-50

1.00000D-50

0.00000000D+00

0.00000000D+00

0.000000000D+00

8.3501789000D+01

0.00000000D+00

0.00000000D+00

0.000000000D+00

FLRATE2

**FLRATE3** 

FI.RATE4

THRRTA1

THRRTA2

THRRTA3

THRRTA4

THRRTA5	0.00000000D+00	1.00000D-50
THRDEC1	7.7018506000D+00	2.731100D-55
THRDEC2	0.00000000D+00	1.00000D-50
THRDEC3	0.00000000D+00	1.00000D-50
THRDEC4	0.00000000D+00	1.00000D-50
THRDEC5	0.00000000D+00	1.00000D-50
ENDBRN1	2.2436700000D+02	7.479700D-01
ENDBRN		1.00000E-50
ENDBRN		1.00000E-50
ENDBRN		1.00000E-50
FNBEND		

# 2.5.23.3 FLRATE

FLRATEO	2+3+4+5+6+7+8 5.6 .100000D+200+0+0+0
COLUMNS FORMAT	DESCRIPTION DEFAULT VALUE & UNITS
1-6 A6	FLRATE - Defines mass flow rate coefficients
7 I1	Type of parameter =0 mass at TO (Kg) =1-4 order of coefficient
25-44 D20.8	A-priori value of derivative of the mass flow (kg/sec**order)
45-59 D15.3	Parameter sigma

# 2.5.23.4 THRRTA

THRRTA1		2+3+4- 0.0	.10000D+	20
COLUMNS	FORMAT	DESCRIPTION		DEFAULT VALUE & UNITS
1-6	A 6	THRRTA - Defines coeff	icients for Thrus	st Right Ascension
7	I1	- Order of coef	ficient	
25-44	D20.8	A-priori value (degrees	3)	
45-59	D15.3	Parameter sigma		

# 2.5.23.5 THRDEC

THRDEC1	-	2+3+4+5+6+7+8 0.007
COLUMNS	FORMAT	DESCRIPTION DEFAULT VALUE & UNITS
1-6	A6	THRDEC - Defines coefficients for Thrust Declination
7	I1	- Order of coefficient
25-44	D20.8	A-priori value (degrees)
45-59	D15.3	Parameter sigma

2.5.23.6	FNBEND		
+ FNBEND	12	+5	+6+8
+	0+0	+0+0	+0+0
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS

1-6 A6 FNBEND - Defines the end of the FNBURN group

## 2.5.23.7 ENDBRN

ENDBRN1	2+5+ 0.007 .10000D- 0+0+0	+20
COLUMNS FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6 A6	FNBEND - Defines burn termination para	ameters
7 Ii	<pre>- index to denote parameter =1 DT =2 DV =0 Additional parameters (3) if 7=2</pre>	(DV)
25-44 D20.8	A-priori value (degrees)	
45-59 D15.3	Parameter sigma	

### 2.5.24 **GPSBXW**

```
---+---1---+---2---+---3----+---4----+---5----+---6----+---7----+---8
GPSBXW
---+---0---+---0
COLUMNS FORMAT
                      DESCRIPTION
                                                         DEFAULT VALUE & UNITS
1-6
       A6
                 GPSBXW - Requests application and/or
                 adjustment of parameters associated with
                 the GPS satellite Box-wing TUM (Technical
                 University of Munich) model
7
       Ι1
                 Parameter Type
                         Solarpanel scale factor \varphi
                                                       SEE NOTE BELOW
                    = 1
                    = 2 Solarpanel rotation lag \theta
                    = 3 Box wing X+ (\alpha + \delta)
                    = 4 Box wing X+ (e)
                    = 5 Box wing Z+ (\alpha + \delta)
                    = 6 Box wing Z+ (e)
                    = 7 Box wing Z- (\alpha + \delta)
                    = 8 Box wing Z-(e)
                    = 9 Y-bias
                 Satellite ID
18-24
       Ι7
25-44
                 A priori value
45-59
                 Sigma
NOTE:
      The presence of one GPSBXW option for one GPS satellite will
      invoke the adjustment of ALL 9 parameters with a sigma = 1.D-30
      A-priori values equal to zero will be overwritten in GEODYN by
      default values (per block) according to the reference:
 "Adjustable box-wing model for solar radiation pressure impacting
 GPS satellites", by C.J. Rodriguez-Solano, U. Hugentobler, P. Steigenberger,
 in "Advances in Space Research, 0273-1177, 2012 COSPAR.
```

### GEODYN A PRIORI VALUES

SPSF0	for	block:	Т	1.2640000000
			_	
dthetasp	for	block:	I	0.000000000
aplusdxp	for	block:	I	0.5660000000
rhoxp	for	block:	I	0.434000000
aplusdzp	for	block:	I	0.355000000
rhozp	for	block:	I	0.6450000000
aplusdzm	for	block:	I	0.6280000000
rhozm	for	block:	I	0.372000000
уb	for	block:	I	0.000000000
SPSF0	for	block:	ΙΙ	1.2350000000
dthetasp	for	block:	ΙΙ	0.000000000
aplusdxp	for	block:	ΙΙ	0.900000000
rhoxp	for	block:	ΙΙ	0.100000000
aplusdzp	for	block:	ΙΙ	0.8880000000

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	_			
rhozp	for	block:	II	0.1120000000
aplusdzm	for	block:	II	0.917000000
rhozm	for	block:	II	0.0830000000
уb	for	block:	II	0.000000000
SPSF0	for	block:	IIA	1.2350000000
${\tt dthetasp}$	for	block:	IIA	0.000000000
aplusdxp	for	block:	IIA	0.900000000
rhoxp	for	block:	IIA	0.1000000000
aplusdzp	for	block:	IIA	0.8880000000
rhozp	for	block:	IIA	0.1120000000
aplusdzm	for	block:	IIA	0.9170000000
rhozm	for	block:	IIA	0.0830000000
уb	for	block:	IIA	0.000000000
SPSF0	for	block:	IIR	1.2783333333
dthetasp	for	block:	IIR	0.000000000
aplusdxp	for	block:	IIR	1.000000000
rhoxp	for	block:	IIR	0.000000000
aplusdzp	for	block:	IIR	1.0000000000
rhozp	for	block:	IIR	0.000000000
aplusdzm	for	block:	IIR	1.0000000000
rhozm	for	block:	IIR	0.0000000000
yb	for	block:	IIR	0.0000000000
SPSF0	for	block:	IIR-M	1.2783333333
dthetasp	for	block:	IIR-M	0.0000000000
aplusdxp	for	block:	IIR-M	1.0000000000
rhoxp	for	block:	IIR-M	0.0000000000
aplusdzp	for	block:	IIR-M	1.0000000000
rhozp	for	block:	IIR-M	0.0000000000
aplusdzm	for	block:	IIR-M	1.0000000000
rhozm	for	block:	IIR-M	0.0000000000
	for	block:	IIR-M	0.0000000000
yb SPSF0	for			1.2193333333
		block:	IIF	
dthetasp	for	block:	IIF	0.0000000000
aplusdxp	for	block:	IIF	0.8880000000
rhoxp	for	block:	IIF	0.1120000000
aplusdzp	for	block:	IIF	0.8880000000
rhozp	for	block:	IIF	0.1120000000
aplusdzm	for	block:	IIF	1.000000000
rhozm	for	block:	IIF	0.000000000
уb	for	block:	IIF	0.000000000

## 2.5.25 HRATEF

+2 HRATEF11100 0000000000			2000039.					
COLUMNS		DESCRIPTION	+0+( )N	)+			+0 & UNITS	
1-6	A6	integration of the contract of	orces to use for lon. For all the innot included for included for HR:	ndices be HR integ	ration			
7	I1	Index for	Drag					
8	I1	Index for	Solar Radiation					
9	I1	Index for	General Accelerat	tion				
10	I1	Index for	Topex Thermal Rac	diation.				
11	I1	Index for	Topex Louver Fore	ces				
12	I1	Index for	Albedo					
13	I1	Lageos The	ermal Drag					
14	I1	Index for	Geopotential					
15	I1	Index for	Earth and Ocean	Tides				
16	I1	Index for	Gravity Anomalies	5				
17	I1	Index for	Central Body Grav	vitation				
18	I1	Index for	Third Bidy Indire	ect Oblat	ion			
19	I1	Index for	Mars Moons Indrre	ect Accel	eration			
20	I1	Index for	General Relativit	ty				
21	I1	Index for	Yukawa Gravity					
22	I1	Index for	Planetary Moon G	ravity				
23	I1	Index for	Yarkovsky/Schach	Accelera	tion			
25-44	D20.8	Satellite	ID					

NOTE 1: This card is not needed to implement DSTATE modeling/estimation

IF CARD OMITTED: No forces will be used for high rate integration

## 2.5.26 HUBER

+1+2	+3+	-4	-5	-6+	7+	8
HUBER	1.5					
00	+0+	-0	-0	-0	0+	0

COLUMNS FORMAT DESCRIPTION

DEFAULT VALUE & UNITS

- 1-6 A6 HUBER Requests HUBER weighting to the observations.
- 15-17 I3 Measurement Type that HUBER weighting will apply.

  If cols 15-17 are blank, HUBER weighting will apply to all Measurement types.
- 25-44 D20.8 C = a factor that multiplies the observation sigma to create a new weight. [NOTE 1]
  - NOTE [1]: With Huber weighing a user can specify a number, C (1.5 is a typical specification for C). If the ratio of DABS(resid\_i)/sig\_i is larger than C, then the original weight that would have been applied to that observation is multiplied by C\*sig\_i/DABS(resid\_i) to obtain the new weight that is actually applied.

    The result of this is to cause what the solution is minimizing to change. The solution now minimizes the combination of the sum of (resid\_i/sig\_i)\*\*2 for all small residuals plus the sum of DABS(resid\_i)/(C\*sig\_i) for all large residuals. This accommodates the larger residuals without bending the solution so much towards them. The norm is L2 for small residuals and L1 for large residuals.

IF CARD OMITTED: Regular GEODYN observation weighting will be applied.

# 2.5.27 MBIAS SUBGROUP

### 2.5.27.1 MBIAS

MBIAS	7063 51	2+3+4 7603901	850731000000.	850831000000. 10.
COLUMNS		DESCRIPTION	-+0+	DEFAULT VALUE & UNITS
1-6	A 6	MBIAS - Requests applica adjustment of measurement timing bias.		
7-14	18	Station Number.		0
15-17	13	Bias Type. See list of bi this option description.	as types after	0
18-24	17	Satellite ID.		0
OR FOR 18-20 21-23 24	VLBI SPLIN I3 I3 I1	E TYPE PARAMETERS: Total number of parameter Parameter number in the s Degree of spline	_	ring
25-44	D20.8	Value of the bias (a prio if adjusted). Units are: Length - M, Time - S, Ang		0.
45-59	D15.3	Date and time at which th begin in form: YYMMDDHHMM		0.
60-72	D13.1	Date and time at which the end in form: YYMMDDHHMMSS		0.
73-80	D8.2	Standard deviation of the bias in the same units as If this value is greater will be adjusted.	bias value.	0.

UNITS: KM =Kilometers; M =Meters ; S =Seconds ; Kg =Kilograms

DEG=Degrees ; RAD=Radians ; AS =Arc seconds; MAS=Milli-arc seconds

NOTE [1]: Please also see EBIAS and PBIAS options.

NOTE [2]: If the measurement type is larger than 99 (3 digit) use MBIASM card to specify it.

### GEODYN-II BIAS TYPES

The GEODYN-II bias types include all of the GEODYN-II measurement types as well as other types which are independent from measurement type or which are an extension of the simple measurement bias.

At present, the only bias types defined include simple constant measurement biases and simple time biases independent of measurement type. Other future bias types will include linearly varying measurement biases, observation scale biases, tropospheric and ionospheric refraction scale biases, station clock polynomials and satellite clock polynomials.

Bias type numbers are assigned as follows:

BIAS DESCRIPTION	BIAS TYPE NU		SATELLIT DEPENDEN	IT D	STATION EPENDENT	
simple measurement biases	measurement		yes	or	yes	
measurement scale bias	meas. type +	100	yes	or	yes	
station clock bias		300	no		yes	sec
station clock linear drift		301	no		yes	sec/hour
station clock quadratic drif	t	302	no		yes	sec/hour 3
station clock cubic drift		303	no		yes	sec/hour
station clock bias (spline)		304	no		yes	sec/hour
altimeter attitude timing bias		399	yes		yes	sec
satellite clock bias		400	yes		no	sec
satellite clock linear drif	t	401	yes		no	sec/hour
satellite clock quadratic d	rift	402	yes		no	sec/hour 3
satellite clock cubic drift		403	yes		no	sec/hour
timing bias (only if MBIAS399 in use; see NOTE 5)		499	yes		yes	sec
tropospheric refraction scale bias		500	no		yes	[NOTE 1]
tropospheric bias zenith (sp	line)	501	no		yes	
tropospheric bias gradient e	ast (spline)	502	no		yes	
tropospheric bias gradient n	orth(spline)	503	no		yes	
ionospheric refraction scale bias		600	no		yes	
a: coefficient to TDRSS 1-way range-rate user frequency correction $\Delta f/I$		[NOTE	3] yes		no	sec

 $F_{\text{ref}}$  => Reference frequency

 $\Delta f$  => Difference between reference frequency and user transmit frequency

frequency correction can be specified to a cubic:

$$\frac{\Delta f}{F_{\text{ref}}} = a + b(t - t_0) + c(t - t_0)^2 + d(t - t_0)^3$$

t => Observation time  $t_0$  => Epoch time defined

by the user [NOTE 2]

	coefficient term)	(to	the	linear			701	yes		no	
	coefficient term)	(to	the	quadrati	С		702	yes		no	1/sec
	coefficient term)	(to	the	cubic			703	yes		no	1/sec**2
VLB	I baseline	bias					800	no	or	yes	
VLB	I baseline	bias			NOTE	[4]	801	no	or	yes	
sim	ple time bi	ases					900	yes	or	yes	

- NOTE [1]: In order to apply a troposphere refraction scale bias the parameter must be adjusted.
- NOTE [2]: Clock polynomials (400) and TDRSS one-way correction (700) cannot be handled by GEODYN simultaneously.
- NOTE [3]: Bias type is measurement type dependent. To specify the measurement type for MBIAS 700 the user should include an MBIASM card immediatlly following the MBIAS 700 card.
- NOTE [4]: Bias type 800 represents a VLBI baseline bias. To define this we need to specify a second station ID. The option MBIAS 801 following immediately MBIAS 800 serves only for defining this station ID.

  NO OTHER information besides the station ID and the bias type 801 is needed on the second card.
- NOTE [5]: A MBIAS900 card should be used for altimeter an observation timing bias. A MBIAS900 card should be used for altimeter an observation timing bias.

  An exception to this is when a MBIAS399 card is being used for an altimeter attitude timing bias. I this case a MBIAS399 card and not a MBIAS900 card should be used.

0.0

MBIAS 7063800 MBIAS 7064801 850731000000. 850831000000.

## 2.5.27.2 MBIASn

MBIAS	7063 51	7603901 0. 850731000000. 0+0+0+	850831000000. 10.
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-5	A5	MBIAS - Requests application and/or adjustment of measurement or station timing bias.	
6	A1	n - Used for estimation of multi-station and/or multi-satellite identification. If "n" is zero or one, then this card takes on the form of the "MBIAS" card. "n" may also take on the values two or three. When this occurs, the station and satellite information on these cards ("MBIAS2" or "MBIAS3") augments that already provided on the "MBIAS" form card.	0
7-14	18	Station number for the "n"th station in the measurement configuration.	0
15-17	13	Bias Type. See list of measurement types after ENDARC description (not used if "n" equals 2 or 3).	0
18-24	17	Satellite ID for the "n"th satellite in the measurement configuration.	0
OR FOR	VLBI SPLIN	E TYPE PARAMETERS:	
18-20 21-23 24	I3 I3 I1	Total number of parameters in the string Parameter number in the sequence of a st Degree of spline	
25-44	D20.8	Value of the bias (a priori estimate if adjusted). Units are: Length - M, Time - S, Angles - RAD	0.
45-59	D15.3	Date and time at which the bias is to begin in form: YYMMDDHHMMSS.	0.
60-72	D13.1	Date and time at which the bias is to end in form: YYMMDDHHMMSS.	0.
73-80	D8.2	Standard deviation of the a priori bias in the same units as bias value. If this value is greater than zero bias will be adjusted.	0.
UNITS:		eters; M = Meters ; S = Seconds ; K es ; RAD=Radians ; AS = Arc seconds; M	-

NOTE: If the time fields on cards with "n" equal to 2 or 3 are non-zero GEODYN will internally generate a complete new set of MBIAS cards having the same configuration with different times. The bias type field will be ignored on MBIAS cards with "n" equal to 2 or 3.

Please also see EBIAS and PBIAS options.

#### EXAMPLE FOR SAT-TO-SAT MEASUREMENTS:

	1+	3+4	+5+	-68
MBIAS1	0577502701 0.		780101000000.	780101000000.
MBIAS2	18570577403901		780101000000.	780101000000.
	0	0		-0+0

BIAS DESCRIPTION	BIAS TYPE NUMBER	DEPENDE	ENT I	DEPENDENT	UNITS
simple measurement biases	measurement type	yes	or	yes	
measurement scale bias	meas. type + 100	yes	or	yes	
station clock bias	300	no		yes	sec
station clock linear drift	301	no		yes	sec/hour
station clock quadratic drif	t 302	no		yes	2 sec/hour
station clock cubic drift	303	no		yes	3 sec/hour
station clock bias (spline)	304	no		yes	sec/hour
altimeter attitude timing bias	399	yes		yes	sec
satellite clock bias	400	yes		no	sec
satellite clock linear drif	t 401	yes		no	sec/hour
satellite clock quadratic d	rift 402	yes		no	sec/hour
satellite clock cubic drift	403	yes		no	3 sec/hour
timing bias (only if MBIAS399 in use; see NOTE 5)	499	yes		yes	sec
tropospheric refraction scale bias	500	no		yes	[NOTE 1]
tropospheric bias zenith (sp	line) 501	no		yes	
tropospheric bias gradient e	ast (spline) 502	no		yes	
tropospheric bias gradient n	orth(spline) 503	no		yes	
ionospheric refraction	600	no		yes	

scale bias

a: coefficient to TDRSS 1-way range-rate user	700 [NOTE 3]	yes	no	sec
<pre>b: coefficient (to the linear term)</pre>	701	yes	no	
<pre>c: coefficient (to the quadratic</pre>	702	yes	no	1/sec
<pre>d: coefficient (to the cubic</pre>	703	yes	no	1/sec**2
VLBI baseline bias	800	no or	yes	
VLBI baseline bias NOTE [4	4] 801	no or	yes	
simple time biases	900	yes or	yes	

- [NOTE 1] In order to apply a tropospheric refraction scale bias the parameter must be adjusted.
- iNOTE [2]: Clock polynomials (400) and TDRSS one-way correction (700) cannot be handled by GEODYN simultaneously.
- NOTE [3]: Bias type is measurement type dependent. To specify the measurement type for MBIAS 700 the user should include an MBIASM card immediately following the MBIAS 700 card.
- NOTE [4]: Bias type 800 represents a VLBI baseline bias. To define this we need to specify a second station ID. The option MBIAS 801 following immediately MBIAS 800 serves only for defining this station ID.

  NO OTHER information besides the station ID and the bias type 801 is needed on the second card.
- NOTE [5]: A MBIAS900 card should be used for altimeter an observation timing bias. A MBIAS900 card should be used for altimeter an observation timing bias.

  An exception to this is when a MBIAS399 card is being used for an altimeter attitude timing bias. I this case a MBIAS399 card and not a MBIAS900 card should be used.

MBIAS 7063800 0.0 850731000000. 850831000000. MBIAS 7064801

#### 2.5.27.3 MBIASM

MBIASM1	00 100	2+5+		10.
COLUMNS	FORMAT	DESCRIPTION	DEFAULT	VALUE & UNITS
1-6	A6	MBIASM - This card is used when the measurement type is larger than 99 (3 digit). The 3 digits in columns 7-10 will overwrite the measurement type in columns 16-17 of the MBIAS card. This option applies to simple measurement biases and measurement scale biases.		
7-10	13	Measurement Type.	0	
15-17	13	Measurement Type for TDRSS one way range correction application. [NOTE 1]	0	

NOTE [1]: If an MBIASM card follows an MBIAS 700 card, columns 15-17 are used to specify the measurement type for which a TDRSS one-way correction will apply. This correction acts like a scale to long link and a constant bias.

IF CARD OMITTED: Measurement type will be read from cols 16-17 of the MBIAS card.

### 2.5.28 MLTARC

MLTARC	_	2+3+4+5+6+7+80+0+0+0+0
COLUMNS	FORMAT	DESCRIPTION DEFAULT VALUE & UNITS
1-6	A6	MLTARC - Requests Multi-arc data file options for arcs 2 through the number of arcs.
7	I1	<pre>IF =0 THEN the current unit for the input 2S observation file will not be advanced.</pre>
		<pre>IF =1 THEN the current unit for the input 2S observation file will be advanced. [NOTE 1]</pre>
8	I1	<pre>IF =0 THEN the current unit for the input 2S observation file will not be rewound.</pre>
		IF =1 THEN the current unit for the input 2S observation file will be rewound.

NOTE [1]: Unit 40 will be advanced from the DD name of FT40F001 to the DD name (or file name) of ARC##, where ## is the number of the arc for which this option is being requested.

Example: If this option is requested for the second arc, then unit 40 will be attached to the DD name  ${\tt ARC02}\,.$ 

IF CARD OMITTED: The unit 40 observation file will be read sequentially for all arcs.

## 2.5.29 NORMPV

NORMPV1 NORMPV5	452361		1.0D+02 1.0D-12	1.0D-14 1.0D-12	1.0D-12	1.0D-12
COLUMNS	FORMAT	DESCRIPTION		DEFA	ULT VALUE	& UNITS
1-6	A6	NORMPV - Normal point control and a priors	_			
7	I1	Specifies elements of a priori diagonal valuatrix being input of	ariance/covari		1	
		1 - a,e,i & asc. noo fields 25-44, 49 respectively. Po order specified	5-59, 60-72 an arameter elimi	d 73-80 nation		
		5 - arg. perigee & rinput in fields respectively. Co	25-44 and 45-	59		
9-14	6I1	Order in which Keple to be eliminated who orbit adjustments to normal point binning	en using singl o edit data be	e pass	452361	
		Parameter Name Nu	ımber Varianc	e Units		
		Semi-major Axis Eccentricity Inclination R. A. of Asc. Node Arg. of Perigee Mean Anomaly	1 M** 2 ()* 3 DEG* 4 DEG* 5 DEG*	*2 *2 *2 *2		
		Order of elimination parameter numbers specified by column	pecified in co f elimination	lumns is		
COLUMNS	FORMAT	DESCRIPTION		DEFA	ULT VALUE	& UNITS
18-24	17	Satellite ID.			0	
25-44	D20.8	Element #1 of var/coby column 7. If .LE be adjusted during n	.0 , parameter	will not	0.	
45-59	D15.3	Element #2 of var/coby column 7. If .LE be adjusted during a	.0 , parameter	will not	0.	

60-72	D13.1	Element #3 of var/cov matrix as determined	0.
		by column 7. If .LE.O , parameter will not	
		be adjusted during normal point editing.	

73-80 D8.2 Element #4 of var/cov matrix as determined 0. by column 7. If .LE.O , parameter will not be adjusted during normal point editing.

Example - Diagonal elements of Keplerian variance/covariance matrix.

1-6	7	9 - 14	18-24	25-44	45-59	60-72	73-80
NORMPV	1	452361	7654321	V 1 1	V22	V33	V44
NORMPV	5		7654321	V55	V66		

NOTES: The parameters specified on this card have no effect on the actual orbit adjustment process. They are used only in single pass orbit parameter adjustments used to edit data prior to the formation of normal points on the last inner iteration of the last global iteration.

The Global Set option card NORMPT must be exercised if normal points are to be generated.

UNITS: KM =Kilometers; M =Meters ; S =Seconds ; Kg =Kilograms
DEG=Degrees ; RAD=Radians ; AS =Arc seconds; MAS=Milli-arc seconds
( )=Unitless

IF CARD OMITTED: Diagonal matrix with values as follows will be used: V11=1.0D2, V22=1.0D-14, V33=1.0D-12, V44=1.0D-12, V55=1.0D-12, V66=1.0D-12

## 2.5.30 VLBI SUBGROUP

# 2.5.30.1 OPVLBI

+1+2	+3+	4	6+	8
OPVLBI				
	+	0	0+	0
COLUMNS FORMAT	DESCRIPTION		DEFAULT VALUE U	

1-6 A6 OPVLBI: Denotes the beginning of the VLBI options

## 2.5.30.2 VLBSTR

VLBSTR	_	2+5+	-	·	_
COLUMNS	FORMAT	DESCRIPTION	DEFAULT	VALUE	UNITS
1-6	A6	VLBSTR			
8	I1	Index for option (1,8)			
11-74	A64	Body of the string			

# 2.5.30.3 ENDOPV

+1+2	+5	+8
ENDOPV		
+0+0	+0+0	+0
COLUMNS FORMAT	DESCRIPTION	DEFAULT VALUE UNITS

1-6 A6 ENDOPV: Denotes the end of the VLBI options.

## 2.5.31 ORBFIL

ORBFIL		6302610 840201080000. 840201090000. 10.
+	0+	0+0+0
COLUMNS	FORMAT	DESCRIPTION DEFAULT VALUE & UNITS
1-6	A6	Requests output of trajectory file(s) on specified unit(s) on the last iteration of the run.
7	I1	Coordinate system of output
		0 - True of date (default) 0 1 - True of reference date 2 - Mean of year 2000
8	I1	Switch indicating whether trajectory file 0
		is for a single satellite or a set of satellites.
		0 - Single satellite
		1 - Set of satellites. This option has meaning only when used in conjunction with sets of satellites (See EPOCH and SLAVE option cards for more details). If satellite ID in columns 18-24 is a master satellite, then the trajectory for all satellites in the set will be output.
9-11	13	Mandatory unit number for trajectory file. 0 All trajectory files within an arc must have unique unit numbers. The suggested unit number starts at 130. [3]
12	I1	Option to convert cgmass in sbf to tor frame 0
		<pre>0 - Center of mass (default) 1 - Center of figure</pre>
18-24	17	Satellite ID. This field must contain 0 a valid ID.
25-44	D20.3	Start date and time for trajectory No Default output (YYMMDDHHMMSS.SS).
45-59	D15.3	Stop date and time for trajectory No Default output (YYMMDDHHMMSS.SS).
60-72	D13.1	Time interval between successive 0 S trajectory outputs.

#### NOTES:

- [1]: In multiple satellite runs (like GPS runs), a separate ORBFIL option card is needed for each satellite. The same start and stop times on each card should be used.
- [2]: If master satellite and its slaves are output as a set (column 8=1), individual members of the set may still be output in separate files by including separate ORBFIL card(s) for the desired satellite(s).
- [3]: The user can place in column 9-11 values between 130 and 230 without over-writting any 2E output units.

IF CARD OMITTED: No trajectory output file will be generated.

### 2.5.32 ORBINF

	1+	25	68
ORBINF1	5	841101000000. 841108000000.	300.
+	0+	0+0+0+	0+0
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A 6	ORBINF - Requests ORBINF file (station/s angles) be generated for the indicated stime period.	
7-8	12	Number of stations (N) to be included in ORBINF file. The first N stations in the GEODETICS file will used first. If N is than the number of stations in the GEODE file then stations from the STAPOS subgribe used. The maximum number of stations allowed is 15.	larger TICS Coup will
18-24	17	Satellite ID.	
25-44	D20.8	Start date and time for ORBINF output. (YYMMDDHHMMSS.SSSSS) (NOTE 1)	
45-59	D15.3	End date and time for ORBINF output. Thi end date MUST be less than or equal to the stop date specified on the epoch car (YYMMDDHHMMSS.SS) (NOTE 1)	
60-72	D13.1	Interval for station/satellite look anglegeneration in integral seconds.	е

NOTE [1]: All times specified on any arc option cards must fall between the earlier of the epoch and start time and end prior to or at the stop time specified on the epoch card.

IF CARD OMITTED: No ORBINF file will be created.

## 2.5.33 ORBTVU

ORBTVU1		2+3+4 6302610650630180000. 0+0+0	650708000000.	3600.	
·					
COLUMNS	FORMAT	DESCRIPTION		DEFAULT	VALUE & UNITS
1-6	A6	Requests trajectory printelements on unit 8 and Ke on unit 10. Use an ORFII trajectory file generation orbit generation runs, or reduction runs. [NOTE 4]	eplerian elements card for on. Required in otional in data	3	
7	I1	Frequency of trajectory	output.	0	
		0 - Trajectory output times specified in and at interval specified of the columns of the colu	columns 25-59		
		1 - Trajectory output times specified in data points only.		t	
		2 - Trajectory output of times specified in data points and at specified in column	columns 25-59 at the interval	t	
8	I1	Coordinate system of outp	out	0	
		0 - True of date 1 - True of reference of 2 - Mean of year 2000	late	0	
9	I1	Trajectory type indicator 0 - Cartesian ephemeris 1 - Keplerian ephemeris 2 - Both Cartesian and ephemerides.	3	0	
COLUMNS	FORMAT	DESCRIPTION		DEFAULT	VALUE & UNITS
10	I1	Iterations on which traje printed.	ectory will be	0	
		<ul> <li>0 - First arc iter of f</li> <li>1 - Last arc iter of la</li> <li>2 - Both first first ar</li> <li>3 - All iterations</li> </ul>	ast global iter	c	
18-24	17	Satellite ID. If not specto all S/C in arc.	cified applies	0	

- 25-44 D20.3 Start date and time for trajectory viewing (YYMMDDHHMMSS.SS). [NOTE 3]

  45-59 D15.3 Stop date and time for trajectory viewing (YYMMDDHHMMSS.SS). [NOTE 3]

  60-72 D13.1 Nominal interval between successive trajectory viewings.
- NOTE 1: Cartesian elements are printed on unit 8 and Keplerian elements are printed on unit 10.
- NOTE 2: Orbit generation mode (ORBGEN mode) is implied by the lack of a DATA card or simulated data generation (SIMDAT) card.
- NOTE 3: Start and stop dates are required. There are no defaults.
- NOTE 4: For an orbit generation run, the users should make sure that all arc parameters have zero sigmas. Orbit generation with arc parameter estimation request are incompatible in GEODYN.

IF CARD OMITTED: No trajectory output will be generated

## 2.5.34 PANEL

PANEL	0 1 1	2+3+ 7521345	1.0E0	0.0EC	)	0.0E0
	0+	0+	0+	-0+	-0+	0+0
COLUMNS	FORMAT	DESCRIPTION			DEFAULT	VALUE & UNITS
1-6	A6	PANEL - Requests a adjustment of parama a flat panel for use force modeling.	eters assoc	iated with		
9	I1	Panel motion indica	tor		0	
		<pre>0 - Panel is fixed body-fixed co 1 - Indicates a m to the spacec:    (ie. the TOPE:    [NOTE 1]</pre>	ordinates ovable Pane raft body f	l with resp ixed frame	ect	
10	I1	Index for the type	of frequen	cy received	by the	panels
	=1	panel receives both spanel receives short panel receives long w	wave radiat	ion	re radiat	ion (default)
11-12	12	Indicates panel num	ber		1	
13-14	12	Parameter type				
		1 - Normal vector coordinates	in body-fi	xed		
		2 - Area			0.	m**2
		3 - Specular refl	•		0.	
		4 - Diffuse refle	ctivity		0.	
		5 - Emissivity 6 - Temperature A			0.	. К
		(cold equilib	rium temper	ature)		
		7 - Temperature C		404107	0.	. K
		(delta temper cold equilib				
		8 - Temperature d (exponential cooling)		for panel	0.	Sec.
		9 - Temperature d (exponential heating)		for panel	0.	Sec.
COLUMNS	FORMAT	DESCRIPTION			DEFAULT	VALUE & UNITS
		10 - Temperature/s (divisor for heating equa	cos(theta)		0.	

18-24	17	Satellite ID (required).	0
25-44	D20.8	Apriori parameter value as dictated in columns 13-14. (X component of normal vector if 1 is specified in columns 13-14)	
45-59	D15.3	Apriori parameter sigma as dictated in columns 13-14. (Y component of normal vector if 1 is specified in columns 13-14)	
60-72	D13.1	Z component of panel normal vector (if 1 is specified in columns 13-14)	

NOTE [1] : Movable panels must be grouped after the non-movable panels, and in ascending order.

## 2.5.35 PBIAS

+ PBIAS		2+3+4+5+6 7603901 1.	-+8
		0+0+0+0	-+00
COLUMNS	FORMAT	DESCRIPTION DEFAU	JLT VALUE & UNITS
1-6	A 6	PBIAS - Requests adjustment of pass by pass measurement or timing biases. [Note 2]	
7-14	18	Station Number.	0
15-17	13	Bias Type. See list of bias types after this option description.	0
18-24	17	Satellite ID.	0
25-44	D20.8	A priori bias value. Units: Length - M, Time - S, Angles - RAD	0.
45-59	D15.3	Start date and time for generating pass by pass biases (YYMMDDHHMMSS). See NOTE 1.	0.
60-72	D13.1	End date and time for generation of pass by pass biases. (YYMMDDHHMMSS). See NOTE 1.	0.
73-80	D8.2	Standard deviation of the a priori bias in the same units. This field must be a non-zero.	
UNITS:	KM =Kilom DEG=Degre	eters; M = Meters ; S = Seconds ; Kg = Kil es ; RAD = Radians ; AS = Arc seconds; MAS = Mil	lograms Lli-arc seconds
NOTE [1]		start and end dates are left blank pass by pass generated for the entire data span.	s biases
NOTE [2	MBIAS c that th setup i	. If the PBIAS option is used with the PUNCH of ards are generated from the PBIAS option. This e original PBIAS cards must be removed from the f the MBIAS cards generated by the PUNCH option the next run.	means

NOTE [3]: If the measurement type is larger than 99 (3 digit) use the PBIASM card to specify it.

BIAS DESCRIPTION	BIAS TYPE NUMBER	SATELLITE STATION DEPENDENT DEPENDENT
simple measurement biases	measurement type	yes or yes
measurement bias varying linearly with time	meas. type + 100	yes or yes
measurement scale bias	meas. type + 200	yes or yes
tropospheric refraction	500	no yes

### scale bias

ionospheric refraction scale bias	600	no		yes
simple time biases	900	yes	or	yes
clock frequency bias	901	yes	or	yes
clock frequency drift	902	ves	or	ves

## 2.5.36 PBIASM

PBIASM 10	2+3+4+5+ 0 0+0+0+	
COLUMNS FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6 A6	PBIASM - This card is used when the measurement type is larger than 99 (3 digit). The 3 digits in columns 15-17 will overwrite the measurement type in columns 16-17 of the PBIAS card. Requests adjustment of pass by pass measurement or timing biases.	
15-17 I3	Measurement Type.	0

UNITS: KM =Kilometers; M =Meters ; S =Seconds ; Kg =Kilograms DEG=Degrees ; RAD=Radians ; AS =Arc seconds; MAS=Milli-arc seconds

IF CARD OMITTED: Measurement type will be read from cols 16-17 of the PBIAS card.

## 2.5.37 PENUMB

PENUMB	-	2+3+4+5+6+7603901 0+0+0+0+0+0+	
COLUMNS	FORMAT	DESCRIPTION DEFAULT VALUE & UNI	TS
1-6	A6	PENUMB - Invoke the R.Robertson's Penumbra model for the satellite on columns 18-24	
18-24	17	Satellite ID. 0	

#### 2.5.38 RELTMC

	1+	2+5+	-6+8
RELTMC			
	0+	0+0+0	-0+0
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A 6	RELTMC - Requests application of the range correction due to relativistic periodic and secular space craft clock drift. [1],[2],[3],[4],[5]	
7	I1	Type of satellite	
		<ul><li>0 - Not GPS - applies secular and periodic correction</li><li>1 - GPS - applies periodic correction</li></ul>	only
25-44	D20.8	Satellite ID	
45-59	D15.3	Satellite ID	
60-72	D13.1	Satellite ID	
73-80	D8.2	Satellite ID	

#### NOTES:

- [1] If no satellite ID's are specified, card applies to all satellites in run.
- [2] The user may request application of this correction for up to four satellites on one card. (A maximum of 30 satellites for this option can be requested)
- [3] The relativistic time correction applies to the following measurement types: 39, 40, 41, 42, 55, 56.
- [4] A PREPRO card must be present in order for the correction to be applied The contribution of this correction is summed into the total relativistic measurement correction which can be printed on unit 16.
- [5] Currently, GEODYN uses the first point in the block to compute the semi-major axis, and also utilizes the time of the block start for the sychronization time. Because of this, you may get slightly different results if you change the blocking of your run.

IF CARD OMITTED: No relativistic time correction will be applied.

## 2.5.39 **RESIDU**

RESIDU	-	2+3+6 0+0+0			
COLUMNS	FORMAT	DESCRIPTION D	EFAULT	VALUE & UNITS	
1-6	A6	RESIDU - Requests residual file output on unit 19 on the last inner of the last global iteration.			
7	I1	Event times/location data switch.	0		
		0 - Residual file will not contain event times or trajectory information.			
		1 - Requests event times and spacecraft trajectory or true pole station locations at those event times be output along with the residual information. Trajectory output is in the true of data coordinate system.	ı		
8		Observation data switch.	0		
		0 - Residual file will not contain observation data.			
		1 - Requests observation data be output along with the residual information			
		2 - Requests observation data and observation correction data be outp along with the residual information			

IF CARD OMITTED: No Residual File will be generated.

#### 2.5.40 ROCK4

ROCK4		2		
COLUMNS	FORMAT	DESCRIPTION	DEFAULT	VALUE & UNITS
1-6	A 6	ROCK4 - Requests application of the ROCK4 or ROCK42 solar radiation model.	[2]	
9-10	I1	Newer thermal model = 0 Older thermal model = 1	0	
25-44	D20.8	Satellite ID		
45-59	D15.3	Satellite ID		
60-72	D13.1	Satellite ID		
73-80	D8.2	Satellite ID		
NOTES :	[1] If no	satellite ID's are specified, card appl	ies to all	satellites

- NOTES : [1] If no satellite ID's are specified, card applies to all satellites in run.
  - [2] The user may request application of this solar radiation model for up to four satellites on one card (A maximum of 30 satellites can be requested).
  - [3] The new version of the rock4 solar radiation model came from a paper given by H.F.Fliegel and T.E.Gallini at the AGU  $30\,\mathrm{May}\,91$  where for BLOCK I satellites X and Z are as follows

```
X = -4.55 SIN(B) + .08 SIN(2B + .9) - .06 COS(4B + .08) + .08

Z = -4.54 COS(B) + .20 SIN(2B - .3) - .03 SIN(4B)
```

For BLOCK II satellites, X and Z are

```
X=-8.96SIN(B)+.16SIN(3B)+.10SIN(5B)-.07SIN(7B)

Z=-8.43COS(B)
```

For BLOCK IIR satellites, X and Z are

```
X=-11SIN(B)-0.2SIN(3B)+0.2SIN(5B)

Z=-11.3COS(B)+0.1COS(3B)+0.2COS(5B)
```

- [4] This card is used in conjunction with SOLRD card.
- [5] The satellite ID for BLOCK I satellites must end in "1", for BLOCK II satellites in "2", and for BLOCK IIR satellites in "3".

IF CARD OMITTED: ROCK4 or ROCK42 models will not be applied.

#### 2.5.41 SLFSHD

```
---+---1---+---2---+---3----+---4----+---5----+---6----+---7----+---8
SLFSHD
---+---0---+---0---+---0----+---0
COLUMNS FORMAT
                                                       DEFAULT VALUE & UNITS
                     DESCRIPTION
1-6
       A6
                SLFSHD Requests application of a self
                shadowing effect to non conservative models
                (drag, solar radiation) due to the shadowing
                of the S/C macromodel plates on each other.
 7
       Ι1
                =0 Use ascii shadowing file
                =1 Use binary shadowing file
 8
                Shadowing application index.
         =0 Do not use shadowing on this satellite
         =1 Use shadowing for this satellite
9-10
                Force application index
         =11 Use shadowing for both Solar Radiation
         and Drag
         =10 Use shadowing for Solar Radiation only
         =01 Use shadowing for Drag only.
18-24
                Satellite ID.
                                                            0
     I7
```

UNITS: KM = Kilometers; M = Meters ; S = Seconds ; Kg = Kilograms

DEG = Degrees ; RAD = Radians ; AS = Arc seconds; MAS = Milli-arc seconds

IF CARD OMITTED: Self shadowing will not be applied on any satellite.

THE DATA STRUCTURE FOR SHADOWING FILE

One shadowing file should be used for each satellite. Each shadowing file should have one general header record and one or two blocker header record. The block header record must be followed by data records. The shadowing file could be either ascii or binary file.

```
ASCII SHADOWING FILE
General Header Record: four integers
1: number of blocks
2: number of panels for this satellite
   This number must match the panel cards in the IIS setup file.
3: model indicator
   0 for ratio
   1 for actual cross-section
4: satellite ID number
Block Header Record: one integer followed by three real numbers
1 (integer): block indicator
             O for solar radiation
             1 for drag
2 (real):
          start time in UTC YYMMDDHHMMSS.SS format
3 (real): stop time in UTC YYMMDDHHMMSS.SS format
           time step in seconds
4 (real):
Data Record:
Each data record includes the time tag followed by the values for each panel.
```

#### BINARY SHADOWING FILE

The binary has exactly the same structure as the ascii shadowing file. One should be careful that the size of integers in binary file should be consistent with the GEODYN versions (i32 or i64).

Note if one wants to use binary shadowing file and has multiple satellites, one must set the 7th columns on all the SLFSHD cards to 1, otherwsie GEODYN IIS will assume ascii shadowing files shall be used for all the satellites. In other words, one could only use either ascii or binary files simultaneously for all those satellites.

## 2.5.42 SIMULATION DATA SUBGROUP

#### 2.5.42.1 SIMDAT

COLUMNS FORMAT

+1	+2	-+	4+	5+6-	+8
SIMDAT O					
+0	+0	-++	0+	0+0-	+0

DEFAULT VALUE & UNITS

1-6	A 6	SIMDAT - Requests output of simulated
		data to UNIT 20 on final iteration.
		Actual data replaced with calculated value
		from final iteration. The simulated data is
		in the GEODYN II binary format expected
		by UNIT 40 of IIS.

8 I1 Simulated data generation indicator.

DESCRIPTION

- =0 Simulated data will be generated by replacing real data with computed data.
- =1 Data will be simulated for measurement types and times as specified on the SIMDAT control cards following this SIMDAT card. The SIMDAT control cards may be input as a separate file in UNIT17.
- 10 I1 Indicator specifying the input file of the SIMDAT control cards.
  - =1 SIMDAT control cards are input on UNIT17

Please see NOTE [1] below

NOTE [1]: This Option is not yet available for all measurement types. For the time being only range and range rate, altimetry and angle measurement types can be simulated. For paired angle measurement types read the note on the SIMTYP card.

IF CARD OMITTED: No data will be simulated.

#### 2.5.42.2 SIMDAT

---+---1---+---2---+---3----+---4----+---5----+---6----+---7----+---8
SIMDAT
----+---0---+---0---+---0---+---0

COLUMNS FORMAT

#### DESCRIPTION

DEFAULT VALUE & UNITS

1-6 A6

SIMDAT: A blank SIMDAT card in the beginning of UNIT17 introduces the simulated data generation mode specification subgroup which may include up to 20 configurations and ends with the ENDSIM card.

Each individual configuration includes the following cards:

SIMSAT

SIMSTA

NOTE: For further details see individual card descriptions in the following pages and Section 3.1.7 and see sample setup in APPENDIX Example 7.

IF CARD OMITTED: If simulated data generation (not replacement) is invoked and this card is missing the run will terminate.

### 2.5.42.3 SIMLIM

SIMLIM	2+5+5+	· ·	·	· ·
COLUMNS FORMAT	DESCRIPTION	DEFAULT	VALUE 8	& UNITS
1-6 A6	SIMLIM: Requests certain limitations for the simulated data generation mode			
7-14 I1	Index to orbit pattern choice:		0	
	=0 all orbits as defined by the following input cards.			
	=1 only ascending orbits	N	/ A	
	=2 only descending orbits	N	/ A	
31-45 D15.8	The interval between "observations" The same interval will be used as counting interval in all measurement types involving distruct Doppler data.		S	
46-60 D15.8	Visibility checking time interval	N	/ A	
61-70 D10.2	Elevation cutoff angle below which no data will be generated.	DI	EG	0

UNITS: KM =Kilometers; M =Meters ; S =Seconds ; Kg =Kilograms DEG=Degrees ; RAD=Radians ; AS =Arc seconds; MAS=Milli-arc seconds

 $\hbox{ IF CARD OMITTED: If simulated data generation (not replacement) is invoked and this card is missing the run will terminate. } \\$ 

#### 2.5.42.4 SIMSAT

SIMSAT	-	2+3+5+6+8 0+0+0+0
COLUMNS	FORMAT	DESCRIPTION DEFAULT VALUE & UNITS
1-6	A6	SIMSAT: Defines a set of satellites involved in the present configuration. SIMSAT and the following SIMSTA cards defines a tracking configuration in simulated data generation mode.
7-14	18	Satellite ID for the tracked satellite in this configuration.
15-22	18	Satellite ID for the relay satellite in this configuration.
23-30	18	Satellite ID for the third satellite in this configuration.

NOTE: For all the satellites involved there must be an equivalent SATPAR card on UNIT 5. For any given configuration (measurement) GEODYN is limited to a maximum of three satellites and three stations.

 $\hbox{ IF CARD OMITTED: If simulated data generation (not replacement) is invoked and this card is missing the run will terminate. } \\$ 

#### 2.5.42.5 SIMSTA

SIMSTA		2+5+5+			
COLUMNS	FORMAT	DESCRIPTION	DEFAULT	VALUE	& UNITS
1-6	A 6	SIMSTA: Defines a set of stations involvin the present configuration.	ed		
7-14	18	Receiving station number No station number is required for altimetry data generation.			
15-22	18	Transmitting station number.			
23-30	18	Third station number.			

NOTE: For all the stations listed on this option card there must be corresponding station coordinates in the STAPOS subgroup.

IF CARD OMITTED: If simulated data generation (not replacement) is invoked and this card is missing for measurement types where stations are involved, the run will terminate.

# 2.5.42.6 SIMTIM

SIMTIM		2+3+4+5+	
+	0+	0+0+0+0	-0+0
COLUMNS	S FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A6	SIMTIM: Start and end time for simulated data generation for the tracking configuration defined on the previous SIMCON card and for the types defined on the previous SIMTYP cards.	
31-45	D12.	Start date in form YYMMDDHHMMSS.	Earliest start date and time from EPOCH cards on UNIT5
46-60	D12.	Stop date in form YYMMDDHHMMSS.	Latest stop date and time from EPOCH card on UNIT5
61-70	I10	Elevation cutoff angle, overrides any values on SIMLIM cards	Value from SIMLIM card.
71-80	I10	Surface elevation for altimetry data	N / A

IF CARD OMITTED: Run will abnormally terminate.

## 2.5.42.7 SIMTYP

2.3.42.7 SIWITT	25	6
SIMTYP		
+0+	0+0+0+	0+
COLUMNS FORMAT	DESCRIPTION	DEFAULT VALUE & UNIT
1-6 A6	SIMTYP: Data for the measurement type described on this card will be generated for the tracking configuration defined on the preceding SIMSAT and SIMSTA cards [NOTE 1]	
12-14 I3	Measurement type. (See Section 5.0 for discussion on measurement types.	
31-45 D15.8	Measurement noise level, for 1st measurement. [NOTE 2]	0
46-60 D15.8	Measurement noise level, for 2nd measurement. (in case of paired measurements) [NOTE 2]	0
61-70 D10.2	Observation time interval	Value from S SIMLIM card.
specif pair 1	mulating paired measurement types the usery the first measurement in the pair i.e. 17-18. GEODYN will automatically simulated the control of	17 for the e both. This
or M/S observ	ne noise level use the same units as the Sec for metric type, DEG RAD or MAS for a vation the noise level will be multiplied on -1 and 1.	ngles). For each
	neters; M = Meters ; S = Seconds ; sees ; RAD=Radians ; AS = Arc seconds;	
IF CARD OMITTED:	If simulated data generation (not replace invoked and this card is missing the run	

terminate

## 2.5.42.8 ENDCON

ENDCON	0+0+0	+0+0	
COLUMNS FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS	

1-6 I6 ENDCON: End of configuration
Will be followed by an ENDSIM card
if this is the last configuration.
Will be followed by a SIMSAT card

if another configuration is requested.

 $\label{eq:card_omega} \text{If card OMITTED: If simulated data generation (not replacement ) is }$ 

invoked and this card is missing the run will

terminate.

# $\boldsymbol{2.5.42.9}\quad \textbf{ENDSIM}$

+2- ENDSIM	+3+5-	+8
	+0+0+0-	0
COLUMNS FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS

1-6 A6 ENDSIM: End of the simulated data input cards

# 2.5.43 SOLRAD

SOLRAD	0	7654321 1.5		
+	0+-	0+0+0+0	-+(	)+0
COLUMNS	FORMAT	DESCRIPTION DEFA	ULT VAI	LUE & UNITS
1-6	A 6	SOLRAD - Requests application and/or adjustment of solar radiation pressure.		
8	I1	Indicates order of radiation pressure coefficients.	0	
		0 - Cr		
		1 - Cr		
		2 - Cr		
9	I1	=1 The UCL solar radiation bus model will be used for JASON		
		=2 The UCL solar radiation model will be us for GFO	ed	
		=3 The UCL solar radiation model will be us for ENVISAT	ed	
10	I1	=1 The UCL solar radiation panel model will be used for JASON		
11	I1	=1 The UCL re-radiation panel model will be used for JASON		
14	I1	Sunlight Ratio input  =1 Sunlight Ratio input will be read from  UNIT 77 and will be used instead of the  ratio being calculated in SOLRAT  =0 SOLRAT will calculate Sunlight Ratio  See NOTE(2) for more details.	0	
18-24	17	Satellite ID.	0	
25-44	D20.8	Solar radiation pressure coefficient. (Units for Cr dot and Cr double dot are per second and per second squared)	0.	none or 1/S or 1/S**2
45-59	D15.3	This field must be left blank unless the time dependent solar radiation option is wanted. To select time dependent solar radiation the end time of the period for which this acceleration coefficient applies is input in the form YYMMDDHHMMSS.SS in this field.  See NOTE(1) for more details.	0.	
60-72	D13.1	Standard deviation of radiation pressure coefficient.	0.	none or 1/S or 1/S**2

UNITS: KM = Kilometers; M = Meters ; S = Seconds ; Kg = Kilograms

DEG = Degrees ; RAD = Radians ; AS = Arc seconds; MAS = Milli-arc seconds

NOTE[1]: Time periods may be specified only for the highest order coefficients used (i.e. if Cr and Cr dot are used, only Cr dot can have time intervals). When using time periods the number of SOLRAD cards required is equal to the number of time periods plus the order (order = the number in column 8 +1). The Cr, Cr dot, and/or Cr double dot without a time interval are used for times beyond the end of the the last interval specified on the SOLRAD cards. See example on next page.

EXAMPLE: SOLRAD cards for using time dependent general accelerations (highest order - Cr ; three time intervals; 4 cards required)

 ${\tt NOTE}\, {\tt [2]}\colon {\tt The}$  input file should have the following structure:

MAIN HEADER

Contains Number of Satellites

FORMAT(I2)

#### SUBHEADERS

One subheader for each satellite.

Contains Satellite ID and Number of observations (observations being when the SC is transitioning to/from the sunlight/shadow) FORMAT(I7,1X,I6)

TITLE HEADER for Satellite 1
Just a title reading SATELLITE 1 DATA

## DATA

Data includes times and sunlight ratios immediately before, during, and immediately after the spacecraft transitions to/from full sunlight.

During the transitions, data should be at regular time intervals. Time is in  $\mbox{UTC}\,.$ 

Data is in the following structure:

Time Ratio

FORMAT (F14.3,1X,F6.4)

This is followed by the TITLE HEADER and DATA of remaining satellites.

EXAMPLE: The following is a portion of input for two satellites, where there are 9245 observations for each satellite 2012001 and satellite 2012002. A line of stars represent a jump in the input.

2 2012001 9245 2012002 9245 SATELLITE 1 DATA 2245277866.791 1.0000 2245277871.791 0.7663 2245277876.791 0.3945 2245277881.791 0.1352 2245277886.791 0.0000 2245280541.886 0.0000 2245280546.886 0.3428 2245280551.889 0.6467 2245280556.889 1.0000 \*\*\*\*\*\* SATELLITE 2 DATA 2245277866.791 1.0000 2245277871.791 0.7663 2245277876.791 0.3945 2245277881.791 0.1352 2245277886.791 0.0000 2245280541.886 0.0000

WARNINGS: If duplicate time periods are specified the run will terminate in IIS with a warning message.

The first time period that exceeds the end time of the run (see EPOCH card) will be accepted. All other end times exceeding the end time of the run will be ignored.

## 2.5.44 STEP

+-	2	-3+	58
STEP	7654321 45.	45.	
+-	0	-0+	0+0

COLUMNS	<b>ГОРМАТ</b>	DESCRIPTION	DEFAULT VALUE &	IINTTO
COLUMNS	FURMAI	DESCRIPTION	DEFAULT VALUE &	UNIIS
1-6	A6	STEP - Changes integrator order and step size.		
7-8	12	Order of integration for orbit.	11	
9-10	12	Order of integration for variational	Same as orbit	
11-12	12	Number of output buffers to save integrinformation.	ration	
14	I1	>1 For this satellite we need to write integration information out for later use	0	
15-17	13	Millions of words devoted to this satelli in dynamic arrays for integration informa		
18-24	17	Satellite ID.	0	
25-44	D20.8	Step size for orbit integration.	60.	S
45-59	D15.3	Step size for variational equations.	Same as orbit	
60-72	D13.1	Small stepsize used for variable size integration step.	1.	S

- NOTE 1: If the DSTATE option is in the setup, variable intergration is required. Columns 60--72 need to be filled in with appropriate step size.
- NOTE 2: Multirate integration should be used for the DSTATE option (note 1) or when it is desired to numarically integrate some forces at a smaller step size than others. When dividing forces into different step sizes the HRATEF card should be used. There is no reason to use the HRATEF card if multi-rate integration is being used only for the purpose of modeling/estimating DSTATE parameters.

UNITS: KM =Kilometers; M =Meters ; S =Seconds ; Kg =Kilograms
DEG=Degrees ; RAD=Radians ; AS =Arc seconds; MAS=Milli-arc seconds

## 2.5.45 TATBIA

TATBIA	•	2+2	0.0	,	0.0	•	•	
COLUMNS	FORMAT	DESCRIPTION			D	EFAULT	VALUE	& UNITS
1-6	A6	TATBIA - Defines biase array pitch and spaced for satellites other t	craft yaw					

I7

25-44 D20.8 Solar array pitch bias [NOTE 1]

Satellite ID

0.0 deg

## NOTE [1]:

18-24

The solar array pitch bias applies to the time period specified on the TATBTM card which immediately follows this card. If no TATBTM card immediately follows this card the time period will be taken from the EPOCH card. Multiple TATBIA cards may be included in a run. The time periods on corresponding TATBTM cards should not overlap. If there is no solar array pitch bias specified for a certain time period the default value applied will be zero.

# 2.5.46 TATBTM

TATBTM	_	0.0 0.0
+	0+	0+0+0
COLUMNS	FORMAT	DESCRIPTION DEFAULT VALUE & UNITS
1-6	A6	TATBTM - Specifies the start and stop times for the solar array pitch bias on the TATBIA card immediately before this card.
25-44	D20.8	Start time for solar array pitch value none specified on TATBIA card immediately before this card. (YYMMDDHHMMSS.SS)
45-59	D15.3	Stop time for solar array pitch value none specified on TATBIA card immediately before this card. (YYMMDDHHMMSS.SS)

IF CARD OMITTED: The time period applied to the TATBIA card will be taken from the EPOCH card.

# 2.5.47 TELEM

TELEM	-	2+5+	
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A6	Requests output of TOPEX telem file on unit 97 on the last iteration of the run.	
18-24	17	Satellite ID of the LEO in a GPS run. This input will invoke a special TELEM output on unit 97 for the LEO on the last iteration. A GPSMOD option is also required in the same setup for this option to work. The absence of the satellite ID will cause TELEM to work normally with attitude and panel output information.	0

IF CARD OMITTED: No TOPEX telem file will be generated.

## 2.5.48 THRDRG

+ THRDRG	1+2 76039	+3+4+5+	6+7-	+8
		+0+0+0	0+0-	+0
COLUMNS		DESCRIPTION	DEFAULT VALU	
1-6	A6	THRDRG - Requests thermal drag application.		
7		Card number indicator (1 or 2). The second card is used to request adjustment of the satellite spin axis.	1	
8	I1	> 0 Indicates that partials are to be calculated but the thermal drag force will not be applied.	0	
18-24	17	Satellite ID (Required)	0	
25-44	D20.8	Scaling parameter D (see model below)	-11.836**-12	MS**-2
45-59	D15.3	Model variable parameter $\zeta$ (see model description below)	1.2	unitless
60-72	D13.1	Colatitude of the satellite's spin axis $\theta$ If there is a 2 in column 7 (if this is the second THRDRG card), then this field is for the standard deviation of the a priori spin axis colatitude. The default standard deviation is zero.	22 DEG	
73-80	D8.2	The right ascension of the satellite's spin axis $\lambda$ If there is a 2 in column 7 (if this is the second THRDRG card), then this field is for the standard deviation of the a priori spin axis right ascencion. The default standard deviation is zero	313 DEG	

Note [1]: The option to adjust the spin axis will not work in a run with more than one satellite or more than one arc.

Note [2]: The thermal drag acceleration is computed from the equation (for detailed description see section 8.12, Vol. 1):

$$\vec{a}_{\mathrm{TD}} = \left[ \frac{-D(\vec{r}_d \cdot \vec{S})}{\sqrt{1 + \zeta^2}} \right] \vec{S}$$

UNITS: KM =Kilometers; M =Meters ; S =Seconds ; M/S=Meters per second

DEG=Degrees ; RAD=Radians ; AS =Arc seconds; MAS=Milli-arc seconds

 $\label{eq:cardinal} \textbf{IF CARD OMITTED} \ : \ \textbf{Thermal drag acceleration will not be included}.$ 

# 2.5.49 TIEOUT

TIEOUT	1	+2+3+4+ 931224000000.00	-5+	-6+	8
	0	+0+	-0+	-0+	0+0
COLUMNS	FORMAT	DESCRIPTION		DEFAUL	T VALUE & UNITS
1-6	A6	TIEOUT - Used to request TIEOUT information be written at end of parameter labels record of emat.			
25-44	D20.8	Output request epoch			YYMMDDHHMMSS.SS
45-59	D15.3	Reference coordinate system date of output defaulted to value in columns 26-44.		ols. 6-44	YYMMDDHHMMDD.SS

# 2.5.50 TOPATT

TOPATT	2	2+3+4+5+9606123 960701112233.00000 960702112233.0	)	90.0
COLUMNS	FORMAT	DESCRIPTION	DEFAULT	VALUE & UNITS
1-6	A 6	TOPATT - Requests override of nominal TOPEX yaw steering algorithm.		
9	I1	Yaw steering mode indicator	0	
		<ul> <li>1 - Sinusoidal yaw</li> <li>2 - Fixed yaw</li> <li>3 - Ramp Up</li> <li>4 - Ramp Down</li> <li>5 - Geocentric pointing</li> </ul>		
18-24	17	Satellite ID (required).	0	
25-44	D20.3	Start time of yaw steering event. YYMMDDHHMMSS.SS [See Note 1]		
45-59	D15.3	Stop time of yaw steering event. YYMMDDHHMMSS.SS [See Note 1]		
60-72	D13.1	Satellite fix yaw angle (mode 2)	0	(deg)

NOTE 1 : TOPATT cards must be in ascending time order.

IF CARD OMITTED: TOPEX will follow nominal yaw steering algorithm.

## 2.5.51 TOPBIA

TOPBIA	-	2+3+4-	0.0	0.0	· ·
	0	0		00	
COLUMNS	FORMAT	DESCRIPTION		DEFAULT VAL	UE & UNITS
1-6	A6	TOPBIA - Defines biases array pitch and spacecr for the TOPEX satellite	aft yaw		
25-44	D20.8	Solar array pitch bias	[1]		0.0 deg
45-59	D15.3	Yaw bias			0.0 deg

## NOTE [1]:

The solar array pitch bias applies to the time period specified on the TOPBTM card which immediately follows this card. If no TOPBTM card immediately follows this card the time period will be taken from the EPOCH card. multiple TOPBIA cards may be included in a run. The time periods on corresponding TOPBTM cards should not overlap. If there is no solar array pitch bias specified for a certain time period the default value applied will be zero.

# 2.5.52 TOPBTM

TOPBTM		2+5+ 0.0 0 0	. 0
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A 6	TOPBTM - Specifies the start and stop times for the solar array pitch bias on the TOPBIA card immediately before this card.	
25-44	D20.8	Start time for solar array pitch value specified on TOPBIA card immediately before this card. (YYMMDDHHMMSS.SS)	none
45-59	D15.3	Stop time for solar array pitch value specified on TOPBIA card immediately before this card. (YYMMDDHHMMSS.SS)	none

IF CARD OMITTED: The time period applied to the TOPBIA card will be taken from the EPOCH card.

# 2.5.53 TOPLOV

TOPLOV	_	2+3+6- 0+0+0+0-		
COLUMNS	FORMAT	DESCRIPTION DE	FAULT	VALUE & UNITS
1-6	A6	TOPLOV - Requests application of an acceleration due to thermal radiation emmitted from the louvers on the TOPEX spacecraft.		
18-24	17	Satellite ID (required).	0	
25-44	D20.8	Magnitude of the acceleration in the TOPEX body fixed frame. component of normal vector if 1 is specified in columns 13-14)	-	m/s**2
45-59	D15.3	X component of normal vector of the acceleration in the TOPEX body-fixed frame.		
60-72	D13.1	Y component of normal vector of the acceleration in the TOPEX body-fixed frame.		
73-80	D15.3	Z component of normal vector of the acceleration in the TOPEX body-fixed frame.		

If card ommitted: no louver accceleration will be applied.

# 2.5.54 TOPYAW

TOPYAW		2+3+4+5+6+ 9601123	80.00	
COLUMNS	FORMAT	DESCRIPTION DEFAULT	VALUE & U	UNITS
1-6	A 6	TOPYAW - Defines betaprime boundaries marking changes in yaw steering mode for the TOPEX satellite.		
10	I1	<ul><li>=1 Default for the following Beta prime values is zero</li><li>=0 Default for the following Beta prime values are indicated below</li></ul>		
18-24	17	Satellite ID		
25-44	D20.8	Betaprime value marking the boundary between fixed yaw at low betaprime and sinusoidal yaw	15.0	deg
45-59	D15.3	Betaprime value marking the boundary between fixed yaw and the flip manuever	0.1	deg
60-72	D13.1	Betaprime value marking boundary between sinusoidal and fixed yaw at high betaprime if 1 is specified in column 10	80.0	deg

# 2.5.55 TSTLOV

TSTLOV	-	2+3+4+5+		
+	0+	0+0+0+0	0+-	0+0
COLUMNS	FORMAT	DESCRIPTION	DEFAULT	VALUE & UNITS
1-6	A6	TSTLOV - Requests application of an acceleration due to thermal radiation emmitted from the louvers on spacecrafts other than TOPEX.		
18-24	17	Satellite ID (required).	0	
25-44	D20.8	Magnitude of the acceleration in the space craft body fixed frame. component of normal vector if 1 is specified in columns 13-14	al	m/s**2
45-59	D15.3	X component of normal vector of the acceleration in the spacecraft body-fixed	frame.	
60-72	D13.1	Y component of normal vector of the acceleration in the spacecraft body-fixed	frame.	
73-80	D15.3	Z component of normal vector of the acceleration in the spacecraft body-fixed	frame.	

IF CARD OMITTED: no louver accceleration will be applied.

# 2.5.56 TUMSOL

TUMSOL	-	2+3+8 0+0+0+0+0
COLUMNS	FORMAT	DESCRIPTION DEFAULT VALUE & UNITS
1-6	A 6	TUMSOL - Introduces the application of the TUM (Tehcnical University of MUNICH) solar radiation model. (only applicable to GPS satellites)
7	I1	<pre>= 0 The model will apply to all satellites = 1 The model will apply to the satellite on     on columns 18-24</pre>
8	I1	= 1 Return complete solar radiation acceleration 1 = 2 Return only the solar orientation bias
9	I1	= 1 Return contribution of Y-bias acceleration 2 = 2 Exclude Y-bias acceleration
18-24	I7	Satellite ID

# 2.5.57 VARCOV

+	1+-	2+3	+4+	5+	68
VARCOV VARCOV	0 10 0 11	7654321 7654321	1.0D+14 1.0D+14	1.0D+14 1.0D+14	1.0D+14 1.0D+14
+	0+-	0+0	++	0+	00
COLUMNS	FORMAT	DESCRIPTI	ON		DEFAULT VALUE & UNITS
1-6	A 6	VARCOV - A-pri matrix for the			
8	I1	Indicator of cadjustment.	oordinate sys	tem of	0
		0 or 1 - Car 2 - Kepleria 3 - Low e an	n.	ular Keplerian	).
9	I1	Indicates whet triangular par	_		0
		•	elements onl	•	
11-12	12	Card identifie			0
18-24	17	Satellite ID. to all satelli		o means apply	0
25-44	D20.8	Element #1 of by column 11.		x as determine	d 0.
45-59	D15.3	Element #2 of by column 11.		x as determine	d 0.
60-72	D13.1	Element #3 of by column 11.		x as determine	d 0.
73-80	D8.2	Element #4 of by column 11.		x as determine	d 0.
Example	1 - Diag	gonal elements of	variance/cov	ariance matrix	
1-6 VARCOV VARCOV	8 9 1	11-12 18-24 10 7654321 11 7654321	25-44 4 V11 V55	5-59 60-72 V22 V33 V66	73-80 V44
Example	2 - Uppe	er triangular var	iance/covaria	nce matrix	
1-6	8 9 1	11-12 18-24	25-44 4	5-59 60-72	73-80
VARCOV	1	10 7654321	V 1 1	V12 V13	V14
VARCOV	1	11 7654321	V15	V16	
VARCOV VARCOV	1 1	20 7654321 21 7654321	V22 V26	V23 V24	V25
V 1110 O V	_	1004021	, 20		

VARCOV	1	30	7654321	V33	V34	V35	V36
VARCOV	1	40	7654321	V44	V45	V46	
VARCOV	1	50	7654321	V55	V56		
VARCOV	1	60	7654321	V66			

Where Vij represent "i"th row and the "j"th column and where the units are as defined by the tables below.

# Cartesian Variance/Covariance Units


\ j	1	2	3	4	5	6
i \						
	1					
1	M**2	M**2	M**2	M*M/S	M*M/S	M*M/S
2	1	M**2	M**2	M*M/S	M*M/S	M*M/S
3	1		M**2	M*M/S	M*M/S	M*M/S
4	1			(M/S)**2	(M/S)**2	(M/S)**2
5	1				(M/S)**2	(M/S)**2
6	1					(M/S)**2

## Keplerian Variance/Covariance Units

-----

\ j i \	1	2	3	4	5	6
	1					
1	M**2	M	M*RAD	M*RAD	M*RAD	M*RAD
2	1	( )**2	( )*RAD	( )*RAD	( )*RAD	( )*RAD
3	1		RAD **2	RAD **2	RAD **2	RAD **2
4	1			RAD **2	RAD **2	RAD **2
5	1				RAD **2	RAD **2
6	1					RAD **2

UNITS: KM =Kilometers; M =Meters ; S =Seconds ; Kg =Kilograms DEG=Degrees ; RAD=Radians ; AS =Arc seconds; MAS=Milli-arc seconds ( )=Unitless

IF CARD OMITTED: Diagonal matrix with values of 1.0D+14 will be used

# 2.5.58 YAWBIA

YAWBIA		+5+		
COLUMNS	FORMAT	DESCRIPTION	DEFAULT	VALUE & UNITS
1-6	A 6	YAWBIAS: Requests for application of yaw bias		
18-24	17	Satellite ID		
25-44	D20.8	Yaw bias	0.0	deg.
45-59	D15.3	Start time	EPOCH st	art time
60-72	D13.1	End time	EPOCH en	d time

IF CARD OMITTED: Yaw bias will not be applied.

# 2.5.59 DATA SELECTION / DELETION SUBGROUP

#### 2.5.59.1 DATA

---+---1----+----3----+----3----+----5----+----6----+----7----+----8
DATA
----+---0---+---0---+---0---+---0---+---0

COLUMNS FORMAT

DESCRIPTION

DEFAULT VALUE & UNITS

1-4 A4 DATA - Indicates data reduction mode utilizing an input data file on unit 41, and introduces the data selection subgroup.

Note: GEODYN II asumes orbit generation type of run unless a DATA or SIMDAT card is present in the arc.

DATA SELECTION SUBGROUP

The DATA card must be present in any arc which requires a data file.

The DATA SELECTION SUBGROUP begins with the DATA card and ends with the ENDARC card. The ENDARC card must be the last card in the arc. The DATA SELECTION SUBGROUP consists of the following keyword cards:

ENDARC METDAT OBSCOR PREPRO SELECT/DELETE SIGMA and must contain at least one SELECT card.

IF CARD OMITTED: Program will operate in orbit generation mode

# 2.5.59.2 SELECT

SELECT		7603901 830901 0+	0	830903	-0	0
COLUMNS		DESCRIPTION		DEFAULT V		
1-6	A6	SELECT - Requests the selects observation data for the data process.  Any OBSCOR, SIGMA, METDAT, as cards that precede the first apply to all data except when overridden.	a reduction nd PREPRO SELECT card			
7-14	18	Station number.		0		
15	I1	Station satellite configurate sequence number.	ion	0		
16-17	12	Measurement Type. 2 digits of See list of measurement types		0		
18-24	17	Satellite ID.		0		
25-30	16	Measurement modulo number.		1		
31-40	D10.8	Measurement Type. 3 digits of Will be converted to an integ See list of measurement types Section 5.0.	ger.	0		
41-46 47-50 51-60	I6 I4 D10.8	Start date in form YYMMDD. : Start time in form HHMM. : Seconds of start time. :		Earliest date and EPOCH ca	time	from
61-66 67-70 71-80	I6 I4 D10.8	Stop date in form YYMMDD.: Stop time in form HHMM.: Seconds of stop time.:		Latest date and EPOCH ca		from
IF CARD	OMITTED:	No data will be selected.				

## 2.5.59.3 **DELETE**

+1-	+3	+5	+6+7+	8
DELETE	517603901	830901	830903	
0-	+0+0	+0+0	+	0

COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A6	DELETE - Requests the deletion of observation data for the data reduction process.	
7-14	18	Station number. [1]	0
15	I1	Station satellite configuration sequence number. [2]	0
16-17	12	Measurement Type. 2 digits or less. See list of measurement types after ENDARC description.	0
18-24	17	Satellite ID. [1]	0
25-30	16	Measurement modulo number.	1
31-40	D10.8	Measurement Type. 3 digits or less. Will be converted to an integer. See list of measurement types after ENDARC description.	0
41-46	16	Start date in form YYMMDD. :	Earliest Start
47-50 51-60	I4 D10.8	Start time in form HHMM. : Seconds of start time. :	<pre>date and time from EPOCH card(s).</pre>
61-66 67-70	I6 I4	Stop date in form YYMMDD.: Stop time in form HHMM.:	Latest Stop date and time from
71-80	D10.8	Seconds of stop time. :	EPOCH card(s).

## NOTES:

- [1] Unless a specific configuration sequence number is specified on he col. 15, all measurements that include this station or this satellite will be deleted (or selected).
- [2] The station/satellite configuration number enables the user to delete (or select) measurement types including more than one station or more than one satellite without affecting other configurations which may include the same station or the same satellite.

## Here are some examples:

In the case of measurement type 51 (two-way range) where only one station and one satelliteare involved, the user does not need to specify anything on col. 15 of the DELETE or the SELECT card.

---+---1----+----3----+----3----+----8

DELETE		5176	50390	01		8309	901			830903
+	0	+	-0	++		-0	-+	0+	(	0+0
	will	delete	all	measurements	51	from	all	stations	to	satellite

---+---1---+---2---+---3---+---4----+---5---+---6---+---7---+----8
DELETE 314 517603901 830901 830903
----+---0----+---0----+---0----+---0

# 7603901 between the times 830901 and 830903.

# 7603901 between the times 830901 and 830903.

will delete all measurements 51 from station 314 to satellite

In the case of measurement type 61 (singly differenced one way ranges where we have one satellite and two stations we will need two DELETE (or SELECT) cards for precise specification of the configuration ie.

---+---1---+---3----+---3----+----5----+---6----+---7----+----8
DELETE 3211618003201
DELETE 4212618003201
----+---0----+---0----+---0----+---0

This means, delete all measurement types 61 from configurations where station 321 is the first station and 421 is the second station in the configuration both observing satellite # 8003201.

Another example is given here for measurement type 87 (doubly differenced one way ranges) (GPS):

1	+3+	4	-6+8
DELETE	3211878003201	870103121956	870103122156
DELETE	4212878307201	870103121956	870103122156
DELETE	5211878003201	870103115956	870103120156
DELETE	4212878307201	870103115956	870103120156
DELETE	3211878509301	870103100156	870103100556
DELETE	4212878307201	870103100156	870103100556
DELETE	5211878509301	870103093756	870103094156
DELETE	4212878307201	870103093756	870103094156
1	3	4	-6+8

This setup will delete all observations of measurement type  $87\,$  with the following cofigurations:

- 1. stat 1 =321, stat 2= 421, sat 1= 8003201 and sat 2= 8307201 between times 870103 121956.000 and 870103 122156.000 (Cards 1 and 2)
- 2. stat 1 =521, stat 2= 421, sat 1= 8003201 and sat 2= 8307201 between times 870103 115956.000 and 870103 120156.000 (Cards 3 and 4)
- 3. stat 1 = 321, stat 2= 421, sat 1= 8509301 and sat 2= 8307201

between times 870103 100156.000 and 870103 100556.000 (Cards 5 and 6)

4. stat 1 =521, stat 2= 421, sat 1= 8509301 and sat 2= 8307201 between times  $870103\ 093756.000$  and  $870103\ 094156.000$  (Cards 7 and 8)

An efficient way for the user to find out how many DELETE (or SELECT) cards he/she needs for each measurement type is to look at section 5.0 in Vol 3 of the GEODYN II documentation (Measuremebnt types). The measurement type description indicates how many stations and satellites are involved.

Assuming we have three stations and three satellites involved, the way to find out which goes on configuration number 1, 2 or 3 is to look at the unit 90 output of TDF. Satellites and stations under trhe label ONE, TWO or THREE correspond to the DELETE (or SELECT) card configuration number.

It is important for these complicated measurement types to specify information on as many DELETE (or SELECT) cards as the maximum number of stations or satellites involved in the measurement type.

IF CARD OMITTED: No data that has been selected will be deleted.

## 2.5.59.4 METDAT

METDAT	_	2+3+4+5+	
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-6	A 6	METDAT - Allows input of meteorological data overriding information on data.	
7-14	18	Station number.	0
18-24	17	Temperature in millidegrees K.	0
25-30	16	Relative humidity in thousandths of perce	nt. 0
31-40	D10.8	Atmospheric pressure in millibars.	0.
41-46 47-50 51-60	I6 I4 D10.8	Start date in form YYMMDD.: Start time in form HHMM.: Seconds of start time.:	Earliest Start date and time from EPOCH card(s).
61-66 67-70 71-80	I6 I4 D10.8	Stop date in form YYMMDD.: Stop time in form HHMM.: Seconds of stop time.:	Latest Stop date and time from EPOCH card(s).

IF CARD OMITTED: Meteorological corrections will be obtained from observation data records if present or will be defined as 293.16 degrees Kelvin, 40 percent relative humidity, and 1013.5 millibars scaled by the altitude in meters (H) of the tracking station, P=1013.5\*(1-1.1138D-4\*H).

## 2.5.59.5 OBSCOR

+ OBSCOR	1+	25	-6+	7		-+	8
+	0+	0+0+0+	-0+	0		-+	0
COLUMNS	FORMAT	DESCRIPTION	DEFAUL	T VAL	UE	& UNI	TS
1-6	A 6	OBSCOR - Constant bias observation corrections as provided on this card are to be subtracted from the data to which this card applies.					
7-14	18	Station number.		0			
15	I1	Station/satellite configuration.		0			
16-17	12	Measurement Type. 2 digits or less. See list of measurement types after ENDARC description.		0			
18-24	17	Satellite ID.		0			
25-30	16	Measurement Type. 3 digits or less. See list of measurement types after ENDARC description.		0			
31-40	D10.8	Observation correction.		0.		M/S RAD	
51-60	D10.8	Observation #2 correction.		0.	RA	D	
71-80	D10.8	Time correction.		0.	S		
UNITS:	KM =Kilom	eters; M =Meters ; S =Seconds ; K	g =Kilo	grams			

UNITS: KM =Kilometers; M =Meters ; S =Seconds ; Kg =Kilograms DEG=Degrees ; RAD=Radians ; AS =Arc seconds; MAS=Milli-arc seconds

IF CARD OMITTED: Constant bias corrections will not be subtracted.

	PREPRO									
+ PREPRO	1+	2+	-3+-	4	-+	-5	+	6+-	7	+8
+	0+	0+	-0+-	0	-+	-0	+	0+-	0	+0
COLUMNS	FORMAT	DESCR	IPTION					DEFAULT	VALUE	& UNITS
1-6	A6	PREPRO - R Provides a observation indicators	means on data p	of overr oreproce	iding ssing					
7-14	18	Station num	mber.							
15	I1	Station/sa	tellite	configu	ration	ı.				
16-17	12	Measurement See list of ENDARC des	f measur	ement t				0		
18-24	17	Satellite	ID.							
25	I1	Indicates which the applies.								
		0 - PREP reco: 9 - PREP reco: n - PREP numb	rds RO words rds	on all	Maste	er hea	der			
COLUMNS	FORMAT	DESCR	IPTION					DEFAULT	VALUE	& UNITS
26	I1	Defines d 27-30, 41 columns r	-50, and	61-70	if the					
		Value	Actio	n						
		0		and ap				d.		
		1		ply cored by th			ready			
		2	words a	the dat and appl	y cor	rectio	ns			
		3	Remove in the	correct data.	ions ]	provid	ed			
		4		correct data an	_					

apply corrections appropriate for this type of data.

27-30	4I1	Switches for bits 1-4 of PREPRO word.	
41-50	10I1	Switches for bits 5-14 of PREPRO word.	
61-70	10 I 1	Switches for bits 15-24 of PREPRO word.	
71-80	D10.8	Measurement Type. 3 digits or less. Will be converted to an integer. See list of measurement types after ENDARC description.	0

- NOTES: [1] Columns 27-30, 41-50, 61-70 have different meanings for different types of data. However, digits specified in these columns result in actions being taken by the program in the same manner as indicated for column 26.
  - [2] If PREPRO is intended to apply to all data a PREPRO card should preceed all SELECT cards otherwise PREPRO applies only to its preceeding SELECT card in the setup.

IF CARD OMITTED: The absence of a PREPRO card is an indicator that the observation data is to be used as it appears on the input data file. With a blank PREPRO card corrections will be applied as indicated by the PREPRO word on all header records.

2.5.59.7	SIGMA
----------	-------

2.5.59.7			_		-
SIGMA	7010 51		. 25		
+	0+	0+0+0-	+0+	0+	0+0
COLUMNS	FORMAT	DESCRIPTION		DEFAULT	VALUE & UNITS
1-6	A 6	SIGMA - Permits specif observation weighting a		· .	
7-14	18	Station number.		0	
15	I1	Station/satellite confi	guration #.	0	
16-17	12	Measurement Type. 2 dig See list of measurement ENDARC description.		0	
18-24	17	Satellite ID.		0	
25-30	16	Measurement Type. 3 dig See list of measurement ENDARC description.		0	
31-40	D10.8	Observation standard de Used in forming normal		Obtained	from data
41-46	16	Flag used to indicate is columns 31-40 will scal observation sigma or redata observation sigma.  0=replaced obssig with	e the data eplace the	0	
		columns 31-40 1=scale obssig by val 31-40			
51-60	D10.8	Observation editing signata editing only. If the left blank the standard specified in columns 31 used for data editing.	his field is deviation	Value in	cols. 31-40
Units f	or sigmas	are : Range - M, Range	Rate - CM/S, Angl	es - AS	
UNITS:		eters; M = Meters ; S es ; RAD=Radians ; A meters			
	preceed al	s intended to apply to a l SELECT cards, otherwis SELECT card in the setup	e SIGMA applies o	only to it	s
SIGMA	40	2.0	2.0	0004440	00000
SELECT SIGMA	40	0.0	0.0	9004140	,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,

SELECT			90041400000.	900415000000
SIGMA	40	1.0	1.0	
SELECT			90041600000.	900417000000
SIGMA	40	0.0	0.0	

In the example above sigma= 0.0 will be applied to all measurements between 900413 and 900414 and between 900416 and 900417, sigma= 1.0 will be applied to all observations between 900414 and 900415, sigma= 2.0 will be applied to all observations between 900415 and 900416. The epoch times in this example are 900413 and 900417.

IF CARD OMITTED: Sigma on data records will be used.

# 2.5.60 ARC SET TERMINATOR

## 2.5.60.1 ENDARC

---+---1---+---2---+---3----+---4----+---5----+---6----+---7----+---8
ENDARC
---+--0---+--0---+---0---+---0---+---0

COLUMNS FORMAT

DESCRIPTION

DEFAULT VALUE & UNITS

1-6 A6

ENDARC - End of DATA SUBGROUP and end of arc indicator. Must be last card in any DATA SUBGROUP section. Only the following option cards can exist between a DATA card and an ENDARC card: SELECT, DELETE, METDAT, OBSCOR, PREPRO and SIGMA.

# 3 GENERAL INFORMATION ABOUT GEODYN-II

In the following section a general description is given concerning INPUT/OUTPUT files used by GEODYN-II. The IN-PUT/OUTPUT files include file names, format specifications for the scratch units and descriptions for both GEODYN-IIS and GEODYN-IIE I/O files including space allocation requirements where needed. Furthermore, the ephemeris files (GEODYN IIS UNIT 1) and the tables file (polar motion, UT1, solar and geomagnetic flux) (GEODYN IIS UNIT 2) are provided. Brief descriptions of the default gravity field file (GEODYN IIS UNIT 12) and the default station geodetics file (GEODYN IIS UNIT 16) are also given.

Unit names on various computers

Computer	Name		
IBM	FTNNF001		
CRAY	fort.N		
HP	ftnNN		
NN = 01,02,,99			
$N = 1, 2, \dots, 99$			

# 3.1 GEODYN-IIS INPUT/OUTPUT FILE NAMES AND BRIEF DESCRIPTIONS

UNIT NO.	FORMAT RECFM LRECL BLKSIZE	DESCRIPTION
1		JPL Planetary Ephemeris input. Necessary for all runs.
2	VBS 196 7294	Polar Motion, UT1, and Solar & Geomagnetic Flux input. Necessary for all runs.
5	FB 80 *	GEODYN-IIS Input Control Data. Necessary for all runs.
6	VBA 137 #	General Printer Output.
8	FB 80	LRA correction file
11 BINARY ASCII	VBS X 15476 FB 132 15048	Interface File output to GEODYN-IIE Allocated default space of 90 tracks will hold 4050 blocks of observations and minimal number of parameters. Recommended space is 30 tracks, and in case additional space is needed, add 1 track per 45 blocks of observations. Space on unit 11 is a function of the number and the type of the parameters used, the number of observations and the number of blocks of observations.
12	FB 80 *	Default Gravity File. See description at the end of this section. (Section 3.5)
15	FB 80 3120	Default Spacecraft Area and Mass Table. (DUMMY) $N/A$
16	FB 80 *	Default Station Geodetics File. See description at the end of this section. (Section 3.6)
UNIT NO.	RECFM LRECL BLKSIZE	DESCRIPTION
17	FB 80 *	Default Simulated Data Control Data See description at the end of this section. (Section 3.7)
18	VBS X 15476	File which contains harmonic expansion coefficients that describe the atmospheric effect on the gravity coefficients (See Vol 3 ATGRAV)
20	VBS X 15476	Proudman functions input file (See Vol 3 OTMOD and Vol 5 Section 2.10)

21	VBS	Х	15476	Sea Surface Topography functions input file (See Vol 3 SSTMOD and Vol 5 Section 2.10)	
23	FB	80	8000	Input file for station height modification (See INSTRMNT card col. 10)	
30	VBS	X	15476	Scratch file used for planetary ephemeris processing. Allocated space of 5 tracks will hold 1135 ephemeris records (32 days each. For additional n output records increase the number of tracks by ((n-1)/227)+1.	
40	VBS	Х	15476	Input Observation Data file (from Tracking Data Formatter). Necessary for all data reduction runs.	
41 BINARY ASCII	VBS FB	X 132	15476 15048	Selected Observation Data file output to GEODYN-IIE. Necessary for all data reduction runs.  One track is required for every 50-100 observations depending on measurement type.  Default allocated space 60 tracks.	
42	υ		32000	Observation Directory scratch file. Allocated space of 200 blocks of 32000 bytes each will hold 200 observation directory records. The number of observation directory records is also the number of output logical blocks of observations. For additional space the maximum number of observations per pass/500 gives the number of output logical blocks and therefore the number of additional blocks required.	
UNIT NO.	RECFM	FORM.	L BLKSIZE	DESCRIPTION	
43	VBS	X	15476	Pass-by-pass bias scratch file. Allocated space of 10 tracks will hold 2020 pass records. For additional passes increase the number of tracks by ((n-1)/202)+1	
50	FB	80	3520	Copy of setup deck scratch file. Allocated space of 6 tracks will hold information from 1428 cards. For additional n input cards increase number of tracks by n*80/BLKSIZE	
52	VBS	Х	15476	Scratch files for ARC information. Allocated space should be less than or equal to the space required for UNIT 11.	
53	VBS	Х	15476	Scratch files for ARC information. Allocated space should be less than or equal to the space required for UNIT 11.	

77				File containing external sunlight ratio data to be used if called for on the SOLRAD card.
88	FB	80	15040	File containing GEODYN warnings for the TOPEX Expert Systems.
89	VBS	X	15476	File containing A1-UTC time differences
90				Debugging output file (DUMMY)
91				Debugging output file (DUMMY)
** exa	t## VBS	X	15476	External attitude file. ## is the number of the arc for which the external attitude applies.

<sup>\*</sup> BLKSIZE FOR FB FILES IS PROVIDED BY THE SYSTEM

\*\* THERE IS NO UNIT NO. FOR THIS FILE. FILENAME IS SPECIFIED EXPLICITLY.

<sup>#</sup> BLKSIZE FOR VBA FILES IS PROVIDED BY THE SYSTEM

# 3.2 GEODYN-IIE INPUT/OUTPUT FILE NAMES AND BRIEF DESCRIPTIONS

UNIT			L BLKSIZE	DESCRIPTION
5	FB	80	3200	Debug print switch input (N/A)
6	VBA	137	#	General printer output
7	FB	80	3200	Updated setup ('punched') output
8	VBA	137	#	Cartesian ephemeris printer output
9	FB	80	8000	Terminal output (see TERMVU description)
10	VBS FB	X 132	15476 15048	Keplerian ephemeris printer output. VBS for unformatted and FB for formatted.
11	VBS FB	X 132	15476 15048	Interface file from IIS. VBS for unformatted (BINARY) and FB for formatted (ASCII).
12	VBS FB	15470 132	15476 15048	Observation file from IIS (DCB for binary and ascii).
13	U	18944	18944	Observation random access scratch file. Allocated space equal or larger than space of data set used on UNIT 12. For additional n observations increase the number of tracks by $((n+5)+(n+5)*4)/18944$
14	FB	80	15040	Copy of setup deck scratch file. Allocated space of 20 tracks will hold 4760 cards. For cards add 1 track per 238 cards.
15	VBA	137	#	Normal point printer report file.
16	VBA	137	#	Preprocessing corrections printer file.
17				PDOUT file.
18				Geolocation Output file.
19	VBS	X	15476	Observation/Residual output file
20	VBS	X	15476	Simulation/Normal Point Generation output file Space requited is the same as in GEODYN IIS UNIT 41.
27	VBS	X	15476	Proudman function Direct access file
30	VBS	Х	15476	Recommended unit for trajectory file.
35	FB	80	*	Shadowing beginning and end times in the form of DELETE cards. Used only by simulated data option.

42-48	VBS X	15476	GPS shadwing
51	VBS X	15476	Work file for partial derivative storage when forming normal equations by parts. Total amount of partial derivative scratch space when forming the normal equations with partitioning will be 2*(20+NUMBER OF ADJUSTED PARAMETERS)* NUMBER OF WEIGHTED MEASUREMENTS*8/15476 tracks.
5 x 5 5 6 x 7 0	VBS X	15476	Optional work files for partial Optical Data Auxillary file derivative storage when forming normal equations by parts.
62	VBS X	15476	Dstate Constraints
66	VBS X	15476	External Ephemeris output. See Volume 3 - XEPHEM input card. See Volume 5 - Section 4.
67	VBS X	15476	File used for the Yarkovksy model
71	VBS X	15476	E-matrix output files (x > 0). Space required is ((NUMBER OF ADJUSTED PARAMETERS)**2)*4/15476 tracks.
80	VBS X	15476	V-matrix output files. (N/A) Space required is ((6*NUMBER OF ADJUSTED FORCE MODEL PARAMETERS +12)*(LENGTH OF ARC/OUTPUT RATE)+5*[(3*NUMBER OF ADJUSTED MODEL PARAMETERS)+12])/15476 tracks.
88	FB 80	15040	File containing GEODYN. warning messages for the TOPEX Expert System.
90	VBS X	15476	Measurement Partial Derivative output. (N/A) Space required is ((50+NUMBER OF ADJUSTED PARAMETERS)*NUMBER OF OBSERVATIONS)+(6*NUMBER OF ADJUSTED PARAMETERS+1))/15476 tracks.
96	FB 132	13200	Unit 96 writes out station position changes due to ocean loading.
97	binary ieee - on Cray	64 bit binary	Spacecraft telemetry file.
98	FB 132	13200	Unit 98 writes out station position changes due to the solid tide.
99	FB 132	13200	Unit 99 writes out station position effects where the station height has been modified

using the INSTRMNT card (col. 10).

400

Unit 400 writes out the troposhere EBIAS and the zenith path delays if the option is selected on the REFRAC card.

- \* BLKSIZE FOR FB FILES IS PROVIDED BY THE SYSTEM
- # BLKSIZE FOR VBA FILES IS PROVIDED BY THE SYSTEM

# 3.3 PLANETARY EPHEMERIS FILES

# 3.4 POLAR MOTION, UT1, SOLAR AND GEOMAGNETIC FLUX FILE

```
CRAY (charney)
----
/u2/z8sgp/geodyn/support/gdntable.data

HP (geodesy2)
--
/geod4/geodyn/SUPPORT/gdntable.data

SUN (Canopy)
---
/home/geodyn/support/gdntable.data
```

# 3.5 DEFAULT GRAVITY FIELD FILE (UNIT 12)

UNIT 12 in GEODYN IIS is used as input unit for gravity field setup. Gravity information on UNIT 12 will be used unless other information is input on UNIT 05. Cards which can be input on UNIT 12 are the following:

- TITLE which contains descriptive information.
- EARTH which introduces earth constants.
- GCOEF which C and S defines coefficients in the geopotential model.
- GCOEFC which defines C coefficients in the geopotential model.
- GCOEFS which defines S coefficients in the geopotential model.

The structure of this file is such that the first card is a TITLE card followed immediately by an EARTH card and then as many GCOEF or GCOEFC and GCOEFS cards as necessary to define all of the spherical harmonic coefficients. All the GCOEFC cards should be together followed by all the GCOEFS cards. Estimation information input on UNIT 12 is ignored.

See GLOBAL SET EARTH, GCOEF, GCOEFC, GCOEFS for detailed description and format. EARTH, GCOEF, GCOEFC and GCOEFS cards on UNIT 05 override information on the same cards on UNIT 12.

# 3.6 DEFAULT CARD IMAGE STATION GEODETICS FILE (UNIT 16)

UNIT 16 in GEODYN IIS is used as input unit for Stations position setup. Station position information included on UNIT 16 will be used unless other information referring to the same stations is included on UNIT 05. Cards to be included as input to UNIT 16 are the following:

- STAPOS which introduces the station position subgroup
- GEODETIC which defines station geodetic information
- EXTRAGEO which specifies planetary shape parameters
- ELCUTOFF which sets station elevation cutoff angle
- INSTRMNT which sets station antenna and provides operating frequency information
- STATION COORDINATE which gives the station positions
- STATION VELOCITY GROUP CARDS (STAVEL, TIMVEL only)
- STATION EARTH TIDES COEFFICIENT CARDS (STATH2 and STATL2)
- ENDSTA which denotes the end of the station position subgroup

The structure of this file is such that the first card should be a STAPOS card and the last card is a ENDSTA card. STATION COORDINATE cards, STATION VELOCITY cards and STATION EARTH TIDES COEFFICIENT cards follow the appropriate cards of other types necessary to specify the conditions that are to apply to those station locations.

Cards which contain a justment information (ADJUSTED, CORREL, CONSTADJ, CONSTEND) are not to be included on UNIT 16.

For detailed description see the GLOBAL SET STAPOS subgroup. Input cards included on UNIT 05 will override input cards on UNIT 16.

#### 3.7 DEFAULT SIMULATED DATA CONTROL FILE (UNIT 17)

UNIT 17 in GEODYN IIS is used as input for the simulated data capability where observations are to be created not from real data replacement with computed observations but from generation of simulated tracking data according to the users instructions. These instructions are to be supplied in UNIT 17 as a set of input cards containing information about measurement types, station/satellite configuration, times and various other limitations. The same information may be specified on UNIT 05 instead, right after the ARC OPTION SIMDAT card. Information on UNIT 05 overrides information on UNIT 17. Data created using this capability after being checked for visibility will be output on UNIT 20 in the internal GEODYN II format and therefore they can be reintroduced later into GEODYN IIS as if they were real data.

The simulated data generation capability is introduced by including a SIMDAT card in the ARC section of UNIT 05. The ARC OPTION SIMDAT card must have a 1 on column 8, otherwise simulated data will be generated by replacing real observations with the computed ones.

The input cards in UNIT 17 (or UNIT 05) must begin with a SIMDAT card. This card is NOT the same as the ARC OPTION SIMDAT card. Its presence requests simulated data generation according to the specifications that follow. The cards which can be included on UNIT 17 are the following:

```
SIMDAT
SIMLIM

SIMSAT .
SIMSTA . These cards are called the simulated SIMTYP . data configuration group.
SIMTIM .
ENDCON .
```

The sequence of the input cards is as given above. The SIMDAT card is mandatory. The SIMLIM card introduces certain limitations for the blocks of data to be generated.

The simulated data configuration group describes in five cards elements and scheduling information for the simulation. These cards are described individually in the next chapter. The last card in this group is the ENDCON card and denotes the end of this particular configuration.

The user may include as many simulated data configuration groups as desired. Finally the ENDSIM card denotes the end of the simulated data generation input.

A detailed description of all the cards on UNIT 17 is given in Section 2.6 right after the SIMDAT arc option card and right before the SOLRAD arc option card.

# 4 TRACKING DATA FORMATTER OPERATIONS

The Tracking Data Formatter program is designed to convert data in a variety of formats to blocked GEODYN-II binary format. (64-bit floating point format). Input to the program consists of BIH tables, a set of users option cards and data of various format types falling into one of the following categories:

- 1. PCE Data Format
- 2. GEOS-A/B Card Image Format N/A
- 3. GEOS-C Card Image Format
- 4. GEODYN Binary Format
- 5. GEODYN Binary Format extended for Altimetry

Chronologically ordered files containing any kind of the above data types can be input and they are block time merged. The output observation file (IUNT41) contains blocks of data sorted by measurement type and by tracking pass.

The Tracking Data Formatter includes input card processing and a limited amount of data preprocessing.

The input cards introduce certain limits and requirements for further data reduction, (special description follows).

During measurement processing input files of data are read and time merged. Furthermore they are converted into internal binary format and each preprocessing word is decomposed. Range data are transit time corrected (all time changed to ground received) and all data are changed to UTC time. Logical file blocks are formed which include one measurement type from one tracking pass per output block. Preprocessing words are composed according to GEODYN-II binary tracking data format and the Header Record as well as the Observation Records are formed for each block.

# 4.1 INPUT / OUTPUT UNITS

# INPUT UNITS

UNIT #		FORMA:	r BLKSIZE	DESCRIPTION
2	VBS	196	7294	BIH tables (same as required for GEODYN-II runs)
5	FB	80		User option cards (see special description)
22-40				Data in the format specified in the FORMAT card. Only six of these units can be used at any one time. See FORMAT card in input option description.

#### **OUTPUT UNITS**

UNIT #	RECF	FORMA M LRECL	T BLKSIZE	DESCRIPTION
6	VBA	137	#	Printer output giving summary of run.
41	VBS	Х	15476	Tracking data in format compatible with GEODYN-II Allocated space 600 tracks.
90	VBA	137	#	Printer output giving summary of each block of data reformatted.

# SCRATCH UNITS

UNIT #	RECFM	FORMAT LRECL	r BLKSIZE	DESCRIPTION
20	VBS	Х	15476	Scratch file for the altimeter crossover identification (could be a dummy file in non-crossover runs)
21	U		15476	Scratch file used when forming new physical blocks. Allocated space of 100 tracks will hold 100 physical blocks. Each physical block contains 236 logical records of length 10
50	FB	80		Copy of setup deck scratch file Allocated space 2 tracks.

#### 4.2 TDF UNIT 5 INPUT CARDS

In the following pages a complete description of all the cards which could be used as input to the Tracking Data Formatter is given.

Information specified by these cards include format of input data, size limitations for logical blocks to be formed, a set of data select cards, an option card for normal point data and a set of cards used for altimetry data.

It is mandatory to use a FORMAT card unless GEODYN binary data are used. PASS and MAXOBS are optional cards. The Tracking Data Formatter input cards may include a DATA SELECTION subgroup similar in editing capabilities to the DATA SELECTION subgroup of GEODYN-IIS. This subgroup must start with a DATA card and end with an ENDDAT card. It may include cards for data selection and deletion limitations and specifications concerning normal points in case this option is desired.

Each individual input UNIT must be represented by its own DATA SELECTION subgroup. If DATA SELECTION conditions are the same for all input UNITS then the DATA SELECTION subgroup nmust be included after the last FORMAT card.

For altimetry data the set of ALTIM and LOCDAT cards should both be input and in the above sequence. The ENDALL card denotes the end of the Tracking Data Formatter input cards.

# **4.2.1 FORMAT**

+ FORMAT	=	+2+3+8
+	0	+0+0+0
COLUMNS	FORMAT	DESCRIPTION DEFAULT VALUE & UNITS
1-6	A6	FORMAT
13-14	12	Unit number for input of data in the 40 format denoted by the number in column 17 (must be greater than 22 and smaller than 41)
16	I1	Type of binary data [4] 0
		0 - Binary data is old IBM format which has I*2's and R*4's.
		1 - Binary data is new IBM format which only has $I*4's$ and $R*8's$ .
17	I1	<pre>Type of Data Format Indicator [3] 0 - Geodyn binary 1 - 90 byte ASCII Range Rate Data 2 - Merit II SLR Data 3 - GEOS-3 card image - 80 byte format 4 - Extended Geodyn binary (altimetry) 5 - PCE [2] 6 - Extended GEOS-3 card image - 120 byte format 7 - GPS metric type format 8 - VLBI binary format 9 - Merit-X SLR Data from CRD reformatter</pre>
18	I1	Extended Geodyn altimetry binary format type  0 - Old extended format         (34, 8 byte words for a I64 machine or             38, 4 byte words for a I32 machine) (default)  1 - Super extended altimetry format         (54, 8 byte words for a I64 machine or         58, 4 byte words for a I32 machine)
24	I1	Time system indicator change 7 Converts time system indicator 7 of modified SSD data to a 3 (UTC BIH). If this is omitted epoch time scale will be considered as being ET
30	I1	<ul><li>1 - Indicates altimetry data on this input unit</li><li>2 - Indicates crossover data on this unit</li></ul>

NOTE [1]: One format card must be present for every type of data used

in the run, unless only Geodyn binary data is used, then the unit will default to 40. The data from the different units will be time merged. One can also use this to merge two sets of the same type data by inputting each set on a different unit. No more than 6 FORMAT cards can be input.

- NOTE [2]: If PCE data is input then the FIRST record in the data stream on the logical unit specified by the pertinent FORMAT card must have a 90 in columns 2 and 3 and the satellite ID is columns 22-45 (format D24.16). All data following this card goes with that satellite. If using PCE data from more than one satellite then insert another FIRST record each time the satellite ID changes.

  See Volume 5 for the format of the FIRST record.
- NOTE [3]: For a detailed description of the various formats see Volume 5, Section 1.0.
- NOTE [4]: This option only applies when the TDF is run on a computer that allows I\*2 and R\*4 data formats.
- NOTE [5]: Column 17 on this card should be 4 for this option.

# **4.2.2** GDYNEP

GDYNEP	-	+3+4+5+		
	0		-0+-	00
COLUMNS	FORMAT	DESCRIPTION	DEFAULT	VALUE & UNITS
1-6	A 6	GDYNEP - specifies a new Geodyn Internal Reference Time for this run		
25-30	16	New Geodyn Internal Reference Time in "YYMMDD". NOTE	410101	(JD = 2430000.5)

NOTE: If YY = [51-99] the year is 19YYIf YY = [00-50] the year is 20YY

IF CARD OMITTED: TDF will use its default reference time

4.2.3 P	ASS		
PASS	_	+2+3+4+5+	
COLUMNS	FORMAT	DESCRIPTION	DEFAULT VALUE & UNITS
1-4	A4	PASS	
31-40	D10.8	Max time difference allowed between contiguous data records before starting new logical block (sec)	900

#### **4.2.4** MAXOBS

MAXOBS		+2+6+			
COLUMNS	FORMAT	DESCRIPTION DEFAUL	LT VALUE & UNITS		
1-6	A6	MAXOBS			
7-14	18	Maximum number of observations allowed in one logical block (limit=9999) (default for PCE data is 500)	1000 [1]		
18-24	17	Maximum number of output buffers per logical block.(MAXBUF) MAXBUF must be less than or equal to 100 and greater than or equal to 10	10		
25-30	16	Maximum number of scratch buffers in core	5		
		must be greater than or equal to 5 and less than or equal to 500. [2]	200		
31-40	D10.3	Bin size in seconds for normal point calculations. If this is present then a value greater than zero must be entered in columns 51-60.			
41-46	16	Maximum internal directories that can be open simultaneously during TDF execution. Must less than 1000. [3]	500		
51-60	D10.3	Minimum spacing between data points for normal point calculation runs. All data closer than minimum spacing will be discarded. If this value is present then a value greater than zero must be entered in columns 31-40. This value can be overridden for specific stations and/or measurement types if NORMPT cards are present.			
NOTE [1]: IIS imposes two restrictions on a block of observations. No block can span a time period that is longer than 10*STEPSIZE, and no single block of observations from the TDF can be subdivided into more than 100 blocks in IIS. The number of IIS subdivisions is given by: IIS subdivisions = Number of observations in a block/ (10*STEPSIZE/time interval between observations). If the number of IIS subdivisions is greater than 100, the maximum number of observations per block must be reduced.					
NOTE [2]		this parameter needs to be greater than 500, please GEODYN Group. The error in TDF if 500 is not enough			

 ${\tt record}$ " . In that case please contact someone from Geodyn

TDF stops with a message "UNIT21" attempt to read past the end of

NOTE [3]: Often when large data sets with many configurations are processed

"Unit 21, attempt to read past end of record"

group to help you.

4.2.5 DA	1.2.5 DATA						
DATA	_	+3+	-			·	
COLUMNS		DESCRIPTION	· ·			VALUE &	·

1--4  $$\operatorname{A4}$$  DATA Introduces the data selection subgroup.

# **4.2.6 SELECT**

SELECT		+2+3+4+5+		
COLUMNS	FORMAT	DESCRIPTION	DEFAULT	VALUE & UNITS
1-6	A6	SELECT - Requests data be selected between specific times.		
25-30	16	Minimum spacing between observations	0	sec
41-46	16	YYMMDD - year, month, day of time to begin data selection.		
47-50	14	HHMM - hours, minutes of time to begin data selection.		
51-60	D10.8	SEC - seconds of time to begin data selection.		
61-66	16	YYMMDD - year, month, day of time to end data selection.		
67-70	14	HHMM - hours, minutes of time to end data selection.		
71-80	D10.8	SEC - seconds of time to end data selection.		

NOTE: [1] For low rate altimetry data to be produced, specify the minimum spacing in seconds in columns 25-30.

NOTE: [2] One can use only one SELECT card per input unit.

# **4.2.7 DELETE**

DELETE	_	+2+3+4+5+6+7+8 +0+0+0+0
COLUMNS	FORMAT	DESCRIPTION DEFAULT VALUE & UNITS
1-6	A6	DELETE - Deletes data by either type or station
7-14	18	Station number - all data for this station will be deleted.
16-17	12	Measurement type - GEODYN I - all data of this type will be deleted.
NOTE:		One can use as many DELETE cards as necessary, however any one card can indicate only one station number or a measurement type.  If both fields 7-14 and 16-17 are non-zero then the card is invalid.

#### **4.2.8 NRMPNT**

+1	+2-	+3	 -4	-5+	6+	78
NRMPNT						
+0	+0-	+0	 -0+	-0+	0+	0+0

COLUMNS FORMAT

DESCRIPTION

DEFAULT VALUE & UNITS

- 1-6 A6 NRMPNT is used when normal point data is being processed to designate the minimum spacing between accepted data for specific station and/or measurement type. All data closer than minimum spacing (col 51-60) will be discarded. Default value on MAXOBS card will be used if no specific NRMPNT card is in for station or type
- 7-14 I8 Station number The minimum spacing in columns 51-60 will be used for all data from this station.
- 16-17 I2 Measurement type GEODYN I the minimum spacing in columns 51--60 will be used for all data of this type.
- 51-60 D10.3 Minimum spacing between data points in seconds.
  All data closer than this spacing will be discarded.

NOTE:

One can use as many NRMPNT cards as necessary. If only the station field is present it refers to all data from that station. If only the measurement type field is present then it refers to all data of that type. If both fields are present then it refers to only data of that type at that station. Cards with both fields present will override cards with only station or only measurement type fields defined. For this function the order of the cards does not play any role.

# 

cards.

# 4.2.10 ALTIM

ALTIM	1	+2+3+4+5+ 0 +0+0+0+		
COLUMNS	FORMAT	DESCRIPTION	DEFAULT	VALUE & UNITS
1-5	A5	ALTIM - Denotes set of LOCDAT cards immediately follow this card.		
13-14	12	Number of LOCDAT cards that follow must be at least 1 and less than or equal to to 5. THIS MEANS THAT AT LEAST ONE LOCDAT CARD IS NEEDED WITH THE ALTIMETRY RUNS. LOCDAT cards are informational only.		

# 4.2.11 LOCDAT

LOCDAT	1	+2+7+8					
+0+0							
COLUMNS	FORMAT	DESCRIPTION DEFAULT VALUE & UNITS					
1-6	A6	LOCDAT - This card is used for information only					
9-16	A8	Alphanumeric name denoting reference ellipsoid used for altimetry data.					
17-24	A8	Alphanumeric name denoting geoid on altimetry data.					
25-32	A8	Alphanumeric name denoting mean sea surface on altimetry data.					
33-40	A8	Alphanumeric name denoting orbits placed on altimetry data.					
41-60	D20.14	Semi-major axis of ellipsoid used to represent earth (meters).					
61-80	D20.14	Inverse of ellipsoidal flattening used in surface height calculations.					
NOTE 1:		At least one LOCDAT card must be included in all altimetry runs					
NOTE 2:		LOCDAT cards follow ALTIM card					
NOTE 3:		The number of LOCDAT cards must equal the value in columns $7\text{-}14$ on the ALTIM card.					

# 4.2.12 ENDALL

+1+2+	3+4+	5+6	+8
ENDALL			
	0+	0+0	+0+0

COLUMNS FORMAT DESCRIPTION

DEFAULT VALUE & UNITS

1-6 A6 ENDALL - Indicates end of unit 5 input no cards may follow this card.

# 5 MEASUREMENT TYPE DESCRIPTIONS

```
TYPE
NO.
       MEASUREMENT DESCRIPTION (REV's 1.5 and later)
       ______
01
       INERTIAL TRUE OF DATE "X" POSITION (S1) [NOTE 1]
       INERTIAL TRUE OF DATE "Y" POSITION (S1)
       INERTIAL TRUE OF DATE "Z" POSITION (S1)
03
 04
       INERTIAL TRUE OF DATE "X" VELOCITY (S1)
       INERTIAL TRUE OF DATE "Y" VELOCITY (S1)
 05
 06
       INERTIAL TRUE OF DATE "Z" VELOCITY (S1)
07
       OSCULATING SEMI-MAJOR AXIS
                                                (S1)
       OSCULATING ECCENTRICITY
 80
                                                (S1)
 09
     OSCULATING INCLINATION
                                                (S1)
 10
       OSCULATING RIGHT ASCENSION OF ASCENDING NODE (S1)
 11
       OSCULATING ARGUMENT OF PERIGEE
       OSCULATING MEAN ANOMALY
 12
                                                (S1)
 13
       INERTIAL RIGHT ASCENSION (S1-->T1) or Laser Illum. (T2-->S1-->T1)
       INERTIAL DECLINATION (S1-->T1) or Laser Illum. (T2-->S1-->T1)
14
 15
       LOCAL HOUR ANGLE (S1-->T1)
 16
       DECLINATION (S1-->T1)
 17
       AZIMUTH (S1-->T1)
      ELEVATION (S1-->T1)
 18
      X-ANGLE (S1-->T1)
 19
20
      Y-ANGLE (S1-->T1)
     1 DIRECTION COSINE (S1-->T1)
21
 22
     m DIRECTION COSINE (S1-->T1)
23
       LANDMARK SCAN ELEMENT (S1-->T1)
24
       LANDMARK SCAN LINE (S1-->T1)
       PLANETARY EDGE SCAN ELEMENT (S1-->T1)
 25
 26
       PLANETARY EDGE SCAN LINE (S1-->T1)
       CELESTIAL LANDMARK S/C B.C.F. RIGHT ASCENSION (S1-->T1)
 27
28
       CELESTIAL LANDMARK S/C B.C.F. DECLINATION (S1-->T1)
       PLANETARY EDGE S/C B.C.F. RIGHT ASCENSION (S1-->T1)
 29
30
       PLANETARY EDGE S/C B.C.F. DECLINATION (S1-->T1)
NOTE [1]: For information on radial PCE data see volume 5, Section 1.2.
TYPE
      MEASUREMENT DESCRIPTION
       ______
       VLBI DELAY (Q1-->T1,T2)
31
       VLBI DELAY RATE (Q1-->T1,T2)
 33
       CAMERA DATA (IMAGES)
 34
       LANDMARK DATA (PLANET/ASTEROID ORBITING MODE OR HELIOCENTRIC MODE)
 35
 36
       DELTA-DOR ((S1-->T1)-(S1-->T2)) - (Q1-->T1,T2)
       ONE-WAY PLANET-EARTH RANGE (P1T1-->T2)
 37
 38
       ONE-WAY PLANET-EARTH DOPPLER (P1T1-->T2)
 39
       ONE-WAY STA-SAT RANGE (T1-->S1)
       ONE-WAY STA-SAT DOPPLER (T1-->S1)
 40
 41
     ONE-WAY RANGE (S1-->T1)
```

```
ONE-WAY DOPPLER (S1-->T1)
 42
 43
        ONE-WAY SAT-SAT RANGE
                                (S2-->S1)
        ONE-WAY SAT-SAT DOPPLER (S2-->S1)
 44
        TWO-WAY GROUND TRANSPONDER RANGE (S1-->T1-->S1)
 45
 46
        TWO-WAY GROUND TRANSPONDER DOPPLER (S1-->T1-->S1)
        TWO-WAY EARTH-PLANET-EARTH RANGE
 47
                                           (T2-->P1T1-->T2)
 48
        TWO-WAY EARTH-PLANET-EARTH DOPPLER (T2-->P1T1-->T2)
        TWO-WAY SAT-SAT RANGE
                               (S2-->S1-->S2)
 49
        TWO-WAY SAT-SAT DOPPLER (S2-->S1-->S2)
 50
        TWO-WAY RANGE
                        (T1-->S1-->T1)
 51
 52
        TWO-WAY DOPPLER (T1-->S1-->T1)
        THREE-WAY RANGE
 53
                          (T2-->S1-->T1)
        THREE-WAY DOPPLER (T2-->S1-->T1)
 54
        THREE-WAY SAT-SAT RELAY RANGE
 55
                                         (S2-->S1-->T2)
        THREE-WAY SAT-SAT RELAY DOPPLER (S2-->S1-->T2)+SF*(T2-->S1-->T2)
 56
        FOUR-WAY SAT-SAT RELAY RANGE
 57
                                       (T3-->S3-->S1-->S2-->T2)
        FOUR-WAY SAT-SAT RELAY DOPPLER (T3-->S3-->S1-->S2-->T2)+SF*
 58
                                        (T2-->S2-->T2)
 59
        FOUR-WAY SAT-TARGET RELAY RANGE
                                           (T3-->S3-->T1-->S2-->T2)
        FOUR-WAY SAT-TARGET RELAY DOPPLER (T3-->S3-->T1-->S2-->T2)+SF*
 60
                                           (T2 --> S2 --> T2)
TYPE
        MEASUREMENT DESCRIPTION
NO.
        SINGLY DIFFERENCED ONE-WAY RANGES
 61
                                           (S1-->T1)-(S1-->T2)
 62
        SINGLY DIFFERENCED ONE-WAY DOPPLERS (S1-->T1)-(S1-->T2)
        SINGLY DIFFERENCED ONE-WAY RANGES
                                             (S1-->T1)-(S2-->T1)
 63
        SINGLY DIFFERENCED ONE-WAY DOPPLERS (S1-->T1)-(S2-->T1)
 64
        SINGLY DIFFERENCED ONE-WAY SAT-SAT RANGES
 65
                                                     (S2-->S1)-(S3-->S1)
        SINGLY DIFFERENCED ONE-WAY SAT-SAT DOPPLERS (S2-->S1)-(S3-->S1)
 66
        SINGLY DIFFERENCED ONE-WAY SAT-SAT RANGES
 67
                                                      (S2-->S1)-(S2-->T1)
 68
        SINGLY DIFFERENCED ONE-WAY SAT-SAT DOPPLERS (S2-->S1)-(S2-->T1)
        SINGLY DIFFERENCED ONE-WAY PLANETRY RANGES
 69
                                                      (P1T1 --> T2) - (P1T1 --> T3)
 70
        SINGLY DIFFERENCED ONE-WAY PLANETRY DOPPLERS (P1T1-->T2)-(P1T1-->T3)
        SINGLY DIFFERENCED ONE- AND TWO-WAY RANGES
 71
                                                       (S2 --> S1 --> T2) -
                                                       (S2-->T2)
 72
        SINGLY DIFFERENCED ONE- AND TWO-WAY DOPPLERS (S2-->S1-->T2)-
                                                       (S2-->T2)
        SINGLY DIFFERENCED THREE-WAY SAT-SAT RANGES
 73
                                                        (S1-->S2-->T2)-
                                                        (S1 --> S3 --> T3)
        SINGLY DIFFERENCED THREE-WAY SAT-SAT DOPPLERS (S1-->S2-->T2)-
 74
                                                        (S1 --> S3 --> T3)
        SINGLY DIFFERENCED THREE-WAY SAT-SAT RANGES
 75
                                                        (S2-->S1-->T2)-
                                                        (S3-->S1-->T3)
        SINGLY DIFFERENCED THREE-WAY SAT-SAT DOPPLERS (S2-->S1-->T2)-
 76
                                                        (S3-->S1-->T3)
        SINGLY DIFFERENCED TWO- AND THREE-WAY RANGES
 77
                                                         (T2-->S1-->T1)-
                                                         (T2-->S1-->T2)
        SINGLY DIFFERENCED TWO- AND THREE-WAY DOPPLERS (T2-->S1-->T1)-
 78
                                                         (T2-->S1-->T2)
        SINGLY DIFFERENCED FOUR-WAY SAT-SAT RANGES (T3-->S3-->S1-->S2-->T2)-
 79
                                                     (T3-->S3-->S1-->S3-->T3)
        SINGLY DIFFERENCED FOUR-WAY SAT-SAT DOPPLERS (T3-->S3-->S1-->S2-->T2)
 80
              +SF2S*(T2-->S2-->T2)-(T3-->S3-->S1-->S3-->T3)-SF3S*(T3-->S3-->T3)
        SINGLY DIFFERENCED FOUR-WAY SAT-TARGET RANGES (T3-->S3-->T1-->S2-->T2)-
 81
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(T3-->S3-->T1-->S3-->T3)
 82
        SINGLY DIFFERENCED FOUR-WAY SAT-TARGET DOPPLERS (T3-->S3-->T1-->S2-->T2)
              +SF2T*(T2-->S2-->T2)-(T3-->S3-->T1-->S3-->T3)-SF3T*(T3-->S3-->T3)
        SINGLY DIFFERENCED FOUR-WAY SAT&TARGET RANGES (T3-->S3-->S1-->S2-->T2)-
 83
                                                       (T3-->S3-->P1T1-->S2-->T2)
        SINGLY DIFFERENCED FOUR-WAY SAT&TARGET DOPPLERS (T3-->S3-->S1-->S2-->T2)
84
             +SF2S*(T2-->S2-->T2)-(T3-->S3-->P1T1-->S2-->T2)-SF2T*(T2-->S2-->T2)
TYPE
        MEASUREMENT DESCRIPTION
NO.
        DOUBLY DIFFERENCED ONE-WAY RANGES
                                             [(S2-->T1)-(S3-->T1)]-
85
                                             [(S2-->S1)-(S3-->S1)]
        DOUBLY DIFFERENCED ONE-WAY DOPPLERS [(S2-->S1)-(S3-->S1)]-
 86
                                             [(S2-->T1)-(S3-->T1)]
        DOUBLY DIFFERENCED ONE-WAY RANGES
                                             [(S1-->T1)-(S2-->T1)]-
 87
                                             [(S1-->T2)-(S2-->T2)]
        DOUBLY DIFFERENCED ONE-WAY DOPPLERS [(S1-->T1)-(S2-->T1)]-
 88
                                             [(S1-->T2)-(S2-->T2)]
        DOUBLY DIFFERENCED ONE - AND TWO-WAY RANGES
                                                      [(S2-->S1-->T2)-(S2-->T2)]-
 89
                                                      [(S3-->S1-->T3)-(S3-->T3)]
        DOUBLY DIFFERENCED ONE- AND TWO-WAY DOPPLERS [(S2-->S1-->T2)-(S2-->T2)]-
 90
                                                      [(S3-->S1-->T3)-(S3-->T3)]
        DOUBLY DIFFERENCED ONE - AND THREE-WAY RANGES [ (P1T1-->S2-->T2)
91
                           -(P1T1-->T2)]-[P1T1-->S3-->T3)-(P1T1-->T3)]
 92
        DOUBLY DIFFERENCED ONE - AND THREE-WAY DOPPLERS (P1T1-->S2-->T2)
                           -(P1T1-->T2)]-[P1T1-->S3-->T3)-(P1T1-->T3)]
        DOUBLY DIFFERENCED TWO- AND THREE-WAY RANGES
 93
                                                      (S1-->S2-->T2)
                        -(T1-->S2-->T2)]-[(S1-->S3-->T3)-(T1-->S3-->T3)]
        DOUBLY DIFFERENCED TWO- AND THREE-WAY DOPPLERS (S1-->S2-->T2)
 94
                        -(T1-->S2-->T2)]-[(S1-->S3-->T3)-(T1-->S3-->T3)]
        DOUBLY DIFFERENCED FOUR-WAY RANGES
 95
                                             [(T3-->S3-->S1-->S2-->T2)-
                   (T3-->S3-->S1-->S3-->T3)]-[(T3-->S3-->T1-->S2-->T2)
                                              -(T3-->S3-->T1-->S3-->T3)
 96
        DOUBLY DIFFERENCED FOUR-WAY DOPPLERS [(T3-->S3-->S1-->S2-->T2)+
                          SF2S*(T2-->S2-->T2)-(T3-->S3-->S1-->S3-->S3)-
                                                   SF3S*(T3-->S3-->T3)]-
                                              [(T3-->S3-->T1-->S2-->T2)+
                          SF2T*(T2-->S2-->T2)-(T3-->S3-->T1-->S3-->S3)-
                                                   SF3T*(T3-->S3-->T3)]
 97
        RESERVED
        RESERVED
98
99
        ALTIMETRY
        ALTIMETRY CROSSOVERS
100
        DYNAMIC CROSSOVERS
101
110 ALTIMETRY CONSTRAINT - LINEAR DISTANCE OF BOUNCE POINT
111 ALTIMETRY CONSTRAINT - RADIAL DISTANCE OF BOUNCE POINT
NOTE:
SYMBOL
        LEGEND
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Sn REFERS TO THE SATELLITE WHICH IS DEFINED ON BLOCK HEADER RECORD NUMBER "n" OF THE GEODYN-II BINARY TRACKING DATA FORMAT.

- Th REFERS TO THE TRACKING STATION WHICH IS DEFINED ON BLOCK HEADER RECORD NUMBER "n" OF THE GEODYN-II BINARY TRACKING DATA FORMAT.
- Pn REFERS TO THE PLANET ON WHICH THE TRACKING STATION To EXISTS.

  AN IMPLICIT REFERENCE TO PLANETARY BODIES IS MADE SIMPLY BY

  SPECIFICATION OF THE TRACKING STATION To. THIS IS BECAUSE EACH

  TRACKING STATION MUST BE REFERENCED TO A PLANETARY BODY WHEN

  INPUT TO GEODYN-II. THE EXPLICIT REFERENCE IN THE ABOVE MEASUREMENT

  DESCRIPTIONS TO PLANETARY BODY Po IS DONE TO REMIND THE READER OF

  THIS FORMAT THAT THESE MEASUREMENTS ARE COMMON OR LIKELY INTER
  PLANETARY MEASUREMENT TYPES. IN SOME INSTANCES, THE MEASUREMENTS

  ONLY MAKE SENSE WITHIN AN INTER-PLANETARY FRAMEWORK. LIKEWISE, IT

  IS POSSIBLE FOR MEASUREMENTS NOT EXPLICITLY DENOTED WITH Po TO BE

  OF AN INTER-PLANETARY NATURE.
- SFnX REFERS TO TWO-WAY DOPPLER SCALING FACTOR FOR SATELLITE n TRACKING USER A SATELLITE (X=S) OR REMOTE RELAY STATION (X=T).
- Qn REFERS TO THE QUASAR WHICH IS DEFINED ON BLOCK HEADER RECORD NUMBER "n" OF THE GEODYN-II BINARY TRACKING DATA FORMAT.